

1986

A SIMULATION PROGRAM FOR MULTIPLE TUBE-ROW TWO-PHASE THERMOSIPHON COIL LOOP HEAT EXCHANGERS.

GURSARAN DAS. MATHUR

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A SIMULATION PROGRAM FOR MULTIPLE TUBE-ROW
TWO-PHASE THERMOSIPHON COIL LOOP HEAT EXCHANGERS

A Dissertation
Submitted to the Faculty of Graduate Studies through
the Department of Mechanical Engineering in Partial
Fulfillment of the Requirements for the Degree of
Doctor of Philosophy at the
University of Windsor

By

Mathur, Gursaran Das

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1986


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ABSTRACT

A computer algorithm has been developed to simulate two-phase thermosiphon coil loop heat exchanger systems in which each coil consists of parallel rows of tubes running between common headers. The coils may be oriented in any arbitrary fashion relative to the gravitational field, subject to the limitation that the condensate must be able to return by gravity to the evaporator and may be subject to different external flow and thermal conditions. The developed program requires the operator to specify the source and sink fluid approach temperature and face velocities; various source, sink, and working fluid properties; the source and sink fluid to coil inner surface heat transfer coefficients; and the size and orientation of the various system components.

The simulated performance was compared to each of the following three very different experimental systems:

1. The evaporator and condenser coils used by Raza (12) each consisted of three coplanar independently water jacketed tubes. The plane of each coil could be inclined from the vertical to simulate the operating conditions of an inclined, one-row coil. They could also be rotated in a vertical plane to simulate the effect of unequal charges in the tube rows of an inclined three-row evaporator coil. The effect

of static charge, temperature difference, unequal heating of each evaporator tube row and unequal charge in each evaporator tube row was studied on the performance of the system. The working fluid was R-11.

2. An air-to-air heat exchanger system which consisted of four thermosiphon loops arranged in a counter flow configuration in the air ducts. Each of the coils in each loop consisted of two rows of tubes with wavy plate fins. In this system, studied by Stauder (17) and by Kosnik & Bertoni (15) the evaporator tubes were vertical and the condenser tubes were inclined at 45 degrees. The evaporator and the condenser coils had the same face area and were installed in warm and cold ducts having the same mass flow rates. The effect on the performance of the system of different air mass flow rates, various static charges and of sequentially increasing and decreasing the overall temperature difference was studied. The working fluid used in the system was R-11.

3. Bergevin et al. (23) studied a system using R-113 in which a single tube evaporator was subject to a constant heat flux boundary condition. The evaporator tube was inclined at an angle of 60 degrees from the horizontal. The condenser was water jacketed. The working fluid was in a subcooled state at the entry to the evaporator tube. The effect of the heat flux was studied on the performance of the system.

For these systems, the results predicted by the computer simulation program agreed very well in most cases with the experimentally determined values of the overall heat transfer rate, the average heat transfer coefficient for each tube, the loop conductance and the wall superheat for either the heating or the cooling mode. A major limitation at present is the need for a correlation to determine the conditions under which boiling will initiate during the system startup.

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To my parents

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NOMENCLATURE

A	- Surface area (sq m)
A_x	- Cross-sectional area (m^2)
Bo	- Boiling number ($q/G h_{fg}$)
C_p	- Specific heat (kJ/kg K)
Csf	- A constant on Rohsenow's pool boiling correlation
D	- Tube diameter (m)
E	- Effectiveness
f	- Friction factor (Btu/hr/sq.ft/F)
F	- Fouling factor
Fl	- Froude number
G	- Mass flux ($Kg/m^2/s$)
Gn	- Graetz number ($Re Pr D/z$)
g	- Gravitational constant
h	- Heat transfer coefficient ($W/m^2/K$)
h_{fg}	- Latent heat of vaporization (kJ/kg)
J	- Superficial velocity (m/s)
K	- Thermal conductivity ($W/m \cdot K$) ; Velocity head factor
L	- Tube length (m)
M	- Entrance effect factor, Mach number , Number of tubes per row
\dot{m}	- Mass flow rate (kg/sec)
N	- Number of tube rows
Nu	- Nusselt number ($h D/K$)
p	- Pressure (N/m^2)
Pe	- Peclet number ($Re Pr$)
Pr	- Prandtl number ($\mu C_p / K$)
Q	- Heat flow rate (W)
q	- Heat flux (W/m^2)

- R - Thermal resistance ($\text{m}^2 \text{K/W}$)
- r - A constant in Rohsenow's pool boiling correlation
- Re - Reynolds number ($\rho V D / \mu$)
- R - Liquid holdup, liquid fraction
- S - Slip ratio, given by equation 3.7
- St - Stanton number ($h/\rho C V$)
- T - Temperature (°C)
- U - Velocity (m/s), Conductance ($\text{W/m}^2 \text{K}$)
- V - Velocity (m/s)
- We - Weber number
- x - Quality
- X - Lockhart-Martinelli parameter
- xc - Critical quality
- z - Distance along the tube axis (m)
- z^* - Dimensionless parameter $(2z/D)/\text{Pe}$

GREEK SYMBOLS:

- σ - Surface tension (N/m)
- ρ - Density (kg/m^3)
- μ - Dynamic viscosity (kg/m s)
- δ - Superheated boundary layer thickness (m)
- ξ - Eddy diffusivity (m^2/s)
- α - Void fraction
- ϕ - Parameter defined in section 3.2.6
- β - Inclination angle of evaporator from the horizontal, degrees
- γ - Inclination angle of condenser from the horizontal, degrees
- λ - Latent heat of vaporization (kJ/kg)

SUBSCRIPT:

- a - acceleration, approach.
- av - average
- b - bulk, boiling, bottom
- c - condenser, cold, critical conditions, condensation
- ce - condenser external
- ci - condenser internal
- co - condenser outside
- cw - condenser wall
- cr - critical conditions
- d - based on diameter, downstream
- e - evaporator, exit, element
- ee - evaporator external
- ei - evaporator inside
- eo - evaporator outside
- ew - evaporator wall
- e3 - effective h in the dryout region, defined by equation 3.61
- eq - equivalent
- f - fluid, frictional
- g - vapour
- gs - vapour superficial
- h - hydrostatic, hot
- he - heat exchanger
- ib - incipient boiling condition
- l - liquid
- lp - liquid phase
- ls - liquid superficial

m - mean, metal
pb - pool boiling
r - recirculation
s - surface condition
sat - saturation condition
scb - sub cooled boiling
t - total, transport
tp - two-phase
v - vapour
vp - vapour phase
w - wall
z - point value
2e - effective h in subcooled boiling region, defined by equation 3.44

CHAPTER I

INTRODUCTION

1.1 GENERAL

Until several years ago, energy sources had been both inexpensive and abundant. Today's situation, however, is drastically different. Fuel costs have risen dramatically and are continuing on the rise. The increase in the cost of the energy has brought about a desire on the part of the consumers to use energy as efficiently as possible. According to the projections made by the Federal Government, 53 % of Canada's total energy consumption in the 1980's will be rejected in the form of waste. Because this waste will generally be in the form of thermal energy, heat reclaiming equipment and systems have become very important and thus are receiving increased attention.

Heat reclamation can be divided into three major categories: (1) heat which is recovered from process flue gases and then recycled directly into the process for reuse; (2) recovered waste heat from process exhausts which is used to preheat make up air during heating seasons; and (3) the use of comfort exhaust to preheat ventilation fresh air during the heating season, and to precool it during the summer months.

Several different classes of air-to-air heat exchangers are currently used for waste heat recovery. If the two air streams are adjacent to one another, compact heat exchangers, rotary regenerative heat exchangers or heat pipe heat exchangers may be effectively used. If the two air streams are not close to each other then liquid coupled forced circulation run around loops may be utilized. The latter system is a highly flexible one but requires a pump to circulate the working fluid.

The waste heat recovery system discussed in this thesis has the features of a combination of heat pipe heat exchangers and liquid coupled forced circulation run around loop systems. The resulting system illustrated in figure 1.1 has been termed a two-phase thermosiphon loop to characterize its shape and to distinguish this system from gravity heat pipes and counter-flowing thermosiphon systems. These systems may also be used to cool electronic and electrical equipment, and to heat hot water in solar systems.

Thermosiphon loop heat exchangers offer a number of significant advantages over other types of heat exchangers. These advantages can be summarized as :

1. Require no external power to circulate the working fluid - the temperature difference between the two air streams acts as the activating mechanism for the fluid circulation.

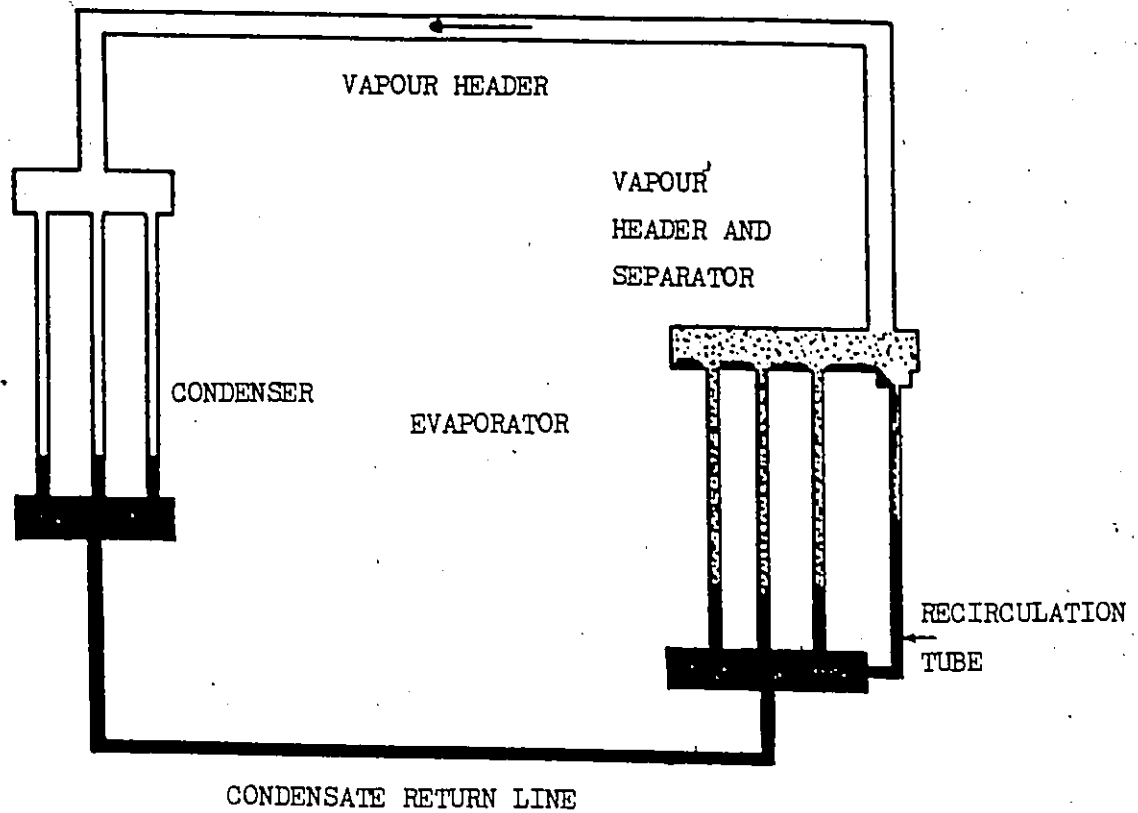


FIG. 1.1 TWO-PHASE THERMOSIPHON LOOP HEAT EXCHANGER

2. No moving parts in the system - only the working fluid circulates in the loop.
3. Long life - since there are no moving parts and the system is sealed, the system can operate indefinitely within the design temperature range.
4. No cross contamination - the two air streams are separated and the system is sealed so there is no problem of contamination of the streams; only heat is transferred from one air stream to another.
5. Reversibility - heat can be transferred in either direction.
6. Less weight - two-phase thermosiphon systems are only partially filled with liquid and hence are less heavy than single-phase circulation heat exchangers.

1.2 PRINCIPLE OF OPERATION OF TWO-PHASE THERMOSIPHON LOOP HEAT EXCHANGERS

A thermosiphon loop heat exchanger transfers energy from a heat source to a heat sink by the natural circulation of an intermediate working fluid between the two regions. A two-phase loop utilizes the difference in the saturation pressure between the evaporator and the condenser to cause the vapour to flow from the hot evaporator coil

through a separator to the cold condenser coil. The geometric configuration must be such that the condensate can flow back to the bottom of the evaporator by gravity.

In the two-phase thermosiphon loop heat exchanger system shown in figure 1.1 the liquid removed in the separator is recirculated to the evaporator coil liquid header. The presence of a separator allows more working fluid to be present in the evaporator, and hence helps to suppress dryout; it also ensures that little liquid is carried over to the condenser and hence improves the condensation film coefficient in the condenser.

If liquid working fluid is present in both coils at all times as shown in figure 1.2(a) then either coil can act as an evaporator, and the system is said to be bidirectional (i.e. it can transfer heat in either direction). For a system to operate in a bidirectional mode, recirculation tubes must be provided in both the heat exchanger coils. If the liquid working fluid drains completely from one of the coils when the system is inoperative as shown in figure 1.2(b), then only the flooded coil can act as an evaporator and heat can be transferred in only one direction. Such a system is said to be unidirectional.

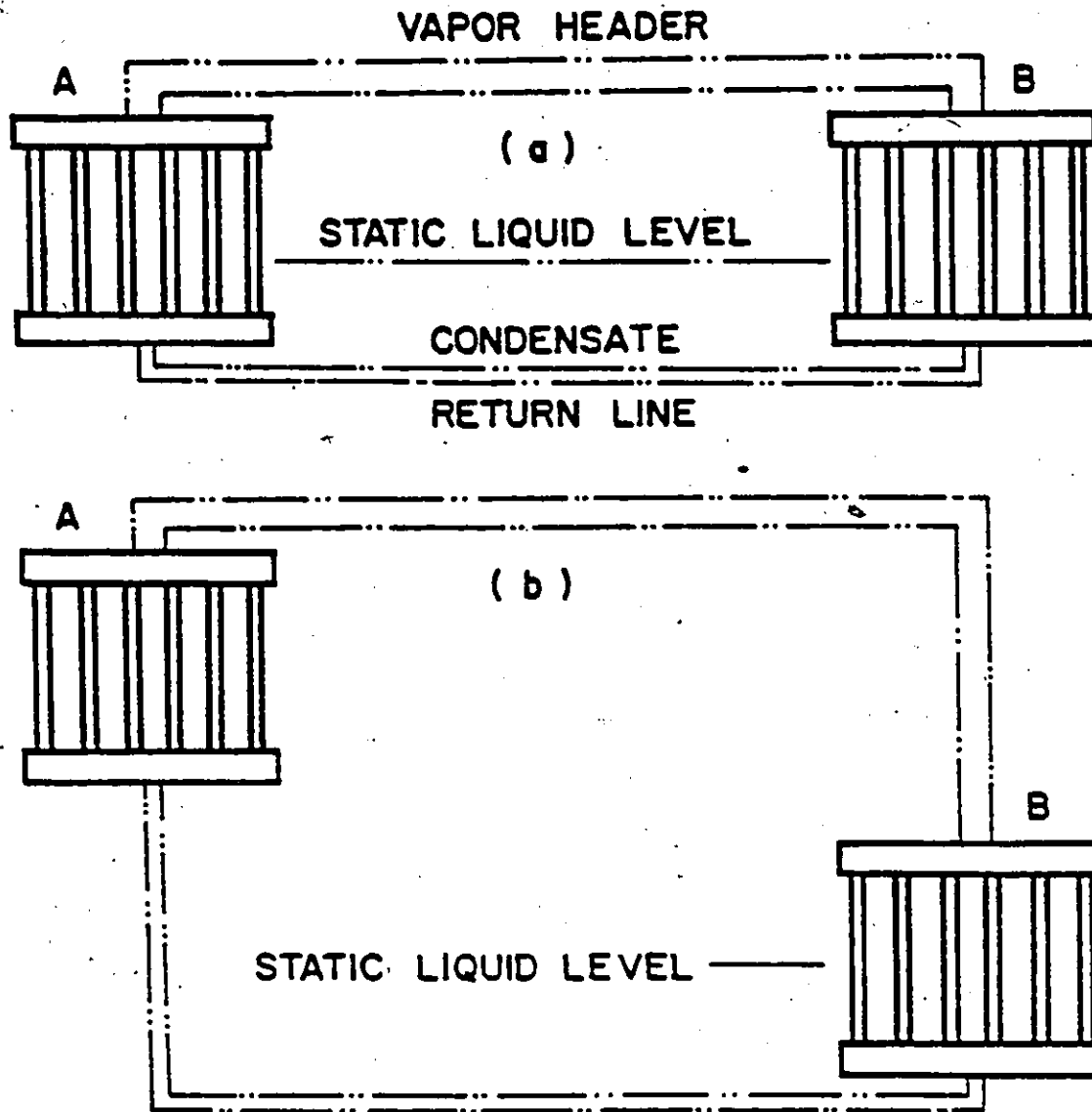


Figure 1-2 SCHEMATIC ELEVATION VIEW OF COIL LOOP THERMOSIPHON HEAT EXCHANGERS

- (a) Bidirectional configuration. Loop can transfer heat in either direction, A to B or B to A.
- (b) Unidirectional configuration. Loop can transfer heat only from B to A.

1.3 FLOW PATTERNS & HEAT TRANSFER REGIONS ENCOUNTERED IN TWO-PHASE THERMOSIPHON LOOP HEAT EXCHANGERS

A two-phase mixture flowing in a pipe can exhibit several flow geometries, such as bubbles, slugs, and films. Large variations in heat transfer occur when the flow geometry changes from one to another e.g. bubbly to slug and slug to annular. The flow pattern and the heat transfer are interrelated and they affect the hydrodynamics of the flow. Figures 1.3 & 1.4 show the various flow patterns and the heat transfer regions encountered in a vertical evaporator tube. The fluid reaches the heat exchanger portion of the evaporator as a liquid in a subcooled state. The liquid enters the evaporator tube from the header and in the entrance region both the thermal and the momentum boundary layers develop as the working fluid flows up the tube. Initially, both the bulk fluid temperature and the wall temperature remain below that necessary for nucleation and the heat transfer process is single-phase convective heat transfer to the liquid phase (region A). At some point along the tube, nucleation takes place in the presence of subcooled liquid (region B) which is known as subcooled nucleate boiling. The bubbles produced in the superheated wall region condense as they move away into the subcooled liquid. In this region, the wall temperature is higher than the saturation temperature and the bulk fluid temperature is still below the saturation temperature. The transition between region B & C is the point at which the bulk fluid temperature

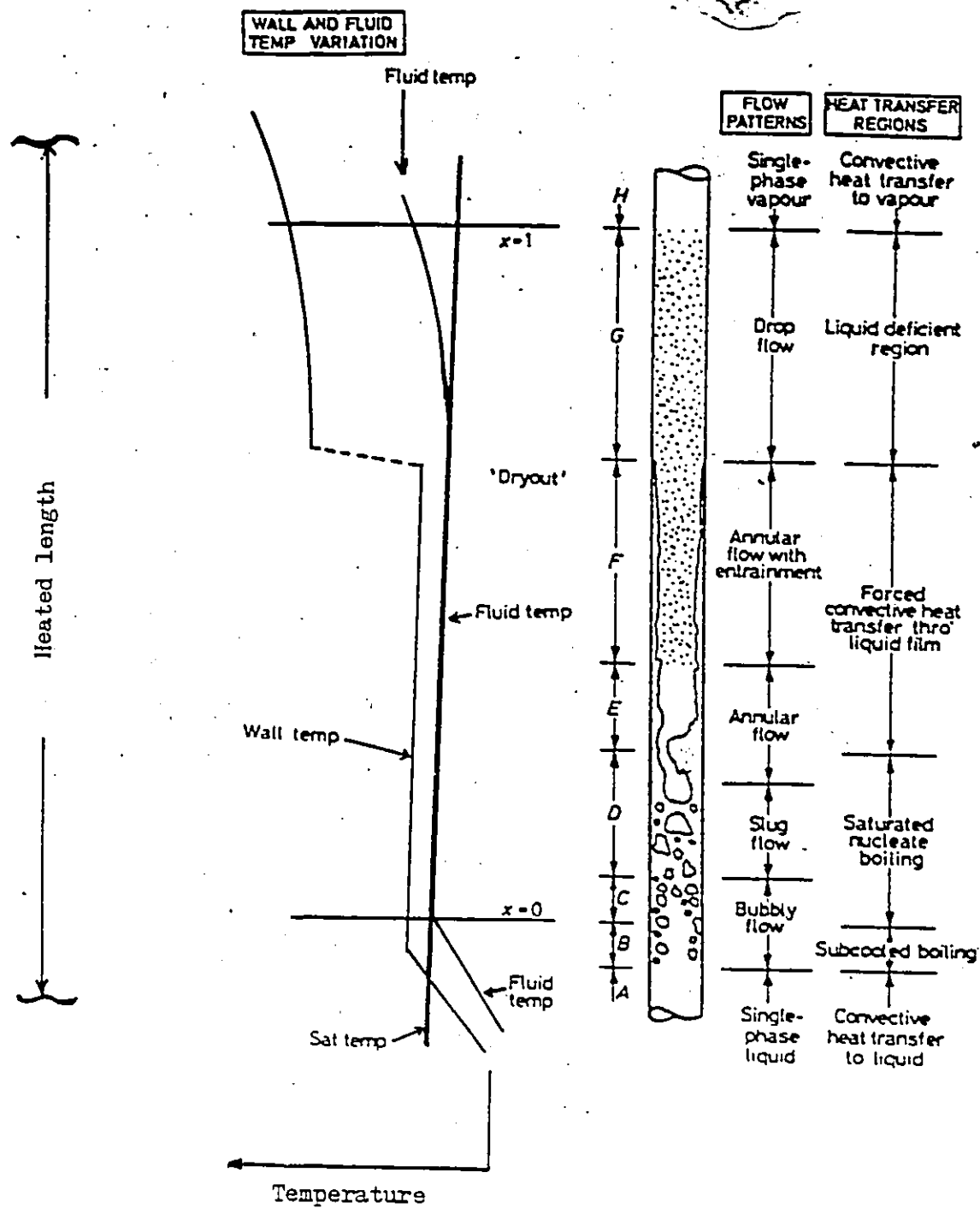


FIG. 1.3 REGIONS OF HEAT TRANSFER IN CONVECTIVE BOILING

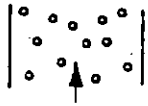
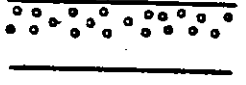

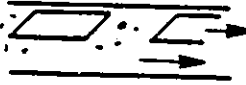

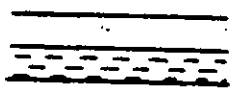

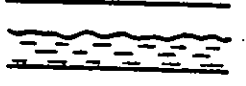



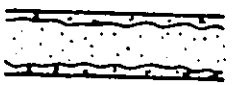
	TUBES	
	VERTICAL FLOW	HORIZONTAL FLOW
BUBBLE FLOW		
PLUG FLOW		
STRATIFIED FLOW		
WAVY FLOW		
SLUG FLOW		
ANNULAR FLOW		

FIG. 1.4 FLOW PATTERNS IN VERTICAL & HORIZONTAL TUBES

equals the saturation temperature and saturated nucleate boiling (also known as bulk boiling) starts. As bubbles accumulate (region C) and coalesce the flow pattern changes from bubbly to slug flow (region D) and then to annular flow (regions E & F). As the quality and the vapour fraction increase, the resulting higher vapour velocities and higher two-phase heat transfer coefficients cause nucleate boiling to be replaced by surface vapourization. At some critical value of the quality dryout starts to occur on the tube wall and causes a significant rise in the wall temperature, an increase which could lead to tube failure in a constant heat flux boundary condition (burnout). Region G, between the dryout point and the transition to single phase vapour is termed the liquid deficient region. In this region, the vapour temperature increases above that of the liquid droplets which are at the saturation state. The heat transfer coefficients for region H are the lowest of all of the heat transfer regions.

The vapour generated in the evaporator reach the condenser either as saturated vapour or as wet mixture through a vapour header. If there is a heat loss from the vapour header, some of the vapour will condense and the condensate is removed in a separator and is circulated back to the evaporator liquid header. If the pressure drop in the vapour header is substantial, the vapour would be in a

superheated state at the inlet to the condenser.

In the condenser the heat transfer can take place by either dropwise condensation or by film condensation. If the condenser is partially flooded, the mode of heat transfer in the flooded region would be by single phase liquid forced convection. The condensed liquid flows by gravity to the evaporator bottom through a liquid header and thus the cycle is completed.

1.4 PROBLEM IDENTIFICATION

The experimental studies of Hwang (1), Diciccio (2) & Sampath (3) and the simulation study of Ali (4) carried out at the University of Windsor established the basic performance characteristics of single tube coil loops and of multiple tube evaporator and condenser coil loops where all of the evaporator and the condenser coil tubes were subjected to the same source and sink temperatures and where the charge present in all the evaporator tubes were the same.

For many applications, the boundary conditions of the evaporator tubes are not the same and there may be different amounts of charge present in the evaporator tube rows. The present work deals with the development of a program which can be used to simulate the performance of two-phase thermosiphon loops in which the evaporator

coil consists of straight tubes running between a common liquid and a common vapour header and where the tubes may be inclined from the vertical and where each tube row may be subject to different thermal boundary conditions. The program can also be used to calculate the effectiveness of commercial finned two-phase thermosiphon loop heat exchanger systems.

The program also takes into consideration the hysteresis effect for boiling of fluids in forced convection vertical flow. This hysteresis occurs due to the fact that a much greater wall superheat is required to initiate nucleate boiling than to maintain it. The boiling characteristics of a surface at low wall superheats is dependent on its past history, i.e. whether the surface nucleation sites are being initiated or quenched. Different correlations must be used for the quenching mode, where a high temperature difference between the surface and the working fluid has already established all the nucleation sites than for the site initiation mode where each successive run is for a greater overall temperature differences.

Figure 1.5(a) shows a thermosiphon loop where the tube rows in the evaporator coil are subjected to unequal heating due to the cooling of the exterior gas as it passes through the coil. Figure 1.5(b) shows an inclined evaporator coil in which each tube row has an unequal charge. Figure 1.6 shows a commercial finned thermosiphon loop heat exchanger.

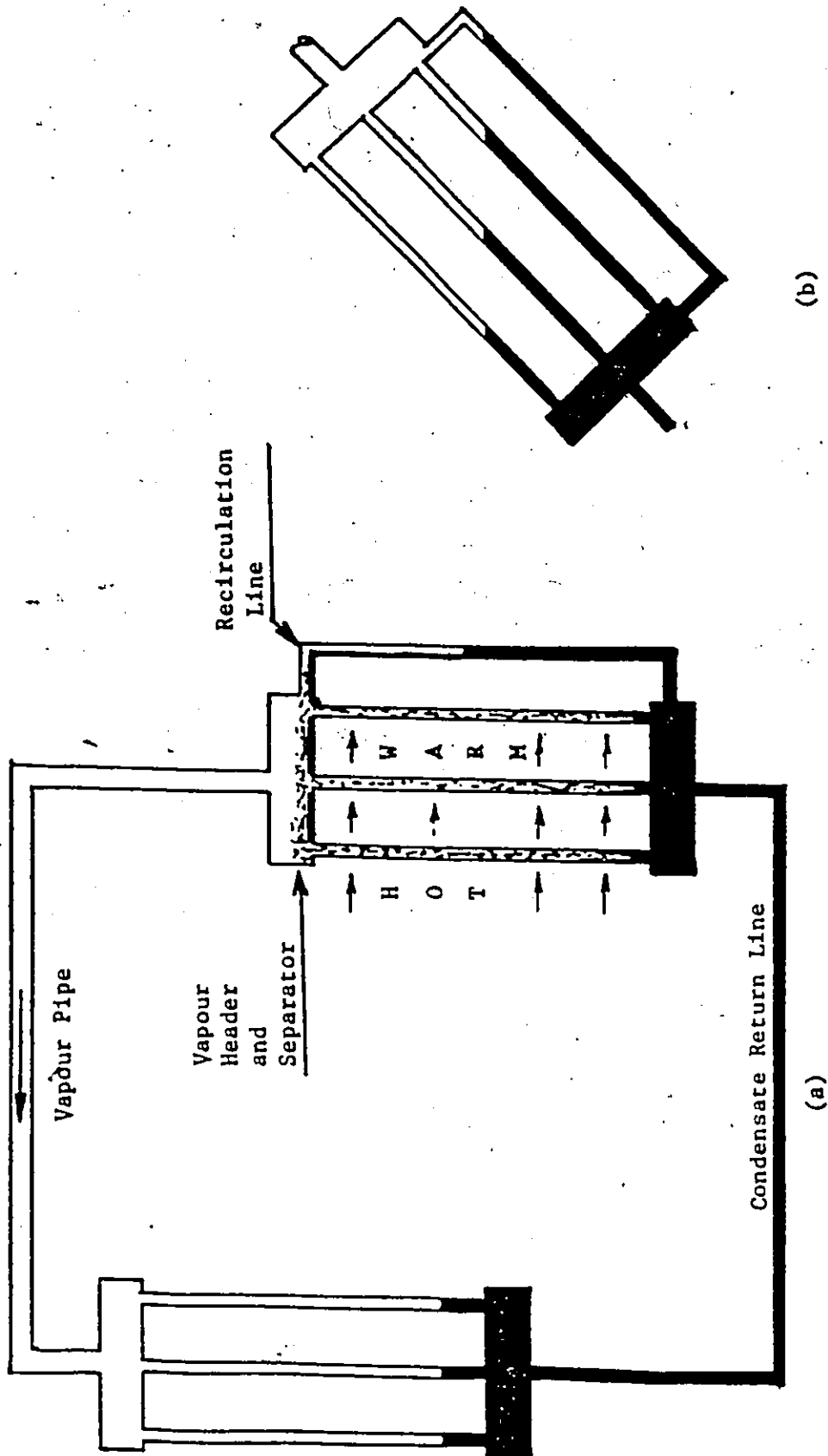


FIG. 1.5 Schematic elevation of coil loop thermosiphon showing evaporation with
 (a) unequal tube heating and (b) unequal tube charge

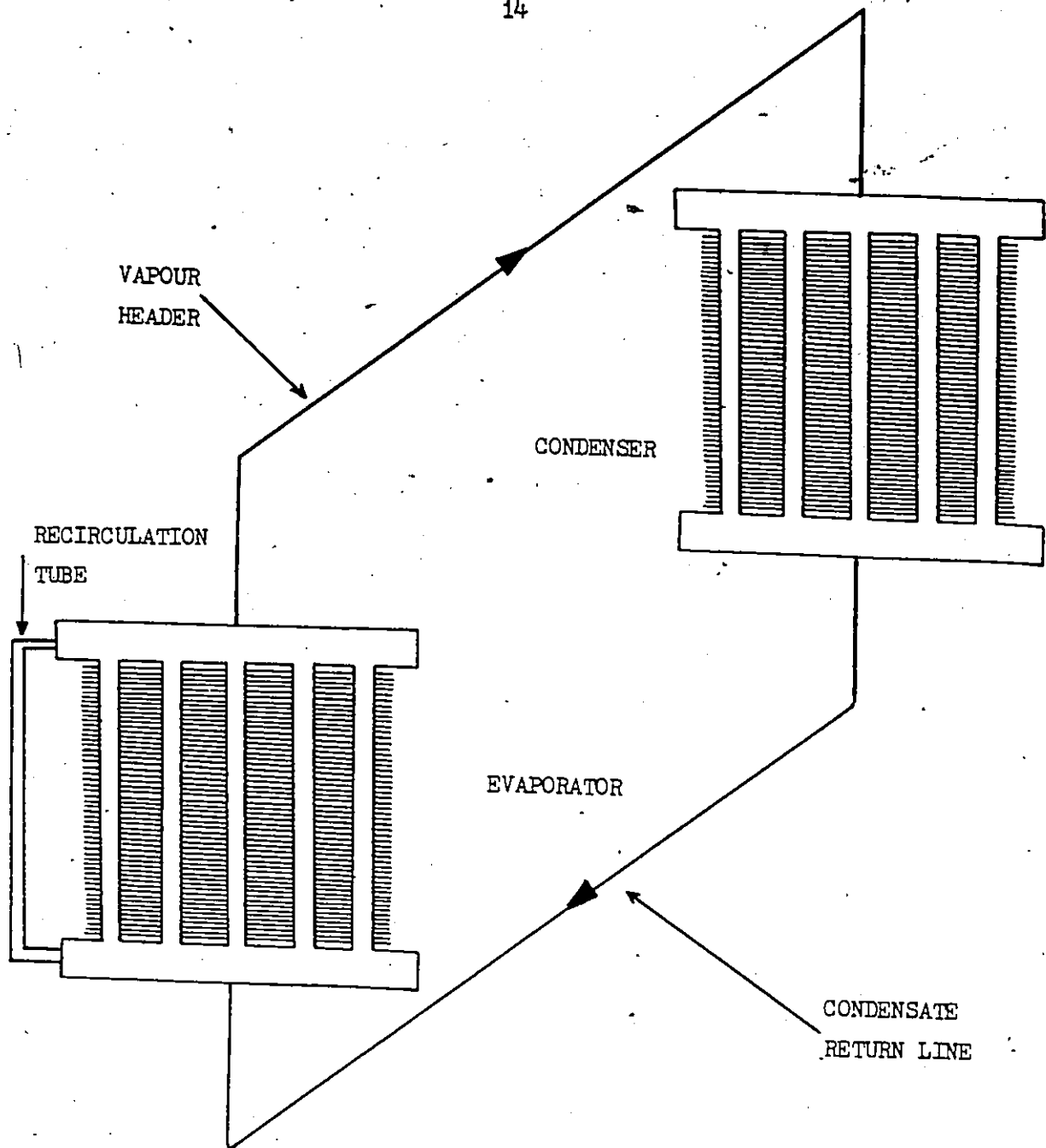


FIG 1.6 A COMMERCIAL FINNED TWO-PHASE THERMOSIPHON LOOP HEAT EXCHANGER

CHAPTER II

LITERATURE SURVEY

2.1 INTRODUCTION

A closed loop thermosiphon is a commonly employed device for heat transfer in various industrial applications. Extensive work has been done on single phase natural circulation loops, where the working fluid does not change phase. For these systems temperature differences around the loop produce the buoyancy effect which causes the working fluid to circulate. On the other hand, in a two-phase thermosiphon loop the vapour generated in the evaporator flow to the condenser due to the difference in the saturation pressure between the evaporator and the condenser. Two-phase thermosiphons are getting a lot of attention these days since they have lot of advantages over the conventional forced circulation heat exchanger systems.

The object of the present work deals with the utilization of two-phase thermosiphon loop heat exchanger systems for waste heat recovery. To the best of author's knowledge the work on this topic has mainly been done at the University of Windsor.

In the following section a literature survey is done on thermosiphons.

2.2 LITERATURE SURVEY ON THERMOSIPHONS

In the following section a literature survey is presented on two-phase thermosiphon loops.

Hwang (1), Diciccio (2), Ali (4), McDonald et al. (5), Ali & McDonald (6) and McDonald & Ali (7) studied the performance of two-phase thermosiphon waste heat recovery systems. Their studies indicated that the optimum performance may be expected from a thermosiphon loop when the evaporator and condenser tubes are oriented in such a way that the condenser is nowhere flooded, no dryout occurs in the evaporator and negligible liquid is carried into the condenser. Sampath (3), McDonald et al. (8) and McDonald & Sampath (9) investigated multiple tube evaporators and condensers with recirculation for both bidirectional & unidirectional two-phase coil loop thermosiphon heat exchangers both experimentally and by computer simulation.. Their results showed that single tube two-phase thermosiphon loops may be designed to provide peak performance over a modest range of operating temperature differences. Unidirectional loops offer a higher performance than the bidirectional loops but are more sensitive to changes in the imposed temperature differences. The

condenser is not flooded in the case of a unidirectional loop which results in a better performance. A two-phase thermosiphon heat exchanger without and with a recirculation tube is shown in figure 2.1. The final reports of these studies were submitted to ASHRAE by McDonald (10,11). The experimental rigs of Hwang, Diciccio & Sampath are shown in figures 2.2 & 2.3. Raza (12) used Sampath's apparatus to investigate the effect of unequal heating and unequal charge conditions for the evaporator tubes, as shown in figure 1.5. Each tube had a separate water jacket to produce the non uniform heating of the evaporator tubes. Raza & McDonald (13), McDonald & Raza (14) found that, for the conditions tested, the overall heat transfer rate and the overall average evaporator heat transfer coefficients were insensitive to these effects for static charges between 40% and 90%. They also noted that the performance of each tube row was changed significantly. Kosnik & Bertoni (15) and McDonald et al. (16) investigated the performance of a finned commercial thermosiphon loop heat exchanger with R-11 as the working fluid. The condenser and the evaporator coils each consisted of 8 rows of tubes. The evaporator tubes were vertical and the condenser tubes were inclined at 45 deg. from the vertical. The test apparatus could be arranged to study 8, 4, 2 or 1 loop system configurations. They used 4 loops for their studies comprising of 2 rows of tubes per loop. The static charge, overall temperature difference and the coil face velocity were varied and their effect on effectiveness was studied. They found that the

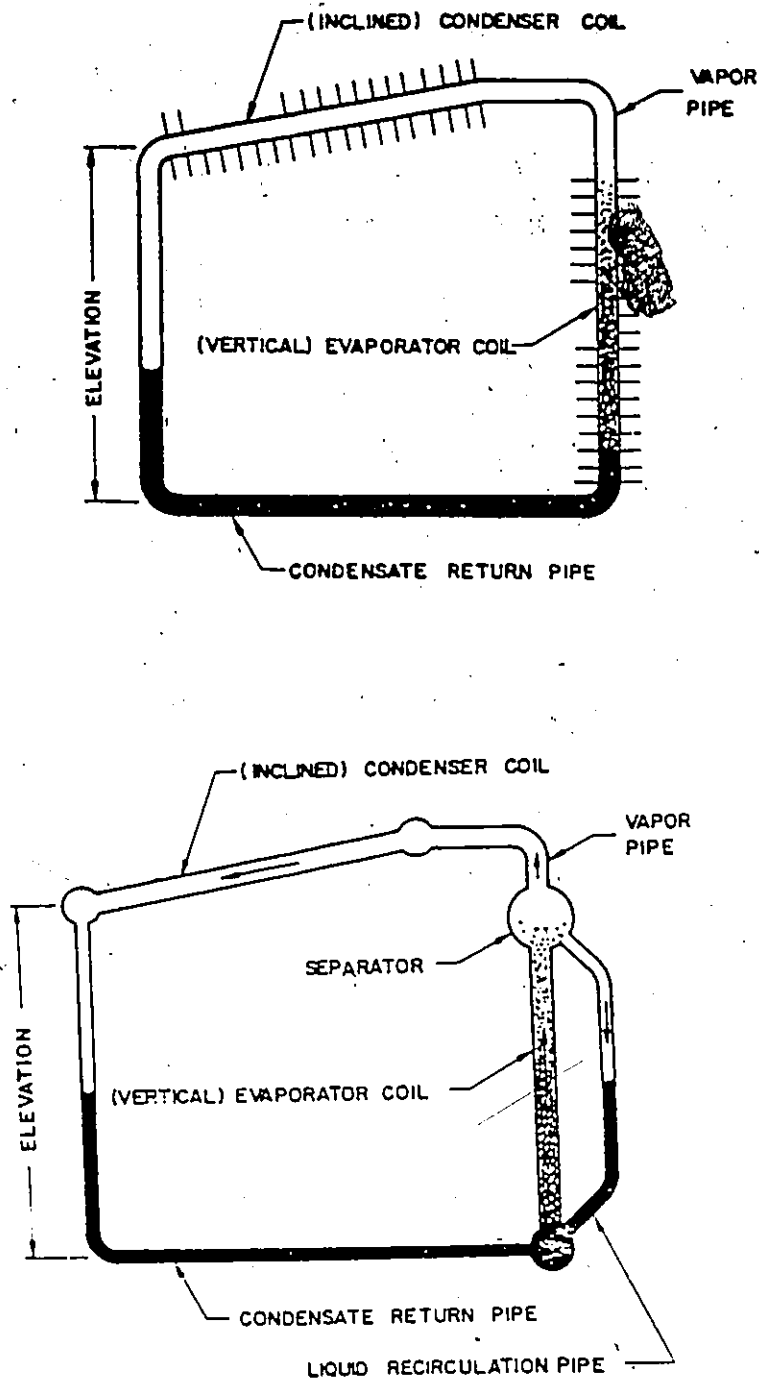


FIG. 2.1 THERMOSIPHON LOOP HEAT EXCHANGERS WITHOUT & WITH A
RECIRCULATION TUBE

1 and 2: Hot water (source) inlet and outlet, respectively.
 3 and 4: Cold water (sink) inlet and outlet, respectively.

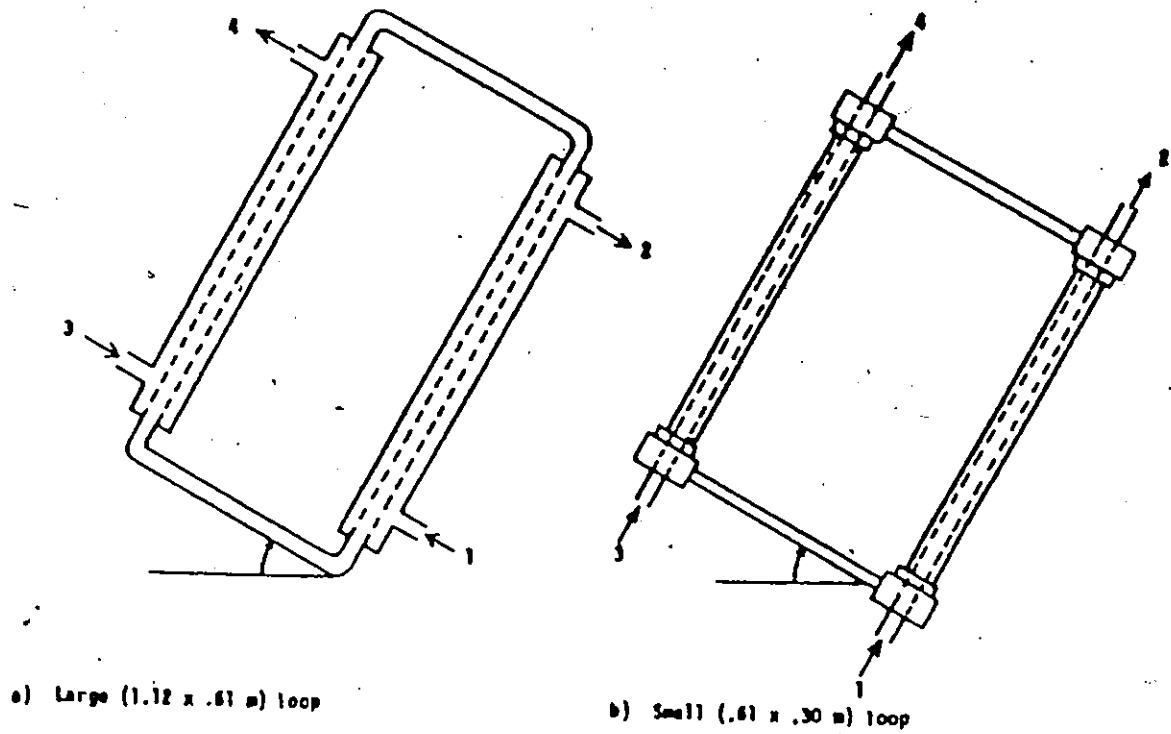


FIG 2.2 EXPERIMENTAL THERMOSIPHON LOOPS OF (A) HWANG (1) &
 (B) DICICCIO (2)

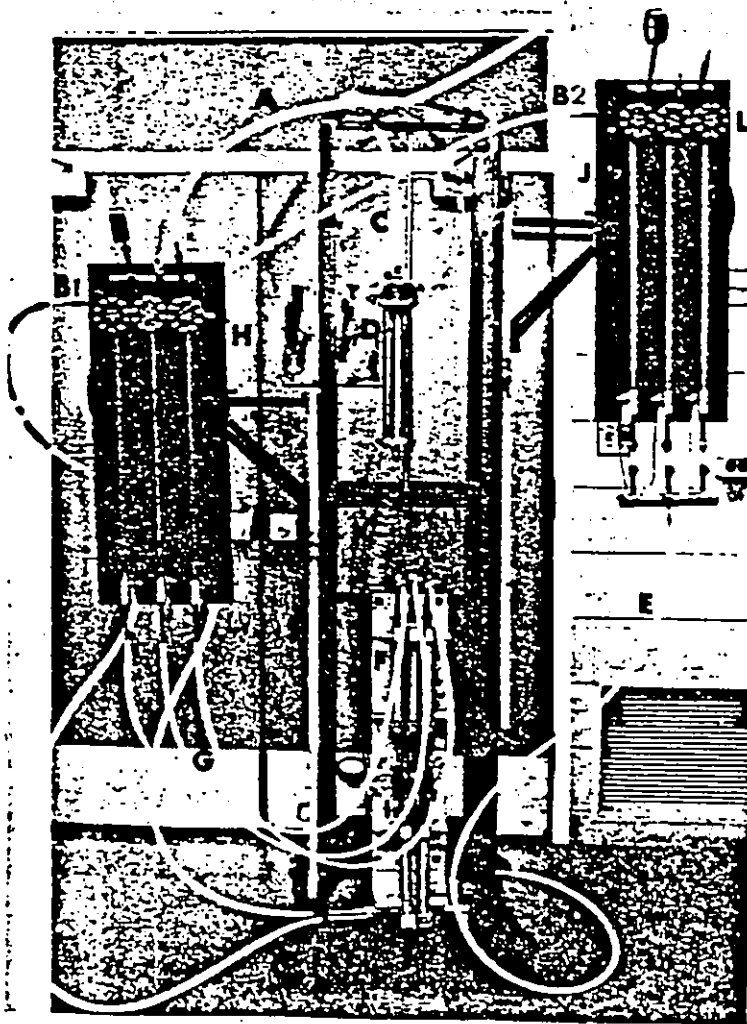


Fig. Uninsulated multiple tube evaporator and condenser apparatus

- A Vapor header
- B1, B2 Evaporator and condenser liquid bypass lines, respectively
- C Vapor line to reservoir
- D Working fluid reservoir
- E Condensate return line
- F Bypass, condensate and evaporator tube flow meters
- G Liquid return lines to the evaporator tubes
- H Evaporator assembly
- J Condenser assembly
- K Liquid line to reservoir (with valve at connection to the loop)
- L Tube wall thermocouple terminals
- X Reservoir vapor line valve

FIG. 2.3 EXPERIMENTAL THERMOSIPHON LOOP OF SAMPATH (REF. 3)

effectiveness of the system was insensitive to the quantity of charge in the evaporator, except when dryout occurred; it increased with the hot to cold air duct temperature difference and decreased as the airflow rates increased. Their 4 loop system operated best between static charge of 60 to 90 % with optimum performance achieved near 80 %. Recently, Stauder (17) made an exhaustive experimental study on the experimental rig used by Kosnik & Bertoni and found that a hysteresis effect occurs in such systems which can severely reduce the performance of two-phase thermosiphon loop heat exchangers. He found that wall superheats of approximately 12-14 C were required to initiate the boiling process for the R-11/copper interface used. Figure 2.4a and 2.4b shows the schematics of the experimental rig used by Kosnik & Bertoni and Stauder. Figure 2.5 shows at a glance, the research work done at the University of Windsor on two-phase thermosiphon loop heat exchangers.

Mendler et al. (18) investigated experimentally and by theoretical model the performance of natural circulation & forced circulation loops with water at 800 to 2000 psia under nonboiling, local boiling and bulk boiling conditions. They used single rectangular channel test sections of three different dimensions and without a liquid-vapour separator. Heat fluxes ranged from 50,000 Btu/hr-sq.ft to burnout with inlet subcoolings of 20, 20 and 100 F. The results showed that single and two-phase pressure drop, burnout heat flux, and riser density measured under natural circulation were

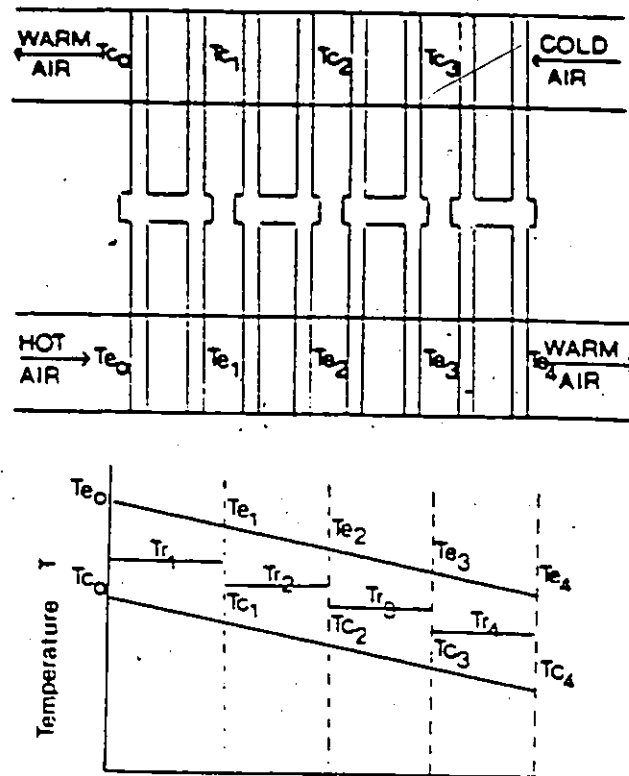


FIG. 2.4a

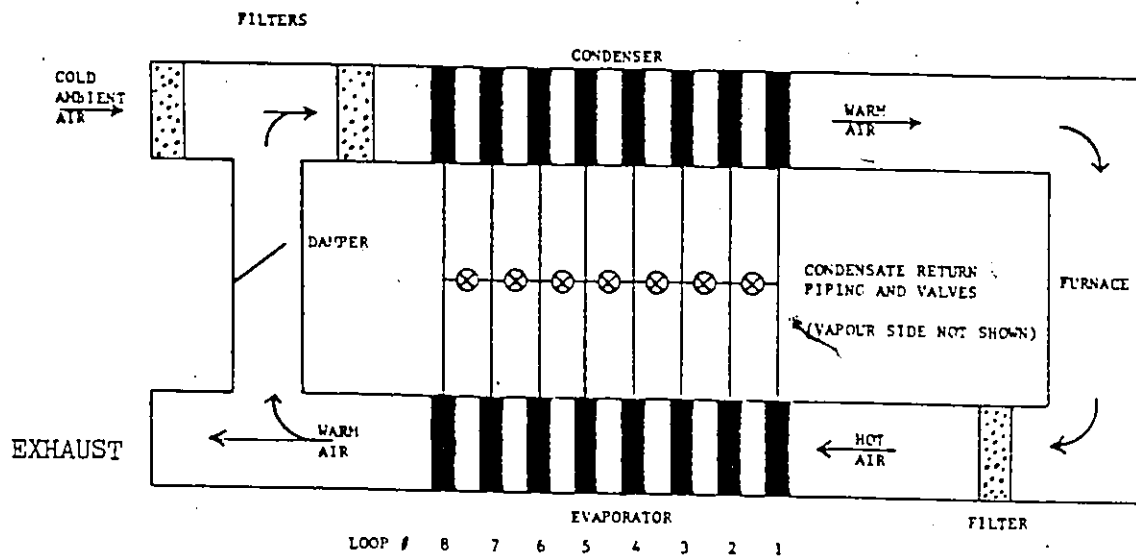


FIG. 2.4b SCHEMATIC REPRESENTATION OF THE THERMOSIPHON TEST FACILITY

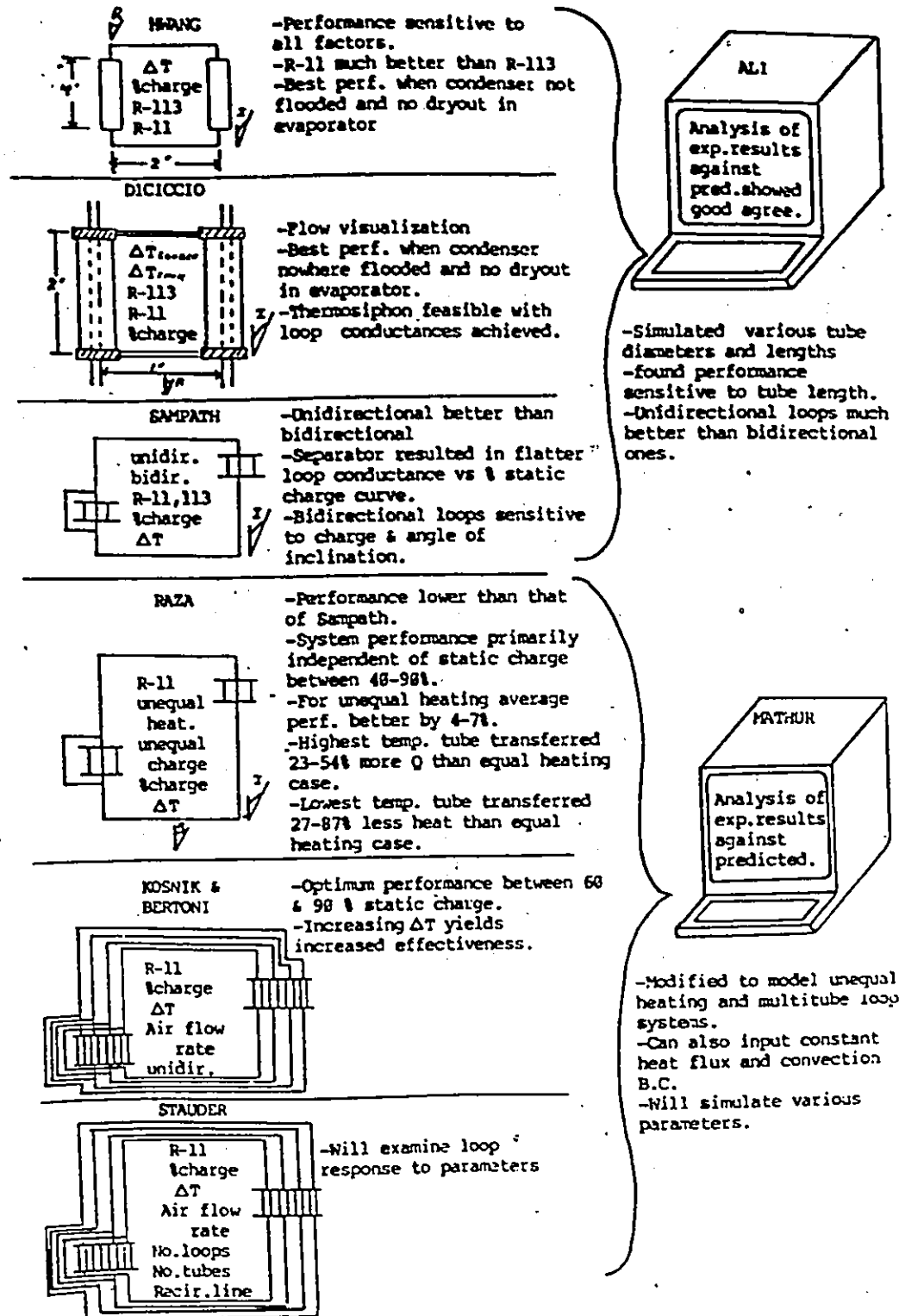


FIG. 2.5 UNIVERSITY OF WINDSOR THERMOSIPHON LOOP RESEARCH HISTORY

no different from those measured with forced circulation at the same thermal and fluid flow conditions. For the loop studied, natural circulation flow rates were predicted within 10%.

Bandy et al. (19) investigated maximization of heat rejection rate to a natural circulation boiling channel by experiments and theoretical model. Water and Freon-113 were used as the working fluid. Variable area and constant area evaporator tubes were used for their studies for non uniform and uniform heat fluxes. The authors found out that the burnout power could be increased significantly by using a variable area heated section. The model was able to predict the theoretical flowrates within 23% and burnout powers within 16%.

Soin et al. (20) investigated the performance of flat plate solar collector with fluid undergoing phase change for natural circulation. They used acetone and petroleum ether (40-60 C) as the working fluids for their experiments. They found that the collector efficiency increases linearly with liquid level measured in the recirculation tube.

Tanger et al. (21) studied heat transfer to sulfur hexafluoride near the thermodynamic critical region in a natural circulation loop without a liquid-vapour separator. Local and average heat transfer coefficients along the test section were calculated from the experimental data. They reported that maximum heat transfer

coefficients were obtained at pressures slightly above the critical state.

Cheng & Rovang (22) investigated the steady state operation and heat transfer characteristics of a two-phase closed thermosiphon system for low grade energy recovery. The working fluid used was water. The parameters which were varied are - the liquid charge level, liquid source and sink temperatures 55-95 C, 14-19 C and their flow rates 30-35 ml/s, 10-70 ml/s, and with or without a liquid-vapour separator. The evaporator pressure was held constant at 34 kPa. The evaporator and condenser coils were fabricated from spiral copper tubes. The flow rates of the working fluid were found to be in the range of 14-36 ml/s and the heat recovered was in the range of 1000-4600 W.

Bergevin et al. (23) investigated the performance of a phase-change solar collector. The solar collector channel was simulated by means of a uniformly heated glass pipe which was supplied with a liquid below its saturation temperature. The channel was inclined at an angle of 60 degrees from the horizontal with a diameter and length of 5 mm and 143 cm respectively. The condenser was water jacketed and was 400 cm long. The water inlet temperature was 25 C with a flow rate of 50-69 gram/minute. The heat flux was varied between 600 to approx. 3400 W/sq.m and the working fluid used was Freon-113. Their

experimental system is shown in figure-2.6. The results of their study is shown on figure 2.7, which is a plot of the average heat transfer coefficient for two-phase and single-phase region as a function of the heat flux. They correlated their local experimental data for both the single- and two-phase regions with the following equations.

$$h_l = q [T_m - T_l]^{-1} \quad \text{--- (1)}$$

$$h_{tp} = 0.023 k_l D^{-1} (\rho_l / \mu_g)^{0.25} [Re(1-x)]^{0.8} Pr^{0.33} \quad \text{--- (2)}$$

References 24 to 43 deal with the studies of two-phase heat pipes operating in a thermosiphon mode. The basic mechanism of operation of a two-phase heat pipe is entirely different than that of a two-phase loop and hence the detailed discussion of these references is not directly relevant to this thesis and will not be presented. References 44 to 63 deal with the studies of single-phase thermosiphon loops. Since the nature of operation of these loops is also different than that of two-phase thermosiphon loops a detailed discussion of these references is omitted. These papers (from 24 to 63) have been included in the list of references in order to provide a more complete literature survey, which the reader might find useful.

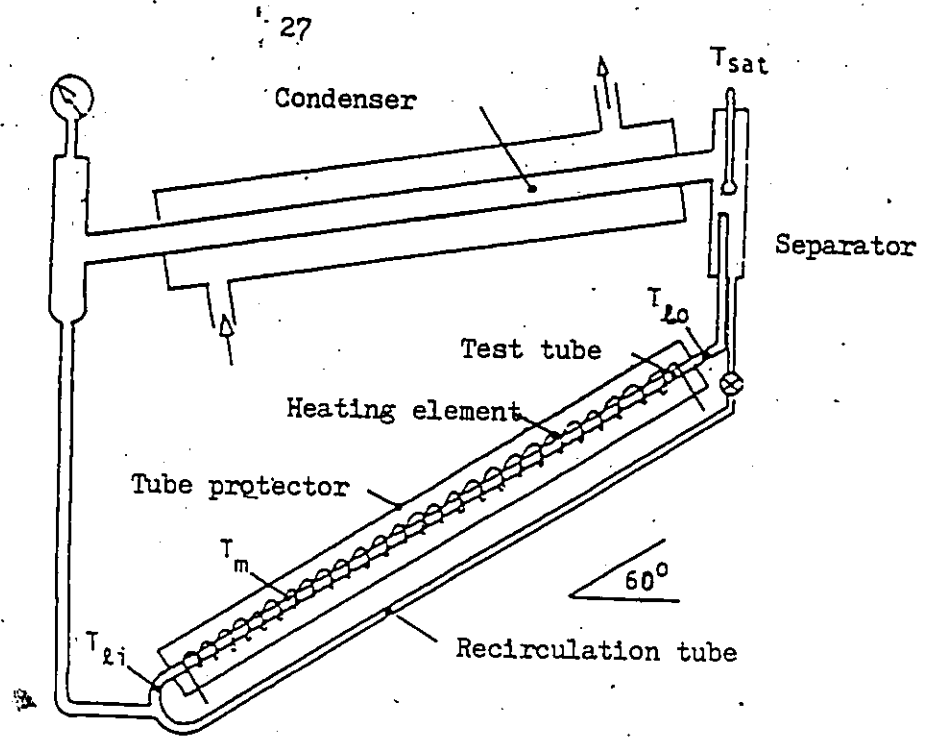


Figure 2.6 Experimental set up of reference 23

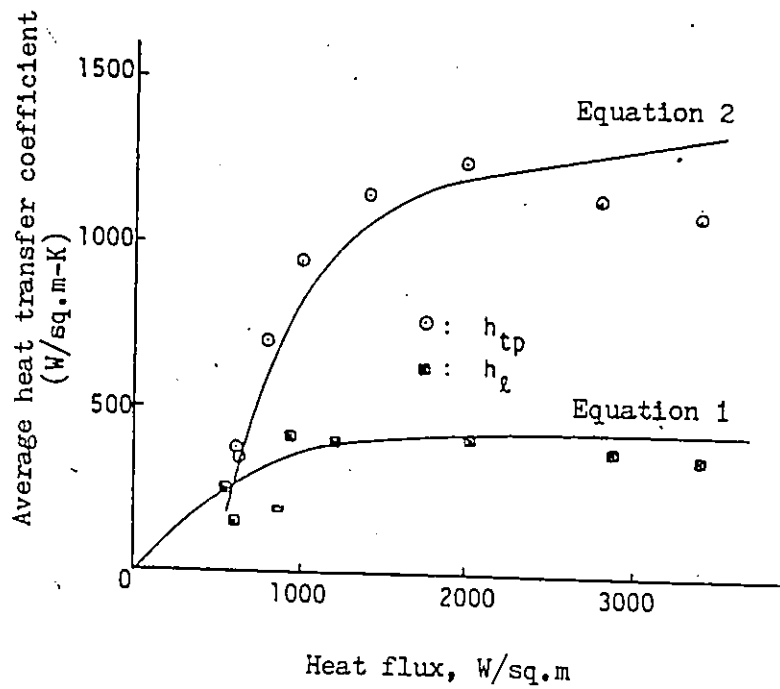


Figure 2.7 Effect of heat flux on the average heat transfer coefficient for two-phase (h_{tp}) and single-phase (h_l) region

2.3 TWO-PHASE FLOW REGIMES

The determination of which flow regime will occur for any given set of conditions is one of the problems facing a designer of two-phase flow devices. The flow regime depends upon the following factors;

1. Volumetric flow rates of each phase
2. Pressure
3. Heat flux at the wall
4. Densities and viscosities of each phase
5. Surface tension
6. Pipe geometry
7. Angle of pipe with respect to the horizontal plane
8. Flow direction (upward, downward, cocurrent, countercurrent)
9. Inlet length
10. Phase injection devices

Traditionally, flow regimes are identified from a flow regime map obtained from experimental observations. Many flow regime maps have been proposed using dimensional coordinates based on the liquid and gas superficial velocities (64-90). Other investigators have attempted to correlate the transition boundaries by non-dimensional groups. There have also been theoretical works to predict the flow regimes. These conventional flow regime maps are based on the liquid and gas superficial velocities or the total mass flux and quality.

Figure 2.8 shows the flow regime map by Hewitt & Roberts (67) for vertical flows. It is the most widely used chart for air-water and steam-water flows. It was originally established for an air-water mixture flowing in a pipe 31.2 mm in diameter at a pressure varying from 1.4 to 5.4 bar. Figure 2.9 shows the modified flow regime map of Baker (89) by Scott (90) for the horizontal flows. Flow pattern map of Mandhane (85) for gas-liquid in horizontal pipes is shown in figure 2.10. Figure 2.11 shows Soliman's (87) flow regime map for transition from mist-to-annular flow condensation.

The correlations to predict the transitions in a two-phase flow are given in table 2.3 of appendix A.

2.4 HEAT TRANSFER IN EVAPORATOR TUBES

In an evaporator tube, the heat transfer may take place in various forms, as the working fluid is transformed from a subcooled liquid at the entrance to the evaporator tubes to either a wet or a superheated vapour at its exit. Each of these mechanisms will be reviewed in order in which they would normally be expected to occur in the evaporator and a summary of the literature survey is presented in tabular forms.

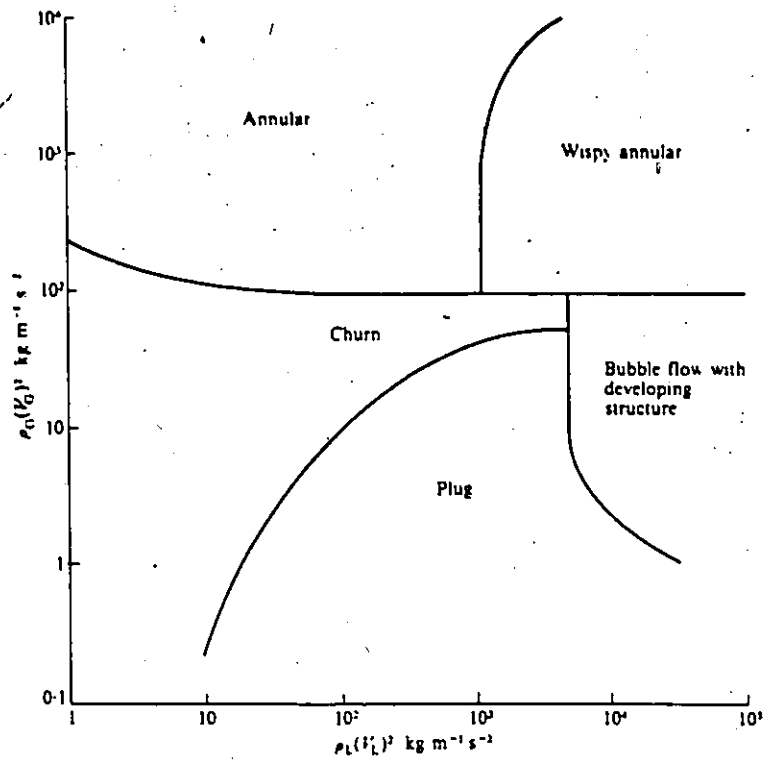
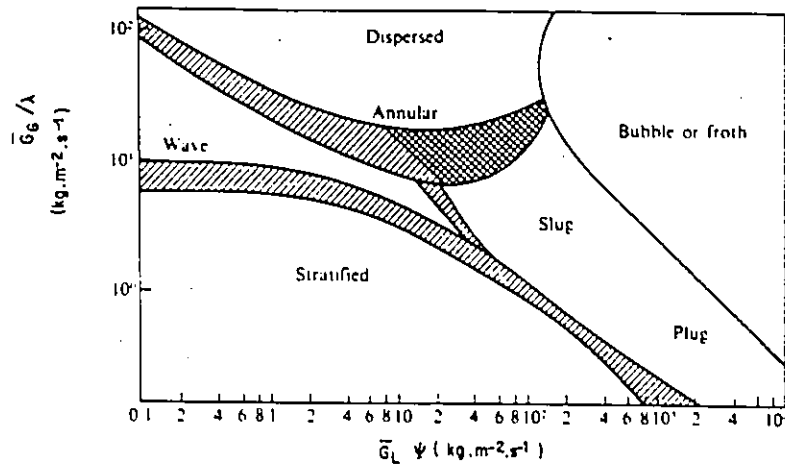


FIG. 2.8. MAP OF HEWITT & ROBERTS FOR VERTICAL TWO-PHASE FLOW (REF. 67)



$$\lambda \triangleq \left(\frac{\rho_G}{\rho_A} \frac{\rho_L}{\rho_W} \right)^{1/2}$$

$$\psi \triangleq \frac{\sigma_{WA}}{\sigma} \left[\frac{\mu_L}{\mu_W} \left(\frac{\rho_W}{\rho_L} \right)^2 \right]^{1/3}$$

FIG. 2.9. FLOW PATTERN MAP OF BAKER AS MODIFIED BY SCOTT (REF. 90)

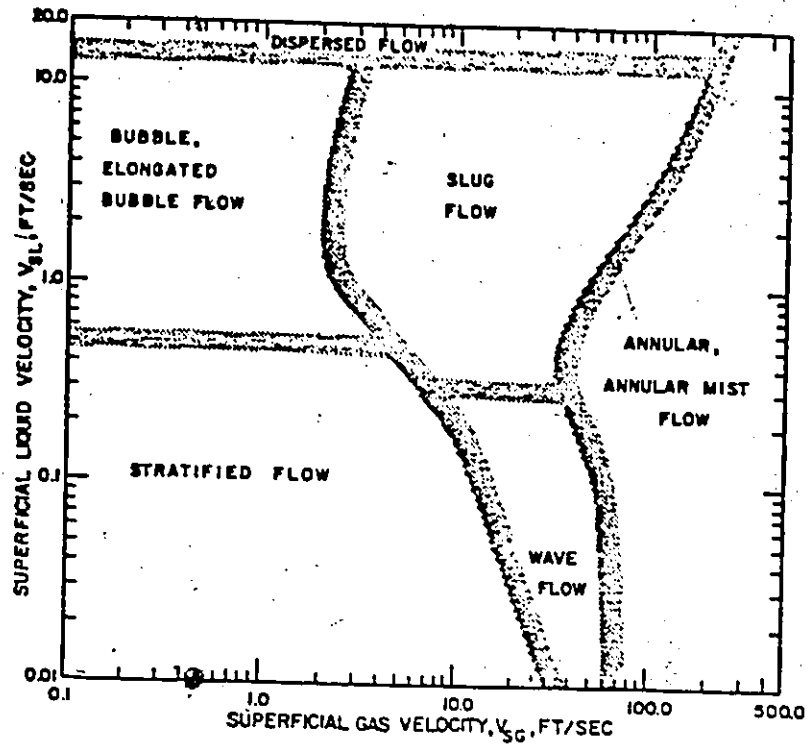


FIG. 2.10 FLOW PATTERN MAP PROPOSED BY MANDHANE (REF. 85)

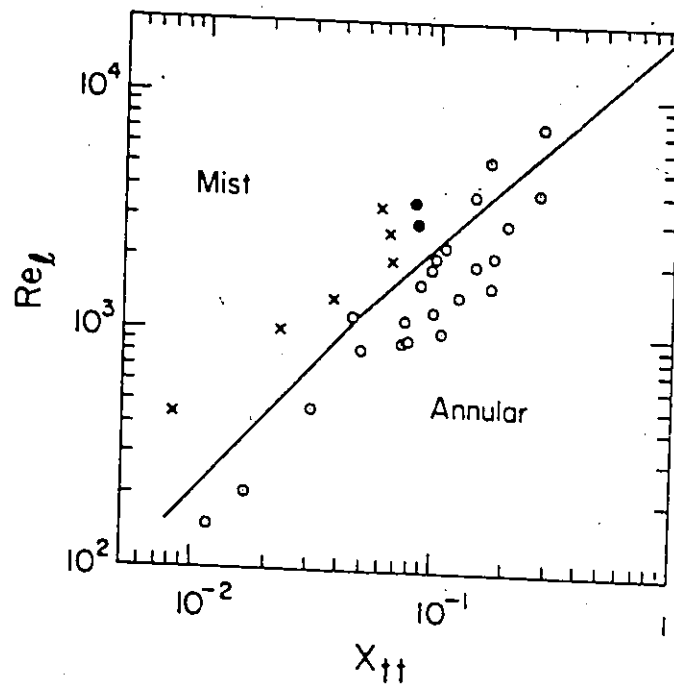


FIG. 2.11 SOLIMAN'S FLOW REGIME MAP FOR TRANSITION FROM MIST-TO-ANNULAR FLOW CONDENSATION (REF. 87). X MIST, ● MIST-ANNULAR, O ANNULAR

2.4.1 Single-phase liquid forced convection

Heat transfer by single-phase convection will occur when the fluid is flowing inside a tube either in laminar or in turbulent flow condition. There are many correlations available for this mode of heat transfer and a summary is presented in table 2.4.1 (appendix A) for laminar fully developed flow, turbulent fully developed flow and for the entrance region.

2.4.2 Transition from single-phase forced convection to nucleate boiling

An understanding of the factors which influence the onset of boiling is of importance in the design and operation of the equipment used for heat transfer in various processes. Transition from single-phase convection to nucleate boiling occurs when the liquid layer adjacent to a heating surface becomes superheated. It is now an accepted fact that the presence of nucleation cavities is required to establish boiling on a metal surface. Many investigators have presented correlations to predict an "incipient boiling" point. Most of these investigators assumed a linear temperature profile in the superheated layer and that the bubbles would grow and detach from the nucleation sites only when the superheated layer is such that a net heat flux into the developing bubble is realized when taking into

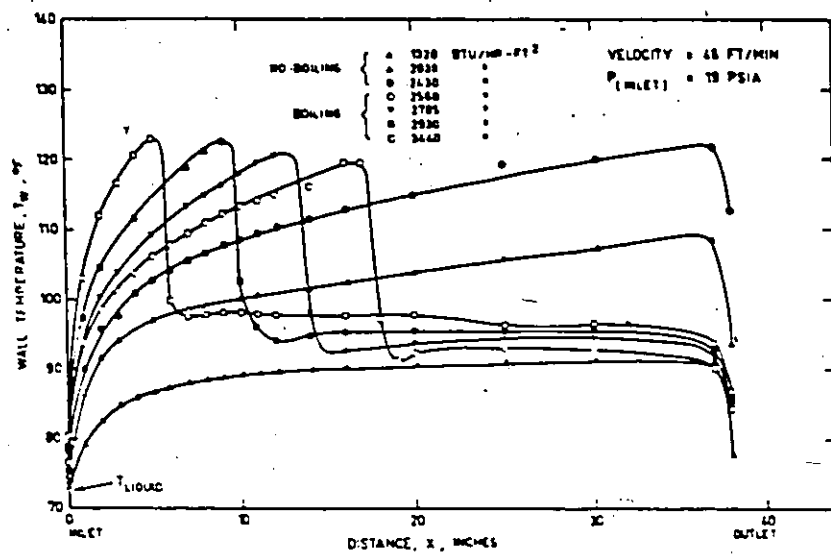
consideration the pressure and hence saturation temperature difference across the bubble boundary due to surface tension. Most of these investigators studied the phenomenon of wall superheat during incipient boiling in tubes. A summary of the correlations to predict this "incipient boiling" point is presented in table 2.4.2 (appendix A). The heat flux or the wall superheat can be calculated from these correlations for this condition.

Corty & Foust (118) observed that the past history of the boiling surface had a pronounced effect on the superheat required to initiate nucleate boiling. They described and classified the phenomenon for ether, normal pentane and R-113 and reported wall superheats far in excess of those required to maintain boiling once initiated. Turton (119) reported large wall superheat - as high as 100 to 120 F for R-11 boiling on a stainless steel tube at pressures above atmospheric and under the influence of increased gravity. Abdelmessih et al. (120) found that a wall superheat of 25 - 30 F was required to initiate boiling of R-11 on a stainless steel tube under forced flow conditions. The superheat was found to be independent of the fluid velocity in the range tested. Hodgson (121) investigated the hysteresis effect in forced convection, subcooled boiling with water. Marto & Rohsenow (122) reported temperatures as high as 135 F for the onset of nucleate boiling for liquid sodium in pool boiling and found that the value of the wall superheat required to initiate boiling depended on the surface finish. Similar overshoots in

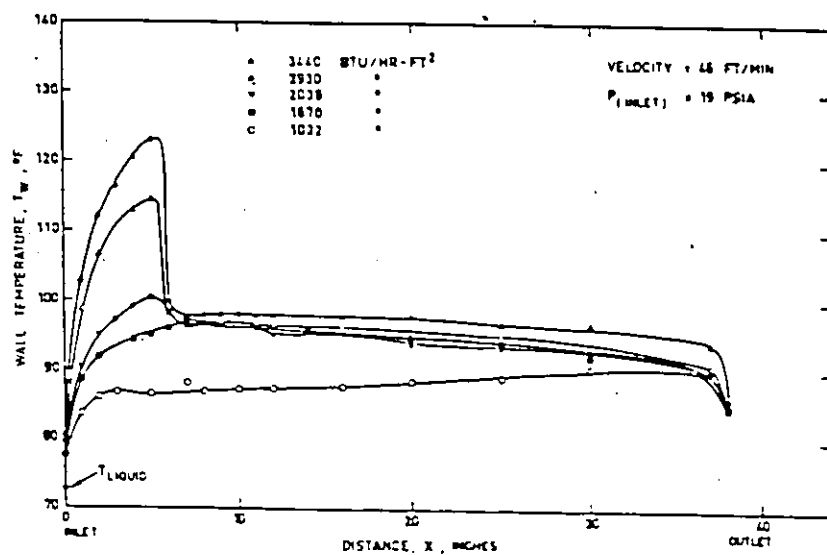
temperatures have been reported by Chen (117) for flow boiling of liquid potassium. Joudi & James (123) reported wall superheats far in excess of normal boiling temperature differences were required to initiate boiling of R-113 and methanol. Clarke & Robertson (124) reported that the onset of convective boiling in a boiling channel of an industrial heat exchanger would be considerably delayed beyond the point where the bulk fluid reached local saturation temperature. Recent work reported by Stauder (17) has shown that the performance of a two-phase thermosiphon loop heat exchanger using R-11 can be seriously affected by this hysteresis phenomenon. Excessive wall superheats were also reported by Bankoff et al. (125) and by Sakurai & Shiotsu (126).

Figure 2.12, a and b shows this effect for the experimental data of Abdelmessih et al. (120) for R-11. Figure 2.12a is for increasing heat flux and 2.12b is for decreasing heat flux. Figure 2.13 shows the hysteresis effect at the inception of flow boiling at two different positions in the test section. Figure 2.14 shows a similar effect investigated by Murphy & Bergles (114). Figure 2.15 shows the hysteresis effect for R-113 and methanol at atmospheric and subatmospheric pressures investigated by Joudi & James (123). Turton's results for Arcton 11 (Fr-11) is shown in figure 2.16.

Yin & Abdelmessih (127) investigated the boiling characteristics



(a) INCREASING HEAT FLUX



(b) DECREASING HEAT FLUX

FIGURE 2.12 WALL TEMPERATURES ALONG A TEST SECTION

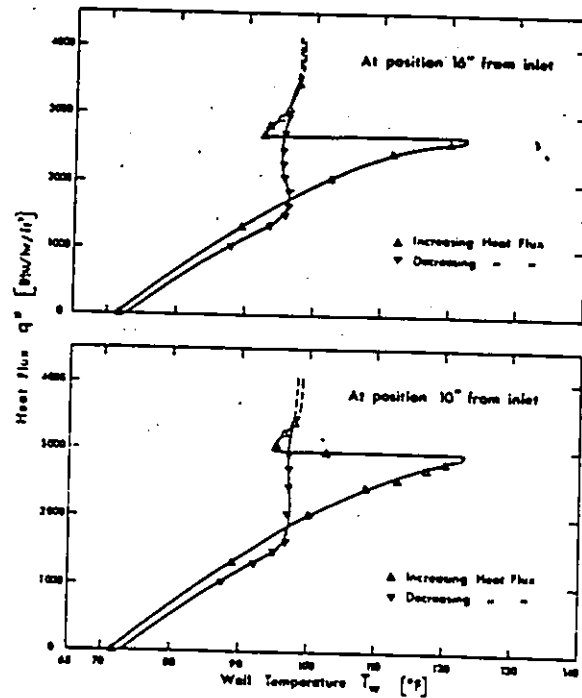


FIGURE 2.13 The hysteresis effect at the inception of flow boiling at two different positions on the test section. $V=46$ ft/min
 $P_{in}=19.0$ psia

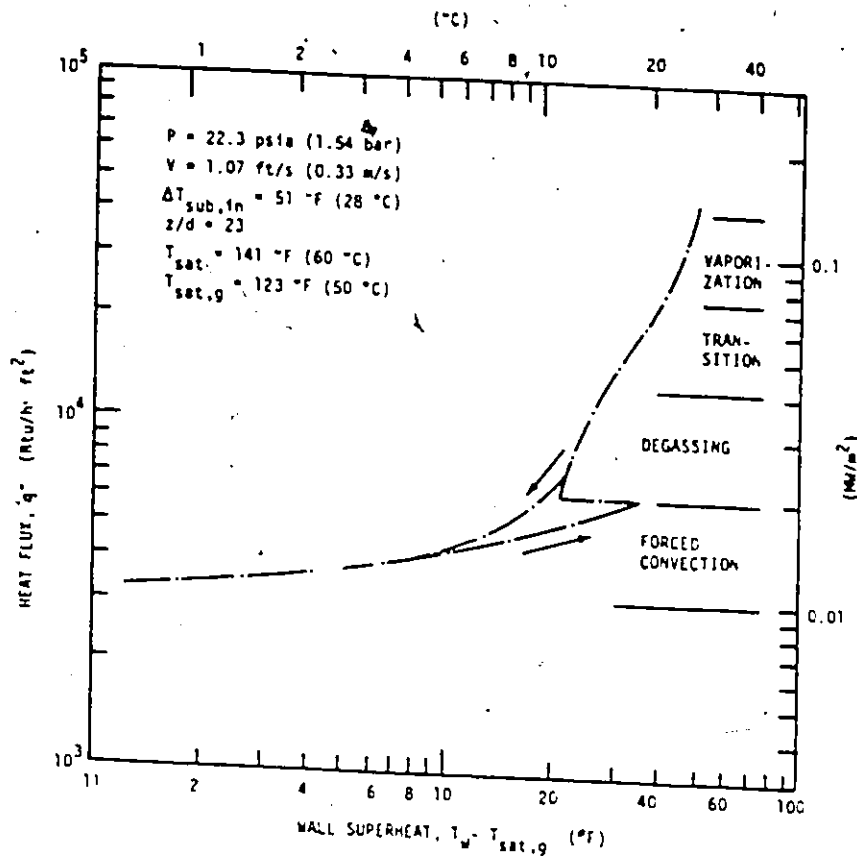
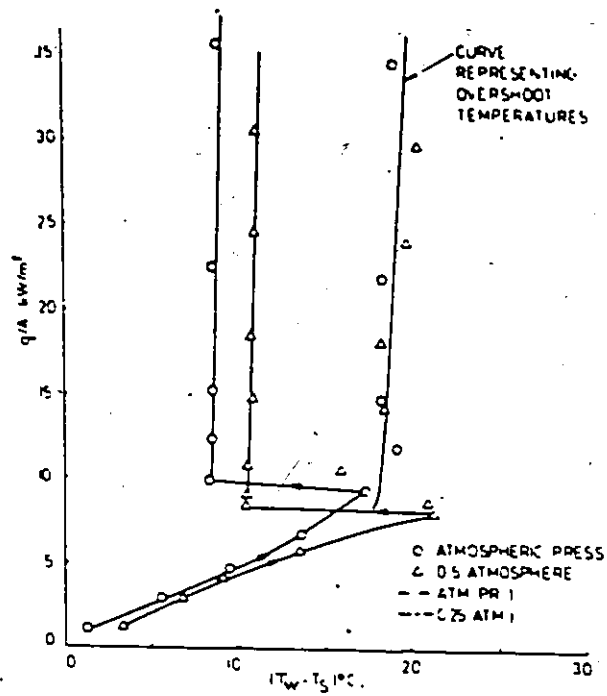
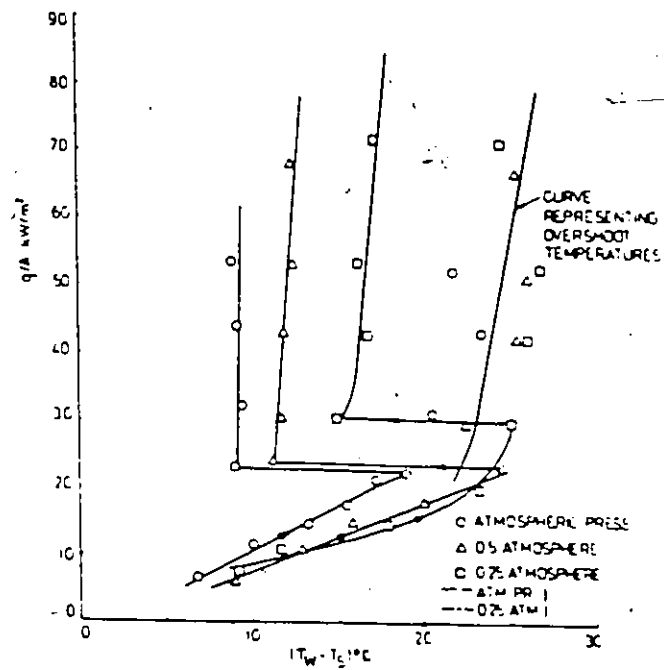


FIGURE 2.14 Boiling curve obtained with gassy coolant by Murphy and Bergles (114) showing hysteresis around the incipient boiling point.



Incipient boiling and temperature overshoots for R-113



Incipient boiling and temperature overshoots for methanol

FIG. 2.15 INCIPIENT BOILING SUPERHEATS FOR R-113 & METHANOL (REF.123)

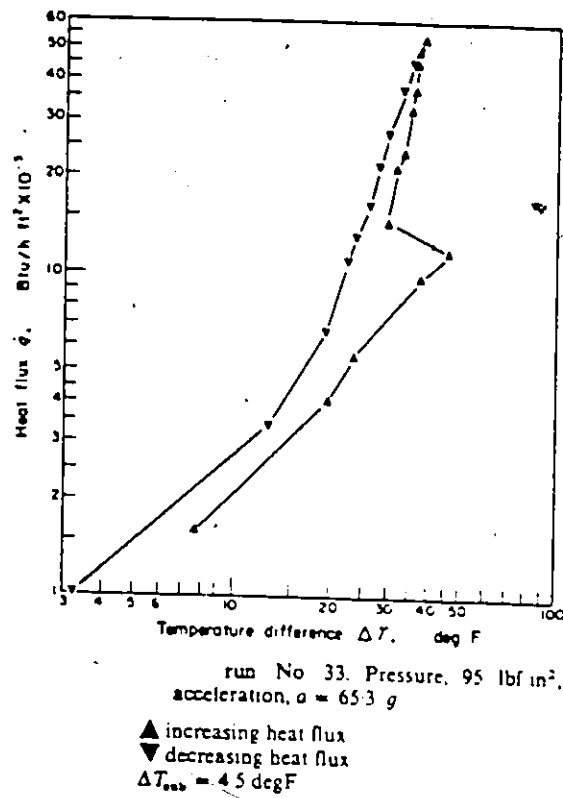
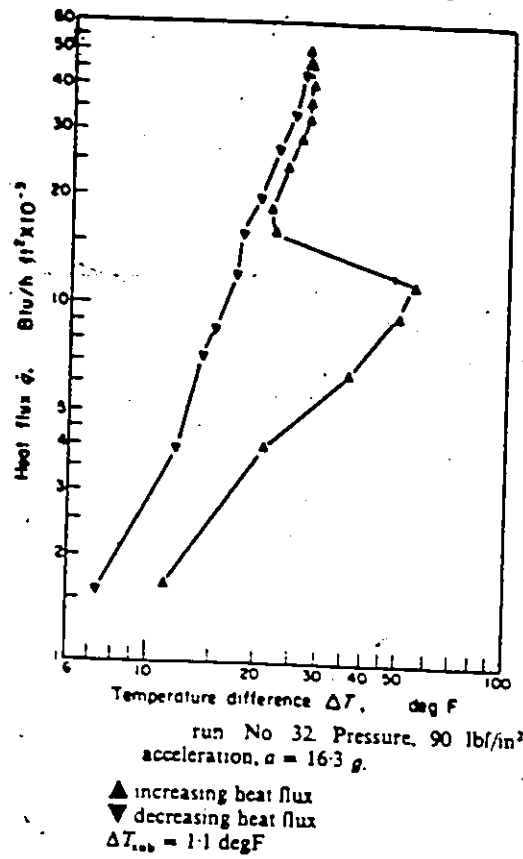


FIG. 2.16 INCIPIENT BOILING SUPERHEATS FOR R-11 UNDER THE INFLUENCE OF INCREASED GRAVITY (REF.119)

of R-11, and they correlated their experimental data with the following equations for increasing and decreasing heat fluxes.

For increasing heat flux (for incipience of boiling process)

$$(T_w - T_{sat})^2 = (7 - q/6500)^2 \frac{2 \sigma T_{sat}}{K_l h_{fg} \rho_v} q$$

For decreasing heat flux (for maintaining the boiling process)

$$(T_w - T_{sat})^2 = 5 \frac{\sigma T_{sat}}{K_l h_{fg} \rho_v} q$$

The values of all the variables in these two equations are in SI units. The experimental data of Yin and Abdelmessih for increasing and decreasing heat fluxes is shown in appendix B.

2.4.3 Nucleate boiling

Various regimes in a typical case pool boiling are shown in figure 2.17, which is the conventional log-log representation of heat flux versus wall superheat. This curve is for pool boiling of water at atmospheric pressure. In the range B-C, the liquid near the wall is superheated forming bubbles. The bubbles transport the latent heat of the phase change and also increase the convective heat transfer by agitating the liquid near the heating surface. The mechanism in this

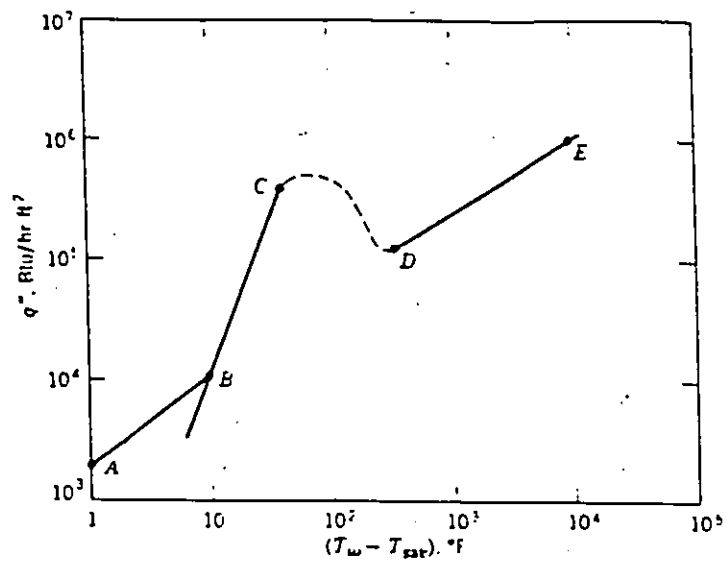


FIG. 2.17 Pool boiling regimes. A-B, natural convection; B-C, nucleate boiling; C-D, partial film boiling; D-E, stable film boiling

range is called nucleate boiling and is characterised by a very high heat transfer rate for only a small temperature difference. There are two subregimes in nucleate boiling;

1. local boiling or subcooled boiling
2. bulk boiling or saturated boiling

Local boiling is nucleate boiling in a subcooled liquid, where the bubbles formed at the heating surface tend to condense locally. Bulk boiling is nucleate boiling in a saturated liquid; in this case, the bubbles do not collapse. When the population of bubbles becomes too high at some high heat flux point C, the outgoing bubbles may obstruct the path of the incoming fluid. The vapour thus forms an insulating blanket covering the heating surface and thereby raising the surface temperature. In the range C-D the surface is alternately covered with a vapour blanket and a liquid layer, resulting in oscillating surface temperature. Boiling in this range is known as the transition boiling or partial film boiling.

There are numerous factors which affect the boiling mechanism e.g. number of active boiling sites, cavity size, mouth radius and depth, heated surface orientation, contact angle, growth and departure of bubbles, fluid-surface combination, initial liquid velocity, fluid properties, subcooling etc. There is a marked change in the heat transfer rate if any of the above variables are changed. Many

investigators (128-142) have studied the effect of the above mentioned variables on nucleate boiling. Detailed information is also available in books on two-phase flow heat transfer (143-147). Table 2.4.3 (appendix A) gives the details of the correlations available to predict the heat transfer rate in nucleate boiling regime.

2.4.4 Transition from nucleate boiling to two-phase forced convection

Nucleate flow boiling occurs in tubes when the quality is low. When the quality increases a transition occurs from nucleate boiling to two-phase forced convection. The quality plays an important role in determining the onset of this transition. Unfortunately, there is no published data to predict the onset of such a transition.

2.4.5 Two-phase forced convection

This is the most important heat transfer regime when the fluid is undergoing a change of phase in an evaporator tube. A literature survey on this topic reveals that there are numerous correlations available for this mechanism of heat transfer. These correlations were derived for different sets of physical conditions and for a variety of working fluids.

As suggested by Martinelli (145), many of the correlations which have been proposed for two-phase convective heat transfer are in the form

$$(h_{tp}/h_{lo}) \text{ or } (h_{tp}/h_l) = a (1/X_{tt})^b \quad \text{--- (I)}$$

where h_{tp} is the two-phase heat transfer coefficient, h_{lo} is the heat transfer coefficient if the total fluid were flowing with liquid properties in the tube, and h_l is the heat transfer coefficient if the liquid were flowing alone in the tube. The parameter X_{tt} is the Lockhart & Martinelli parameter and is given by

$$X_{tt} = ((1-x)/x)^{0.9} (\rho_g/\rho_l)^{0.5} (\mu_l/\mu_g)^{0.1}$$

A number of authors have extended the above equation to the following form

$$(h_{tp}/h_l) = C1 [Bo + C2 (1/X_{tt})^{n1}]^{n2} \quad \text{--- (II)}$$

where Bo is the boiling number and is given by $(q/G h_{fg})$. This equation gives the heat transfer coefficient for saturated convective heat transfer with simultaneous nucleate boiling. Tables 2.4.5 A and B (appendix A) show empirical values for the multipliers and exponents in equations I and II respectively and table 2.4.5C (appendix A)

gives the literature survey of the other correlations available to predict the two-phase forced convection heat transfer coefficient.

2.4.6 Transition from two-phase forced convection to liquid deficient region

During the process of evaporation inside tubes several flow pattern occur. Generally annular flow prevails over most of the tube length. Prior to the complete evaporation, the liquid film vanishes from the heating surface due to a complex hydrodynamic and heat transfer phenomenon causing dryout. This marks the change from annular to mist flow. The mass vapour fraction at the onset of dryout may be termed critical fraction or critical quality. Table 2.4.6 (appendix A) gives the details of the correlations which may be used to predict the onset of dryout in flow boiling.

2.4.7 Heat transfer in the liquid deficient region

In this region the magnitude of the heat transfer coefficients continue to drop as the quality increases. Table 2.4.7 (appendix A) gives a list of the correlations which may be used to calculate the heat transfer coefficients in this regime. Some of the investigators (201-206) have studied this region by experiments.

2.4.8 Heat transfer in single-phase vapour region

When the tube wall ceases to be wetted (no droplets impinging on the walls), the heat transfer coefficient becomes a minimum. This results in an increase in the wall temperature of the heated surface. After this point onwards, the heat transfer coefficient is governed by the correlations listed in table 2.4.1 (appendix A). The vapour properties should be used to evaluate the heat transfer coefficients.

2.5 HEAT TRANSFER IN THE CONDENSER

Condensation occurs when the temperature of a vapour is reduced below its saturation temperature. In practice, the process generally occurs when the vapour comes in contact with a cool surface. The latent energy of the vapour is released, and there is heat transfer to the surface. Condensation may be divided into two categories as follow;

1. Film condensation
2. Dropwise condensation

Film condensation is generally characteristics of clean, uncontaminated surfaces and is very common in any engineering process. In the film condensation process the surface is blanketed by the film, which grows in thickness as it moves down the plate or a tube. A

temperature gradient exists in the film, and the film represents a thermal resistance to heat transfer. In dropwise condensation a large portion of the area of the plate or a tube is directly exposed to the vapour; there is no film barrier to heat flow, and higher heat transfer rates are experienced. The heat transfer rates in dropwise condensation may be as much as ten times higher than in film condensation. Because of the higher heat transfer rates, dropwise condensation would be preferred to film condensation, but it is extremely difficult to maintain since most surfaces become wetted after exposure to a condensing vapour over an extended period of time. Various surface coatings and vapour additives have been used in attempts to maintain dropwise condensation, but these methods have not met with general success to date.

A literature survey on this topic reveals that there are a number of correlations available to predict the heat transfer coefficients for the condensation process. However, most of these correlations are for falling film i.e. for film condensation. A number of investigators (207-233) have studied film and dropwise condensation and a summary of the correlations to predict the condensation coefficients inside a tube are presented in table 2.5 of appendix A.

2.6 PRESSURE DROP & VOID FRACTION CORRELATIONS

In any evaporating and condensing heat exchanger system, the working fluid undergoes a phase change and it may exist in either single or two-phase. In the following section a literature survey is presented on the correlations available to predict the pressure drop and the void fraction for single-phase and two-phase regions.

2.6.1 Single-phase region

In single-phase flow the frictional pressure drop can be calculated by the following relation

$$\Delta P_f = f \frac{L v^2}{2 g D} \rho_l$$

where f is the Blasius (234) friction factor and is a function of the Reynolds number.

$$f = 64/Re \quad \text{for laminar flow}$$

$$f = 0.316/Re^{0.25} \quad \text{for turbulent flow}$$

Hydrostatic pressure drop can be calculated from the following relation

$$\Delta P_h = \rho g Z \sin \gamma$$

where γ is the inclination angle from the horizontal.

The void fraction for this region is either 1 or 0 depending on the type of flow.

$a = 0$ for single-phase liquid flow

$a = 1$ for single-phase vapour flow

2.6.2 Two-phase region

The correlations of two-phase pressure drop and void fraction for isothermal and non-isothermal flow is of importance in many technical flow problems and has been the subject of numerous investigations (235-271) in the past. A literature survey on the correlations available is presented in table 2.6.2 (appendix A) for the calculation of two-phase frictional pressure drop, total pressure drop and the void fraction. The methods proposed have varied from semi-theoretical to empirical correlations without any one being completely successful, although the proposed method of Lockhart & Martinelli (250) has perhaps been the most widely used, where the use of two-phase frictional multipliers, whose actual values have been

determined empirically, have allowed helpful predictions to be made for certain flow regimes. Detailed checks with extensive data have shown that the correlation overpredicts the frictional pressure drop or stratified flow regime; is quite reasonable for slug and plug flow; and for annular flow it underpredicts for small diameter pipes but overpredicts for larger pipes (ref 241).

The hydrostatic pressure drop for this region may be calculated by first evaluating the density of two-phase mixture which is a function of void fraction and densities of both phases.

$$\Delta P_{tp,h} = \rho_m g Z \sin \gamma$$

where

$$\rho_m = a \rho_v + (1-a) \rho_l$$

The third kind of pressure loss occurs because of the acceleration of fluids due to vapour generation. The acceleration pressure drop can be calculated from the following relations.

For homogenous flow model

$$\Delta P_a = G_t^2 / \rho_l \left[(1-x) + x (\rho_l / \rho_v) - 1 \right]$$

For separated flow model

$$\Delta P_a = G_t^2 / \rho_l \left[(1-x)^2 / (1-a) + (x^2/a) (\rho_l / \rho_v) - 1 \right]$$

CHAPTER III

MODEL DEVELOPMENT

3.1.1 INTRODUCTION

In the last chapter, various aspects of a two-phase flow systems were discussed and various correlations were presented for the flow regimes that are encountered in a such a system. In this chapter, a model will be developed by using the correlations listed in chapter 2.

The original model (see appendix C) developed by Ali (4) showed good agreement with earlier experiments (6,8,9). However, Ali's program was not able to simulate the more recent experiments done by Raza (12), where the overall performance was much lower than the results reported by Sampath (8). The lower performance obtained by Raza could be due to the fact that at the time he carried out his experiments, he was not aware of the presence of a hysteresis effect resulting from fluids boiling on a heated surface. Thus Raza's results may be for the heating process instead of cooling where a larger wall superheat is needed to initiate the boiling process.

In order to obtain better agreement with the data obtained by

Raza several correlations had to be changed. In addition some mistakes found in Ali's program were corrected. These mistakes were of a minor nature and would not have affected the previous results. The new model developed not only agrees with the data of Raza but also does a good job in simulating the experimental data obtained by Stauder (17), Kosnik & Bertoni (15) and Bergevin et al. (23). The reasons for selecting the various correlations are also discussed in this chapter.

3.1.2 Definitions

In this section some of the important parameters are defined as follows.

Vapour quality

By assuming thermal equilibrium in two-phase flow, the vapour quality is defined as

$$x = \frac{(h - h_f)}{h_{fg}} = \frac{\dot{m}_v}{(\dot{m}_v + \dot{m}_l)} \quad (1)$$

Void fraction

The void fraction is defined as the ratio of the flow area occupied by vapour to the total flow area.

$$\alpha = \frac{A_v}{A_v + A_l} \quad (2)$$

Mean density

Mean density of a two-phase mixture is given by the following relation.

$$\rho_m = \alpha \rho_v + (1-\alpha) \rho_l \quad (3)$$

Phase velocities

For one-dimensional flow the area-averaged velocity of liquid and vapour phases are defined as follows.

$$V_l = \frac{\dot{m}_l}{\rho_l A_x (1 - \alpha)} \quad \text{--- (4)}$$

$$V_v = \frac{\dot{m}_v}{\rho_v A_x \alpha} \quad \text{--- (5)}$$

Slip ratio

Slip ratio is defined as the ratio of the vapour phase velocity to the liquid phase velocity.

$$S = V_v / V_l \quad \text{--- (6)}$$

from equations 4, 5 & 6

$$S = [x / (1 - x)] (\rho_l / \rho_v) [(1 - \alpha) / \alpha] \quad \text{--- (7)}$$

3.1.3 Conservation equations

In this section the conservation equations are given, which should be satisfied for the loop.

Mass conservation

For steady state the conservation of mass equation must be satisfied.

$$\frac{dm}{dz} = 0$$

(8)

Momentum conservation

For steady state the momentum equation is

$$-\frac{dp}{dz} = \frac{dG}{dt} + \frac{1}{A} \frac{d}{dz} \left[\frac{G^2 A}{\rho} \right] + \frac{P}{A} \tau_w + \rho g \cos \theta \quad (9)$$

Equation (9) represents the acceleration pressure drop, frictional pressure drop, hydrostatic pressure drop and the pressure drop in bends, fittings, etc.

$$-dP_t = dP_a + dP_f + dP_h + dP_l \quad (10)$$

The terms dP_a , dP_f , dP_h and dP_l are acceleration pressure loss, frictional pressure loss, hydrostatic pressure loss and pressure drop in valves, bends, fittings, etc. respectively.

For a closed system the summation of the pressure drops around the loop must be zero.

$$\Delta P_t = 0 \quad (11)$$

Energy conservation

The amount of heat transferred to a mixture per unit length, q' , is equal to the rate of change of enthalpy of mixture with respect to z , neglecting the changes in potential and kinetic energies.

$$q' = \frac{d}{dz} (\dot{m}_l h_l) + \frac{d}{dz} (\dot{m}_v h_v) \quad \text{--- (12)}$$

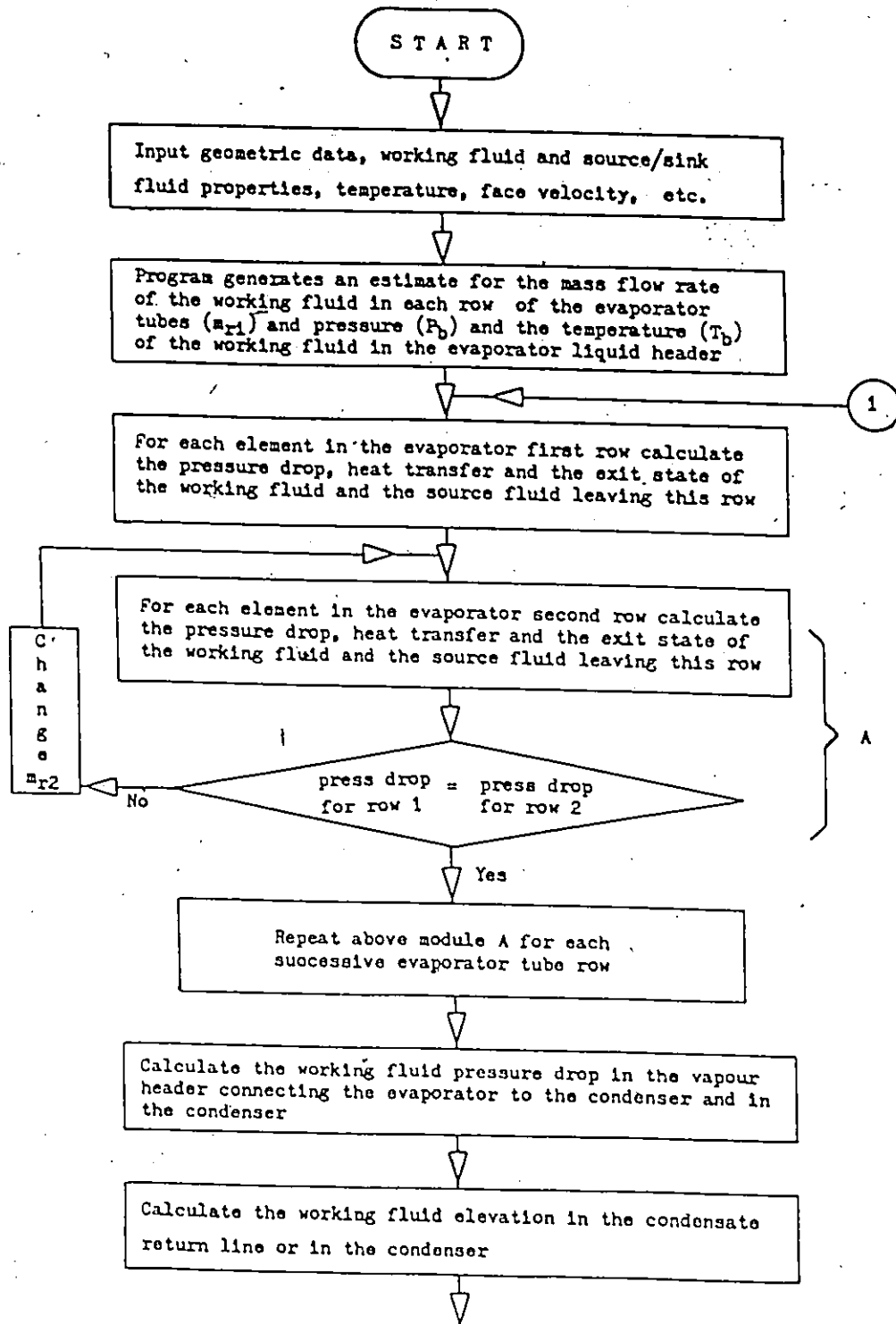
$$q' dz = d (\dot{m}_l h_l) + d (\dot{m}_v h_v) \quad \text{--- (13)}$$

3.1.4 Program logic

In this section the model used for the development of a computer simulation program is described which is capable of simulating the performance of fluid-to-fluid, two-phase thermosiphon loop heat exchanger systems, using any working fluid. The model developed is based on the assumption of one-dimensional steady state equilibrium flow in the tubes. The equations used in the model were taken from the literature discussed in chapter 2, and the detailed description of the correlations is presented in sections 3.2 to 3.6. Due to many possible modes of heat transfer that can occur along the length of the evaporator tubes, as discussed in chapter 2, the evaporator was analyzed by subdividing each tube into shorter elemental lengths for which average values provide a good approximation.

Figure 3.1 shows a flow diagram for the program which illustrates the program logic and the type of information which the operator must supply. Before using the program the operator must ensure that the appropriate property subroutines for the working fluids and the source and the sink fluids are incorporated (see appendix D). To run the program, the operator must specify the hot

FIG. 3.1 SIMULATION PROGRAM FLOW DIAGRAM - CONT. ON NEXT PAGE



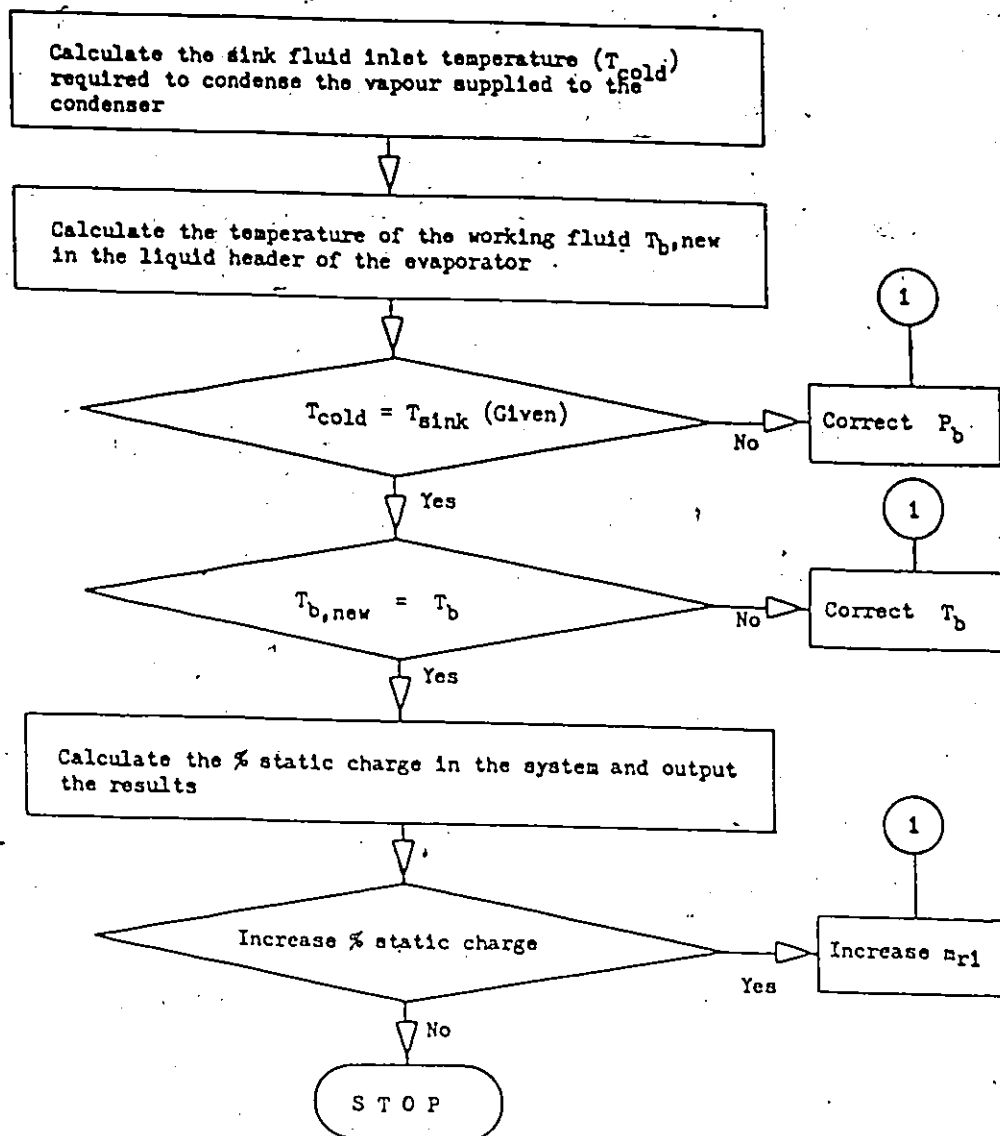


FIG. 3.1 SIMULATION PROGRAM FLOW DIAGRAM

and cold supply duct temperatures (T_h) and (T_c); the geometric configuration and the size information necessary to describe the system and the number of elements into which the length of each evaporator tube is to be subdivided (the default value is 10). Appendix H2 "User's manual" gives the precise form of the subroutines which should be used.

The program estimates an initial pressure and temperature in the evaporator liquid header and the mass flow rate per tube in the first row of tubes, then proceeds to calculate the heat transfer, the exit state of the working fluid and the source fluid temperature downstream of the evaporator element for each tube element in the first row. Similar calculations are done for each subsequent row of evaporator tubes. For each of these rows the flow rate is varied until its exit pressure agrees with that from the first row of tubes. The pressure drop across and the heat loss from the vapour header is calculated. Assuming a negligible pressure drop across the condenser tubes, and by knowing the inlet and the exit states of the condenser, the heat transfer in condenser and the sink fluid temperature is calculated. The inlet pressure of the evaporator liquid header is varied until the given cold duct supply temperature is achieved. Once this convergence is reached an energy balance applied to the evaporator liquid header yield a new value for the fluid temperature at the bottom of the evaporator. This new value is used to generate final data including the static liquid level which would exist in the loop upon shutdown.

The program outputs the overall results and the calculated static charge in the evaporator as a percentage of the total tube volume. To generate a set of results for a higher or a lower static charge in the evaporator, the mass flow rate of the working fluid per tube in the first row of evaporator tubes must be increased or decreased respectively.

3.1.5 Heat transfer relationships

In order to calculate the rate of heat transfer through any element, a relationship must be developed between the heat flux, the temperature difference between the source fluid temperature and the working fluid and the thermal properties of the element. This may be easily done with the help of an electrical analog of the system.

Figure 3.2 shows such an electrical analog for an evaporator heat exchanger transferring heat from a source fluid at a constant temperature T_h to a working fluid at a temperature of T_f . The various resistances to the flow of heat from the source to a working fluid are shown and identified below:

1. Evaporator resistance, R_e , the thermal resistance between evaporator wall and the working fluid.
2. Evaporator external resistance, R_{ee} , the resistance between the evaporator tube wall and the source fluid.
3. Evaporator wall resistance, R_{ew}
4. Evaporator fouling factors; Inside, F_{ei} & Outside, F_{eo}

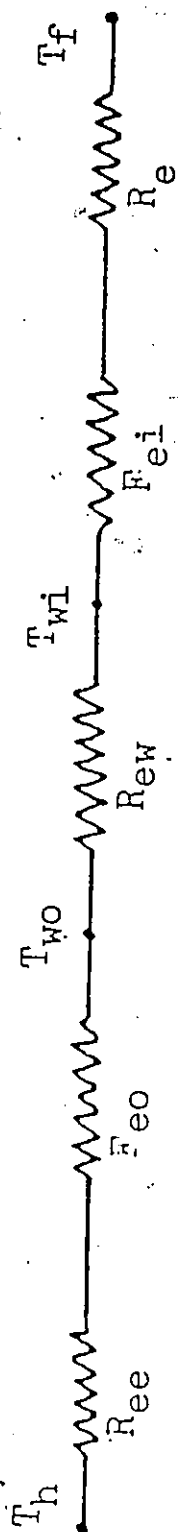


FIGURE 3.2 ELECTRICAL ANALOG FOR AN EVAPORATOR ELEMENT

exchanger depends largely on the resistances listed 1 and 2.

The heat transfer rate at any point can be calculated as follows:

$$Q_z = \frac{(T_h - T_f)}{R_{tot}} \quad \text{--- (14)}$$

where T_h is the average source fluid temperature for an element.

$$\text{and } R_{tot} = R_{ee} + F_{eo} + R_{ew} + F_{ei} + R_e \quad \text{--- (15)}$$

$$R_{tot} = 1/(U_z A_i) \quad \text{--- (16)}$$

$$1/U_z = 1/U_o (A_i/A_o) + F_{eo} + [\Delta A_i \ln(r_o/r_i)/2\pi k z] + F_{ei} + 1/h_{lp} \quad \text{--- (17)}$$

The first term on the right hand side of equation 17 is the evaporator outside resistance (including the effect of external fins etc.), the second and fourth terms are the outside and inside fouling factors respectively, the third term is the wall resistance and the last term is the evaporator inside resistance. The values of F_{ei} & F_{eo} must be obtained from the literature. Once U_o is supplied for a particular geometry and fin configuration, and h_{lp} is calculated in the program then heat transfer rate may be evaluated. See appendix E & F for the evaluation of U_o for annulus and cross flow finned tube configurations.

For the case where the source fluid temperature is varying (as in the case of air flowing over finned tubes) the effectiveness may be calculated from

$$E_z = 1 - e^{(-NTU_z)} \quad \text{--- (18)}$$

$$\text{where } NTU_z = ((U_z A_i)/(\dot{m} Cp)_{air}) \quad \text{--- (19)}$$

\dot{m} is the mass flow rate of hot air flowing past the tube element, and $(U_z A_i)$ may be calculated using equation 17.

The rate of heat transfer for the element may be calculated by

$$Q_z = E_z (\dot{m} Cp)_{air} (T_a - T_f) \quad \text{--- (20)}$$

where T_a is the approach air temperature for the element.

For a prescribed heat flux boundary condition

The rate of heat transfer for the element was calculated by

$$Q_z = q_z A_i \quad \text{--- (21)}$$

and the wall temperature was calculated by

$$T_w = T_f + Q_z/(h_i A_i) \quad \text{--- (22)}$$

3.2 EVAPORATOR CALCULATIONS

In this section, the correlations used for the various flow regimes are discussed in details. In the beginning of this section, the estimation of the working fluid state in the evaporator liquid header is discussed.

3.2.1 Estimation of an initial state of the working fluid in evaporator liquid header

Input data is used to estimate the state of the working fluid (pressure & temperature).

(a) Estimation of pressure & temperature

The source and the sink temperatures T_h & T_c govern the maximum and minimum pressure limits in a two-phase thermosiphon loop. The maximum pressure, P_{max} , would correspond to the saturation pressure at T_h and the minimum pressure, P_{min} , would correspond to the saturation pressure at T_c . The initial starting pressure in the evaporator liquid header was set equal to the average of these two pressures plus the pressure head corresponding to the evaporator tube filled with liquid. Hence, the following relation was used in determination of the first estimation of pressure in the evaporator liquid header.

$$P_{evap,b} = (P_{max} + P_{min})/2 + H_e \rho_l g \quad \text{--- (23)}$$

where H_e is the difference in elevation between the vapour and the liquid headers of the first row of evaporator tubes.

The working fluid enters the bottom of the evaporator from the condenser in a subcooled state and hence the temperature of the working fluid in the evaporator inlet would be less than the saturation temperature corresponding to $P_{evap,b}$. The following relation was used to estimate the temperature of the working fluid.

$$T_{\text{evap},b} = T_{\text{sat}}(P_{\text{evap},b}) - C [T_{\text{sat}}(P_{\text{evap},b}) - T_{\text{cold}}] \quad \text{--- (24)}$$

where T_{cold} is the cold fluid temperature. A value of 0.045 for C was found to agree with the converged results obtained by the computer program.

The estimated values of pressure and temperature from equations 23 and 24 resulted in convergence in 2 or 3 iterations.

(b) Estimation of mass flow rate in the evaporator tube

The mass flow rate for the evaporator tube may be estimated by the following method.

The heat added in the evaporator tube may be estimated by

$$Q_e = h_{e,av} A_e (T_{ew,av} - T_{eb,av}) = \dot{m} h_{fg} x_e \quad \text{--- (25)}$$

where $h_{e,av}$ is the average heat transfer coefficient for the evaporator, $T_{ew,av}$, $T_{eb,av}$, x_e & A_e are the average evaporator wall temperature, average bulk fluid temperature, the evaporator exit quality and the evaporator surface area. Solving for \dot{m}

$$\dot{m} = \frac{h_{e,av} A_e (T_{ew,av} - T_{eb,av})}{h_{fg} x_e} \quad \text{--- (26)}$$

From the experience of running the computer program it was found that when x_e is taken as 0.45, a good initial estimate is produced.

Once the evaporator inlet pressure, temperature and the mass flow rate are estimated, the program proceeds to calculate the heat transfer & pressure drop for each evaporator element. In the following sections the correlations which were used for the calculations are discussed.

3.2.2 Single-phase flow

The working fluid enters the evaporator in a subcooled state and hence is all liquid at the entrance. Heat transfer takes place by single phase forced convection and the pressure decreases as the fluid progresses up the tube. The heat transfer and the pressure drop for the first element were determined as follows:

3.2.2.1 Heat transfer calculations

The correlations of Sieder & Tate (96) were used to determine the local heat transfer coefficient for single-phase developing flow. The following correlations were used to determine the heat transfer coefficient for this single-phase liquid region.

For laminar flow

$$Nu_z = 1.24 [Re Pr / (L/D)]^{1/3} (\mu/\mu_s)^{0.14} \quad \text{--- (27)}$$

For turbulent flow

$$Nu_D = 0.027 Re^{0.8} Pr^{0.33} (\mu/\mu_s)^{0.14} \quad \text{--- (28)}$$

Once the value of h_{1p} is known, the heat transfer rate into the element can be calculated from equation 14.

3.2.2.2. Pressure drop calculation

The frictional and the hydrostatic pressure drops were calculated from the following relations.

$$\Delta P_{fz} = f \frac{\Delta z v_1^2}{2 D} \rho_1 \quad \text{--- (29)}$$

$$\Delta P_{hz} = \Delta z \rho g \sin \gamma \quad \text{--- (30)}$$

where, for Laminar flow ($Re < 2000$)

$$f = 64/Re \quad \text{--- (31)}$$

and for Turbulent flow ($Re > 2000$)

$$f = 0.316/Re^{0.25} \quad \text{--- (32)}$$

As the flow progresses up the evaporator tube, the pressure drops and hence the saturation temperature decreases. The bulk fluid temperature and the wall temperature increase along the length of this section of the evaporator tube. When the wall temperature exceeds the local saturation temperature by a sufficient amount (even when the bulk fluid temperature is below the saturation temperature) the first nucleation site becomes activated. If boiling takes place when the

bulk fluid temperature is below the saturation temperature, it is termed subcooled nucleate boiling.

3.2.3 Subcooled boiling

The correlations available to predict the conditions necessary for boiling are listed in table 2.4.2. These correlations may be used to predict the minimum heat flux required to maintain subcooled boiling in a flow boiling situation. The correlation of Sato & Matsumura (112) for turbulent flow, and their modification introduced by Ali for laminar flow were used to predict the minimum heat flux necessary to sustain boiling.

For laminar flow

$$q = \frac{k_L \lambda}{8cT_s (v_g - v_L)} \left[\frac{(Dv_L Z)^{1/3} q}{1.86 v_L^{0.33} Pr_L^{0.33} k_L} - (T_s - T) \right]^2 \quad (33)$$

For turbulent flow

$$q = \frac{k_L \lambda}{8cT_s (v_g - v_L)} \left[\frac{D^{0.2} v_L^{0.8} q}{0.023 Pr_L^{0.4} k_L v_L^{0.8}} - (T_s - T) \right]^2 \quad (34)$$

However, when no vapour bubbles are present as in the case of the system startup, a much greater wall superheat is required to initiate the bubble growth. Abdelmessih et al. (120) investigated this incipient boiling effect for R-11 flowing in a heated stainless steel 10.92 mm inside diameter tube with a mass flow range of 2.44

kg/min to 18.29 kg/min. They found that for the startup process the wall superheats were in the range of 13.9 - 16.7 C. For the present studies for R-11 boiling in copper tubes, Stauder (17) found that wall superheats of 12 - 14 C were required to initiate the boiling for the system startup. To investigate this effect a flag can be set in the program which requires the system to have a wall superheat of 13.3 C before any boiling was allowed to take place.

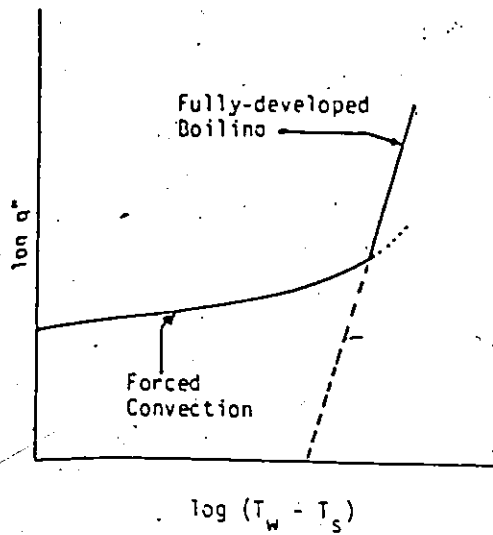
3.2.3.1. Effective heat transfer coefficient in the subcooled boiling region

Once subcooled nucleate boiling has been initiated, vapour bubbles formed at the wall detach and move into the bulk of the fluid. Since the bulk fluid temperature is less than the saturation temperature, the bubbles condense in the fluid and give up their latent heat. Figure 3.3 illustrates different models suggested to estimate the effective heat transfer coefficient for this region. In this region highly localized nucleate boiling is initially superimposed on the heat flux due to forced convection at the wall. The effective heat transfer coefficient for this region was evaluated from the following relation, which was developed by Ali (4)

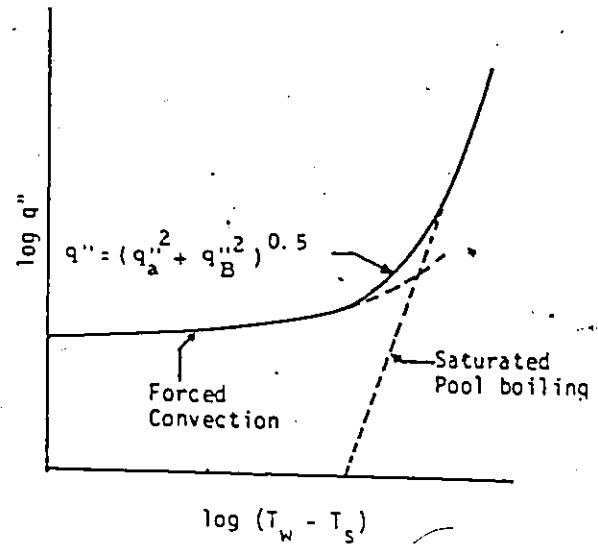
$$h_{el} = [h_b^p + h_{lp}^p]^{(1/p)} \quad \text{--- (35)}$$

where the power p is given by

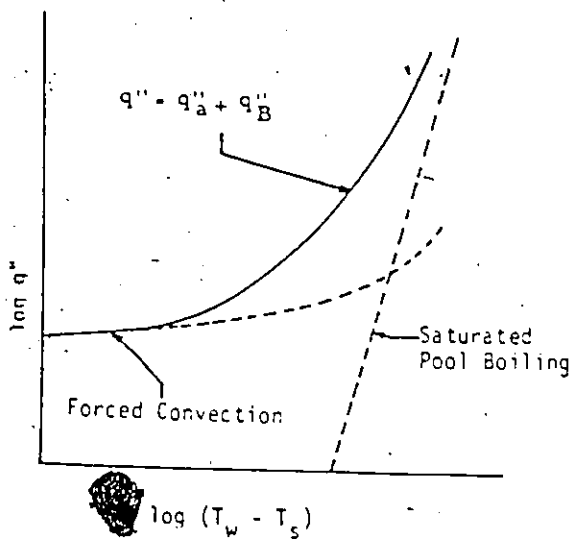
$$p = 2 - \frac{(T_s - T_f)}{(T_s - T_f)_{ib}} \quad \text{--- (36)}$$



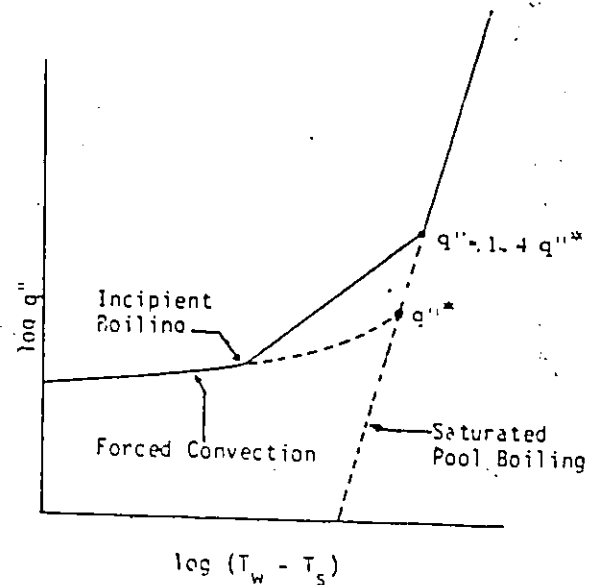
Procedure of McAdams et al.



Procedure of Kutateladze



Superposition Procedure of Rohsenow



Procedure of Forster and Grief

FIG. 2.3 ESTIMATION PROCEDURE IN TRANSITIONAL BOILING

p is unity at the point of incipient boiling, therefore additive (Rohsenow), then approaches the value two as the degree of subcooling $(T_s - T_f)$ approaches zero (agreeing with Kutateladze).

In the subcooled nucleate boiling region the vapour quality is zero (based on the enthalpy), even though the void fraction may be sufficient to cause an increase in the velocity of the fluid.

For this region, the single-phase forced convection coefficient was calculated from equation 27 or 28 with the modified Reynolds number as follows;

$$Re_{scb} = V_1 D \rho_m / \mu_1 \quad \text{--- (37)}$$

where ρ_m is given by equation 3.

In the subcooled nucleate boiling region it is presumed that the fluid is thoroughly mixed due to condensing of the vapour bubbles and hence a void fraction was calculated based on the homogenous model.

$$\alpha = 1 / [1 + (1 - x_{scb}) / x_{scb} (\rho_v / \rho_l)] \quad \text{--- (38)}$$

The following quality was used in equation 38 to calculate the void fraction.

$$x_{scb} = q_b / (m_t h_{fg}) \quad \text{--- (39)}$$

where q_b is the heat transfer due to boiling. Once q_b is calculated from Rohsenow's (152) pool boiling correlation (discussed in 3.2.4),

the Reynolds number was calculated from equation 37 by using equations 38 & 39 and thus h_{el} was calculated from equation 35.

3.2.3.2. Pressure drop calculation

Even though the void fraction and the quality in the subcooled region are of very small magnitude, the pressure drop would be lower in comparison to that of single-phase liquid flow. The method of estimation of the pressure drop is discussed in 3.2.5.2.

3.2.4. Flow boiling

As the fluid progresses up the evaporator tube, the bulk fluid temperature would eventually be equal to the saturation temperature, when saturated nucleate boiling starts.

3.2.4.1. Determination of boiling heat transfer coefficient h_b

Rohsenow's pool boiling correlation is the most widely used correlation in the determination of the nucleate boiling coefficient, even though numerous correlations are available in the literature. The following is Rohsenow's correlation (152)

$$q_b = \mu_L \lambda \left[\frac{g(\rho_L - \rho_g)}{\sigma} \right]^k \left[\frac{C_L(T_w - T_s)}{\lambda(CSF) Pr^{1.7}} \right]^r \quad \text{--- (40)}$$

where Csf and r are constant for a given working fluid/heating surface combination.

Once the heat flux q_b is known, the boiling heat transfer coefficient can be calculated from

$$h_b = q_b / \Delta T_b \quad \text{--- (41)}$$

where $\Delta T_b = T_w - T_s$

The following values of C_{sf} and r were used by Ali.

$$\begin{array}{l|l} C_{sf} = 0.01 & \\ r = 1.31 & \end{array} \quad \text{--- (42)}$$

Initially, the program was run with the values used by Ali. But it was found that the boiling heat transfer coefficient h_b was greater than the two-phase heat transfer coefficient even at high vapour qualities, where nucleate boiling should have been suppressed. Literature shows that various values for r in Rohsenow's boiling equation have been used with a range of 0.33 to 3.0. In order to accurately model the experimental data of Raza, so that the simulated wall temperature and the heat transfer rate agreed with the corresponding experimental data it was found necessary to correct the value of r . The following values of C_{sf} and r were used in the present studies.

$$\begin{array}{l|l} C_{sf} = 0.01 & \\ r = 1.80 & \end{array} \quad \text{--- (43)}$$

The same values of these constants were also used to simulate

the experimental data provided by Stauder, Kosnik & Bertoni and the experimental data of Bergevin et al.

3.2.4.2 Boiling suppression

The quality and the vapour fraction increases with the heat addition resulting in higher vapour velocities and higher two-phase heat transfer coefficients. As the quality and vapour fraction continue to increase, the flow inside the evaporator becomes annular and nucleate boiling is suppressed. The following relation, which was used to calculate the effective heat transfer coefficient in two-phase flow region, shows the domination of two-phase heat transfer coefficient at higher qualities (when $h_{tp} > h_b$), where the effect of h_b diminish.

$$h_{2e} = [h_{tp}^p + h_b^p]^{(1/p)} \quad \text{--- (44)}$$

3.2.4.3 Pressure drop calculations

In flow boiling region the working fluid is in two-phase state and the procedure for calculation of the pressure drop is discussed in section 3.2.5.2.

3.2.5 Two-phase forced convection

3.2.5.1 Heat transfer calculations

At the point where the bulk fluid temperature reaches the

saturation temperature, a two-phase flow coefficient h_{tp} , based on the correlation of Davis & David (161) was calculated. Ali used Sachs & Long (173) correlation for the determination of the two-phase flow coefficient which was used to simulate the experimental data of Hwang (1) & Sampath (3). The agreement between the simulated results and the experimental data was excellent. Since, Raza used the same experimental rig which was used by Sampath, the author used the correlation used by Ali to simulate the experimental studies done by Raza. Surprisingly, the overall heat transfer rates and the wall temperatures did not match with the experimental values provided by Raza. The simulated overall heat transfer rate, on average, was 20 - 25 % greater than that was obtained from the experiments. The temperatures of the evaporator wall on the average were 2 C lower than the experimental values. One of the major cause for this lower performance obtained by Raza could be due to the fact that the experimental set up was sitting in the lab for more than 3 years and the boiling characteristics could have been changed due to deposition of foreign matter resulting in destroying some of the nucleation sites. The other reason could be that Raza was not aware of the presense of a hysteresis effect resulting from fluids boiling on a heated surface. Hence, it was necessary to increase the inside evaporator resistance which would result in a lower heat transfer rate and a higher wall temperature for the evaporator. It was logical to use a correlation which would yield a lower heat transfer

coefficient in comparison to that of Sachs & Long. It was found that the correlation proposed by Davis & David yielded a good agreement of the simulated and the experimental results. However, it should be noted that their correlation was derived for water-steam mixtures flowing in a horizontal duct. The following is the correlation of Davis & David.

$$h_{tp} = 0.0333 \left(\frac{K_1}{D} \right) \left(\frac{D G_t}{\mu_{tp}} \right)^{0.87} Pr_1^{0.4} \quad (45)$$

where

$$\frac{1}{\mu_{tp}} = \frac{x}{\mu_v} + \frac{1-x}{\mu_l} \quad (46)$$

In the two-phase region the effective heat transfer coefficient was calculated from equation 44 with the power $p = 4.0$. Ali used a value of 2 for p for his model, which had to be changed to 4.0 to bring the wall temperatures in to agreement with the experimental values. The model used by the author in this region to combine h_b and h_{tp} is shown in figure 3.4. By using $p = 4$ in equation 44, the effective heat flux in the transitional boiling region follow very close to the true boiling curve, as can be seen from figure 3.4.

3.2.5.2 Pressure drop & void fraction calculations

The Lockhart & Martinelli (249) correlations were used to relate the two-phase frictional pressure drop to the single-phase frictional pressure drops for the vapour and liquid phases flowing separately.

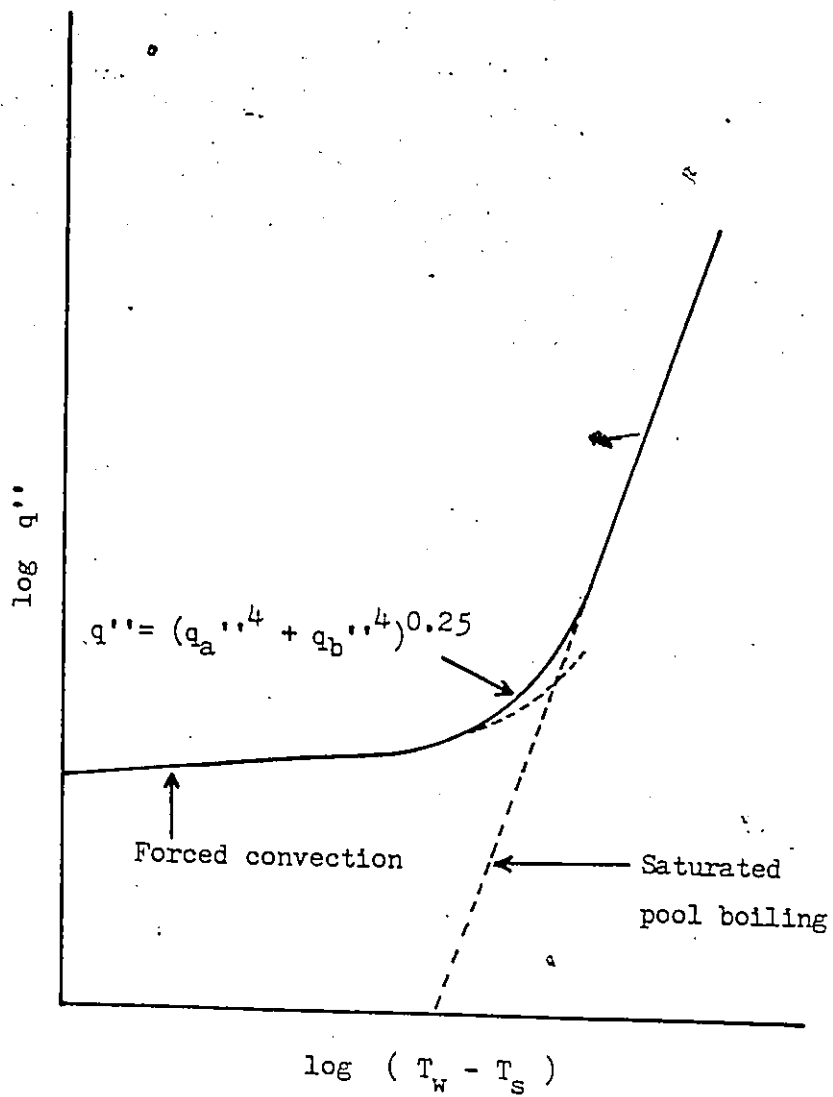


Figure 3.4 Author's proposed method of estimation effective heat flux in transitional boiling

8

$$\Delta P_{tp}/\Delta P_{lp} = \phi_{ij}^2 \quad \text{--- (47)}$$

The parameter ϕ is empirically determined in terms of Lockhart-Martinelli parameter X . Subscript $i j$ represents the conditions of liquid and the vapour phases respectively. Four such conditions exist for the mixture.

1. liquid phase laminar and vapour phase laminar
2. liquid phase laminar and vapour phase turbulent
3. liquid phase turbulent and vapour phase laminar
4. liquid phase turbulent and vapour phase turbulent

For case 1

$$X_{ll} = \left(\frac{1-x}{x} \right) \left(\frac{\rho_v}{\rho_l} \right) \left(\frac{\mu_l}{\mu_v} \right)^{0.5} \quad \text{--- (48)}$$

$$\phi_{ll} = (1 + X_{ll}^{-0.70})^{1.43} \quad \text{--- (49)}$$

For case 2

$$X_{lt} = 18.65 Re_v^{-0.4} \left(\frac{1-x}{x} \right) \left(\frac{\rho_v}{\rho_l} \right) \left(\frac{\mu_l}{\mu_v} \right)^{0.5} \quad \text{--- (50)}$$

$$\phi_{lt} = (1 + X_{lt}^{-0.56})^{1.80} \quad \text{--- (51)}$$

For case 3

$$X_{t1} = 0.054 \text{ Re}_1^{0.4} \left(\frac{1-x}{x} \right) \left(\frac{\rho_v}{\rho_l} \right) \left(\frac{\mu_l}{\mu_v} \right)^{0.5} \quad (52)$$

$$\phi_{t1} = (1 + X_{t1}^{-0.56})^{1.80} \quad (53)$$

For case 4

$$X_{tt} = \left(\frac{1-x}{x} \right)^{0.9} \left(\frac{\rho_v}{\rho_l} \right)^{0.5} \left(\frac{\mu_l}{\mu_v} \right)^{0.1} \quad (54)$$

$$\phi_{tt} = (1 + X_{tt}^{-0.49})^{2.05} \quad (55)$$

The hydrostatic pressure drop was calculated from the following relation

$$\Delta P_{h,tp} = \Delta z \rho_m \sin \gamma \quad (56)$$

where ρ_m is given by equation 3 and γ is the inclination angle from the horizontal.

The acceleration pressure drop was calculated (249) from the following relation.

$$\Delta P_a = G_t^2 / \rho_l \left[(1-x)^2 / (1-a) + (x^2/a) (\rho_l / \rho_v) - 1 \right] \quad (57)$$

The empirical correlation of Lockhart & Martinelli (66) was used for the calculation of the void fraction.

$$a = (1 + X^{0.8})^{-0.378} \quad (58)$$

3.2.7 Transition from two-phase forced convection to the liquid deficient region

The correlation of Lavin & Young (180), which was used by Ali, was used at first for the present studies. But it was found that it did not predict dryout even at lower static charges of approximately 25 %, which was clearly observed from the experimental data of Raza. Hence correlations proposed by Sthapak et al. (193), Roko et al. (194), Rhee & Young (181) and Jones & Altman (195) were tried. It was found that the correlation proposed by Rhee & Young predicted this transition point accurately. Dryout occurred if the quality exceeded the critical quality defined as

$$x_c = \left[\frac{We \cdot \sigma \cdot \rho_g}{G_L \mu_L \{Re_L / (1-\alpha)\}^{0.125}} \right]^{0.5} \quad \text{--- (59)}$$

where We is given by Rhee & Young

$$We = 1.87 \times 10^{-7} Re_g \left(\frac{\rho_l}{\rho_g} \right)^{0.6} \left(\frac{D}{0.0608} \right)^{-0.48} \quad \text{--- (60)}$$

where the diameter, D , is in feet.

3.2.7.1 Effective heat transfer coefficient in the dryout region

In this region the heat transfer coefficient decreases as the quality increases and reaches a minimum equal to that of single phase vapour when $x = 1$. The effective heat transfer for this region was calculated as follows;

$$h_{e3} = A h_v + B h_{e2} \quad \text{--- (61)}$$

$$\text{where } A = [(x - x_c)/(1 - x_c)]^m \quad \text{--- (62)}$$

$$B = (1 - A) \quad \text{--- (63)}$$

At first, the author used a linear variation ($m=1$) for the heat transfer coefficient for the dryout region. It was found that for $m = 1$; the simulated wall temperatures and the heat transfer rate did not agree with the corresponding experimental results. A value of 0.5 for m was tried, which yielded good agreement with the experimental data.

The single phase vapour heat transfer coefficient was calculated under the assumption that vapour alone occupied the tube.

3.2.7.2 Pressure drop calculations

The pressure drop in this region decreases as the quality increases. The following method was adopted from reference 9, where the frictional pressure drop was calculated as a function of x_c . The following is the relation used in the program to calculate the frictional pressure drop for the dryout region.

$$\Delta p_{f,do} = (C \Delta p_{f,g} + D \Delta p_{f,tp}) \quad \text{--- (64)}$$

$$\text{where } C = (x - x_c)/(1 - x_c) \quad \text{--- (65)}$$

$$D = (1 - C) \quad \text{--- (66)}$$

$$\text{at } x = x_c ; \Delta p_{f,do} = \Delta p_{f,tp}$$

and at $x = 1$; $\Delta p_{f,do} = \Delta p_{f,g}$

The pressure drop decreases linearly with the increase of x_c between the above two limits.

The hydrostatic and acceleration pressure drops were calculated in a manner similar to that described in section 3.2.5.2.

3.2.8 Single-phase vapour flow

3.2.8.1 Heat transfer calculation

When the magnitude of the quality reaches 1, the flow is all vapour and the heat transfer coefficient for this region was calculated from the relations described in section 3.2.2.1 by using vapour properties instead of the liquid properties.

3.2.8.2 Pressure drop calculation

Since the flow is all vapour, the pressure drop was calculated from the relations as indicated in section 3.2.2.2 by using vapour properties.

3.2.9. Calculations for evaporator second and successive tube rows

3.2.9.1 Estimation of the mass flow rate of the working fluid through the tubes

The source fluid temperature decreases as it passes over the evaporator tube rows and hence the evaporator tube rows are subjected

to unequal heating. Hence, for a given mass flow rate for the evaporator coil first row, the mass flow rate in the second tube row is calculated, which would be different from the first row (due to unequal heating). The flow rate in the second row is varied until the pressure drop for the second row equals to the pressure drop of row 1. Similar calculations may be done for the successive evaporator tube rows. The mass flow rate and the pressure drop are related by the following (see appendix G for the derivation).

$$\dot{m} = f(\Delta P)^n \quad \text{--- (67)}$$

where $n = 1.0$ for laminar flow

& $n = 0.571$ for turbulent flow

Equation 67 was used to calculate the mass flow rate in the second and the successive rows

$$\dot{m}_{2 \text{ row}} = \dot{m}_{1 \text{ row}} [\Delta P_{2 \text{ row}} / \Delta P_{1 \text{ row}}]^n \quad \text{--- (68)}$$

3.2.9.2 Heat transfer calculations

Since the source fluid temperature decreases as it passes over the evaporator tube rows, the downstream source fluid temperatures for the elements of the evaporator first row were stored in a dimensional form. The variation of the downstream source fluid temperature for the evaporator first row along the length of the tube is shown in figure

3.5 (assuming that the first row of the evaporator is not dried up). These temperatures were used as the source fluid temperatures for the calculations of the heat transfer rate for the evaporator second tube row. Similar calculations could be done for each successive evaporator tube rows.

3.3 SEPARATOR CALCULATIONS

When the fluid reaches the evaporator top, it goes into the vapour header and then through a separator from which the liquid is recirculated to the evaporator liquid header by gravity. The recirculation flow rate and the vapour header flow rate were calculated assuming perfect separation and equilibrium conditions.

Hence

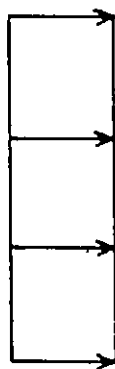
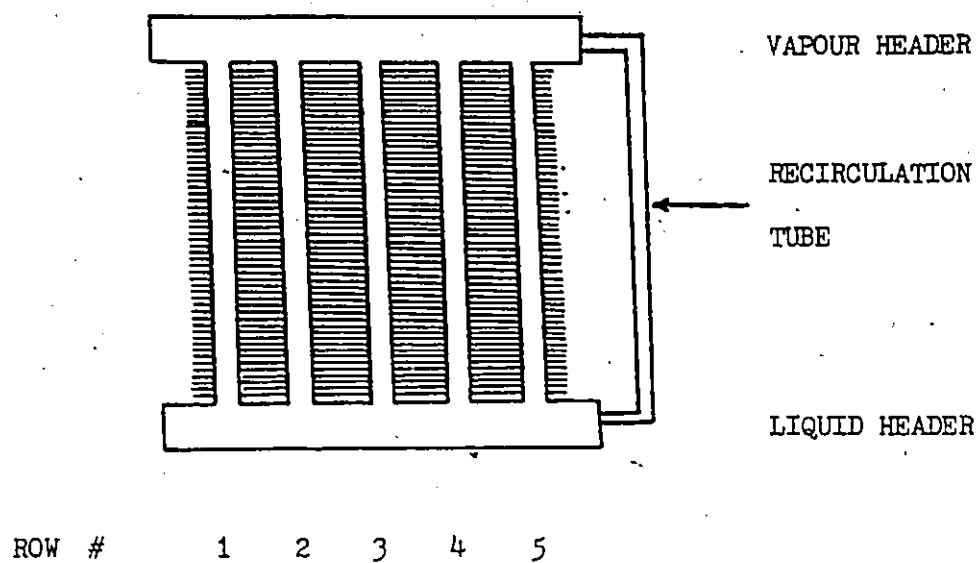
$$\dot{m}_r = \dot{m}_t (1 - x) \quad \text{--- (69)}$$

$$\dot{m}_v = \dot{m}_t x \quad \text{--- (70)}$$

and thus the quality of the vapour entering the vapour header is one.

3.4 VAPOUR HEADER CALCULATIONS

The vapour header interconnects the top of the evaporator to the top of the condenser. Heat loss from the vapour header was calculated from the following equations.



(a)



(b)

FIGURE 3.5 Temperature profiles for the source fluid approaching evaporator (a) row # 1 (b) row # 2

$$E = 1 - \exp(-L_{vh}/R \dot{m}_v C_{p_v}) \quad \text{--- (71)}$$

$$Q = E \dot{m}_v C_{p_v} (T_f - T_s) \quad \text{--- (72)}$$

where E , L_{vh} , R , \dot{m}_v , C_{p_v} , T_f and T_s are the effectiveness, length of the vapour header (m), the resistance per unit length (m K/W), the mass flow rate of vapour (kg/s), the specific heat of the vapour (kJ/kg K), the inlet working fluid temperature (C) & the surrounding temperature (C) respectively. The amount of insulation on the vapour header must be accounted for by the user when he specifies the external thermal resistance per unit length, R , (m K/W) for the vapour header.

In order to select suitable properties for the pressure drop calculations, an average state in the header was estimated by calculating an exit quality (or superheat) assuming no pressure drop. This average state was used to calculate the frictional and hydrostatic pressure drops for the vapour header either by the method discussed in section 3.2.5.2, when the working fluid is in a wet state at the exit of the vapour header, or by the method discussed in section 3.2.8.2, when the working fluid is in single phase vapour state. The entrance and exit pressure losses and the pressure drop due to bends and elbows were also calculated using velocity head factors for each which are incorporated in the program.

3.5 CONDENSER

3.5.1 Condensation

If the vapour is in a superheated state at the entrance to the condenser, vapour desuperheating takes place and the wall temperature is calculated using single-phase vapour heat transfer correlations. When the wall temperature drops below the saturation temperature of the vapour, heat transfer takes place by condensation.

Numerous correlations were presented in table 2.5 to determine the condensation coefficient. The correlation proposed by Ali & McDonald (228) was used in the determination of the condensation coefficient. The following is their correlation.

$$\bar{h}_c = B_1 \left[\frac{K_1^3 \rho_1^2 h_{fg} g}{\mu_1 D (T_s - T_w)} \right]^{1/4} \quad \text{--- (73)}$$

$$\text{for } 0^\circ < \beta < 40^\circ ; \quad B_1 = 0.727$$

$$\text{for } 40^\circ < \beta < 90^\circ ; \quad B_1 = 0.727 \left[\cos \left(\frac{\beta - 40}{50} \right) 90 \right]^{0.25}$$

For partly flooded condensers, the heat transfer and the pressure drop were calculated as indicated in section 3.2.2.2.

3.5.2 Calculations for condenser tube rows

The sink fluid temperature increases as it passes over the condenser tube rows, as shown in figure 3.6 (assuming saturated state of vapour in the vapour header), and hence the tube rows are subjected to unequal heating. The vapour flow rate for the condenser tube rows were calculated as follows:

The flow rate of vapour for one tube in the condenser first row was initially estimated by taking the average of the total vapour ($\dot{m}_{t,e}$) generated in the evaporator.

$$\dot{m}_{l,row} = \dot{m}_{t,e} / (N * M) \quad \text{--- (74)}$$

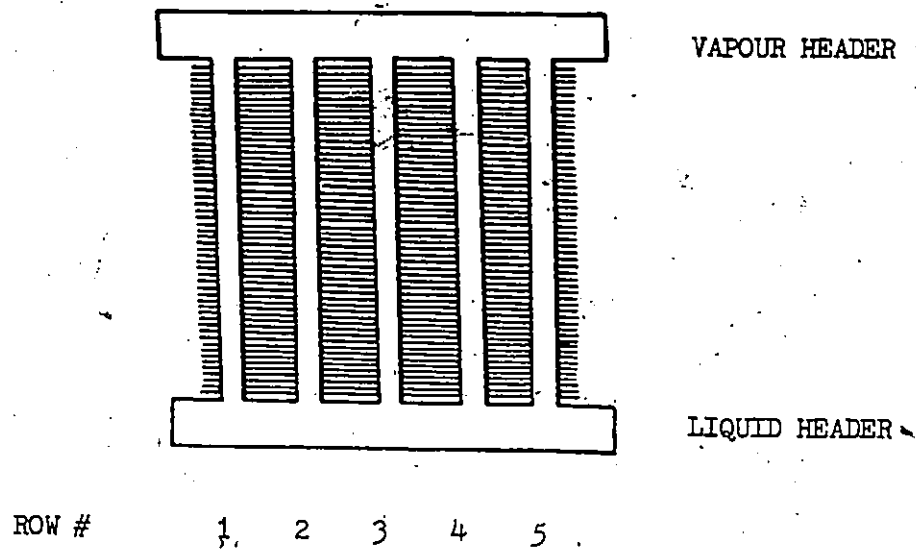
where N and M are the number of tube rows and tubes per row for the condenser.

Calculations for the heat transfer were made for the condenser first row by using this vapour flow rate by the following equation

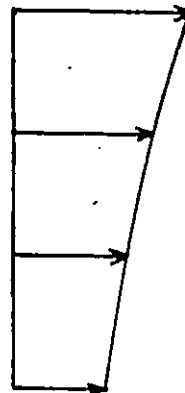
$$Q_{l,row} = \dot{m}_{l,row} * (h_{c,in} - h_f) \quad \text{--- (75)}$$

where $h_{c,in}$ and h_f are the enthalpies (Btu/lb) at the top and the bottom of the condenser.

The cold fluid (sink) temperature approaching this row and the downstream sink temperatures were calculated as follows.



(a)



(b)

FIGURE 3.6 Temperature profiles for the sink fluid approaching condenser (a) row # 1 (b) row # 2

$$T_a = T_s - Q_{1,row} / (\dot{m} Cp)_{air} * E \quad \text{--- (76)}$$

$$T_d = T_a + Q_{1,row} / (\dot{m} Cp)_{air} \quad \text{--- (77)}$$

This downstream temperature was used to estimate the rate of heat transfer for the second row by initially assuming $h_{c,2} = h_{c,1}$.

$$Q_{2,row} = E (\dot{m} Cp)_{air} (T_s - T_d) \quad \text{--- (78)}$$

$$\dot{m}_{2,row} = Q_{2,row} / h_{fg} \quad \text{--- (79)}$$

By knowing the value for $Q_{2,row}$, the condenser wall temperature was calculated followed by the condensation coefficient $h_{c,2}$. This new value of condensation coefficient was used to calculate the overall "U" value and hence the effectiveness, E. Convergence was obtained for $h_{c,2}$ and equations 78 and 79 were used to calculate the heat transfer and the mass flow rate for the vapour.

Similar calculations were done for each of the successive condenser rows. The total flow rate of the vapour through the condenser ($\dot{m}_{t,c}$) was calculated and it was then compared to $\dot{m}_{t,e}$. If both these flow rates did not match, the flow rate for the condenser first row was varied until $\dot{m}_{t,c}$ equalled to $\dot{m}_{t,e}$. The flow diagram shown in figure 3.7 explains this logic with the for a condenser consisting of two rows of tubes.

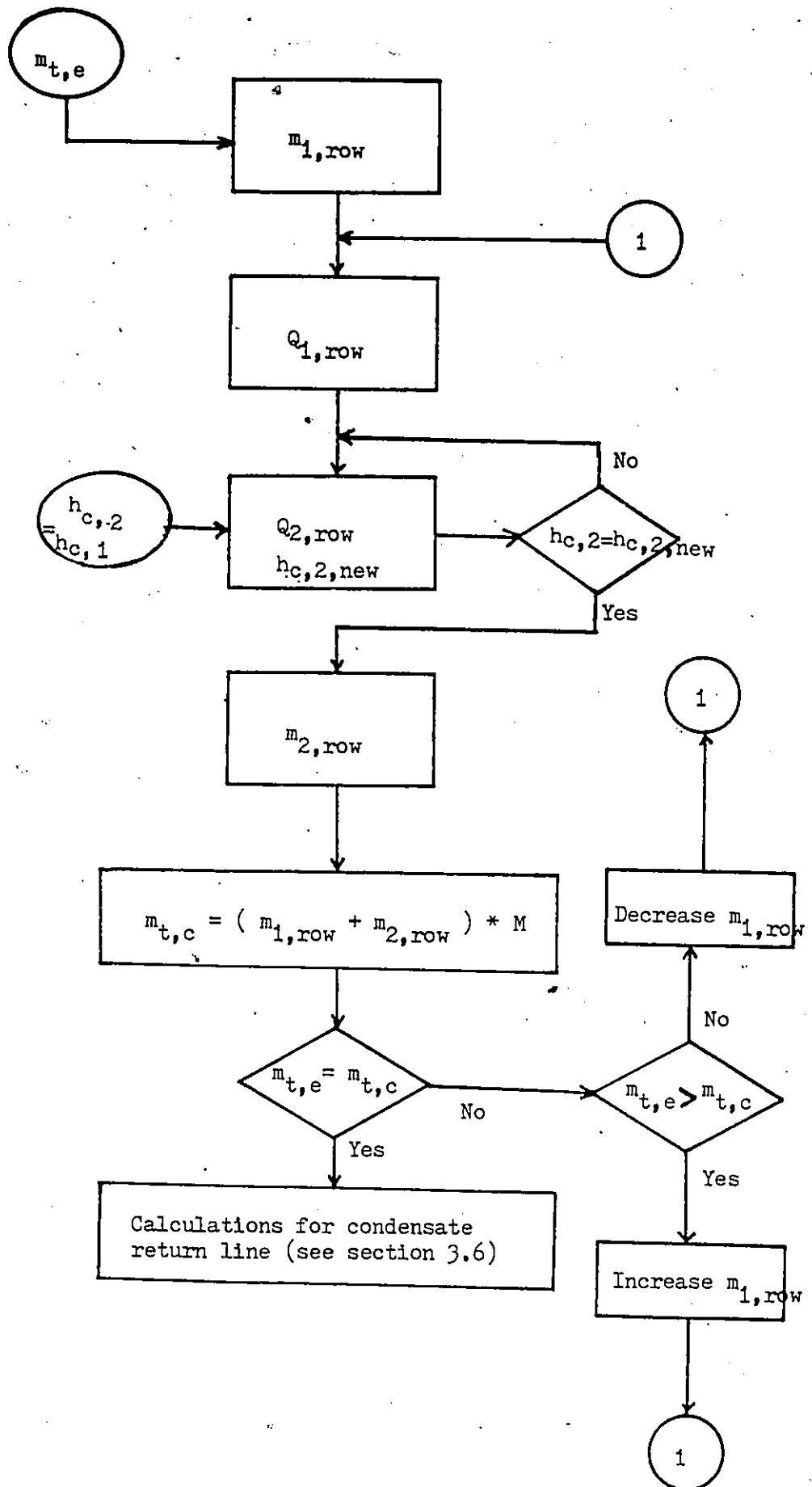
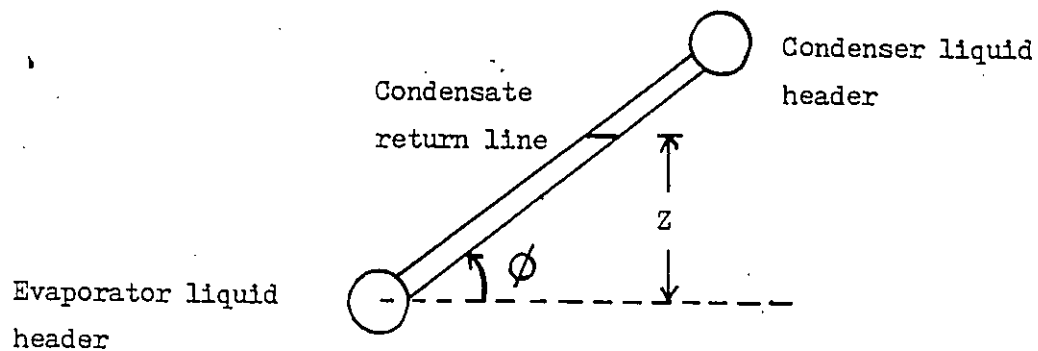


FIGURE 3.7 Flow diagram for condenser calculations

3.6 CONDENSATE RETURN LINE

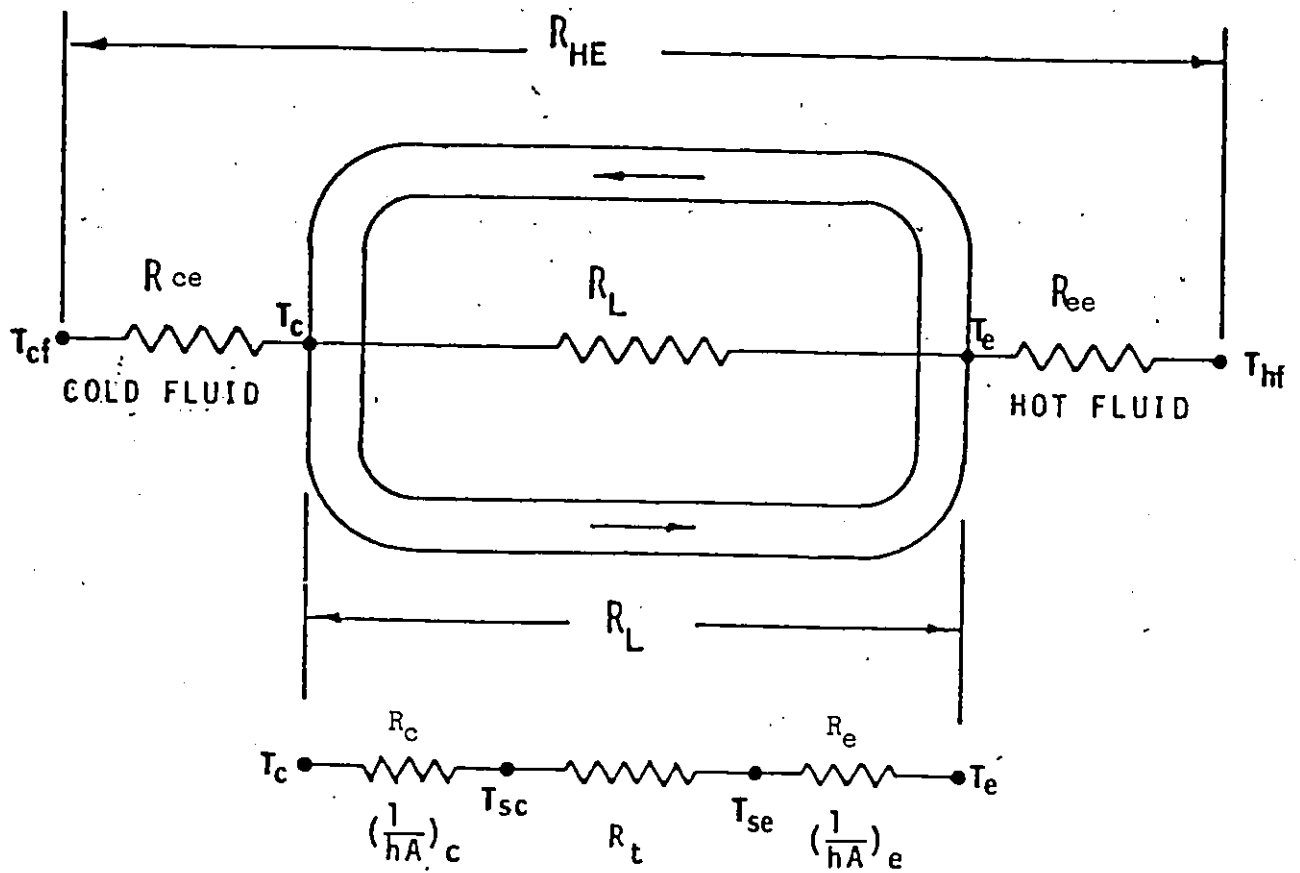
The condensate coming out of the condenser liquid header flows into the condensate return line. The amount of insulation on the condensate return line may be accounted for by the user when he specifies the external thermal resistance per unit length (m K/W) for the condensate return tube. The heat transfer was calculated in a manner similar to that discussed in section 3.4 for the vapour header. The frictional pressure drop in the condensate return line was calculated by the method described in 3.2.8.2. The configuration model used in the program for the condensate return line is shown below. The elevation of the liquid vapour interface in the condensate return line can be calculated from the following, by assuming a negligible pressure drop in the condenser.



$$(P_{\text{evap},b} - P_{\text{cond},b}) = \rho g Z + \left[\frac{f (Z/\sin \phi) + K_1}{D} \right] (V^2/2g)$$

--- (80)

For flooded condensers, different lengths and diameters should be used in equation 80 to calculate Z.



$$\text{HEAT EXCHANGER CONDUCTANCE } (UA)_{HE} = R_{HE}^{-1}$$

$$\text{THERMOSIPHON LOOP CONDUCTANCE } (UA)_L = R_L^{-1}$$

FIGURE 3.8 : ELECTRICAL ANALOG OF IDEALIZED THERMOSIPHON LOOP

The flow is all liquid in the condensate return line , and the fluid then enters the evaporator liquid header, thus completing a cycle.

3.7 PERFORMANCE RATING

Figure 3.8 shows an idealized electrical analog of a two-phase thermosiphon loop heat exchanger and identifies the various resistances.

The performance for a thermosiphon loop heat exchanger is characterized by defining its effectiveness as the ratio of the heat transferred to the maximum possible heat transfer.

The effectiveness of a heat exchanger is dependent on the outside source, outside sink and the loop resistances which are identified as R_{ee} , R_{ce} and R_L respectively.

In order to isolate the effect of the outside resistances in studying the effect of variable parameters on the performance of the loop it is convenient to concentrate on $R_L = (UA)$, and ignore R_{ee} and R_{ce} . The overall loop conductance (UA) for a two-phase thermosiphon heat exchanger is defined as the amount of heat transferred per unit temperature difference between the source and the sink fluids.

The loop resistance has three components: evaporator, condenser and the transport resistances given by R_e , R_c and R_t respectively. The

resistance of the evaporator and condenser can be expressed in terms of their average heat transfer coefficients which are defined as the heat transfer rate per unit tube inner surface area and per unit temperature difference between the average wall temperature and the saturation temperature in the evaporator vapour header.

In some cases it is useful to study the effect of the local heat transfer coefficient.

CHAPTER IV

RESULTS & DISCUSSIONS

4.1 INTRODUCTION

This chapter presents a comparison of experimental and corresponding simulated results for a number of experimental investigations (12, 15, 17, 23) which were carried out on three very different systems.

R-11 was used by reference 12, 15 & 17; and R-113 was used by reference 23 as the working fluid for their respective systems , and the systems were oriented in such a way that they could only operate in an unidirectional mode.

In all of the following figures, the curves shown represent the simulated values whereas the points shown represent experimental results.

4.2 COMPARISON WITH WATER-TO-WATER SYSTEM

For this study, the simulated values of the heat transfer rate, average heat transfer coefficients, conductance etc. have been

compared with the corresponding results obtained by Raza (12).

The system used in this study was a well instrumented three tube evaporator and three tube condenser system especially designed for studying thermosiphon loop behaviour. Each of the three coplanar tubes in each heat exchanger was independently water jacketed, 0.61 m long copper tubes. Each heat exchanger could be translated vertically, rotated in a plane about a horizontal axis, and be inclined from the vertical with its header remaining horizontal. The inside diameter of the evaporator and condenser coil tubes was 7.9 mm and 4.8 mm respectively. The tube spacing in the evaporator coil was 8.0 cm. A complete description of the system is given by Raza and McDonald (13).

Figures 4.1 and 4.2 show, respectively, a comparison of the total heat transfer rate and the average evaporator inner tube surface heat transfer coefficient versus the evaporator static charge when all three evaporator tubes are vertical and equally charged. The system was run with source to sink temperature differences of 20 C and 10 C and with an average source/sink temperature of 30 C. Both the figures show a good agreement with the experimental results.

A comparison of plots of the heat transfer rate and the average evaporator inner tube surface heat transfer coefficient versus evaporator static charge when the tubes are inclined from the vertical and are equally heated is shown in figures 4.3, 4.4, 4.5 & 4.6 for 20

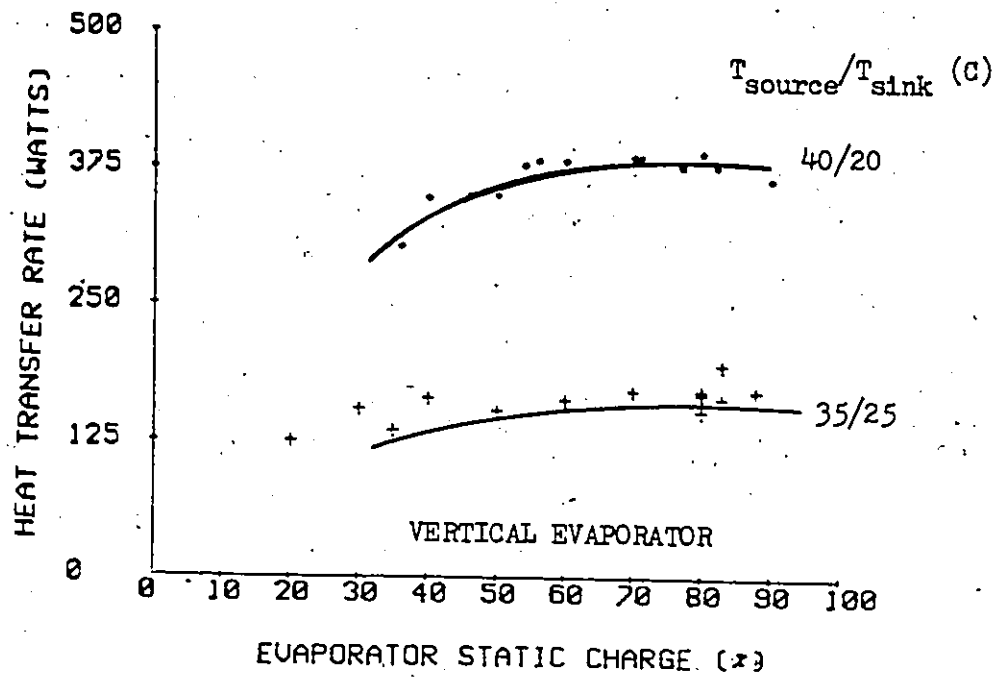


FIG. 4.1 HEAT TRANSFER RATE VERSUS STATIC CHARGE FOR
VERTICAL EVAPORATOR, $T_{\text{mean}} = 30^\circ\text{C}$

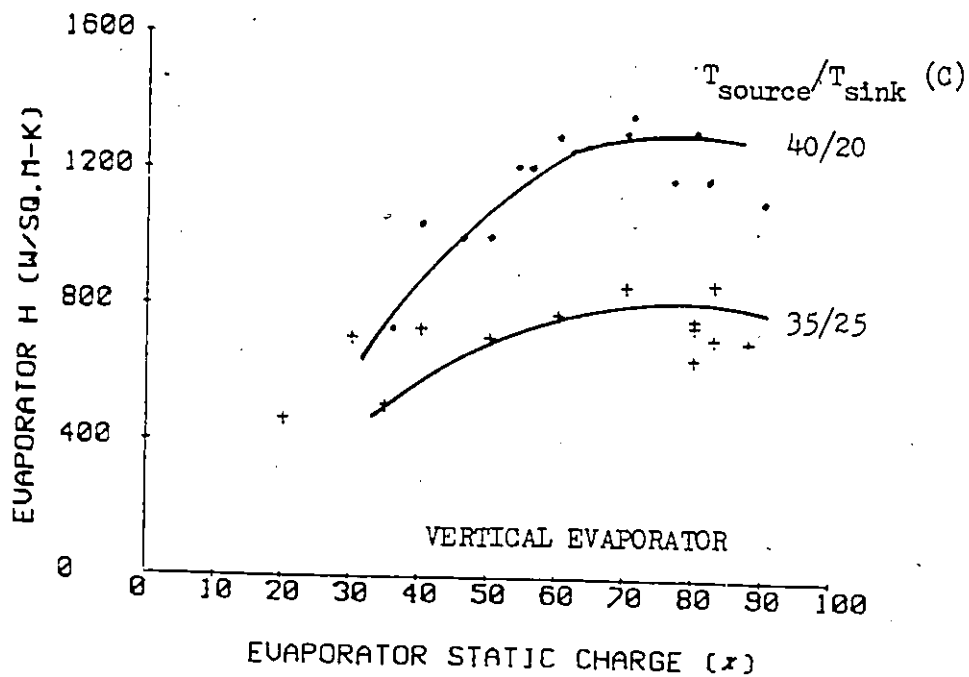


FIG. 4.2 AVERAGE EVAPORATOR HEAT TRANSFER COEFFICIENT
VERSUS STATIC CHARGE, $T_{\text{mean}} = 30^\circ\text{C}$

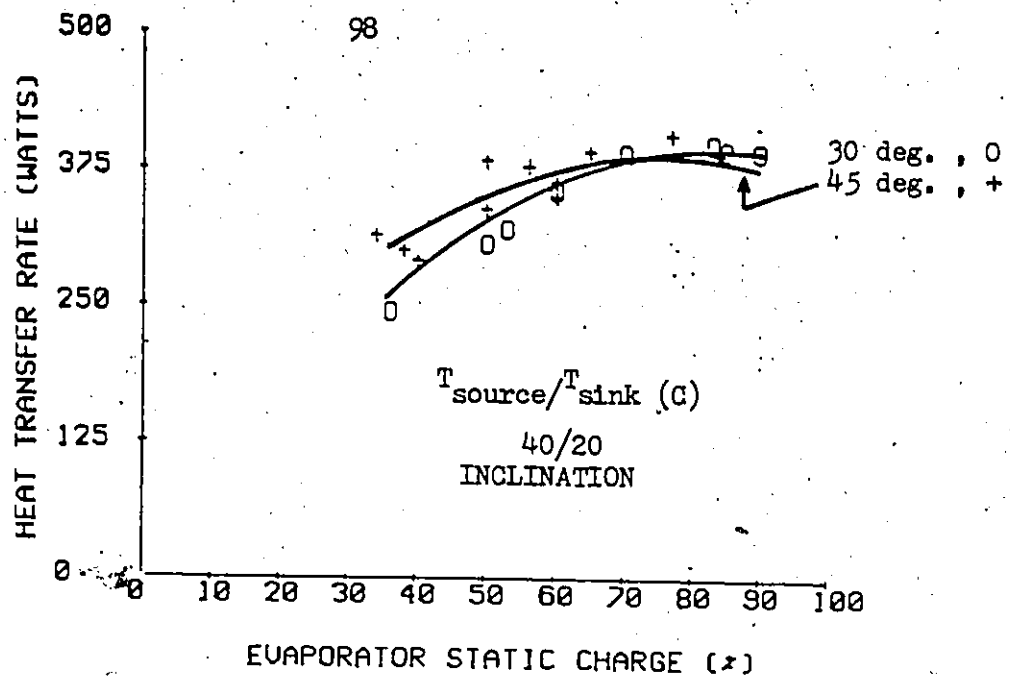


FIG. 4.3 HEAT TRANSFER RATE VERSUS STATIC CHARGE FOR INCLINED EVAPORATOR TUBES, $T_{\text{mean}} = 30 \text{ C}$

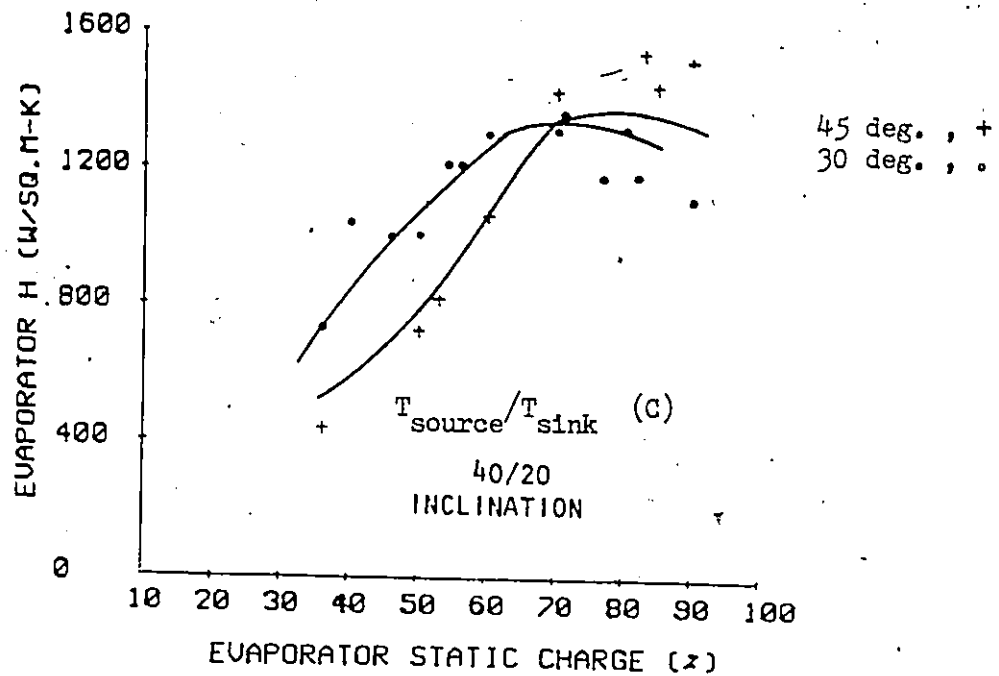


FIG. 4.4 AVERAGE EVAPORATOR HEAT TRANSFER COEFFICIENT VERSUS STATIC CHARGE FOR INCLINED EVAPORATOR TUBES, $T_{\text{mean}} = 30 \text{ C}$

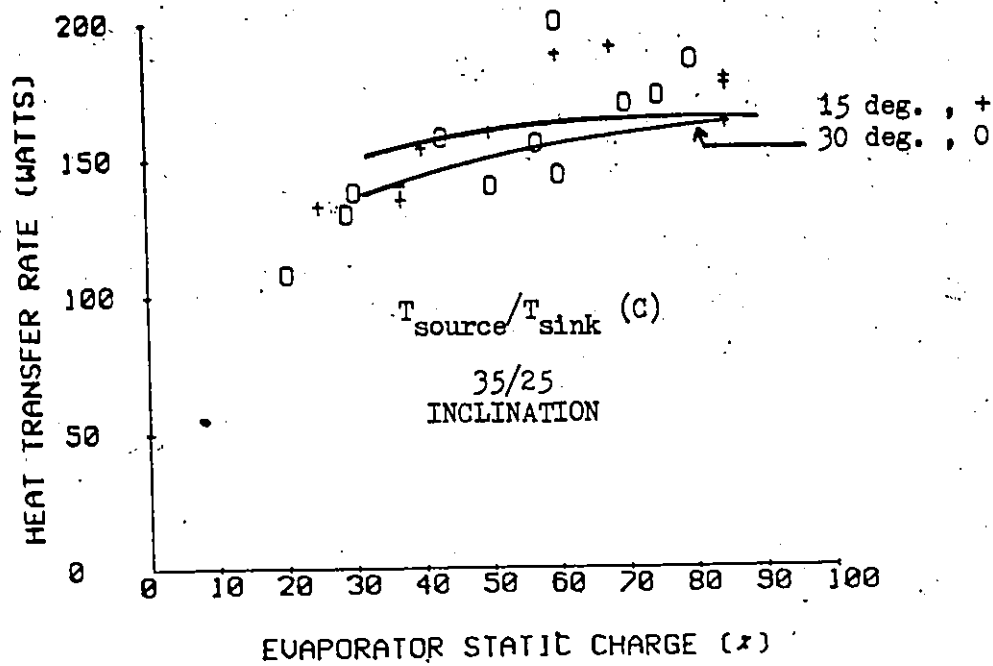


FIG. 4.5 HEAT TRANSFER RATE VERSUS STATIC CHARGE FOR INCLINED EVAPORATOR TUBES, $T_{\text{mean}} = 30\text{ C.}$

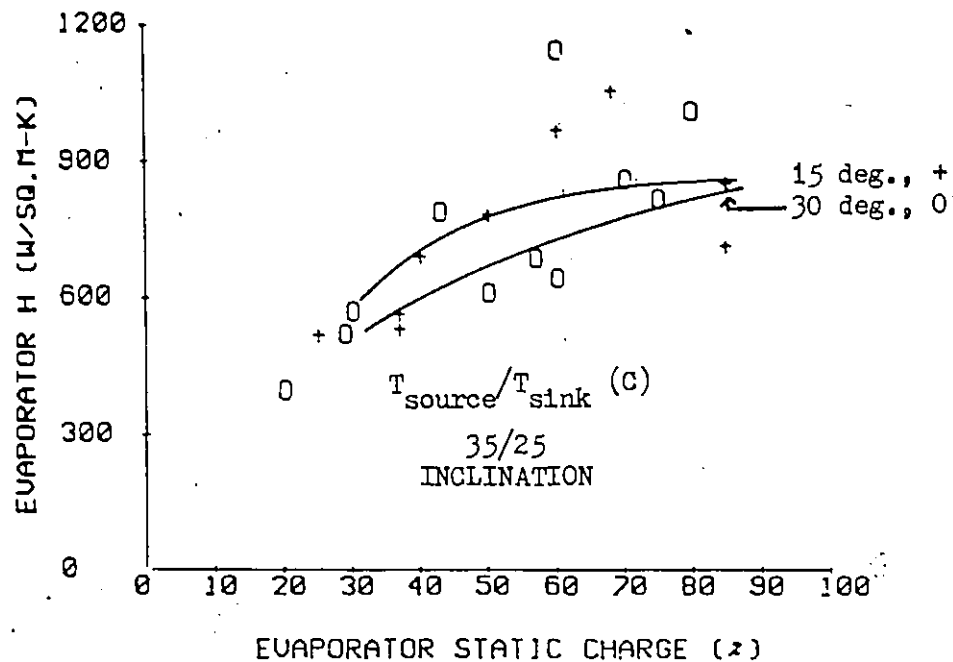


FIG. 4.6 AVERAGE EVAPORATOR HEAT TRANSFER COEFFICIENT VERSUS STATIC CHARGE FOR INCLINED EVAPORATOR TUBES, $T_{\text{mean}} = 30\text{ C.}$

C and 10 C source to sink temperature differences. For the 20 C source to sink temperature difference tests the evaporator tubes were inclined at 30 and 45 degrees from the vertical. For the 10 C source to sink temperature difference they were inclined at 15 and 30 degrees from the vertical. The simulated results accurately predicted the effect of tube dryout for source and sink temperature of 40C & 20C with increasing angle for charges below 70 % (figures 4.3 & 4.4), but failed to predict any enhancement above this value. Due to the scatter in the experimental data for 35C/25C (figures 4.5 & 4.6), the simulated data does not agree very well with the experimental data. The improved performance obtained experimentally for higher charges with increased angle of inclination, is a result of the reduced hydrostatic head in the evaporator tube and causes a reduction in the evaporator inlet pressure and hence promotes boiling.

The performance of the evaporator when rotated from the vertical, and hence with unequal charges present in each evaporator tube, are shown in figures 4.7, 4.8, 4.9 and 4.10 for the 40C/20C and 35C/25C source/sink temperatures. Rotation angles of 15, 30 & 45 degrees from the vertical resulted in static charge differences between adjacent tubes of 3 %, 8 % and 13 % respectively. The results are plotted against the average static charge (the value for the center tube). In this case it may be seen that the experimental results show little effect as compared to the inclination of the

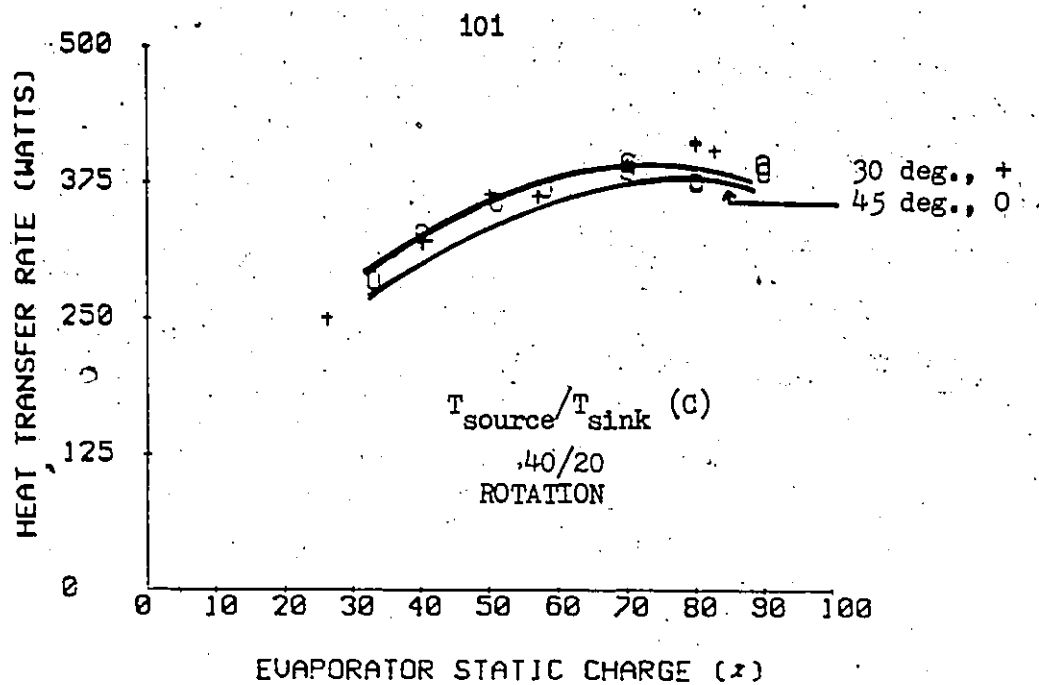


FIG. 4.7 HEAT TRANSFER RATE VERSUS STATIC CHARGE FOR
ROTATED EVAPORATOR TUBES, $T_{\text{mean}} = 30$ C.

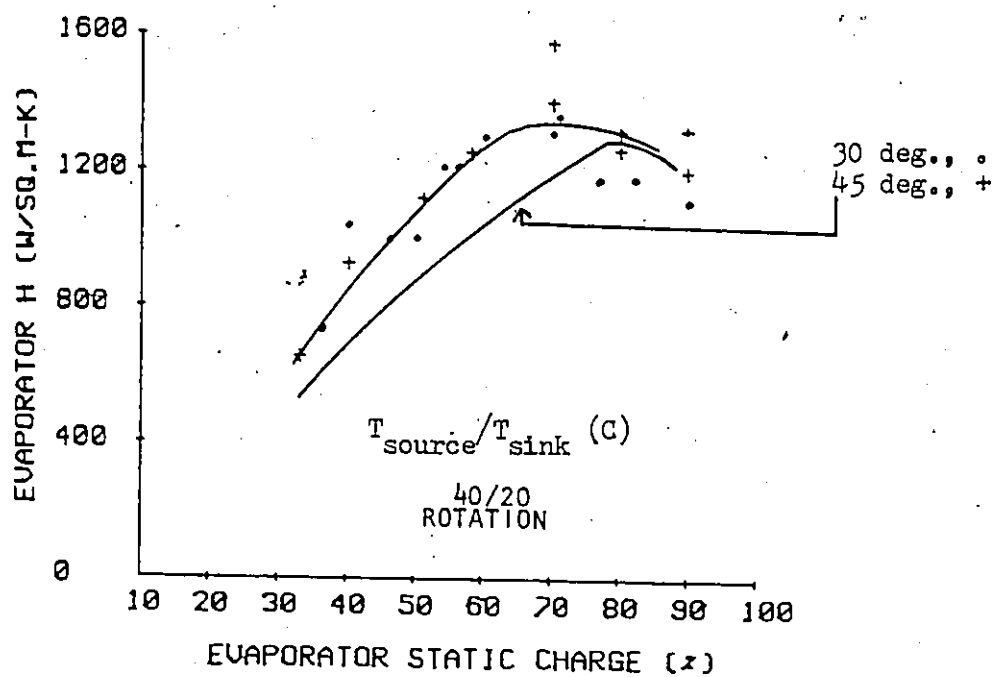


FIG. 4.8 AVERAGE EVAPORATOR HEAT TRANSFER COEFFICIENT
VERSUS STATIC CHARGE FOR ROTATED EVAPORATOR TUBES, $T_{\text{mean}} = 30$ C.

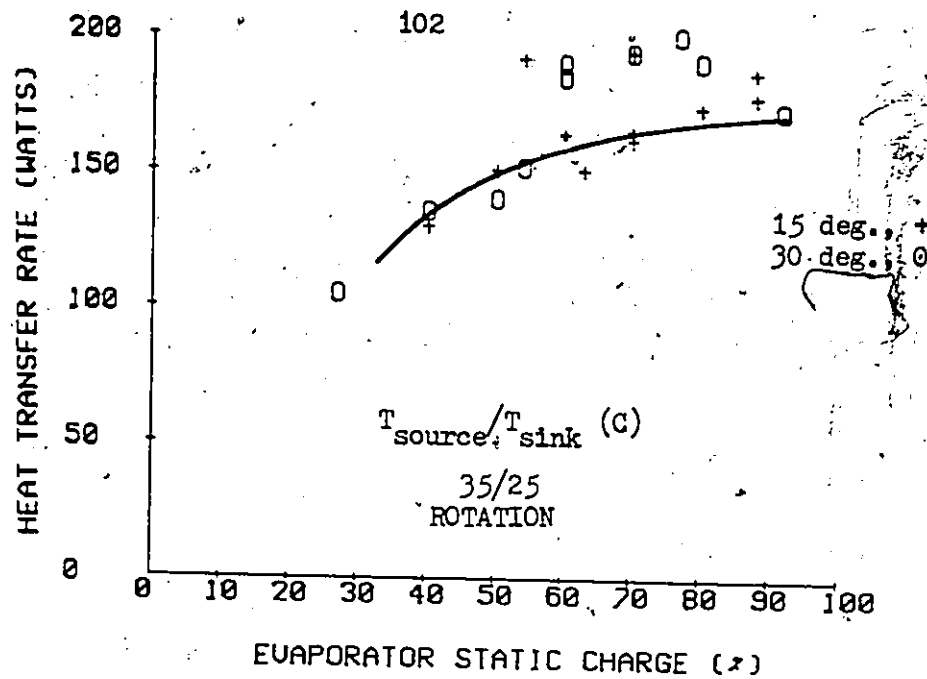


FIG. 4.9 HEAT TRANSFER RATE VERSUS STATIC CHARGE FOR ROTATED EVAPORATOR TUBES, $T_{mean} = 30C$.

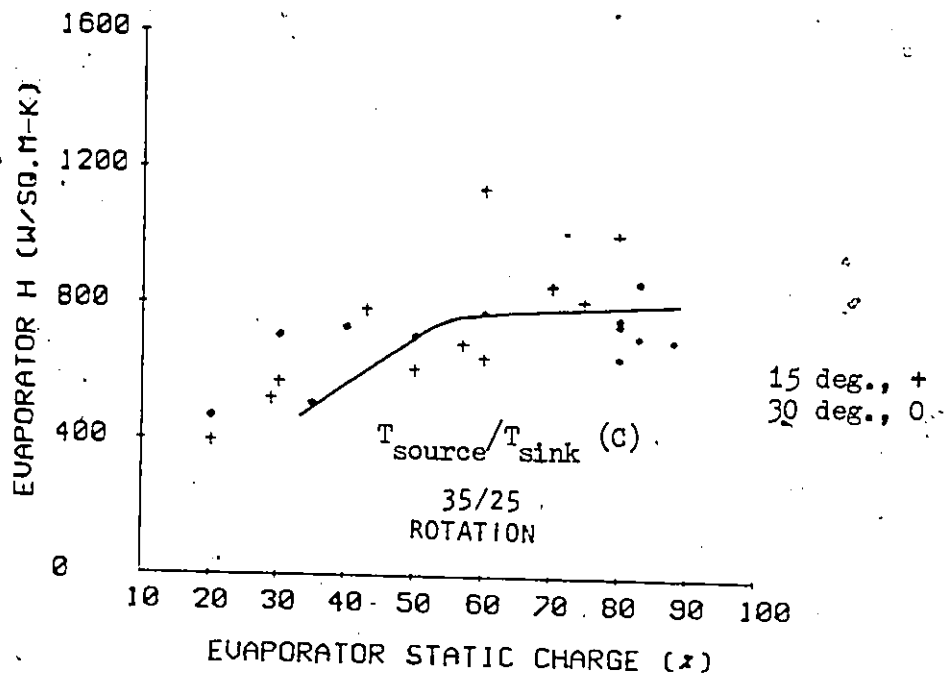


FIG. 4.10 AVERAGE EVAPORATOR HEAT TRANSFER COEFFICIENT VERSUS STATIC CHARGE FOR ROTATED EVAPORATOR TUBES, $T_{mean} = 30C$.

tubes. The simulated results for 40C/20C (figures 4.7 & 4.8)* indicate that dryout should be much more severe than was found. The only explanation offered for this behaviour is that liquid carried into the evaporator vapour header from the tube with greatest charge may have splashed into the top of the tubes with less charge which would otherwise have dried out. For 35C/25C source/sink temperatures, the simulated data for 15 deg. and 30 deg. is shown by a single curve, since there was very little difference in the values for 15 deg and 30 deg. rotation. The agreement is better for this case.

Figure 4.11 shows a comparison of the values of the overall loop conductance "U" and for the mean wall superheat. For the loop conductance, curves 1 and 2 are for source water jacket temperatures of 40 C and (44, 40, 36 C) respectively, are in very good agreement but curve 3, for source and sink temperatures of 35C and 25C respectively, falls along the lower edge of the experimental scatter. The temperature differences are also in reasonably good agreement with the simulated values.

Figure 4.12 shows the average heat transfer coefficient versus the static charge. The simulated curves are in good agreement with the experimental points within the accuracy of the data. Curve "2" shows the variation of the average evaporator heat transfer coefficient versus static charge for source fluid temperatures of 44, 40 & 36 C with a condensing temperature of 20C. Although, this performance is

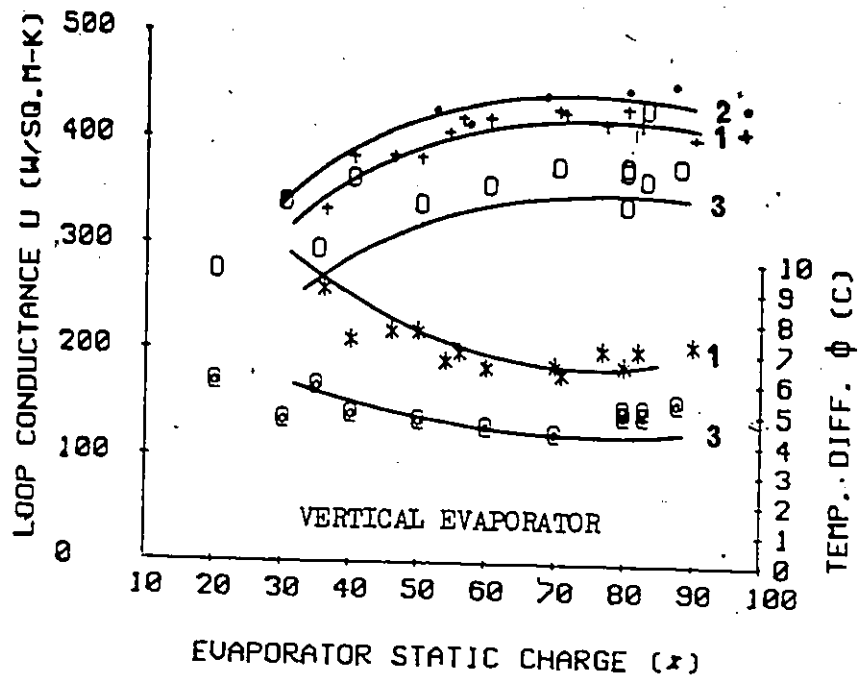


FIG. 4.11 EFFECT ON CONDUCTANCE & WALL SUPERHEAT OF STATIC CHARGE & SOURCE-SINK TEMPERATURES. (1) 40 C SOURCE & 20 C SINK; (2) 44 C, 40 C, 36 C SOURCE & 20 C SINK & (3) 35 C SOURCE & 25 C SINK.

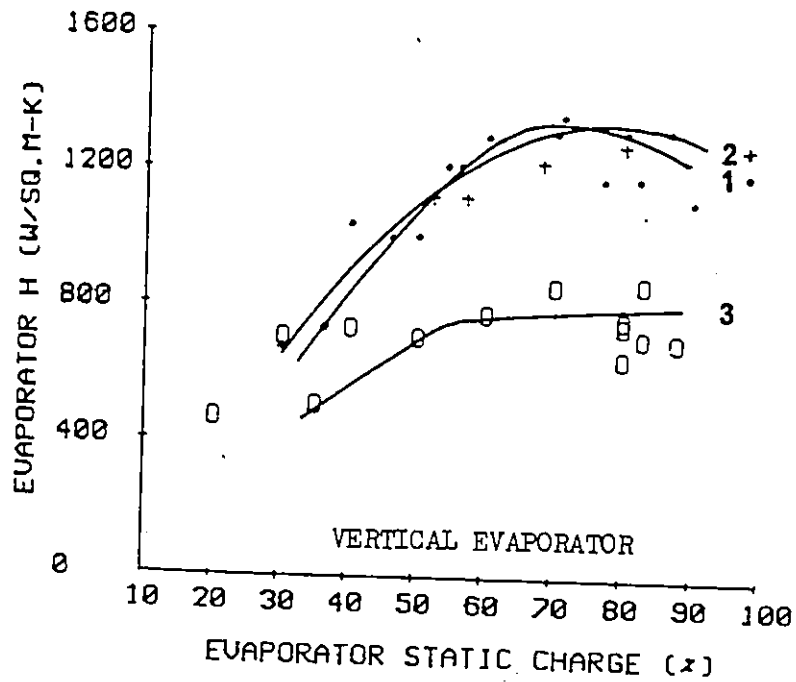


FIG. 4.12 EFFECT OF SOURCE-SINK TEMPERATURES ON THE AVERAGE EVAPORATOR HEAT TRANSFER COEFFICIENTS

very similar to curve "1", in which all three evaporator tubes ~~are~~ equally heated at 40 C, the performance of the individual tubes varies a great deal. Figure 4.13. shows that the simulated curves for the individual tubes were, initially, not in good agreement with the experimental data although the trends are similar. This discrepancy was caused by the hysteresis effect which occurs in real systems due to the fact that a greater wall superheat is required to initiate boiling than to maintain it. As a result, the actual performance of a surface when subjected to low wall superheats is dependent on its past history. To illustrate the effect of this phenomenon the simulation program was rerun with the additional constraint that the fluid could not boil until the wall superheat was equal to 13.3 C (24 F). The result obtained is plotted in figure 4.14. This simulation was in much better agreement with the experimental data. It indicated that the fluid would not boil in the coolest tube (subjected to a source fluid temperature of 36 C), since wall superheats of only 10-12 C occurred in that tube. The fact that the fluid did boil in the coolest tube for some of the experimental runs is most likely due to the fact that for those tests the thermal history of the coolest tube was not noted and hence nucleation sites may have already been activated. As a result, the experimental behaviour should fall somewhere between those predicted in figure 4.13 and figure 4.14. With this understanding it may be seen that the simulation program can account for the measured behaviour.

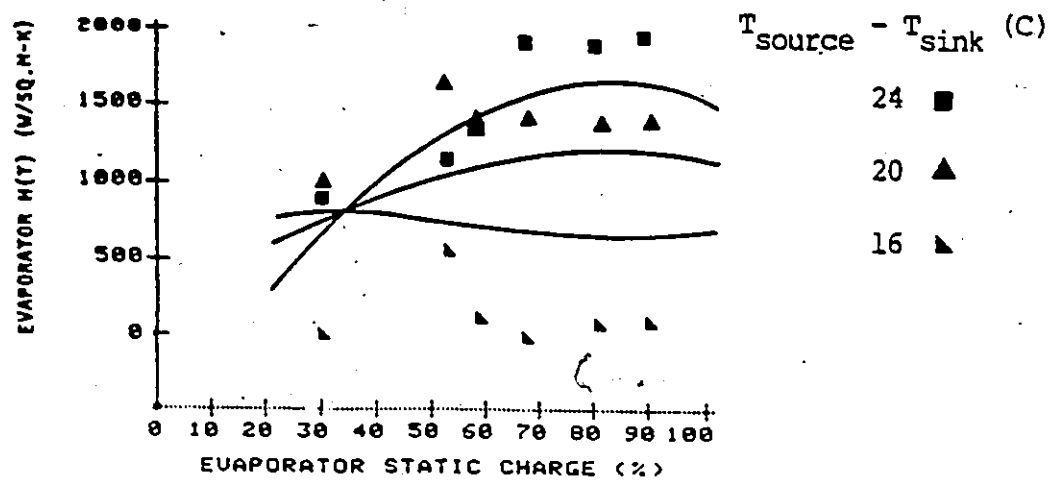


Figure 4.13

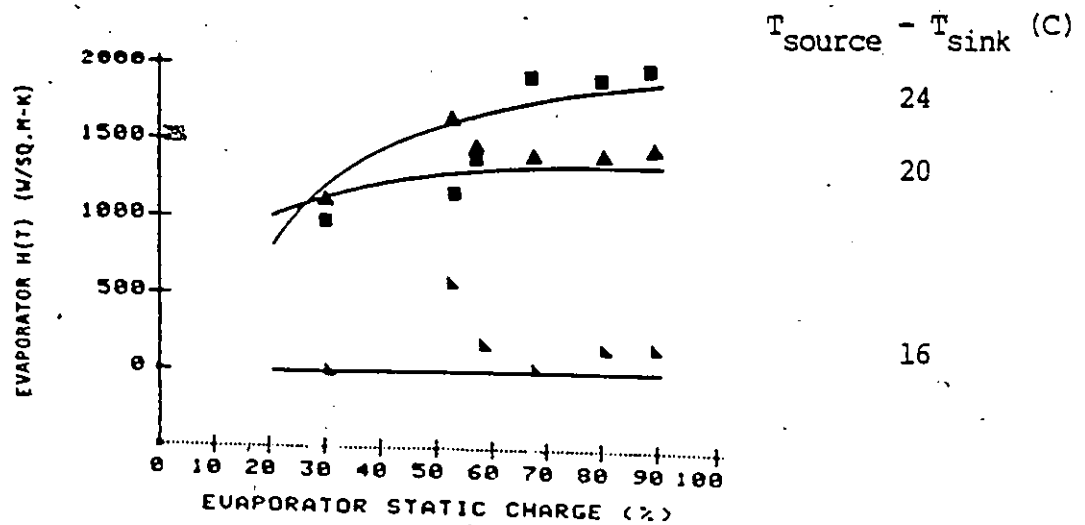


Figure 4.14

Average evaporator heat transfer coefficient in individual tubes versus static charge with source to sink temperature difference of 24, 20 and 16 C and $T_{sink} = 20$ C.

4.3 COMPARISON WITH AIR-TO-AIR SYSTEM

For this study, the simulated effectiveness of an air-to-air thermosiphon loop heat exchanger is compared with the corresponding experimental results obtained by Kosnik & Bertoni (15) and Stauder (17).

The air-to-air heat exchanger system of Kosnik & Bertoni and Stauder consisted of a number of thermosiphon loops mounted in a counter flow configuration in the warm and cold air ducts. A schematic diagram of the system is shown in figure 2.4. The heat exchanger coils were fabricated using commercially manufactured finned tube evaporator and condenser coils with wavy aluminum plate fins at a spacing of 12 fins per inch (473/m). The condenser and the evaporator coils each consisted of 8 rows of tubes of 7.9 mm I.D. each with its own separate liquid and vapour header. The evaporator tubes were vertical and 0.915 m long. The condenser tubes were inclined at 45 degrees and were 0.305 m long. The evaporator and the condenser coils had the same face area and were installed in warm and cold ducts having the same air mass flow rates. A complete description of this system is given by Kosnik & Bertoni (15) and Stauder (17). Kosnik & Bertoni used 4 loops for their study of the effect of the static charge on the effectiveness of the system for two different overall temperature differences and two air flow rates. Stauder studied the performance of one and four loop

systems as a function of the hot to cold duct air temperature difference for three different duct air mass flow rates. This temperature difference was first increased (heating process) then decreased (cooling process) in order to study the hysteresis effect caused by the boiling characteristics of R-11. It has been shown by Abdelmessih et. al (120) and others that it requires a much greater wall superheat to initiate nucleate boiling than it does to sustain it. Once initiated, this large temperature difference is not required.

Figure 4.15 shows the performance of single loop systems for both heating and cooling for a coil face velocity of 2.2 m/s. The simulated results are in excellent agreement with the experimental data, both for the heating and cooling processes. The effectiveness is considerably lower for the heating portion of the curve since a larger amount of wall superheat is needed to initiate bubble growth for the system startup. This initiation of the bubble growth for this system started when the overall temperature difference exceeds approximately 13C. When the temperature difference exceeds approximately 40C, the simulated performance curves for heating and cooling merge and become essentially constant over the range studied (up to 65C). When the overall temperature difference is decreased from 40C the effectiveness of the system remains high since the nucleation sites which were established at the higher temperatures are initially not destroyed. Eventually, as the overall temperature difference continues to decrease, some of the nucleation sites are deactivated. All were

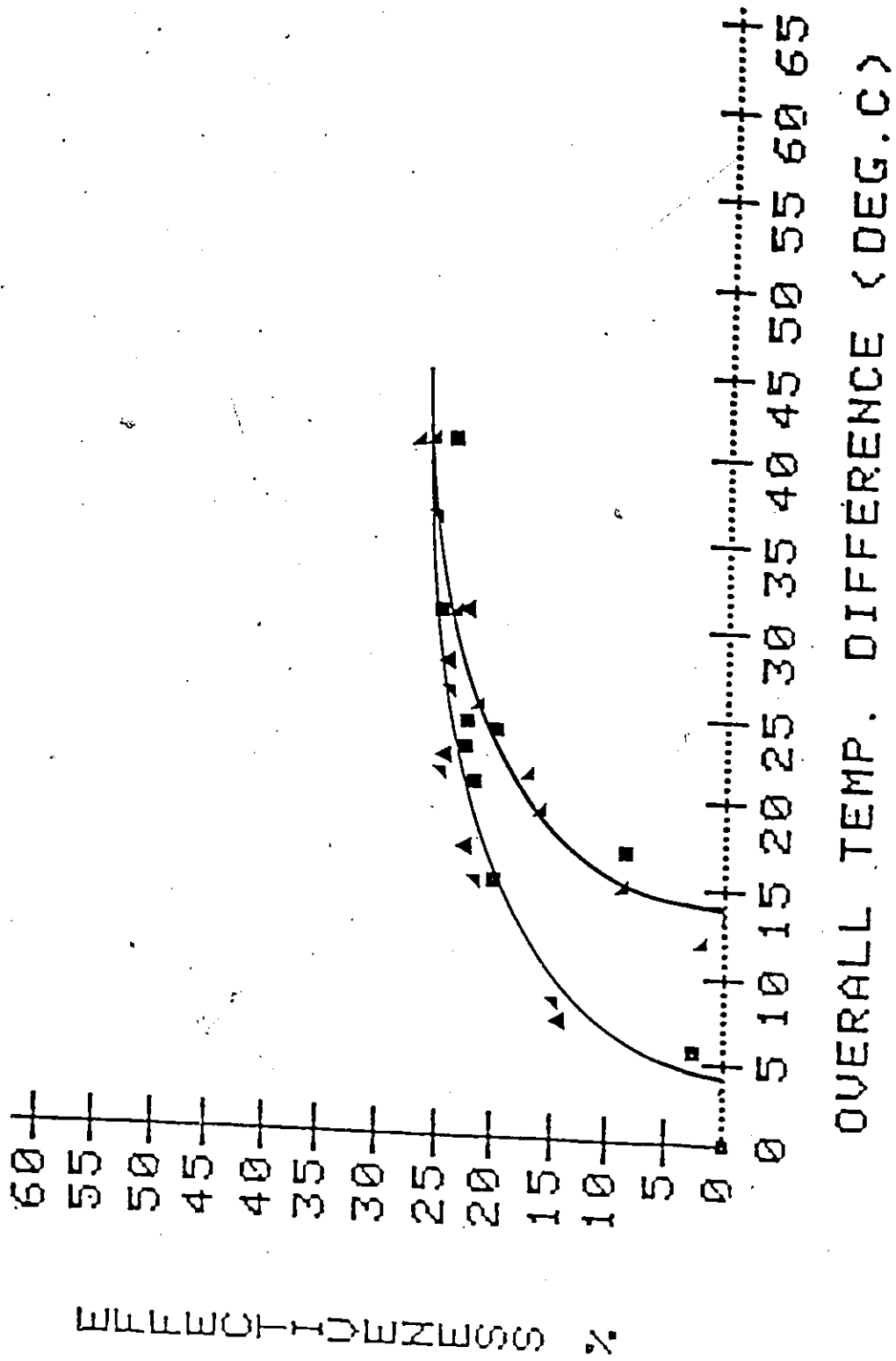


FIGURE 4.15 EFFECTIVENESS VERSUS OVERALL TEMPERATURE DIFFERENCE FOR A SINGLE LOOP

deactivated when the overall temperature difference was decreased below 4C.

In order to use the simulation program to predict the performance of 4 loops it was necessary to make the following relationship. The effectiveness, $E(N)$, for a heat exchanger consisting of "N" loops in a counterflow configuration can be related to one having "M" loops - provided each loop has the same temperature differences (see figure 2.4a). See appendix I for the derivation.

$$E(N) = \frac{N E(M)}{M + (N - M) E(M)} \quad \text{--- (1)}$$

$$\theta(N) = \theta(M) [M + (N - M) E(M)] \quad \text{--- (2)}$$

To simulate the experimental data for 4 loops, the program was used to simulate the performance for a range of temperature differences for one loop. The effectiveness and the overall temperature difference for 4 loops were then calculated from equation 1 and 2.

Figures 4.16, 4.17 and 4.18 show the effect of the heating and cooling processes on the performance of the system for the 4 loop configuration for coil face velocities of 1.3, 2.2 and 3.1 m/s respectively. The simulation program is able to predict the

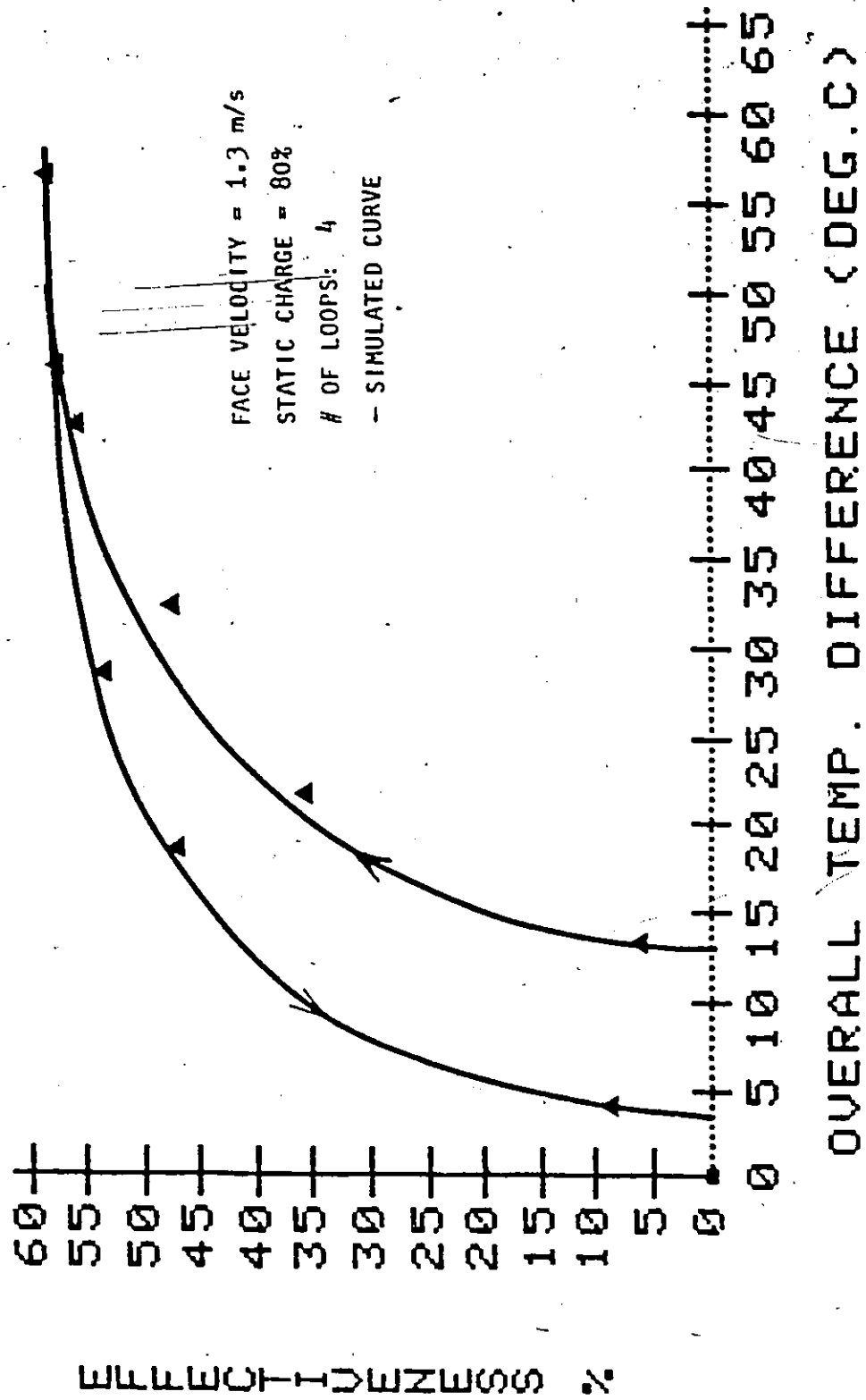


FIG. 4.16 HYSTERESIS ENVELOPE FOR A COIL FACE VELOCITY OF 1.3 M/S

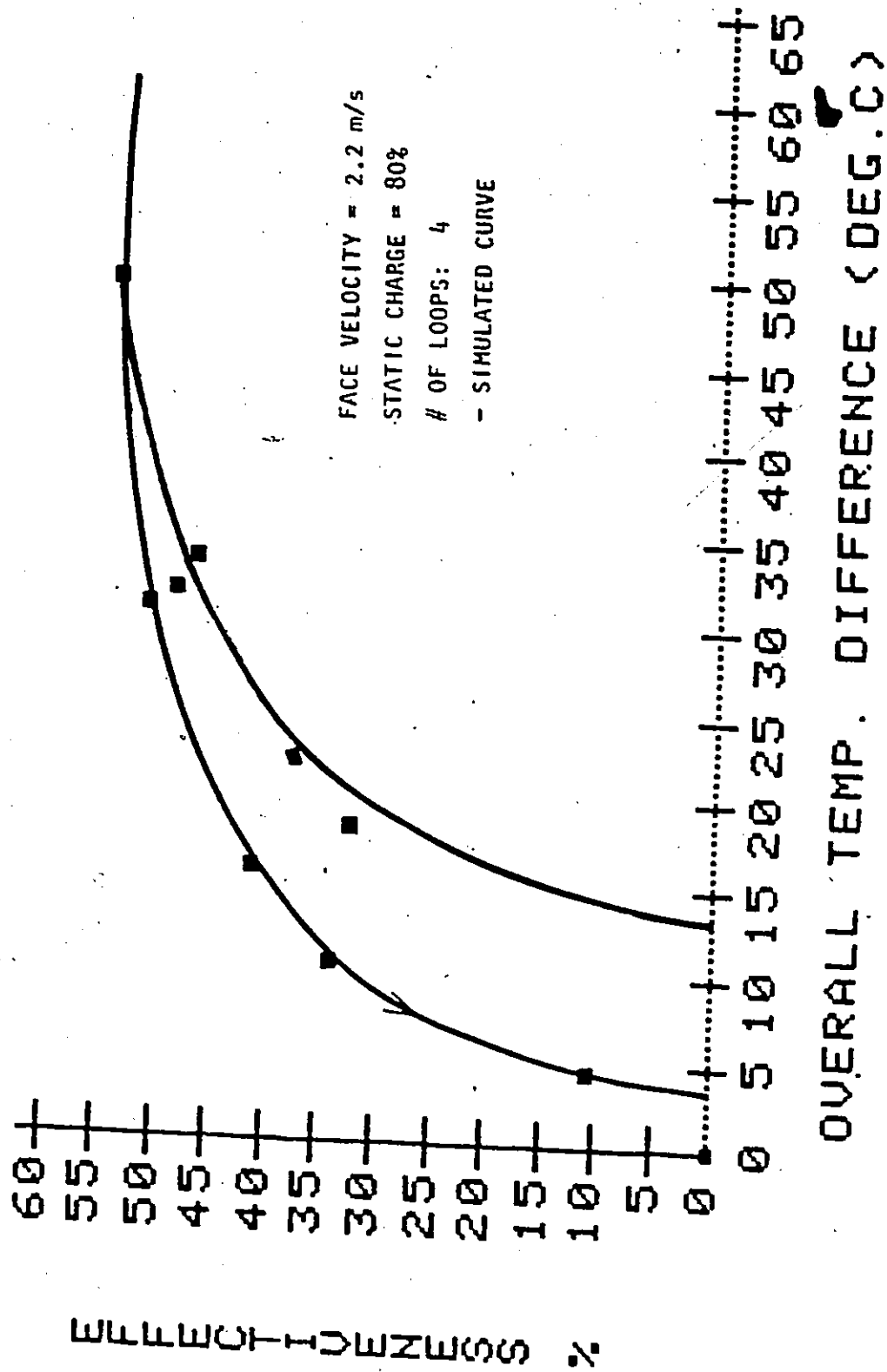


FIG. 4.17 HYSTERESIS ENVELOPE FOR A COIL FACE VELOCITY OF 2.2 M/S

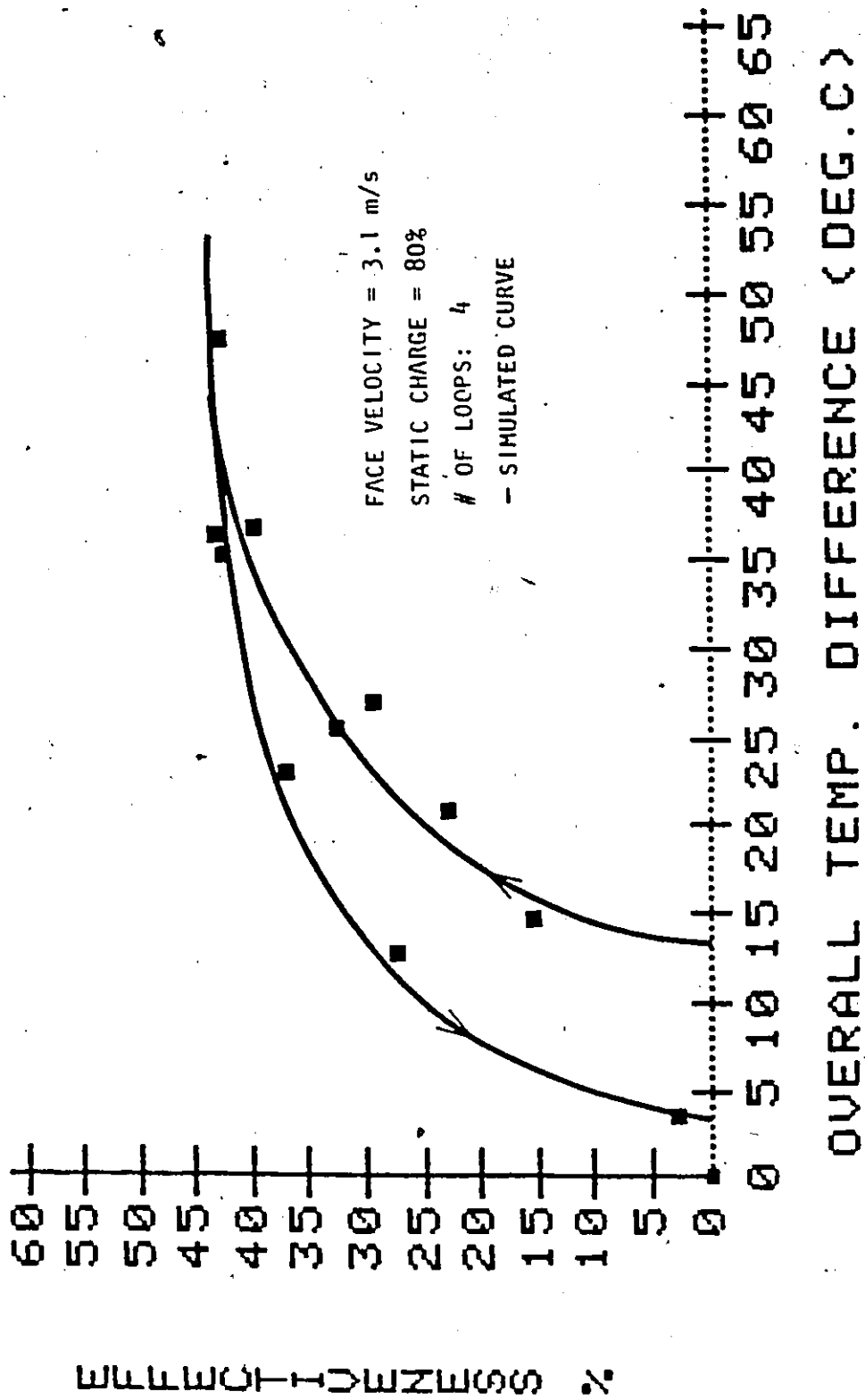


FIG. 4.18 HYSTERESIS ENVELOPE FOR A COIL FACE VELOCITY OF 3.1 M/S

performance of the system very well provided that the external thermal resistance is modeled accurately.

Figure 4.19 shows a comparison of the system effectiveness versus the overall temperature difference between the experimentally obtained results with the corresponding results obtained by the simulation program. These results are for cooling process for three different coil face velocities. The simulation program is able to predict the experimental cooling results very well. The simulated results, which have been extended to temperature differences much higher than could be achieved experimentally, show the effect of dryout which occurs with large temperature differences. This effect is clearly visible for the coil face velocities of 2.2 and 3.1 m/s. The curve also shows that the boiling process is quenched when the overall temperature difference is lowered to approximately 4 C, since below this temperature difference all the nucleation sites are destroyed. The performance of the heat exchanger becomes independent of the temperature difference for temperature differences greater than 40 C, which indicates that nucleation has been initiated at all the readily available sites.

Figures 4.20 and 4.21 show the effect of static charge on the effectiveness of the system for overall temperature differences of 35 and 25C and for a coil face velocities of 2.2 and 3.4 m/s. The

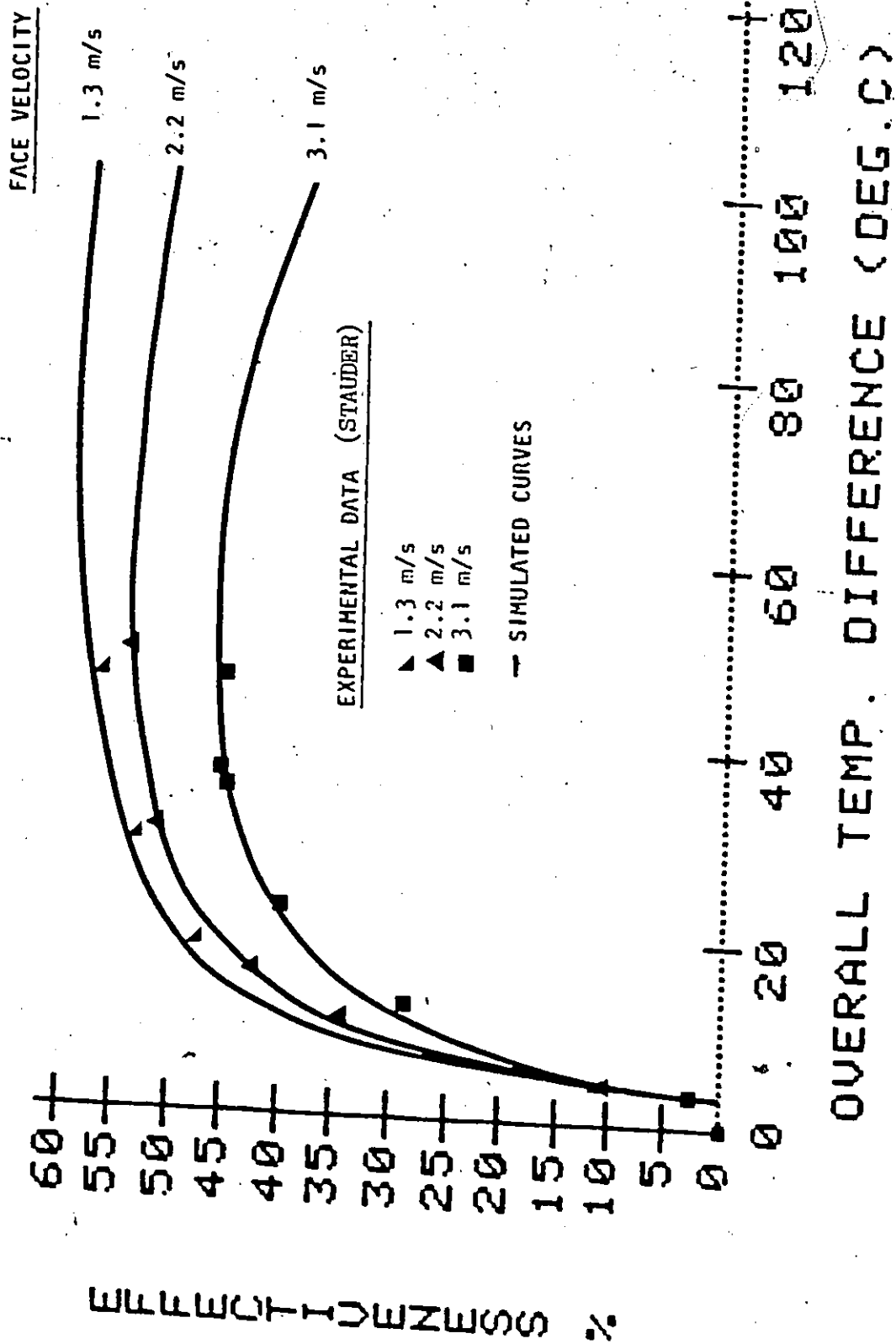


FIG. 4.19 COMPARISON OF EXPERIMENTAL & SIMULATED LOOP PERFORMANCE FOR THE COOLING PROCESS

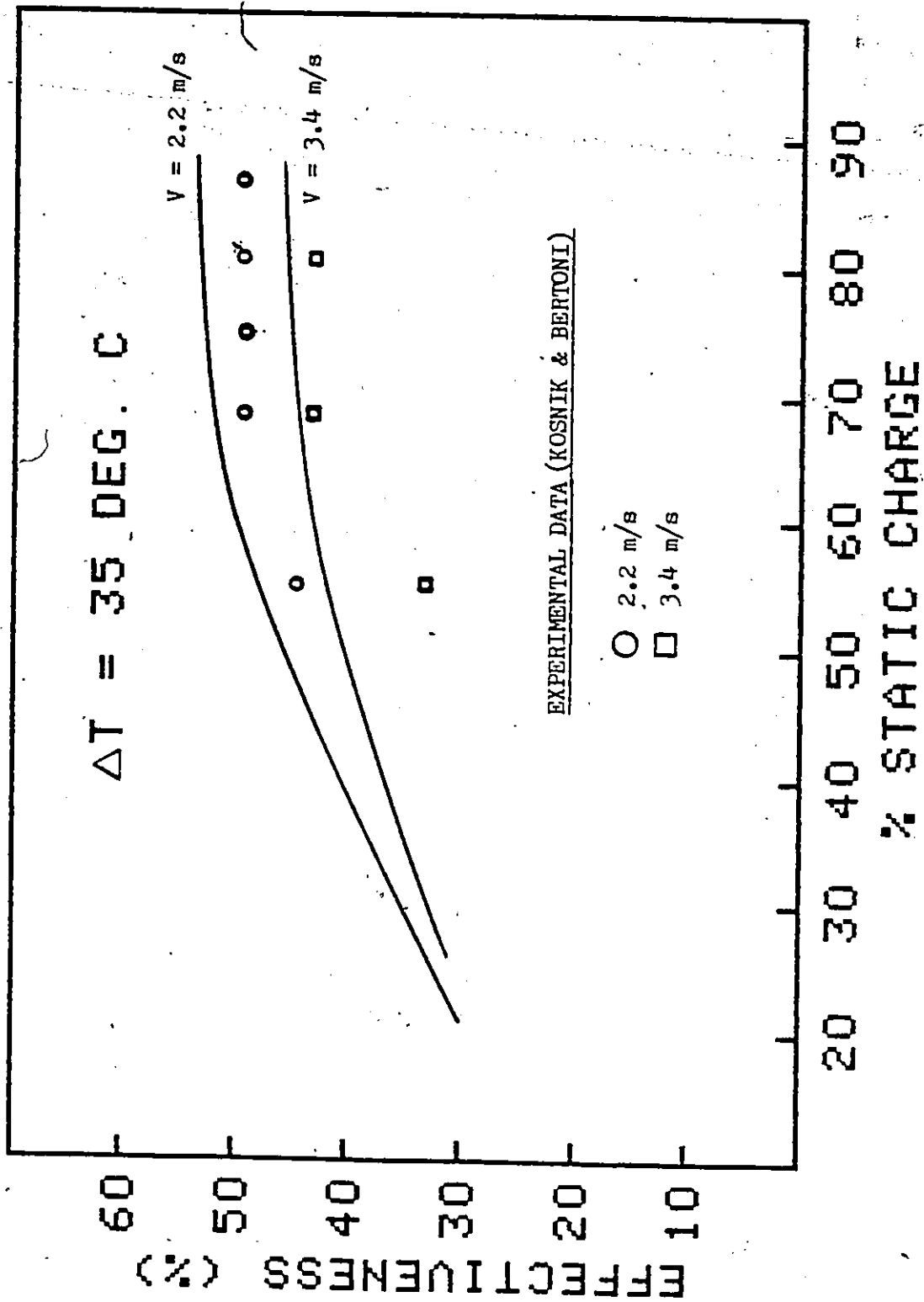


FIG. 4.20 SYSTEM EFFECTIVENESS VERSUS STATIC CHARGE FOR $\Delta T = 35 \text{ C}$ & FACE VELOCITIES OF 2.2 & 3.4 M/S

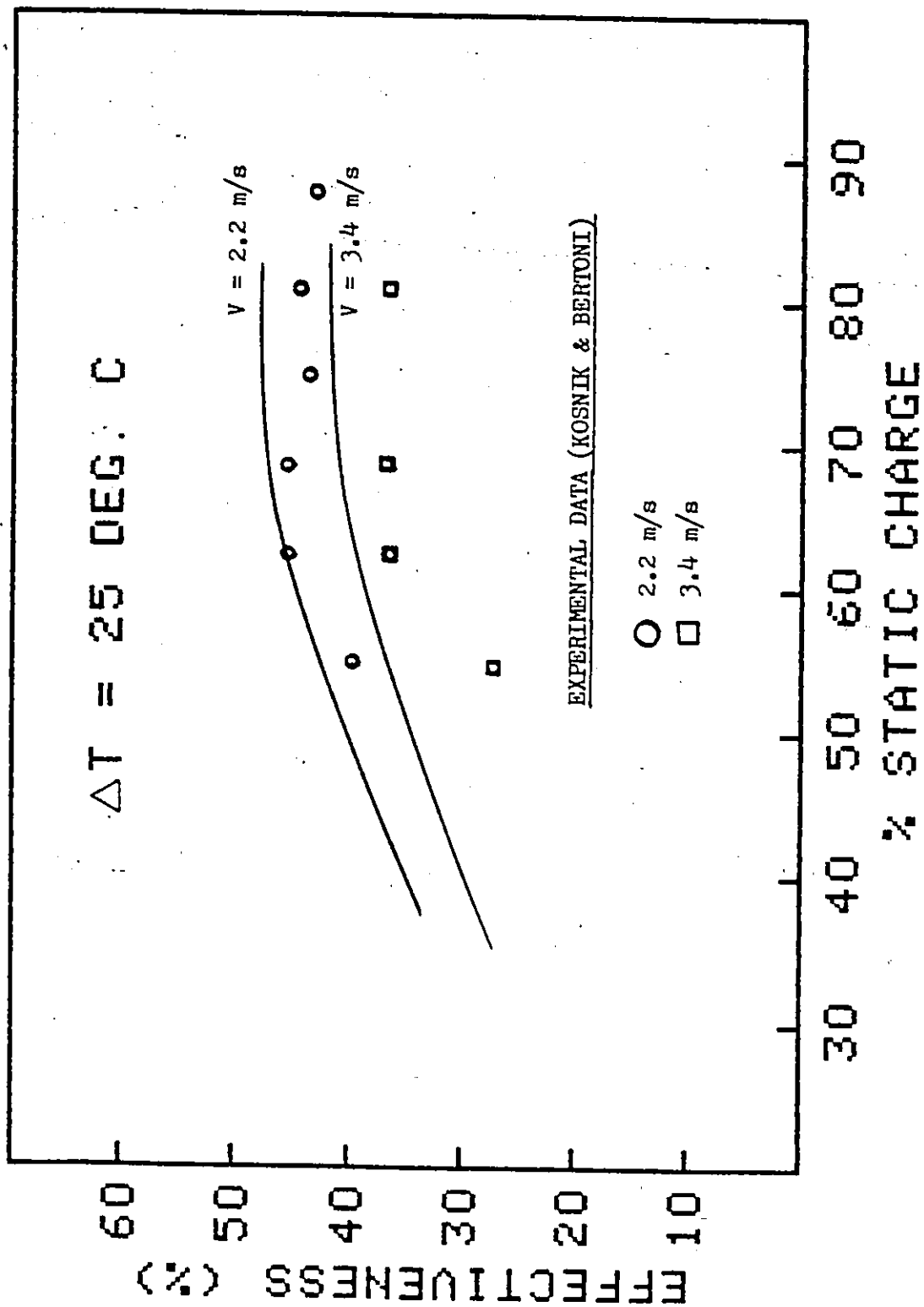


FIG. 4.21 SYSTEM EFFECTIVENESS VERSUS STATIC CHARGE FOR $\Delta T = 25 \text{ C}$ & FACE

VELOCITIES OF 2.2 & 3.4 M/S

agreement of the simulated curves with the corresponding experimental data (15) is not as good as for those of Stauder (17), but do generally show the same trends. This difference in the performance between the experimental results and the simulated results could be due to the presence of some air in the system of reference (15). The decrease in performance, noted in both the experimental and simulated results when the charge decreases below 60 %, is caused by dryout.

4.4 COMPARISON WITH CONSTANT HEAT FLUX BOUNDARY CONDITION IMPOSED ON THE EVAPORATOR TUBE

For this study, the simulated values of the average heat transfer coefficient for the boiling region have been compared with the corresponding results obtained by Bergevin et al. (23).

The system used in this experimental study was made of glass. The evaporator tube was inclined at an angle of 60 degrees from the horizontal. The condenser was water jacketed. The lengths of the evaporator and the condenser tubes were 143 and 400 cm respectively, and the internal diameters of both the tubes were 5 mm. The heat flux was varied from 600 to approximately 3400 W/sq.m. The working fluid

was in a subcooled state at the inlet to the evaporator and the inlet temperature was 45 C.

Figure 4.22 shows their experimentally measured values of the average heat transfer coefficient over that portion of the tube where boiling is present. The experimental results can be explained as follows: At low heat flux, the performance is governed by pool boiling, where the magnitude of the heat transfer coefficient is low. As the heat flux is increased, the two-phase heat transfer mode starts to dominate and the flow becomes annular. With further increase of the heat flux, the two-phase heat transfer dominates over the most of the tube length. When the heat flux is increased sufficiently high, dryout starts, and the performance starts to deteriorate.

To compare the simulated performance with the above experimental data, the computer program was run with the original value of "m" = 0.5 and with m = 0.1 and 0.05. Exponent "m", used in the calculation of the effective heat transfer coefficient in the dryout region, governs how quickly dryout influences the heat transfer coefficient as shown by the following relations.

$$h_{e3} = A h_v + B h_{e2} \quad \text{--- (1)}$$

where the proportion of dry area $A = ((x - x_c)/(1 - x_c))^m$

and the proportion of wet area $B = (1 - A)$.

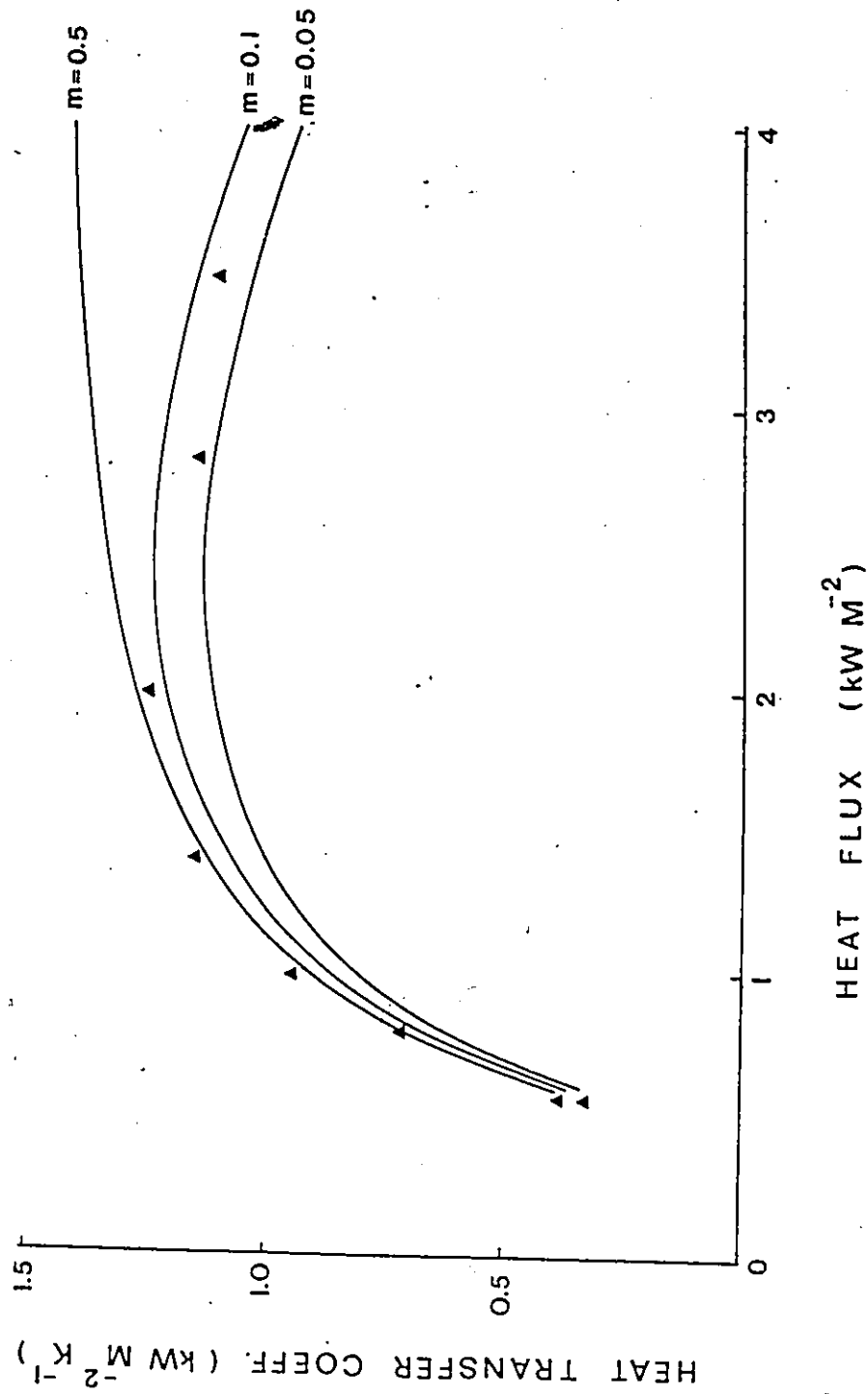


FIG. 4.22 AVERAGE EVAPORATOR HEAT TRANSFER COEFFICIENT FOR THE BOILING REGION VERSUS HEAT FLUX (EXPERIMENTAL DATA OF BERGEVIN ET AL., REF. 23).

For $m = 0.5$, the value established for prescribed-wall-temperature tubes, comparison with the experimental data was excellent up to 2000 W/sq.m, but failed to predict the drop in performance for higher heat fluxes. This means that the effective heat transfer coefficient, h_{e3} , given by equation 1 was much too high for larger heat fluxes. This would indicate that, for this prescribed-heat-flux boundary condition, the proportion of the surface area subject to dryout increases much more rapidly than is the case for dryout within prescribed-wall-temperature tubes. Figure 4.23 shows the variation of the area proportions "A" and "B" for $m = 0.5$, 0.01 and 0.05 when $x_c = 0.05$. The simulated performance curves for the system with $m = 0.1$ and 0.05 show that the decrease in the value of the average heat transfer coefficients as the heat flux increases above 2000 W/sq.m can be achieved by reducing m .

A value of 0.1 for m indicates that for large inclined evaporator tube lengths (1.43 m for the present case), and for high heat fluxes, the dryout is much more severe in comparison to inclined evaporator tubes of smaller lengths (0.62 m in Raza's system).

It is very important to note that the critical quality, x_c , the quality where dryout starts, has a significant impact on the effective heat transfer coefficient, h_{e3} , for the dryout region. An erroneous value of x_c could change the results drastically.

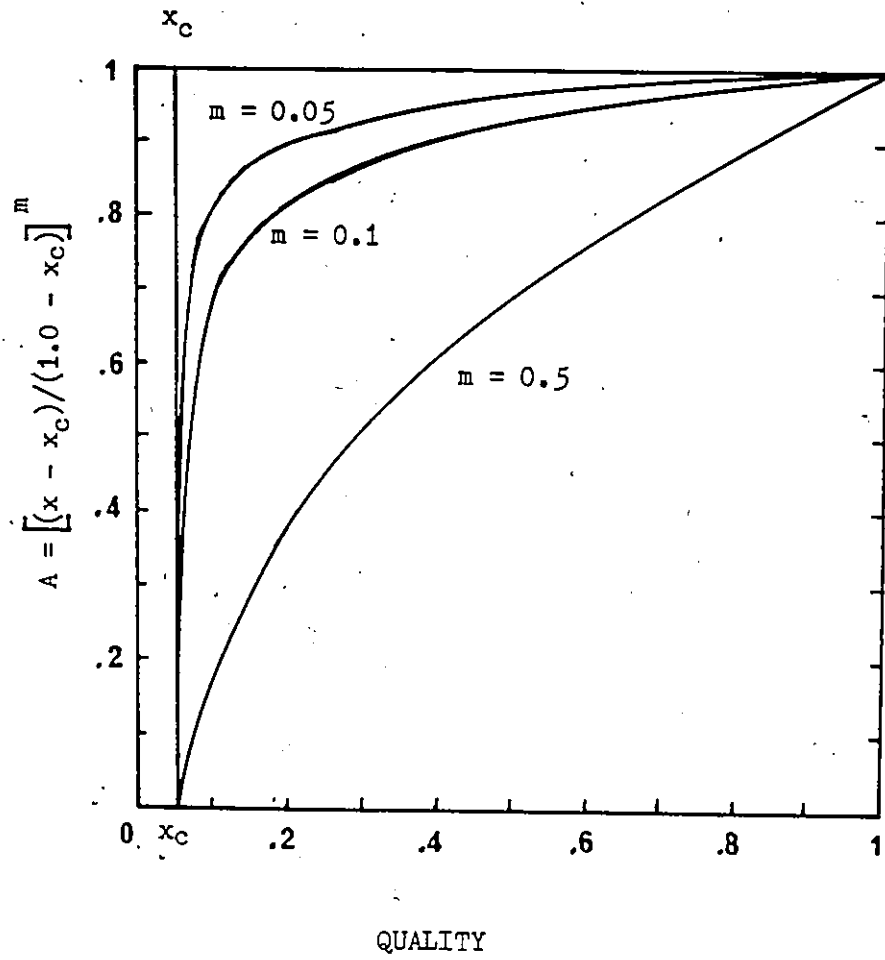


FIGURE 4.23 EFFECT OF POWER "m" ON VARIABLE "A"
 $B = (1.0 - A)$

CHAPTER V.

SIMULATED THERMOSIPHON EVAPORATOR PERFORMANCE

5.1 INTRODUCTION

In the last chapter, a comparison between simulated and experimental data was made for three two-phase thermosiphon loop heat exchanger systems having very different geometries and sizes. By comparing the experimental data of Raza (12), Kosnik & Bertoni (15), Stauder (17) and Bergevin et al. (23) with the corresponding simulated results, it was found that they agreed very well over a wide range of operating conditions. As a result, the model developed in chapter 3 could be used as a tool to simulate the performance of fluid-to-fluid two-phase thermosiphon-loop heat exchanger systems of different geometries and sizes.

In a typical application the temperatures and the mass flow rates of the exhaust gas, T_h and m_h and of the intake air T_c and m_c respectively, are known to the designer. The designer has control over the geometry of the exhaust and intake air ducts, the size of the evaporator and condenser tubes, the number of tube rows, and the tube and fin spacing. There are infinite numbers of combinations for which

the system could be studied. This chapter illustrates how a sequence of simulations could be carried out to assist the designer in selecting an optimum length and diameter of a given evaporator coil design for a given overall temperature difference.

Since the factors influencing condenser performance have received much attention in the past and are well documented, attention will be focused on the effect of various parameters on thermosiphon evaporator performance. This study is necessary as the performance of an evaporator in a thermosiphon system, unlike most evaporators is influenced by the complex interaction which occurs between the flow, pressure drop and heat transfer which governs the behaviour of the working fluid within the loop as the fluid is transformed from a subcooled—liquid at entry to either a wet or a superheated vapour at the exit.

The following section describes the simulations which were carried out to study the behaviour of these systems.

5.2 PARAMETERS INVESTIGATED

The following table shows the range over which each parameter was investigated in the course of this study.

S.N.	PARAMETERS INVESTIGATED	
1.	Length of the coils	$L_c = 0.5 L_e$; $0.61 \text{ m} < L_e < 2.44 \text{ m}$
2.	Evap. and cond. tube diameters	$D_c = D_e - 1/4", 3/8" \text{ and } 1/2"$ nominal size
3.	Overall temp. diff.	- varied between 30 - 160 C
4.	Static charge	30 - 90 %
5.	Pressure drop in vapour header	0, 3.45, 6.90 Kpa

The ambient air temperature T_c was fixed at 20 C. Both coil face velocities and fin spacing were also held constant at 3.1 m/s (600 fpm) and 473/m (12 fins/inch) respectively. For all the simulations, the diameter of both the coil tubes were set equal and the evaporator and the condenser coils were assumed to be in vertical position. The areas of both the coils were taken equal and it was assumed that the condenser coil height was half the size of the evaporator coil (condenser coil width = 2 * evaporator coil width).

The airside resistances for 1/4" and 1/2" tubes were calculated

in a manner similar to that discussed in section 3.2.2.1. The coefficients used for the evaluation of airside resistance in equation 1 (appendix F) for tube diameters of 1/4" and 1/2" are given in Table F1 of appendix F.

The evaporator conductance, U_e , is defined by

$$U_e = \frac{Q}{A_e (T_{w,av} - T_{sat})}$$

where Q , A_e , $T_{w,av}$, T_{sat} are the heat transfer rate (W), area of the evaporator (sq.m) average evaporator wall temperature (C) and saturation temperature in the evaporator vapour header (C), respectively. The saturation temperature, T_{sat} , in the evaporator vapour header is a function of the pressure drop in the evaporator tube.

In the following section, the simulated thermosiphon evaporator performance is presented in graphical form. For these simulations, the hysteresis was not accounted for and the performance is based on the cooling curve, i.e. it was assumed that the heated surface was at a higher temperature before reducing to the current level and thus nucleation sites were already established at a higher temperature.

5.3 RESULTS

The effect of evaporator static charge and tube length on the evaporator performance for a tube diameters of 0.635 cm (1/4") and 0.953 cm (3/8") and 1.270 cm (1/2") are shown in figures 5.1 to 5.8. For low static charges, the evaporator performance is poor due to the presence of dryout in the tubes. With the increase of the static charge and hence the flow rate, the dried out portion of the evaporator tubes decreases which results in better performance. However, when larger evaporator tubes are used, the evaporator conductance starts decreasing (see figures 5.3, 5.4, 5.8) at higher charges. This happens because as the charge and hence flow rate is increased, the increase in the frictional pressure drop in the evaporator causes a decrease in the saturation temperature in the evaporator vapour header and hence an increase in the temperature differences across the evaporator. For these high charges the temperature difference increases more rapidly than the heat flux and hence causes a decrease in the evaporator conductance. The decrease in the evaporator conductance for a tube diameter of 0.635 cm, with the increase of the overall temperature difference is due to dryout. However, the performance increases for larger tube diameters with the increase of the overall temperature difference, since the two-phase heat transfer coefficient and the vigour of the boiling increases. The decrease in the evaporator conductance for the overall temperature

difference lower than 20 C is due to the boiling suppression over large portion of the evaporator tube.

It was found that generally the evaporator performance increased with the increase of the percent static charge except for large evaporator tube lengths and temperature differences where excessive pressure drops occurred. It was seen from these results that the evaporator performance is fairly good at static charges greater than 75 %. Hence, in order to analyze these results, the evaporator conductance versus evaporator tube length and overall temperature difference curves shown in figures 5.9 to 5.12, for a static charge of 80 % were constructed by cross-plotting the curves shown in figures 5.1 to 5.8.

Figure 5.9 shows the variation of the evaporator conductance with the length of the evaporator tube for a static charge of 80 % for $D_e = 0.635$ cm and 0.953 cm. For any source fluid temperature investigated for $D_e = 0.635$ cm, the peak performance is obtained at an evaporator tube length of 1.22 m. For an evaporator tube diameter of 0.953 cm and for 60 and 70 C overall temperature differences, the peak performance is obtained at 1.83 m. For an overall temperature difference of 90C the peak occurs at 2.44 m. For overall temperature differences larger than 90C, the peak would be obtained at tube lengths greater than 2.44 m. For the overall temperature difference of

100 C, the evaporator performance for 0.61 and 1.22 m tube length is lower in comparison to 90 and 70C overall temperature differences due to higher pressure drop at 80 % static charge, and dryout. Figure 5.10 shows similar results for a tube diameter of 1.270 cm.

Figure 5.11 shows the variation of the evaporator conductance with the overall temperature difference for a static charge of 80%, $D_e = 0.635$ and 0.953 cm, $L_e = 0.61, 1.22, 1.83$ and 2.44 m. For $D_e = 0.635$ cm, the decrease in the performance with the increase of the overall temperature difference is due to dryout. The decrease in the evaporator conductance for the overall temperature difference lower than 20C is due to the boiling suppression over large portion of the evaporator tube. For $D_e = 0.953$ and 1.270 cm (Figure 5.12), the performance increases with the increase of the overall temperature difference and peaks off at some higher overall temperature difference. Further increase of the overall temperature difference yields a lower performance due to increased dryout and increased pressure drop. For a given evaporator tube length, the optimum temperature difference can be determined from these graphs - i.e. the optimum temperature difference for $L_e = 0.61$ m and $D_e = 1.270$ cm is 100 C (see figure 5.12). However, the optimum length for an overall temperature difference of 100 C is 2.44 m.

In all the graphs shown so far (figure 5.1 - 5.12), the pressure

drop in the vapour header was set equal to zero. The significant effect that a pressure drop in the vapour header can have on the performance is illustrated in figure 5.13. The figure shows the variation of the performance for vapour header pressure drops of 0, 3.45 and 6.90 kPa (0, 0.5 and 1.0 psi). It is very important to keep this pressure drop to a minimum.

Figures 5.9 - 5.12 are plotted for a static charge of 80%. The performance of the thermosiphon evaporator could be very different for charges greater than 80%, especially for larger tube lengths. Hence, care should be taken for proper selection of the static charge, for a given set of condition.

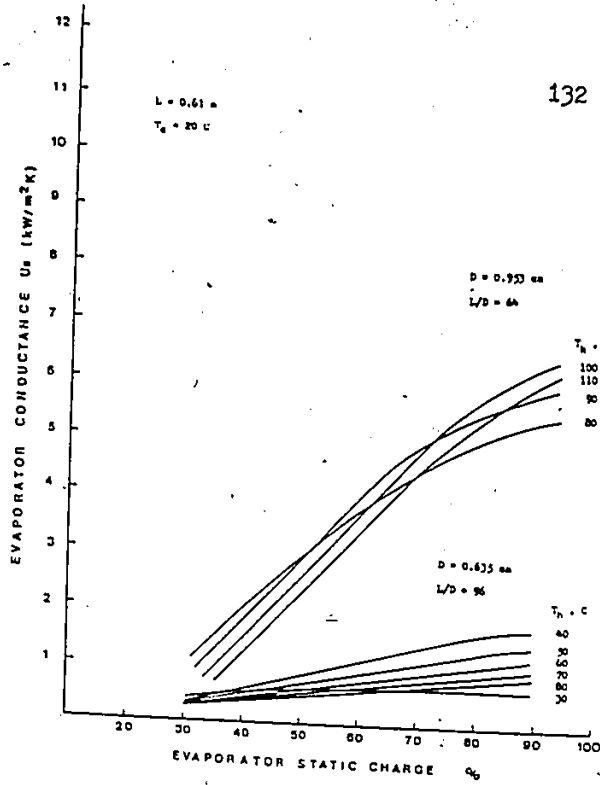
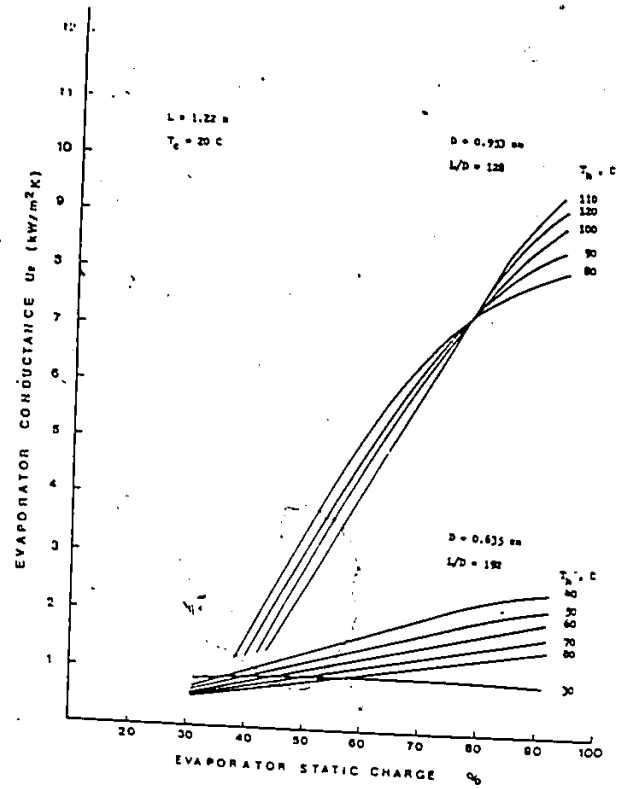
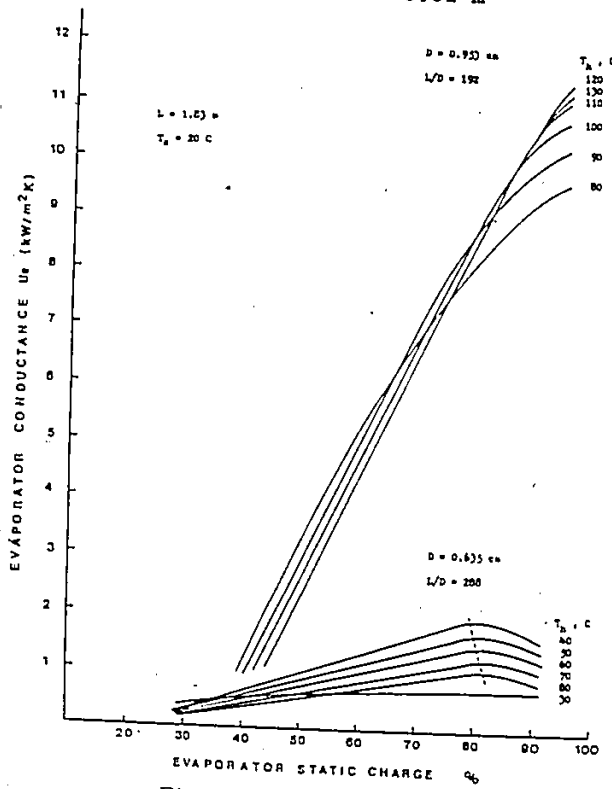
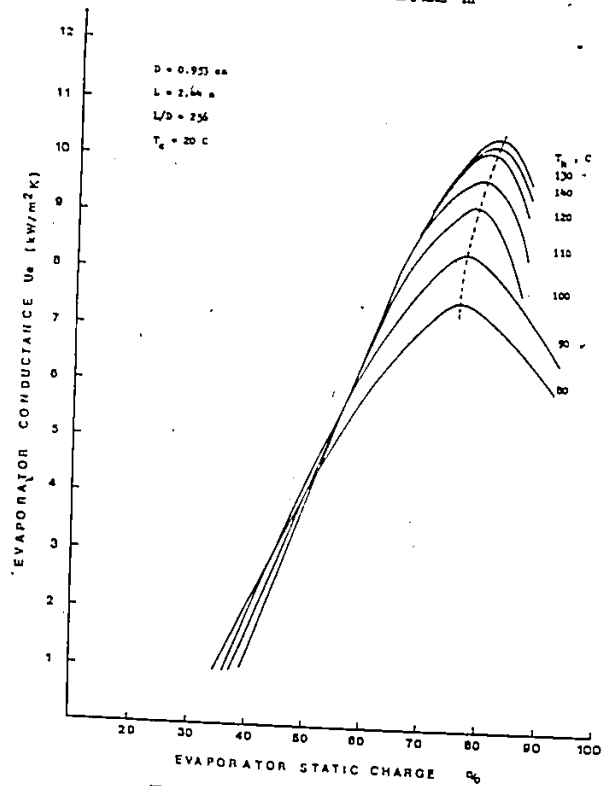
5.4 CONCLUSIONS

Through a careful selection of the coil tube diameter and length as well as the static charge, finned multiple-tube thermosiphon evaporators can provide very good conductance values over a wide range of temperature differences. The range is limited by the evaporator tube dryout for large temperature differences and by boiling suppression for small temperature differences.

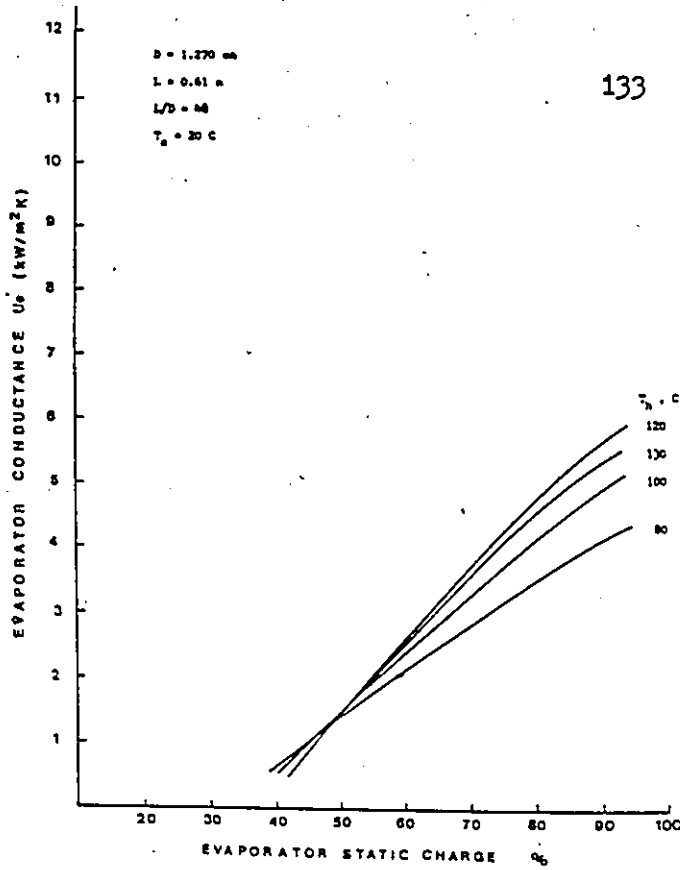
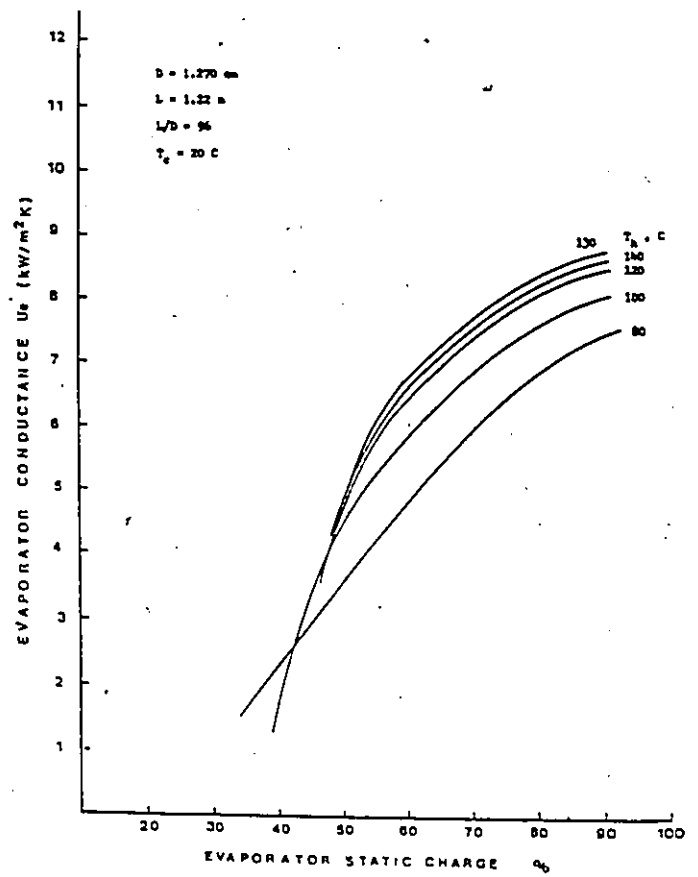
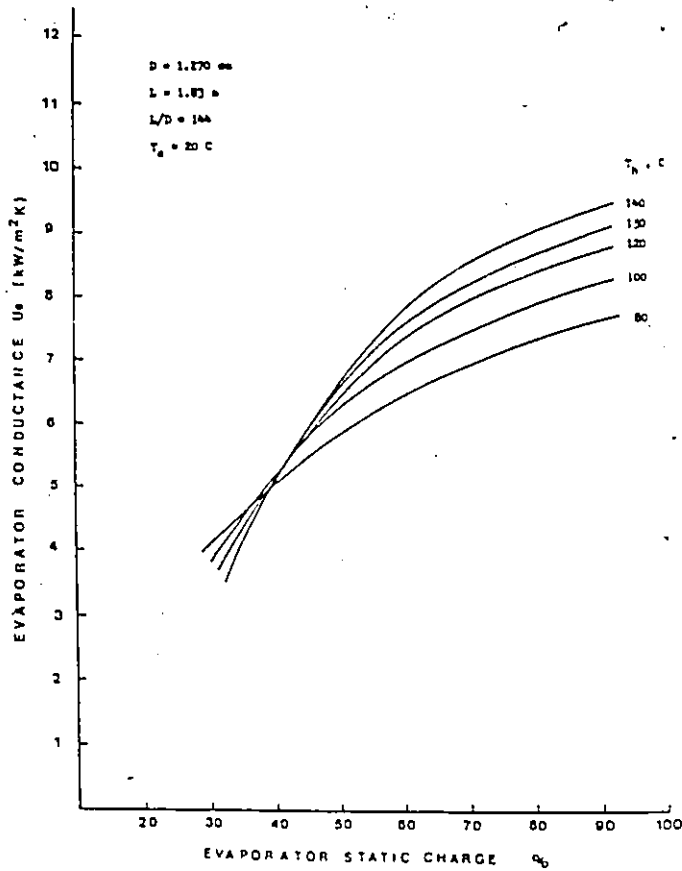
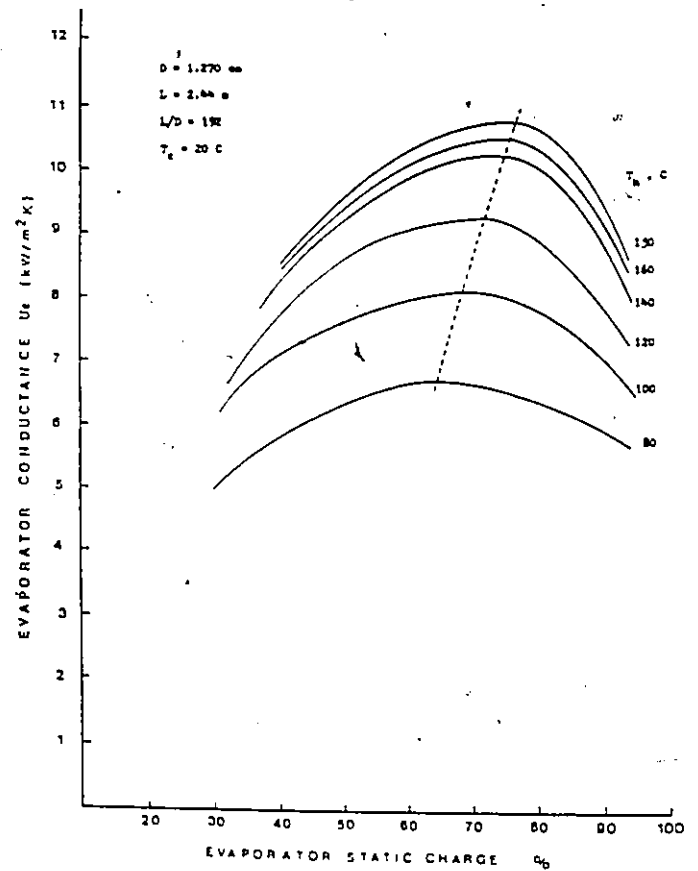
For a given overall temperature difference, there exists a tube diameter and length which yield a maximum evaporator performance. The

optimum diameter and length increase with increasing temperature difference as shown in the following table.

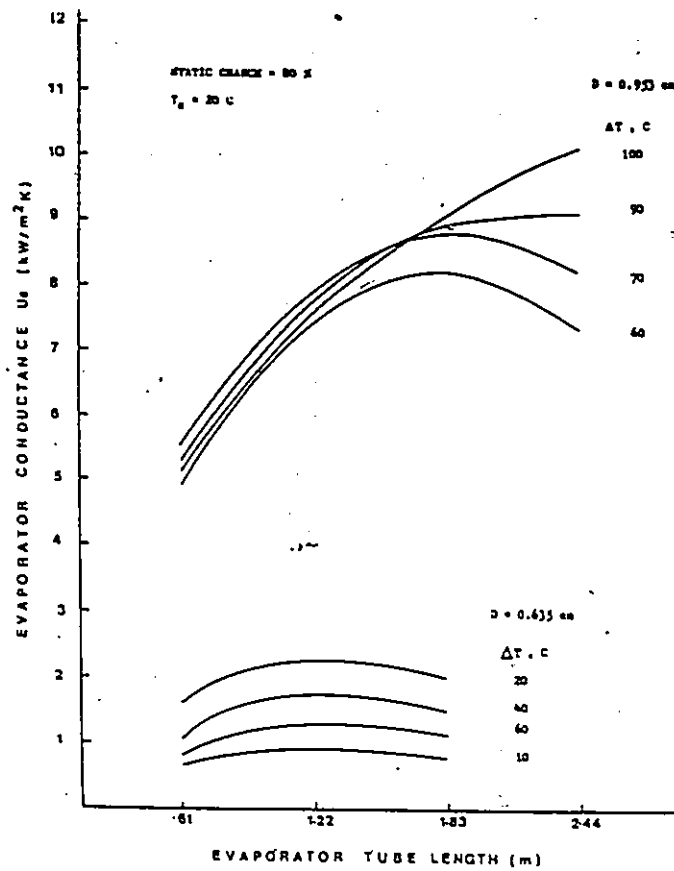
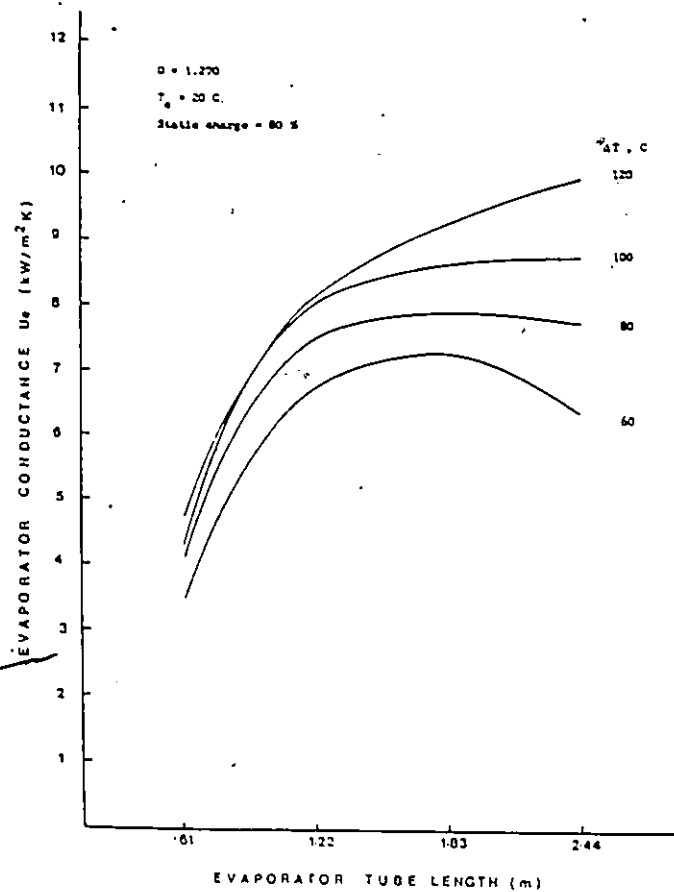
TEMPERATURE DIFFERENCE (C)	OPTIMUM	
	DIAMETER (cm)	LENGTH (M)
10	0.635	1.22
20-70	0.953	1.83
80	0.953	1.83/2.44
90-110	0.953	2.44
120-130	1.270	2.44

Figure 5.1 $L = 0.61$ mFigure 5.2 $L = 1.22$ mFigure 5.3 $L = 1.83$ mFigure 5.4 $L = 2.44$ m

Effect of tube length, static charge and source fluid temperature on the evaporator conductance for $D = 0.635$ and 0.953 cm

Figure 5.5 $L = 0.61 \text{ m}$ Figure 5.6 $L = 1.22 \text{ m}$ Figure 5.7 $L = 1.83 \text{ m}$ Figure 5.8 $L = 2.44 \text{ m}$

Effect of tube length, static charge and source fluid temperature on the evaporator conductance for $D = 1.270 \text{ m}$

Figure 5.9 $D = 0.635, 0.953 \text{ cm}$ Figure 5.10 $D = 1.270 \text{ cm}$

Effect of overall temperature difference and evaporator tube length on the evaporator conductance for a static charge of 80 %

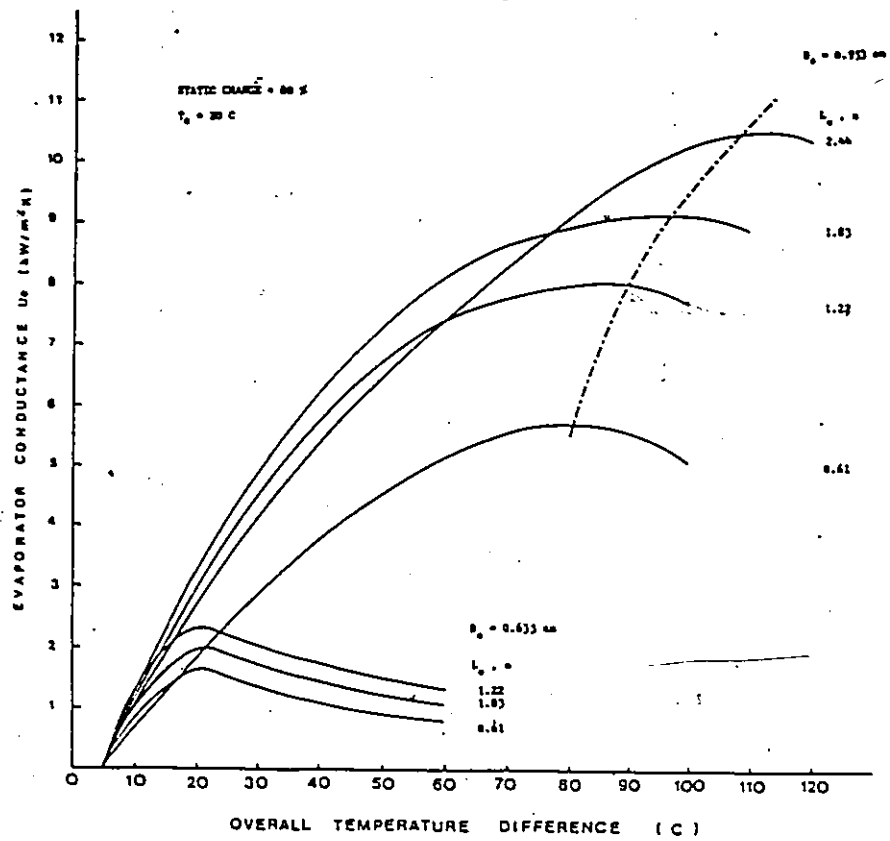


Figure 5.11 $D = 0.635, 0.953$ cm

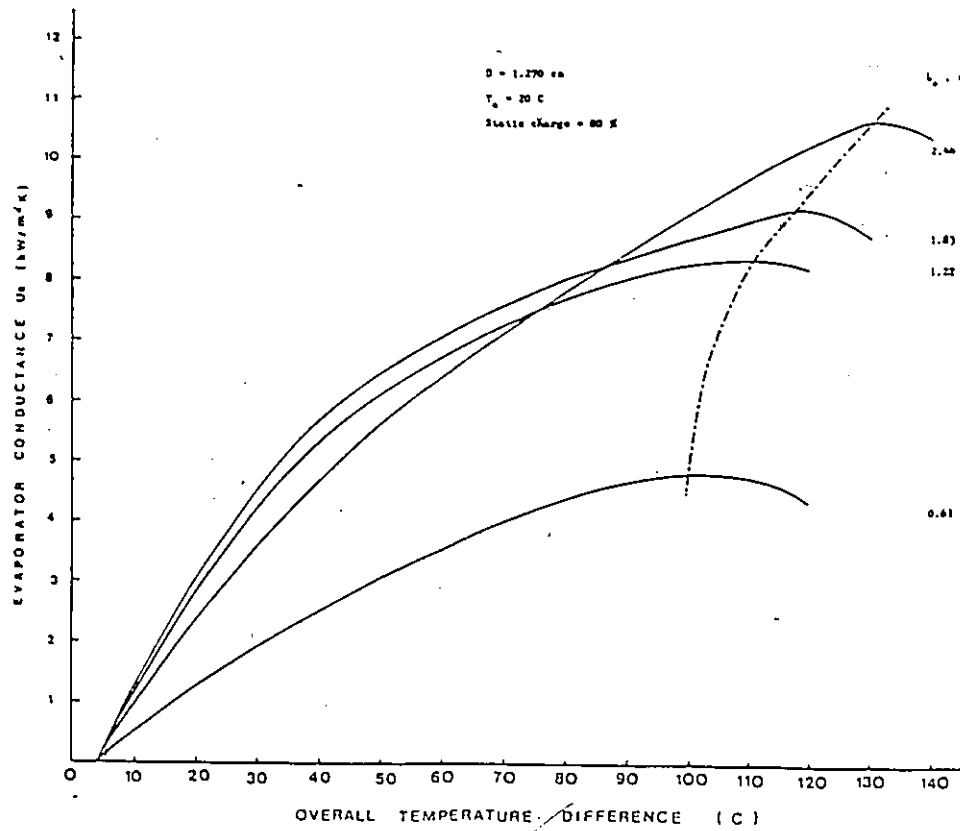


Figure 5.12 $D = 1.270$ cm

Effect of evaporator tube length and overall temperature difference on the evaporator conductance for a static charge of 80 %

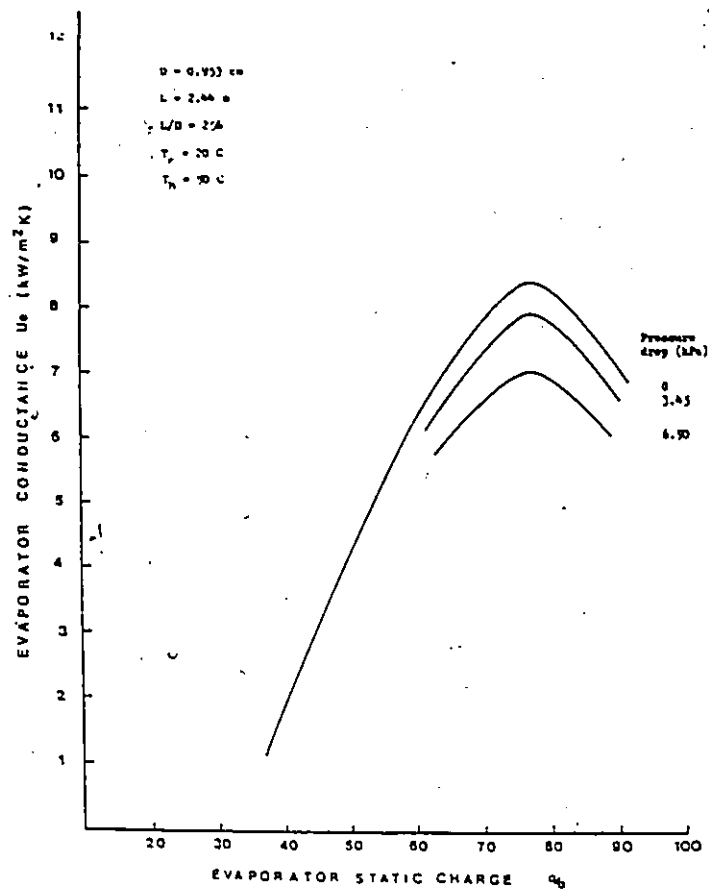


Figure 5.13 Effect of vapour header pressure drop on the evaporator conductance for $D = 0.953 \text{ cm}$, $L_e = 2.44 \text{ m}$, $T_h = 90 \text{ C}$ & $T_c = 20 \text{ C}$

CHAPTER VI

CONCLUSIONS AND RECOMMENDATIONS

The computer program developed is capable of simulating the performance of fluid-to-fluid two-phase thermosiphon-loop heat exchanger systems in which the evaporator and condenser coils consist of parallel rows of straight tubes running between a common liquid header and a common vapour header. The model developed is based on the assumption of one-dimensional equilibrium flow in the tubes. The program is modular in form and hence may be easily modified to simulate any external boundary conditions. The following external flow and thermal conditions are built into the program:

1. Prescribed surface temperature
2. Source and or sink fluid flowing in an annular jacket
3. Source and or sink fluid flowing over finned tubes
4. Prescribed heat flux

The validity of the program was checked by comparison with experimental results generated in three different systems, one with water annuli jackets on all tubes (12), and one with air flowing over finned evaporator and condenser tubes (15), (17), and one with a constant heat flux boundary condition imposed on the evaporator tube (23). It was found that the computer program is capable of simulating

the behaviour of these two-phase thermosiphon loops with good accuracy.

Since computer program is based on the assumption of one-dimensional equilibrium flow in the tubes it will not be valid for large diameter tubes where two-dimensional effects will be present. The program has accurately simulated the performance of two-phase thermosiphon loops having coils consisting of 1- to 4.7-ft tubes with diameters of 1/4 and 3/8 in. A major limitation at present is the need for a correlation to determine the conditions under which the boiling will initiate during the system startup. Experimental studies should also be carried out for the dryout region so that better correlations can be developed for the estimation of the heat transfer coefficients.

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APPENDIX - A

TABLE OF CORRELATIONS

Table 2.3	Correlations to predict the flow transitions
Table 2.4.1	Correlations to predict Nusselt number for the single-phase region
Table 2.4.2	Correlations to predict the wall superheat to sustain boiling
Table 2.4.3	Correlations to predict nucleate boiling heat transfer coefficients
Table 2.4.5A } Table 2.4.5B } Table 2.4.5C }	Correlations to predict two-phase heat transfer coefficients
Table 2.4.6	Correlations to predict the critical quality
Table 2.4.7	Correlations to predict the heat transfer coefficients in the liquid deficient region
Table 2.5	Correlations to predict the condensation coefficients
Table 2.6.2	Correlations to predict the pressure drop and void fraction for two-phase region

TABLE 2.3 Correlations to predict the flow transitions

Investigators

Correlation

Barnea

et al. (76)

For bubble to slug flow transition

$$U_{1s} = \frac{1-\alpha}{\alpha} U_{gs} - 1.53 (1-\alpha) *$$

$$\frac{g (\rho_l - \rho_g) \sigma^{1/4}}{\rho_l^2} \sin \beta$$

where $\alpha = 0.25$; U is the superficial velocity.

Mishima &

Ishii (77)

For bubble to slug flow transition

$$J_f = \left(\frac{3.33}{Co} - 1 \right) J_g - \frac{0.76}{Co} \left(\frac{g \Delta \rho}{\rho_f^2} \right)^{1/4}$$

$$\text{where } Co = 1.2 - 0.2 (\rho_g / \rho_l)^{0.5}$$

Taitel &

Dukler (236)

For bubble to slug flow transition

$$\frac{J_l}{J_g} = 2.34 - 1.07 \frac{[g (\rho_l - \rho_g) \sigma]^{1/4}}{J_g \rho_l^{1/2}}$$

For slug to annular flow transition

$$\frac{J_g \rho_l^{1/2}}{[g (\rho_l - \rho_g) \sigma]^{1/4}} = 30.9 X^{-1}$$

$$\text{for } X \gg 1 ; X = \left[(dp/dz)_l / (dp/dz)_g \right]^{1/2}$$

TABLE 2.3 CONTINUED

Investigators

Correlation

Soliman (87)

For mist to annular flow transition during
condensation

$$Re_1 = 72.0 \left(\frac{\mu_1^2}{\rho_1 \sigma D} \right)^{-0.38} \left(\frac{\rho_1}{\rho_g} \right)^{-0.073} *$$

$$\left(\frac{\mu_g}{\mu_1} \right)^{0.141} x_{tt}^{0.9} \phi_g^{0.506}$$

for $Re > 1250$

$$Re_1 = 37.6 \left(\frac{\mu_1^2}{\rho_1 \sigma D} \right)^{-0.469} \left(\frac{\rho_1}{\rho_g} \right)^{0.0867} *$$

$$\left(\frac{\mu_g}{\mu_1} \right)^{0.173} x_{tt}^{1.11} \phi_g^{0.625}$$

for $Re \leq 1250$

TABLE 2.4.1 Correlations to predict Nusselt number for the single-phase region

Investigators	Correlation
<u>A LAMINAR FLOW</u>	
Graetz (91)	$Nu_z = \frac{\sum_{n=0}^{\infty} G_n \exp(-f_n Z^*)}{2 \sum_{n=0}^{\infty} G_n / f_n^2 \exp(-f_n^2 Z^*)}$ <p> $Nu_d = 4.36$ for constant heat flux for fully developed flow </p> <p> $Nu_d = 3.66$ for constant wall temperature for fully developed flow </p>
Leveque (92)	<p>For uniform wall temperature</p> $Nu_z = 1.3565 (Z^*)^{-1/3}$ <p>For constant heat flux</p> $Nu_z = 1.6393 (Z^*)^{-1/3}$ <p>For linearly varying wall temperature</p> $Nu_z = 2.0348 (Z^*)^{-1/3}$
Lipkiss (93)	<p>For constant wall temperature</p> $Nu_z = \frac{1.3565 (Z^*)^{-1/3}}{1.0 - 4.07 (Z^*)^{2/3}}$ <p>for $Z^* \leq 0.001$</p>

TABLE 2.4.1 CONTINUED

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Investigators	Correlation
Hausen (104)	$Nu_m = 3.66 + \frac{0.0668 (D/L) Re Pr}{1 + 0.04 [(D/L) Re Pr]^{2/3}}$
<u>B TURBULENT FLOW</u>	
Nusselt (97)	$Nu_m = a Re^n Pr^m$ <p>where $a=0.036$, $n=0.764$, $m=0.355$</p>
Colburn (98)	$Nu_d = 0.023 Re^{0.8} Pr^{0.33}$
Dittus & Boelter (99)	$Nu_d = 0.023 Re^{0.8} Pr^n$ <p>where $n=0.4$ for heating $n=0.3$ for cooling</p> <p>$0.7 \leq Pr \leq 160$ $Re \geq 10,000$ $L/D \geq 60$</p>
Sieder & Tate (96)	$Nu_d = 0.027 Re^{0.8} Pr^{0.33} (\mu / \mu_s)^{0.14}$ <p>$0.7 \leq Pr \leq 16,700$ $Re \geq 10,000$ $L/D \geq 60$</p>
McAdams (103)	$Nu = 0.0272 (1 + 50 D/L) Re^{0.8}$ <p>for $34 \leq L/D \leq 100$</p> <p>for developing flow</p>
Mills (105)	$Nu = 0.0397 Re^{0.73} Pr^{0.33}$ <p>$10^5 < Re < 10^6$ (turbulent flow)</p>

TABLE 2.4.1 CONTINUED

Investigators

Correlation

$$Nu_z = 1.3565 (Z^*)^{-1/3 - 1.7}$$

$$\text{for } Z^* \leq 4 \times 10^{-3}$$

Norris & Stried (94)

$$Nu_m = 2.035 (Z^*)^{-1/3}$$

$$\text{for } Z^* < 0.1$$

Kay & London (95)

$$(h_m/Gc) Pr = 3.66/Re + 0.05 Pr/(L/D)$$

where Gc is the Graetz number = Re Pr D/L

Sieder & Tate (96)

$$Nu_m = 1.86 [Re Pr/(L/D)]^{1/3} (\mu/\mu_s)^{0.14}$$

$$0.48 < Pr < 16,700$$

$$0.0044 (\mu/\mu_s) < 9.75$$

Kay (102)

For constant wall temperature

$$Nu_m = 3.66 + \frac{0.208/Z^*}{1 + (0.028/Z^*)}$$

For constant heat flux

$$Nu_m = 4.36 + \frac{0.072/Z^*}{1 + (0.0022/Z^*)}$$

both correlations valid for Pr=0.7

TABLE 2.4.2 Correlations to predict the wall superheat to sustain boiling

Investigators	Correlation
Hsu (108), Hsu & Graham (109)	$[q\delta/k_1 - (T_{\text{sat}} - T_b)]^2 > \frac{8\sigma T_{\text{sat}} C_1}{h_{fg} \rho_v k_1} q$
Bergles & Rohsenow (110)	$q = 15.6 p^{1.156} (T_w - T_{\text{sat}})^{2.3/p^{0.0234}}$
Han & Griffith (111)	$(T_w - T_{\text{sat}})^2 = \frac{12\sigma T_{\text{sat}}}{h_{fg} \rho_v k_1} q$
Sato & Matsumura (112)	$(T_w - T_{\text{sat}})^2 = \frac{8\sigma T_{\text{sat}}}{h_{fg} \rho_v k_1} q$
Davis & Anderson (113)	$(T_w - T_{\text{sat}})^2 = \frac{8\sigma T_{\text{sat}} C_1}{h_{fg} \rho_v k_1} q$
Murphy & Bergles (114)	$q = (k_1/8\sigma) * [T_w - T_{\text{sat}}(p_1)]^2 \left. \frac{dp}{dT} \right _{p_1, \text{sat}}$
McAdam et al. (115)	$(T_w - T_b)_q = \left[\frac{0.026 k_1 (T_w - T_b)}{C_1 D} q \right]^{1/4} *$ $(DG/\mu)^{0.8} Pr^{1/3} (\mu/\mu_w)^{0.14}]^{0.259}$

TABLE 2.4.2. CONTINUED

Investigators	Correlation
Buchberg et al. (116)	$q = 0.0027 \frac{G^{0.8} T_b^{0.32}}{D^{0.2}} (646 - T_b)$ $q = 0.00083 \frac{G^{0.8} T_b^{0.53}}{D^{0.2}} (487 - T_b)$

The first equation is valid for pressures upto 140 ata. and the second upto 35 ata.

Chen (117)

$$P_s = \frac{\sigma(P' - P_v')}{\sigma'} - \frac{G_o}{r_d^3} *$$

$$[T + \frac{\sigma}{\sigma'} T']$$

where ' represents the deactivation conditions, G_o is the dimensional variable measuring the amount of gas (lb f.in/R), r_d is the deactivation radius (inches)

TABLE 2.4.3 Correlations to predict nucleate boiling heat transfer coefficients

Investigators	Correlation
Shah (148, 149)	A graphic correlation for subcooled nucleate boiling. See ref. 122, 123.
Stephan & Auracher (150)	<p>For refrigerants in forced convection</p> $\frac{h_b D}{K_l} = 207 \left(\frac{q D}{K_l T_s} \right)^{0.745} \left(\frac{\rho_g}{\rho_l} \right)^{0.581} \left(\frac{\mu}{K_l} \right)^{0.55}$ <p>for $0.003 \leq p/p_{cr} \leq 0.78$</p> <p>see ref. 124 for correlations for water, hydrocarbon, cryogenics fluids.</p>
Stephan & Abdelsalam (151)	<p>For refrigerants in natural convection boiling.</p> $h = C_4 q^{0.745}$ <p>C_4 is calculated from a plot in ref. 124.</p> <p>See ref. for correlations for water, hydrocarbons and cryogenics fluids.</p>
Rohsenow (152)	$\frac{C_{p_l} \Delta T}{h_{fg}} = C_{sf} \left[\frac{q}{\mu_l h_{fg} \sqrt{\frac{g_c \sigma}{g (\rho_l - \rho_v)}}} \right]^r Pr_l^{1.7}$
Chang (153)	<p>For feeble boiling</p> $q = 0.145 \frac{K_l}{D} \left(1 + \frac{\epsilon}{\alpha_l} \right)^{2/3} \Delta T^* (Gr - Pr)^{2/3}$

TABLE 2.4.3 CONTINUED

Investigators

Correlation

Nishikawa (154)

$$\frac{h D}{K_1} = 8.0 \left[F_s^{0.5} \left(\frac{P}{P_{atm}} \right) \left(\frac{C_{p1} \rho_1^2}{C_5^2 P K_1 h_{fg} \rho_v} \right)^{0.5} \right]^{2/3} * D^{1.5} q$$

where F_s is the foamability factor $F_s = 1$ for smooth pipes; $= 3.6$ for rough pipes C_5 is a constant & is equal to 900 m^{-1} .Forster &
Grief (155)

$$q = 4.3 * 10^{-5} \left(\frac{\alpha_1 C_{p1} \rho_1 T_s}{\sigma^{0.5} (h_{fg} \rho_v)^{1.5}} \right) *$$

$$\left(C_{p1} T_s \alpha_1^{1/2} \right)^{0.25} \left(\frac{\rho_1}{\rho_v} \right)^{5/8} Pr^{0.33} \Delta P^2$$

Levy (156)

$$q = \frac{K_1 C_{p1} \rho_1^2}{\sigma T_s (\rho_1 - \rho_v)} \left(\frac{1}{B_1} \right) (T_w - T_s)^3$$

where B_1 is a dimensionless parameter, a function of $\rho_v h_{fg}$.

McNelly (157)

$$\frac{h D}{K_1} = 0.225 \left(\frac{D q}{h_{fg} \mu_1} \right)^{0.67} \left(\frac{P D}{\sigma} \right)^{0.31} *$$

$$\left(\frac{\rho_1}{\rho_v} - 1 \right)^{0.33} \left(\frac{C_{p1} \mu_1}{K_1} \right)^{0.69}$$

TABLE 2.4.3 CONTINUED

Investigators

Correlation

Piret &
Isbin (158)

$$\frac{h D}{K_1} = 0.0086 \left(\frac{D V_m \rho_1}{\mu_1} \right)^{0.8} Pr_1^{0.6} \left(\frac{\sigma_w}{\sigma} \right)^{0.33}$$

Gilmour (159)

$$\left(\frac{h}{C_{p1} G} \right) Pr_1^{0.6} \left(\frac{\rho_1 \sigma}{p^2} \right)^{0.425} = \text{Const} *$$

$$\left(\frac{D G}{\mu} \right)^{-0.3}$$

$$\text{where } G = \frac{V_v \rho_1}{A \rho_v}$$

Const term depends on the characteristics
of the heating surface.

Mikic &
Rohsenow (160)

$$q = C_1 \frac{C_2 C_3^{1/2} r_s^m}{\pi 2^{m-1}} (K_1 C_{p1} \rho_1)^{1/2} *$$

$$\left(\frac{h_{fg} \rho_v}{T_{sat} \sigma} \right)^m \left[\frac{\sigma g_0 g (\rho_1 - \rho_v)}{\rho_1^2} \right]^{1/8} *$$

$$Ja^{15/8} \Delta T^{m+1}$$

where r is the radius of an active cavity
 Ja is the Jacob number. For C_1 , C_2 , C_3 &
 m see ref 134.

Lavin &
Young (180)

$$\frac{h_b D}{K_1} = 0.225 \left(\frac{q D}{h_{fg}} \right)^{0.69} Pr_1^{0.69} *$$

TABLE 2.4.3 CONTINUED

Investigators

Correlation

$$\left(\frac{\rho_l}{\rho_v} - 1 \right)^{0.31} \left(\frac{P D}{\sigma} \right)^{0.31}$$

Rhee &
Young (181)

$$Nu = 0.023 Re_l^{0.8} Pr_l^{0.4} + 0.3 Re_{th}^{0.69} *$$

$$\left(\frac{\rho_l}{\rho_v} - 1 \right)^{0.31} \left(\frac{P D}{\sigma} \right)^{0.31}$$

TABLE 2.4.5A Correlations to predict two-phase heat transfer coefficients

Values of the coefficients a and b found by various workers

$$(h_{tp}/h_{lo}) \text{ or } (h_{tp}/h_l) = a (1/X_{tt})^b$$

Authors	a	b	h_l base	Working fluid	Maximum exit x(%)
Dengler & (162) Addoms	3.5	0.5	h_{lo}	Water	70
Guerrierri (163) & Talty	3.4	0.45	h_l	varius organics	12
Schrock & (164) Grossman	2.5	0.75	h_{lo}	Water	59
Wright(165)	2.721	0.581	h_l	Water	19
Somerville (166)	7.55	0.328	h_l	n-Butanol	31
Collier & (167) Pulling	2.167	0.699	h_l	Water	66
Pujol & (168) Stenning	4.0	0.37	h_{lo}	Fr-113	70

TABLE 2.4.5B Correlations to predict two-phase heat transfer coefficients

Values of constants C_1 , C_2 , n_1 and n_2 found by various workers
$$(h_{tp}/h_1) = C_1 [Bo + C_2 (1/x_{tt})^{n_1}]^{n_2}$$

Authors	C_1	C_2	n_1	n_2	h_1 base	Fluid/ Max.exit x(%)
Schrock & (164) Grossman	7.39	1.5	0.67	1	h_{10}	Water/59
Sani(169)	14.8	1.5	0.67	1	h_1	Water/14
Wright (165)	6.7	3.5	0.67	1	h_1	Water/19
Somer- (166) ville	24.5	1.5	0.67	1	h_1	n-Butanol/31
Chaddock & (170) Brunemann	19.1	1.5	0.67	0.6	h_{10}	various chlorinated paraffins/75
Pujol & (168) Stenning	9.0	4.45	0.37	1	h_{10}	Fr-113/70
Pujol & (168) Stenning	5.3	7.55	0.37	1	h_{10}	Fr-113/70

TABLE 2.4.5C Correlations to predict two-phase heat transfer coefficients

Investigators	Correlation
Kvamme (171)	$h_{tp} = 1.5 \left[\frac{(1-x)}{R_1} \right]^{0.8}$ <p>R_1 is the Martinelli holdup</p>
Groothuis & Hendl (172)	$h_{tp} = 0.029 \left(\frac{K_1}{D} \right) \frac{D G_t}{\mu_{tp}}^{0.87} Pr_1^{0.33} \left(\frac{\mu_b}{\mu_w} \right)^{0.14}$
Sachs & Long (173)	<p>Working fluid R-11</p> $h_{tp} = 0.023 \left(\frac{K_1}{D} \right) Re_{tp}^{0.8} Pr_1^{0.33}$ <p>where $Re_{tp} = (V_v D_v \rho_1 / \mu_1)$</p>
Shah (174)	$h_{tp} = h_1 \left(1 + \frac{V_g}{V_l} \right)^{0.25}$ <p>V_g & V_l are superficial vapour & liquid velocities.</p>
Chaddock & Noeranger (175)	<p>Working fluid R-12</p> $h_{tp} = 1.85 \left[10^4 Bo + 1.5 (1/X_{tt}) \right]^{2/3} h_1$ <p>(modified correlation of Schrock & Grossman)</p> $h_{tp} = 3.0 h_1 (1/X_{tt})^{0.667}$ <p>(modified correlation of Dangler & Addoms)</p>

TABLE 2.4.5C CONTINUED

Investigators

Correlation

Coulson &
McNelly (176)

$$h_{tp} = \left(1.3 + \frac{D}{A} \right) \left(\frac{K_1}{D} \right) \left[Pr_1^{0.9} Re_1^{0.23} * \right. \\ \left. Re_v^{0.34} \left(\frac{\rho_1}{\rho_v} \right)^{0.23} \left(\frac{\mu_y}{\mu_l} \right) \right]$$

A is a constant & is equal to 12 m^{-1} .

Bo Pierre (177)

$$h_{tp} = 0.0009 \left(\frac{K_1}{D} \right) \left(\frac{D G}{\mu_1} \right) A^{0.5}$$

valid for $x > 90 \%$

$$h_{tp} = 0.0082 \left(\frac{K_1}{D} \right) \left(\frac{D G}{\mu_1} \right)^{0.8} A^{0.4}$$

can be used for superheats upto 6 C at
the exit of the evaporator.

$$A = \frac{h_{fg} J \Delta x}{L}$$

both equations valid for $10^9 < Re_1^2 A < 7 \times 10^{11}$.

Altman (178)

Working fluid R-22

$$h_{tp} = h_l \left[0.0225 \left(\frac{K_1}{D} \right) (Re_1^2 A)^{0.375} \right]$$

TABLE 2.4.5C CONTINUED

Investigators	Correlation
Ananiev (179)	$h_{tp} = h_l \left(\frac{\rho_l}{\rho_{av}} \right)^{0.5}$ $\rho_{av} = \frac{1.0}{\frac{x}{\rho_g} + \frac{(1-x)}{\rho_l}}$ <p>The average coefficient for complete evaporation is approximately given by;</p> $\bar{h}_{tp} = (h_l/2) \left[1 + \left(\frac{\rho_l}{\rho_g} \right)^{0.5} \right]$
Lavin & Young (180)	<p>Working fluid R-12 & R-22</p> $h_{tp} = A h_l Bo^{0.1} \left[\frac{1+x}{1-x} \right]^{1.16}$ <p>where A = 3.79 for vertical tubes & A = 6.59 for horizontal tubes</p>
Rhee & Young (181)	<p>Working fluid R-12 & R-22</p> <p>For annular-mist flow regime</p> $h_{tp} = 59.03 h_l \left[\frac{1+x}{1-x} \right]^{0.81} Fl^{-0.30}$

TABLE 2.4.5C CONTINUED

Investigators

Correlation

For stratified flow regime

$$h_{tp} = 313.76 h_l \left[\frac{1+x}{1-x} \right]^{-0.542} Fl^{-0.448}$$

For annular (no mist) flow regime

$$h_{tp} = 140.80 h_l \left[\frac{1+x}{1-x} \right]^{0.565} Fl^{0.41}$$

where $Fl = 1/Bo$

Chen (182)

$$h_{tp} = h_{mac} + h_{mic}$$

where $h_{mac} = h_l F (1/X_{tt})$, &

$$h_{mic} = \frac{0.00122 K_1^{0.79} C_{p1}^{0.45} \rho_l^{0.47} g_c^{0.25}}{\sigma^{0.5} \mu_l^{0.29} h_{fg}^{0.24} \rho_v^{0.24}}$$

$$* \Delta T_s^{0.24} \Delta P_s^{0.75} S Re_{tp}$$

S is the boiling suppression factor

Bennett (183)

Presented a relation for S for Chen's correlation

$$S = \phi_{l_{tt}}^{0.89} \left[\frac{Pr_l + 1}{2} \right]^{0.44}$$

$$\phi_{l_{tt}} = 1 + 20/X_{tt} + 1/X_{tt}^2$$

TABLE 2.4.5C CONTINUED

Investigators	Correlation
Bennett & Chen (184)	<p>Working fluids: distilled water & ethylene glycol</p> $h_{tp} = h_{mac} + h_{mic}$ $h_{mac} = 0.023 Re_1^{0.8} Pr_1^{0.4} \frac{K_1}{D} \left[\frac{(dp/dz)_{tp}}{(dp/dz)_1} \right]^{0.44}$ $* F(Pr_1) \left(\frac{\Delta T}{\Delta T_s} \right)_{mac}$ <p>where h_{mic} is same as in Chen's correlation</p> $\left(\frac{\Delta T}{\Delta T_s} \right)_{mac} = 1.0 \text{ for pure components}$ $F(Pr_1) = Pr_1^{0.296} \text{ for pure components}$
Uchida & Yamaguchi (185)	<p>Working fluid R-12</p> $h_{tp} = 17.5 Bo^{0.2} (1/X_{tt})^{1.67}$
Chawla (186)	<p>Working fluid R-11</p> $h_{tp} = \left(\frac{K_1}{D} \right) C_1 (Re_1 Fr_1)^{C_2} \left(\frac{x}{1-x} \right) *$ $\left(\frac{\rho_1}{\rho_g} \right)^{0.8} Re_1^{0.35} Pr_1^{0.45}$ <p>see reference 160 for C_1 & C_2.</p>

TABLE 2.4.5C CONTINUED

Investigators	Correlation
Steiner & Schlunder (187)	<p>Working fluid nitrogen</p> $Nu_{tp} = 1.05 * 10^{-2} K_1^a K_2^b K_3^c K_4^d K_5^e K_6^f F_{wet}$ <p>where $K_1 - K_6$ are dimensionless group & a, b, c, d, e, f & F_{wet} are constants. See reference 161 for these values.</p>
Shah (188)	<p>Working fluids: water, refrigerant cyclohexane</p> <p>Graphic correlation presented in paper.</p>
Kandlikar & Thakur (189)	<p>Working fluids: water, refrigerant cyclohexane</p> $h_{tp} = B1 CO^{B2} (25 Fr_1)^{B5} h_l + B3 Bo^{B4} Fr_1^{B6} h_l$ $\text{where } CO = \left(\frac{1-x}{x} \right)^{0.8} \left(\frac{\rho_g}{\rho_l} \right)^{0.5}$ <p>$B1 - B6$ are constants. See reference 163.</p>
Purcupile et al. (190)	<p>Working fluids: R-11, R-12 & R-113</p> $T_w - T_s = 22.8 \left(\frac{\rho_l}{\rho} \right)^{-0.25} q^{0.3} D^{0.6}$ <p style="text-align: center;">v</p>
Elamvaluthi & Srinivas (191)	$h_{tp} = 0.5 \left(\frac{K_1}{D} \right) Re_m^{0.7} Pr^{0.33} \left(\frac{\mu_g}{\mu_l} \right)^{0.25}$ $* \left(\frac{\mu_b}{\mu_w} \right)^{0.14}$

TABLE 2.4.5C CONTINUED

Investigators	Correlation
	<p>where Re_m is the Reynolds number of mixed flow</p> $Re_m = \frac{D V_l \rho_l}{\mu_l} + \frac{D V_g \rho_g}{\mu_g}$
Mobarak et al. (192)	<p>For $x < 0.1$</p> $Nu_{tp} = 0.8 \left(\frac{Re_{tp}}{Re_l^{0.75}} \right)^{1.15}$ <p>For $x > 0.1$</p> $Nu_{tp} = 2.75 \left(\frac{Re_{tp}}{Re_l^{0.75}} \right)^{0.75}$ <p>where $Re_{tp} = G D / \mu_{tp}$</p> $\& \frac{1}{\mu_{tp}} = \frac{x}{\mu_v} + \frac{1-x}{\mu_l}$
Davis & David (161)	<p>Working fluid; water & steam</p> <p>For separated annular flow model</p> $h_{tp} = 0.06 \left(\frac{K_1}{D} \right) \left(\frac{D G_t}{\mu_l} \right)^{0.87} Pr_l^{0.4} \left(\frac{\rho_l}{\rho_g} \right)^{0.28}$ <p>For homogenous flow model</p> $h_{tp} = 0.0333 \left(\frac{K_1}{D} \right) \left(\frac{D G_t}{\mu_{tp}} \right)^{0.87} Pr_l^{0.4}$

TABLE 2.4.6 Correlations to predict the critical quality

Investigators	Correlation
Sthapak et al. (193)	<p>Working fluid: R-12</p> $x_{cr} = 7.943 \left[Re_g (2.03 * 10^4 Re_g^{-0.81} \Delta T - 1) \right]^{-0.161}$ $\Delta T = T_{wall} - T_{sat}$
Lavin & Young (180)	<p>Working fluids: R-12, R-22</p> $We = 3.07 * 10^{-8} Re_g (1/Bo)^{0.06} \left(\frac{\rho_l}{\rho_v} \right)$
Rhee & Young (181)	<p>Working fluids: R-12, R-22</p> $We = 1.89 * 10^{-7} Re_g \left(\frac{\rho_l}{\rho_v} \right)^{0.6} \left(\frac{D}{0.0608} \right)^{-0.48}$ <p>D is in ft.</p>
Roko et al. (194)	<p>Working fluid: steam & water</p> $\frac{1 - x_{cr}}{x_{cr}} = \left[12.01 \left(\frac{p}{p_{cr}} \right)^2 - 5.628 \left(\frac{p}{p_{cr}} \right) + 0.8623 \right] \left(\frac{G}{1000} \right)$
Jones & Altman (195)	$x_{cr} = 1.04 - 1.14 * 10^{-5} \left[\left(\frac{D G}{\mu_l} \right) * \left(\frac{J \Delta x h_{fg}}{L} \right)^{3/8} \right]$

TABLE 2.4.6 CONTINUED

Investigators

Correlation

Biasi

et al. (196)

$$\frac{dG_{lf}}{dx} = -G + 2.3 \frac{\rho_g^{1.5} \tau_c^{0.5} S_{mc} h_{fg}}{\rho_l \phi x}$$

$$* (G_{lfeq} - G_{lf})$$

By integrating this equation, the dryout point can be determined as a point in which the liquid film flowrate reduces to zero.

lf = liquid in the film

c = film core interface

mc = mean in the core

Chien &

Ibele (197)

$$\left(\frac{4 \dot{m}_g}{D_g} \right) \left(\frac{4 \dot{m}_l}{D_l} \right)^{0.301} = 1.199 * 10^6$$

TABLE 2.4.7 Correlations to predict the heat transfer coefficients in the liquid deficient region

Investigators	Correlation
Unal & Gasselt (198)	$Nu = 0.0091 a_1 a_2 a_3 a_4 a_5 a_6 a_7 a_8$ <p>where $a_1 - a_8$ are constants. see ref. 172</p>
Lavin & Young (180)	$h = 0.0162 \left(\frac{K_g}{D} \right) Re_g^{0.84} Pr_g^{0.33} (1 - x)^{0.1}$
Rhee & Young (181)	<p>Working fluid: R-12, R-22</p> $Nu = 0.023 Re_g^{0.8} Pr_g^{0.4} x^{-1.23}$ <p>(for R-12)</p> $Nu = 0.038 Re_g^{0.8} Pr_g^{0.4} x^{-1.23}$ <p>(for R-22)</p>
Heinman (199)	$Nu = 0.0157 Re_l^{0.84} Pr_l^{0.33} \left(\frac{L}{D} \right)^{-0.04}$ <p>for $6 < L/D < 60$</p> $Nu = 0.0133 Re_l^{0.84} Pr_l^{0.33}$ <p>for $L/D > 60$</p>
Varma (200)	$\frac{h}{h_g} = f(\eta, \dot{m})$ <p>where $\eta = \frac{x - x_{cr}}{1 - x_{cr}}$; \dot{m} is the mass velocity</p>

TABLE 2.5 Correlations to predict the condensation coefficients

Investigators	Correlation
Soliman et al. (210)	<p>For annular flow condensation</p> $\frac{h_z \mu_1}{K_1 \rho_1^{0.5}} = 0.036 \text{ Pr}_1^{0.65} \text{ Fo}^{0.5}$ <p>Fo is the wall shear stress</p>
Altman et al. (211)	<p>For horizontal tubes</p> $\frac{h_z \mu_1}{K_1 \rho_1^{0.5}} = 0.057 \text{ Pr}_1^{0.5} \text{ F}^{0.5}$ <p>F is the shear stress in laminar condensate sublayer</p>
Akers & Rosson (212)	$\text{Nu} = 13.8 \text{ Pr}_1^{0.33} \left(\frac{h_{fg}}{C_{p1} \Delta T} \right)^{1/6} \text{Re}_g^{0.2}$ <p>for $1000 < \text{Re}_g < 20,000$</p> $\text{Nu} = 0.1 \text{ Pr}_1^{0.33} \left(\frac{h_{fg}}{C_{p1} \Delta T} \right)^{1/6} \text{Re}_g^{0.67}$ <p>for $20,000 < \text{Re}_g < 100,000$</p>
Akers et al. (213)	$\text{Nu} = 0.0265 \text{ Pr}_1^{0.33} \left(\frac{D G_t}{\mu_1} \right)^{0.8}$ <p>for $\text{Re} > 5 \times 10^4$</p>

TABLE 2.5 CONTINUED

Investigators

Correlation

$$Nu = 5.03 Pr_1^{0.33} \left(\frac{D G_t}{\mu_1} \right)^{1/3}$$

for $Re < 5 \times 10^5$

Chato (214)

For horizontal tubes

$$h_m = 0.357 C_5 \left[\frac{K_1^3 \rho_1 (\rho_1 - \rho_v) h_{fg} g X}{\mu_1 D \Delta T} \right]^{0.25}$$

where $X = 1 + 0.68 (Cp_1 \Delta T / h_{fg})$ C_5 is a constant & is a function of angle.see reference for C_5 (in graphical form).

Matsuki

For horizontal annulus

et al. (216)

$$Nu_m = 1.69 \times 10^{-3} Pr_1^{0.33} Re_{eq}^{1.25}$$

where $Re_{eq} = (Re_g + Re_l)/2$

Honda &

See reference for details.

Nozu (217)

Meyers &

Rosson (232)

$$Nu = C_4 Re^{-0.14} \left(\frac{D^3 \rho_1^2 g}{\mu_1^2} \right)^{1/3}$$

where $C_4 = 1.3$ for methanol. For other fluids see the paper.

TABLE 2.5 CONTINUED

Investigators	Correlation
Traviss et al. (227)	$Nu_z = F_2^{-1} 0.15 C_{P1} \mu_g^{0.1} K_1 x^{0.9} \left(\frac{\rho_1}{\rho_g} \right)^{0.5} *$ $\left(\frac{G^{0.9}}{D^{1.1}} \right) A$ <p>where $A = 1 + 2.85 \left[\left(\frac{\mu_1}{\mu_g} \right)^{0.1} \left(\frac{1-x}{x} \right)^{0.9} * \right.$</p> $\left. \left(\frac{\rho_g}{\rho_1} \right)^{0.523} \right]$ $F_2 = 0.707 Pr_1 Re_1^{0.5} \quad \text{for } Re_1 < 50$ $F_2 = 5 Pr_1 + 5 \ln \left[1 + Pr_1 (0.09636 Re_1^{0.585} - 1) \right]$ <p>for $50 < Re_1 < 1125$</p>
Ali & McDonald (228)	$\bar{h}_c = B_1 \left[\frac{K_1^3 \rho_1^2 h_{fg} g}{\mu_1 D (T_s - T_w)} \right]^{1/4}$ <p>for $0^\circ < \beta < 40^\circ$; $B_1 = 0.727$</p> <p>for $40^\circ < \beta < 90^\circ$; $B_1 = 0.727 \left[\cos \left(\frac{\beta - 40}{50} \right) 90 \right]^{0.25}$</p>
Nusselt (229)	<p>For horizontal tube</p> $h_m = 0.725 \left(\frac{K_1^3 \rho_1^2 h_{fg} g}{\mu_1 D \Delta T} \right)^{1/4}$

TABLE 2.5 CONTINUED

Investigators	Correlation
	<p>For vertical plates (also valid for vertical tubes for small film thickness)</p> $h_m = 0.943 \left[\frac{\rho_l (\rho_l - \rho_v) K_l^3 h_{fg} g}{L \mu_l \Delta T} \right]^{1/4}$ <p>For inclined surfaces (also valid for vertical tubes for small film thickness)</p> $h_m = 0.943 \left[\frac{\rho_l (\rho_l - \rho_v) K_l^3 h_{fg} g \sin \beta}{L \mu_l \Delta T} \right]^{1/4}$ <p>where β is the inclination angle.</p>
Bromley et al. (230)	<p>For horizontal tubes</p> $h_m = 0.728 \left[\frac{K_l^3 \rho_l (\rho_l - \rho_v) h_{fg} g \left(1 + 4 \frac{C_{p_l} \Delta T}{h_{fg}} \right)^{0.25}}{D \mu_l \Delta T} \right]$ <p>For vertical tubes</p> $h_m = 0.943 \left[\frac{K_l^3 \rho_l (\rho_l - \rho_v) h_{fg} g \left(1 + 4 \frac{C_{p_l} \Delta T}{h_{fg}} \right)^{0.25}}{D \mu_l \Delta T} \right]$

TABLE 2.5 CONTINUED

Investigators	Correlation
Rohsenow (231)	$h_m = 0.943 \left[\frac{K_1^3 \rho_1 (\rho_1 - \rho_v) h_{fg} X}{L \mu_1 \Delta T \left(1 - \frac{C_{p1} \Delta T}{10 h_{fg}} \right)^4} \right]^{1/4}$ <p>where $X = 1 - \frac{C_{p1} \Delta T}{h_{fg}} - 0.0328 \left[\frac{\Delta T}{h_{fg}} \right]^2$</p> $h_{fg} = h_{fg} + 3/8 C_{p1} \Delta T$ $h_m = 0.943 \left[\frac{K_1^3 \rho_1 (\rho_1 - \rho_v) h_{fg} g \left(1 + 0.68 \frac{C_{p1} \Delta T}{h_{fg}} \right)}{L \mu_1 \Delta T} \right]^{1/4}$ <p>for $0 < (C_{p1} \Delta T / h_{fg}) < 1.0$</p>
Hassan & Jakob (233)	<p>For inclined cylinders</p> $h_m = 0.727 \left[\frac{\rho_1^2 h_{fg} K_1^3 \cos \beta}{\mu_1 D \Delta T} \right]^{1/4}$
Rao & Murthy (209)	<p>For dropwise condensation</p> <p>See reference for details</p>

TABLE 2.6.2 Correlations to predict the pressure drop and void fraction for two-phase region

Investigators

Correlation

Lockhart &

For two-phase frictional pressure drop

Martinelli (249)

$$\phi^2 = \frac{(\text{dp/dz})_{tp}}{(\text{dp/dz})_l}$$

For turbulent - turbulent flow condition

$$X_{tt} = \left(\frac{1-x}{x} \right)^{0.9} \left(\frac{\rho_v}{\rho_l} \right)^{0.5} \left(\frac{\mu_l}{\mu_v} \right)^{0.1}$$

$$\phi_{tt} = (1 + X_{tt}^{-0.49})^{2.05}$$

For turbulent - laminar flow condition

$$X_{tl} = 0.054 \text{Re}_l^{0.4} \left(\frac{1-x}{x} \right) \left(\frac{\rho_v}{\rho_l} \right) \left(\frac{\mu_l}{\mu_v} \right)^{0.5}$$

$$\phi_{tl} = (1 + X_{tl}^{-0.56})^{1.80}$$

For laminar - laminar flow condition

$$X_{ll} = \left(\frac{1-x}{x} \right) \left(\frac{\rho_v}{\rho_l} \right) \left(\frac{\mu_l}{\mu_v} \right)^{0.5}$$

$$\phi_{ll} = (1 + X_{ll}^{-0.70})^{1.43}$$

For laminar - turbulent flow condition

$$X_{lt} = 18.65 \text{Re}_v^{-0.4} \left(\frac{1-x}{x} \right) \left(\frac{\rho_v}{\rho_l} \right) \left(\frac{\mu_l}{\mu_v} \right)^{0.5}$$

$$\phi_{lt} = (1 + X_{lt}^{-0.56})^{1.80}$$

TABLE 2.6.2 CONTINUED

Investigators	Correlation
	$\alpha = (1 + x^{0.8})^{-0.378}$
Levy (254)	<p>For two-phase frictional pressure drop</p> $\left(\frac{dp}{dz}\right)_{tp} = \left(\frac{dp}{dz}\right)_l \left[\frac{1}{(1 - \alpha)^2} \right]$
Larsen & Tong (262)	<p>For two-phase frictional pressure drop</p> $f_{tp} = 16.5 \left[\frac{V D \rho_1}{(1 - \alpha) \mu_1} \right]^{-0.6}$
Owens & Schrock (251)	<p>For two-phase frictional pressure drop</p> $\Delta P_{tp} = f \left(\frac{L}{D} \right) \left(\frac{V_{tp}^2}{2 g_c} \right) \rho_{tp}$ <p>where $V_{tp} = G_t v_{tp}$</p> $v_{tp} = v_l + x (v_v - v_l)$
Bonnecaze et al. (268)	<p>For total pressure drop in slug flow</p> $\left(\frac{dp}{dz}\right) = \left(\frac{R_l - R_{lf}}{1 - R_{lf}}\right) \left[g \rho_l \sin \theta + \right. \\ \left. + \left(\frac{2 \rho_l f_{sf} V_t^2}{D} \right) \right]$ <p>where $f_{sf} = 0.0048 + 3980/Re^{1.285}$</p>

TABLE 2.6.2 CONTINUED

Investigators	Correlation
	$R_{lf} = (1/\pi) \left[\cos^{-1}(1-n_s) - (1-n_s) \sqrt{(n_s(2-n_s))} \right]$ <p> n_s = ratio of liquid depth to pipe radius R_l = liquid holdup </p>
Beggs (267)	<p>For two-phase frictional pressure drop</p> $f_{tp} = \left(\frac{2 D}{G_t V_t} \right) \left[\frac{dp}{dz} \left(1 - \frac{\rho_{tp} V_t V_{sg}}{P} \right) - (\rho_{tp} g \sin \beta) \right]$
Beattie & Whalley (248)	<p>For two-phase frictional pressure drop</p> $\left(\frac{dp}{dz} \right)_{tp} = f \left(\frac{2 G^2}{\rho_{av} D} \right)$ <p>Reynolds number is based on ρ_{tp}.</p>
Rouet (235)	<p>For turbulent - turbulent flow condition (for Lockhart - Martinelli correlation)</p> $\alpha = 1 - (m/2) X_{tt}^{2/(2-b)} - \left[(1+(m/2) X_{tt}^{2/(2-b)})^2 - 1 \right]^{0.5}$ <p>where $m = 0.054$</p> $\alpha = 1 - (1/\phi_{tt}) ; \phi_{tt}^2 = 1 + 2/X_{tt} + 1/X_{tt}^2$

TABLE 2.6.2 CONTINUED

Investigators	Correlation
Spedding & Chen (259)	<p data-bbox="630 464 1013 497">For vertical upward flow</p> $\frac{\bar{R}_g}{\bar{R}_l} = \frac{1}{(0.2 + K1 Q_l/Q_g)}$ <p data-bbox="639 726 894 772">for $\bar{R}_g/\bar{R}_l \leq 4.0$</p> $\frac{\bar{R}_g}{\bar{R}_l} = K2 \left[1 - \exp(-K3 Q_g/Q_l) \right] (Q_g/Q_l)^{0.65}$ <p data-bbox="651 995 1000 1041">for $\bar{R}_g/\bar{R}_l > 4.0 - 275$</p> <p data-bbox="654 1077 956 1110">For horizontal flow</p> $\frac{\bar{R}_g}{\bar{R}_l} = 0.45 (Q_g/Q_l)^{0.65}$ <p data-bbox="667 1346 1377 1455">where Q_g and Q_l are the flow rates of liquid and gas respectively.</p> $\ln K1 = -1.44 \ln V_{sl} - 0.007$ $K2 = 0.14 \ln V_{sl} + 1.0$ $\ln K3 = 0.97 \ln V_{sl} - 3.0$

TABLE 2.6.2 CONTINUED

Investigators

Correlation

Smith (271)

1

$$\alpha = \frac{1 + \frac{1-x}{x} \frac{\rho_g}{\rho_l}}{K - (1-K) \left\{ \frac{\frac{\rho_l}{\rho_g} + \frac{1-x}{x}}{1 + \frac{1-x}{x}} \right\}^{1/2}}$$

where K is a function of ρ_l/ρ_g

Hashizume (79),

$$\alpha = \frac{\rho_l - \frac{\rho_{sh} (V1 + V2 + Vo) - \rho_l V2}{V1}}{\rho_l - \rho_g}$$

where ρ_{sh} is the density of refrigerant in superheated gas phase under ambient condition. V1, V2 & Vo are the volumes of different test sections for Hashizume's apparatus.

Fujie (258)

$$x = \frac{\alpha}{m(1-\alpha)} \left[1 + \frac{1/4}{1 - \sqrt{\alpha}} \sqrt{K1} \right]$$

$$\text{where } m = \rho_l/\rho_g ; K1 = \frac{f \rho_l A}{2 D C1}$$

A & D are the cross sectional area and diameter of the tube & C1 is a constant.

TABLE 2.6.2 CONTINUED

Investigators	Correlation
Armand (269)	For bubble & slug flow $\frac{\bar{R}_g}{\bar{R}_l} = \frac{1}{0.2 + 1.2 (Q_l/Q_g)}$
Madsen (261)	$\alpha = \frac{1}{1 + \left(\frac{\rho_l}{\rho_g}\right)^{0.5} \left(\frac{1-x}{x}\right)^m}$ <p>where $m = \frac{0.5 \log(\rho_l/\rho_g) - \log[\alpha_o/(1-\alpha_o)]}{\log(\rho_l/\rho_g) - \log[\alpha_o/(1-\alpha_o)]}$</p> <p>where α_o is the value of void fraction at $V_l = V_g$</p>
Bankoff (270)	$\alpha = \frac{B}{1 + \frac{1-x}{x} \frac{\rho_v}{\rho_l}}$ <p>where $B = 0.71 + 0.001 p$</p>
Thom (253)	$\alpha = \frac{1}{1 + \frac{1-x}{x} \frac{1}{A}}$ <p>where A is a function of ρ_l/ρ_g.</p>

TABLE 2.6.2 CONTINUED

Investigators

Correlation

Chen &

For horizontal annular flow

Spedding (241)

$$\bar{R}_g = \frac{1}{1 + X^{1.5}}$$

$$\bar{R}_l = \frac{X^{1.5}}{1 + X^{1.5}}$$

For two-phase frictional pressure drop

$$\left(\frac{dp}{dz}\right)_{tp} = Re_{sg}^{1.8} f \left(\phi_g^2 - 1 \right) + \left(\frac{dp}{dz}\right)_{sg}$$

Tandon

For annular two-phase flow

et al. (260)

$$\alpha = \left[1 - 1.928 Re_1^{-0.315} F(X_{tt})^{-1} + 0.9293 Re_1^{-0.63} F(X_{tt})^{-2} \right]$$

for $50 < Re_1 < 1125$

$$\alpha = \left[1 - 0.38 Re_1^{-0.088} F(X_{tt})^{-1} + 0.0361 Re_1^{-0.176} F(X_{tt})^{-2} \right]$$

for $Re_1 > 1125$

APPENDIX B

Experimental data of Yin & Abdelmessih (reference 127) for increasing and decreasing heat fluxes.

q (W/m ²)	$\left[\frac{(T_w - T_{sat})^2}{\left(\frac{2 \sigma T_{sat}}{K_l h_{fg} \rho_v} \right)} \right] \text{ (W/m}^2\text{)}$	
	Increasing	Decreasing
1000	46870	2500
3000	128250	7500
5000	194112	12500
7000	245580	17500
9000	283793	22500
11000	309888	27500
13000	325000	32500
15000	330266	37500
17000	326822	42500
19000	315804	47500
21000	298349	52500
23000	275591	57500
25000	248669	62500
27000	218715	67500
29000	186870	72500

APPENDIX C

EQUATIONS USED IN THE SIMULATION PROGRAM OF ALI (4)Heat Transfer Coefficient CalculationsSingle-phase forced Convection (point value). (1).

For laminar flow

$$h_a = 1.24 \frac{kF_1}{D} (Re \ Pr \ D/Z)^{1/3} \left(\frac{\mu}{\mu_w}\right)^{0.14} \quad (A1)$$

For turbulent flow

$$h_a = 0.027 \frac{kF_2}{D} Re^{0.8} Pr^n \left(\frac{\mu}{\mu_w}\right)^{0.14}$$

where -

$$n = 0.4 \text{ for heating, } n = 0.3 \text{ for cooling} \quad (A2)$$

Two-phase forced convection (point value)

$$Re_f = \frac{V_g D_g \rho_m}{\mu_L} \quad (A3)$$

where

V_g is the actual vapor velocity, D_g is the diameter of the vapor core, μ_L is the viscosity of liquid, and

$$\rho_m = \alpha \rho_g + (1-\alpha) \rho_L \quad (A4)$$

For laminar flow

$$h_f = 1.24 \frac{k_L}{D} (Re_f \ Pr_L \ \frac{D}{Z})^{1/3} \quad (A5)$$

For turbulent flow

$$h_f = 0.023 \frac{k_L}{D} Re_f^{0.8} Pr_L^{0.4} \quad (A5)$$

Entrance effect factors (3,4).

$$F_1 = 1 + 5 \exp\left(\frac{-Z}{100}\right) \quad (A7)$$

$$F_2 = 1 + 7 \frac{D}{Z} \quad (A8)$$

$$F_3 = 0.344 Re^{-0.078153} \left(\frac{D}{Z}\right)^{1.93/Re^{0.3}} \quad (A9)$$

$$F_4 = 1 + 3.5 \exp\left(\frac{-Z}{100}\right) \quad (A10)$$

Boiling (5).

$$q = \mu_L \lambda \left[\frac{g(\rho_L - \rho_g)}{\sigma} \right]^{1/2} \left[\frac{C_L(T_w - T_s)}{\lambda(CSF) Pr^{1.7}} \right]^r \quad (A11)$$

where

CSF and r are dependent on the fluid and surface properties

$$h_b = q / (T_w - T_s) \quad (A12)$$

Combined boiling and two-phase convection.

$$h_{fb} = (h_f^2 + h_b^2)^{1/2} \quad (A13)$$

Combined boiling and two-phase forced convection with dryout (6).

Criterion for transition to dryout:

$$x > x_c \quad (A14)$$

where

$$x_c = \left[\frac{We \sigma \rho_g}{G_L \mu_L \left(\frac{Re_L}{1-\alpha} \right)} \right]^{0.5} \quad (A15)$$

and

$$We = 3.07 \times 10^{-8} \text{ Reg } \left(\frac{G_L \lambda}{q} \right)^{0.06} \left(\frac{\rho_L}{\rho_g} \right) \quad (A16)$$

$$h_{bfd} = C_1 h_d + C_2 h_{fb} \quad (A17)$$

where

$$C_2 = \left(\frac{1-x}{1-x_c} \right)^a \quad (A18)$$

and

$$C_1 = 1 - C_2 \quad (A19)$$

and $h_d = h_a$ for complete dryout at $T = T_s$ Subcooled boiling (7).If $T \leq T_s$ and if $q > q_{IB}$, then

$$h_{fb} = (h_f^p + h_b^p)^{1/p} \quad (A20)$$

where

$$p = 2 - (T_s - T_x) / (T_s - T_x)_{IB} \quad (A21)$$

and for turbulent flow,

$$q_{IB} = \frac{k_L \lambda}{8 \sigma T_s (v_g - v_L)} \left[\frac{0.02 v_L^{0.2} q^{0.8}}{0.023 \text{ Pr}_L^{0.4} k_L v_L^{0.8}} - (T_s - T) \right]^2 \quad (A22)$$

for laminar flow,

$$q_{IB} = \frac{k_L \lambda}{8 \sigma T_s (v_g - v_L)} \left[\frac{(D v_L Z)^{1/3} q}{1.86 v_L^{0.33} \text{ Pr}_L^{0.33} k_L} - (T_s - T) \right]^2 \quad (A23)$$

Condensation (8,9).

For an inclined tube

$$h_c = B \left(\frac{k_L^3 \rho_L^2 \lambda}{\mu_L D (T_s - T_w)} \right)^{1/4} \quad (A24)$$

where

B is a function of tube diameter and angle.

For a vertical tube

$$h_c = \frac{k_L}{\delta} \quad (A25)$$

where

$$\delta = \left\{ \frac{3\mu_L \Gamma}{\rho_L^2} \right\}^{1/3} \quad (A26)$$

Pressure Drop Calculations

Friction factor (10).

For laminar flow

$$f = 64/Re \quad (A27)$$

For turbulent flow

$$f = .316 Re^{-.25} \quad (A28)$$

Frictional pressure drop.

For single-phase flow

$$\Delta P_a = f \frac{Z}{D} \frac{\rho V^2}{2} \quad (A29)$$

For two-phase flow (11)

$$\Delta P_f = \Delta P_{1p} \phi^2 \quad (A30)$$

For liquid-phase laminar and vapor-phase laminar

$$\phi = [1 + \chi^{-.70}]^{1.43} \quad (A31)$$

where

$$\chi = \left\{ \frac{1-x}{x} \right\} \left\{ \frac{\rho_g}{\rho_L} \right\} \left\{ \frac{\mu_L}{\mu_g} \right\}^{0.5} \quad (A32)$$

For Liquid-phase laminar and vapor-phase turbulent

$$\phi = [1 + \chi^{-.56}]^{1.80} \quad (A33)$$

where

$$\chi = 348 Re_g^{-0.4} \left[\left\{ \frac{1-x}{x} \right\} \left\{ \frac{\rho_g}{\rho_L} \right\} \left\{ \frac{\mu_L}{\mu_g} \right\} \right]^{0.5} \quad (A34)$$

For liquid phase turbulent and vapor phase laminar

$$\phi = [1 + \chi^{-.56}]^{1.80} \quad (A35)$$

where

$$\chi = .054 Re_L^{0.4} \left[\left\{ \frac{1-x}{x} \right\} \left\{ \frac{\rho_g}{\rho_L} \right\} \left\{ \frac{\mu_L}{\mu_g} \right\} \right]^{0.5} \quad (A36)$$

For liquid phase turbulent and vapor phase turbulent

$$\phi = [1 + \chi^{-.49}]^{2.05} \quad (A37)$$

where

$$x = \left\{ \frac{\mu_L}{\mu_G} \right\}^{0.1} \left\{ \frac{\rho_g}{\rho_L} \right\} \left\{ \frac{1-x}{x} \right\}^{0.9} \quad (A38)$$

Acceleration pressure drop (11).

For separated flow

$$\Delta P = \frac{G_L^2 \Delta x}{\rho_g \alpha} (2x + \Delta x) - \frac{G_L^2 \Delta x}{\rho_L (1-\alpha)} (2-2x-\Delta x) \quad (A39)$$

For homogeneous flow

$$\Delta P = \frac{G_t^2}{\rho_L} (1-x) + x \left(\frac{\rho_L}{\rho_g} \right) - 1 \quad (A40)$$

Hydrostatic pressure drop

$$\Delta P = \rho g Z \quad (A41)$$

Liquid Fraction Calculation

$$R = (1 + x^{-.42})^{-2.38} \quad (A42)$$

NOMENCLATURE

- c - Specific heat, J/kgK
- CSF- Surface-liquid combination constant in Rohsenow's boiling equation
- D - Tube diameter, m
- F₁ - Entrance effect factor for laminar flow in the evaporator as defined in Eq. 7
- F₂ - Entrance factor for turbulent flow in the evaporator as defined in Eq. 8
- F₃ - Entrance effect factor for turbulent flow in the condenser as defined in Eq. 9
- F₄ - Entrance effect factor for laminar flow in the condenser as defined in Eq. 10
- f - Fanning friction factor
- G - Mass flow rate at a point per unit of flow cross-sectional area, kg/m²_s
- g - Acceleration due to gravity, m/s²
- h - Heat transfer coefficient, W/m²K
- k - Thermal conductivity, W/mK
- P - Pressure, N/M²
- p - exponent used to combine boiling heat transfer coefficient with single-phase convection as defined in Eq. (20)
- Pr - Prandtl number
- q - Heat flux, W/m²
- R - Liquid fraction

- r - Exponent used in Rohsenow's boiling equation (function of surface-liquid combination)
- Re - Reynolds number
- Reg - Superficial vapor Reynolds number defined as GxD/μ_g
- Rel - Superficial liquid Reynolds number defined as $G(1-x)D/\mu_L$
- T - Temperature (with no subscript, bulk fluid temperature (K))
- V - Velocity, M/s
- v - Specific volume, M^3/kg
- We - Weber number
- x - Quality
- xc - Quality at which partial dryout starts to occur
- Z - Distance along the evaporator or condenser from its entrance, m

-
- λ - Latent heat of vaporization, J/kg
 - μ - Dynamic viscosity, kg/m-s
 - σ - Surface tension, N/m
 - α - Void fraction
 - ρ - Fluid density, kg/m^3
 - δ - Liquid film thickness, m
 - Γ - Condensate mass flow rate per unit perimeter, kg/s-m
 - ν - Kinematic viscosity, m^2/s
 - x - Lockhart-Martinelli parameter defined in Eq. (A32)

Subscript

- a - Single-phase liquid
- b - Nucleate pool boiling
- c - Condensation
- d - Single-phase vapor
- f - Two-phase flow
- g - Saturated vapor state
- IB - Incipient boiling
- L - Liquid
- s - Saturation
- t - Total
- w - Wall

APPENDIX D

LIST OF PHYSICAL PROPERTIES AND CONSTANTS USED IN THE PROGRAM

In order to calculate the performance characteristics of a thermosiphon loop, the computer program requires subroutines to estimate the following physical properties of the working fluid:

1. Saturation pressure as a function of temperature
2. Saturation temperature as a function of pressure
3. Enthalpy of saturated liquid as a function of temperature
4. Density of liquid as a function of temperature
5. Latent heat of vaporization as a function of temperature
6. Surface tension for the liquid-vapor interface
7. Viscosity of liquid as a function of temperature
8. Thermal conductivity of liquid as a function of temperature
9. Specific heat of liquid as a function of temperature
10. Viscosity of vapor as a function of temperature
11. Specific heat of vapor as a function of temperature

The constants supplied to the program are dependent on the surface-liquid combination. They are

1. CSF: A constant used in Rohsenow's boiling equation (Eq. A11).
2. r : An exponent used in Rohsenow's boiling equation (Eq. A11).

The accuracy with which these subroutines calculate the required properties should be compatible with the tolerances which are specified by the user in the main program.

Also see appendix H2.

APPENDIX E

EQUATIONS FOR ESTIMATING HEAT TRANSFER COEFFICIENT IN ANNULI

To estimate the convective heat transfer coefficient in annulus, the correlations by Sieder-Tate and Monrad & Pelton (also used by Ali - see reference 2) for laminar and turbulent flow respectively were utilized.

For laminar flow:

$$Nu_m = 1.86 (Re Pr D/z)^{1/3} \quad \text{--- (C1)}$$

For turbulent flow:

$$Nu_m = 0.02 Pe^{0.8} Pr^{0.33} (D_o / D_i)^{0.53} \quad \text{--- (C2)}$$

In both cases Nu_m and Re were based on the equivalent diameter. However, it was also necessary to enhance the heat transfer coefficient in the entrance region. It was observed that if equations C1 and C2 were multiplied by the following entrance effect factors, satisfactory estimation of the heat transfer data in this region was possible.

For laminar flow:

$$M = 1 + 3.5 \exp (-z/90D) \quad \text{--- (C3)}$$

For turbulent flow:

$$M = 1 + 3.0 \exp (-z/30D) \quad \text{--- (C4)}$$

APPENDIX FESTIMATION OF AIR SIDE RESISTANCE

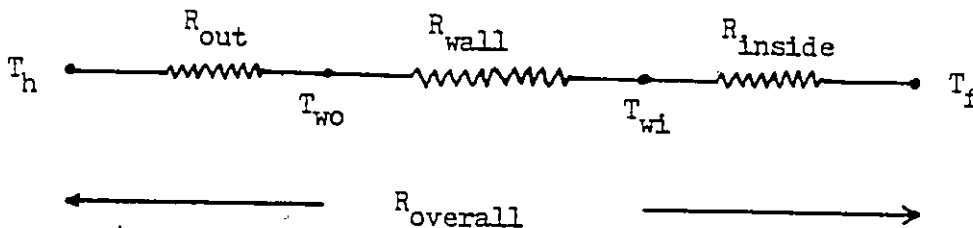
The airside resistance was calculated by using data from McQuay coil catalog. (McQuay Inc., Minneapolis, Minnesota)

A plot of effectiveness versus the face velocity for $3/8$ " O.D. copper tubes is shown in figure F1, which was taken from the McQuay coil catalog. For a specified number of fins/inch (12) and number of row deep (1), the effectiveness for the whole heat exchanger can be found for a given coil face velocity.

The data from this graph and other similar ones for $1/4$ " and $1/2$ " O.D. copper tube were correlated by the following equation:

$$E_{he} = B_0 + B_1 V + B_2 V^2 + B_3 V^3 \quad \text{--- (1)}$$

where V is the velocity of air in ft/sec and the values of B_0 , B_1 , B_2 , B_3 are shown in the Table F1.



Using an electrical analogy for the heat transfer resistance through the tube, as shown, the overall resistance, $R_{overall}$, can be calculated from

$$R_{overall} = R_{out} + R_{wall} + R_{inside}$$

The outside resistance, R_{out} , can be determined from the above equation if the other three resistances are known. The wall and the inside resistances can be determined easily by standard relations of heat transfer. R_{out} can be determined if the magnitude of $R_{overall}$ is known. The overall resistance, $R_{overall}$, was evaluated as follows.

Effectiveness is related to the number of transfer units by the following relation;

$$E_{he} = 1 - \exp(-NTU)$$

$$= 1 - \exp\left[-(U A)_{he}/(\dot{m} C_p)_{he}\right]$$

$$= 1 - \exp\left[-(U dA)_e/(\dot{m} C_p)_e\right]$$

$$\exp(-U dA/\dot{m} C_p)_e = 1 - E_{he}$$

$$\exp(U dA/\dot{m} C_p)_e = 1/(1 - E_{he})$$

$$(U dA/\dot{m} C_p)_e = \ln[1/(1 - E_{he})]$$

$$(U dA)_e = (\dot{m} C_p)_e \ln[1/(1 - E_{he})] \quad \text{--- (2)}$$

where subscript he and e stands for the heat exchanger and element respectively.

$$R_{overall} = 1/(U dA)_e$$

$$R_{wall} = \ln(D_o/D_i) \left[1/(2 \pi dL K_m)\right]$$

$$R_{inside} = 1/(h_{cond} D_i dL)$$

$$R_{out} = R_{overall} - (R_{wall} + R_{inside}) \quad \text{--- (3)}$$

R_{inside} was calculated by using the properties of R-22, since the graph shown (Figure F1) is for condensation of R-22.

Equation 3 was used for the calculation of the air side resistance.

TABLE F1

The following data was used in the evaluation of airside resistance.

Description	D I A M E T E R (N O M I N A L S I Z E)		
	1/2"	3/8"	1/4"
Pattern	1.25 * 1.082 Staggered	1.0 * 0.866 Staggered	1.0 * 0.866 Staggered
I.D.	0.436 "	0.311 "	0.190 "
Thickness	0.032 "	0.032 "	0.030 "
Coefficients for equation ($\overline{D1}$) (appendix F)			
B0	0.619453164	0.59718992	0.673042579
B1	-0.082210817	-0.08137122	-0.096060295
B2	$5.8274575/10^3$	$5.8313175/10^3$	$6.52356854/10^3$
B3	$-1.5016166/10^4$	$-1.5054689/10^4$	$-1.56466564/10^4$

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Section B, Page 3.2

DRY SURFACE CONDENSER COIL RATINGS - R22 **3/8 TUBE STANDARD COILS - 3CX**

10, 12, 14 & 15 FPI

TUBES: 3/8 O.D. Copper
 PATTERN: 1 x .866 Staggered

FINS: Rippled & Corrugated
 10, 12, 14 & 15 FPI-.006 Alum.

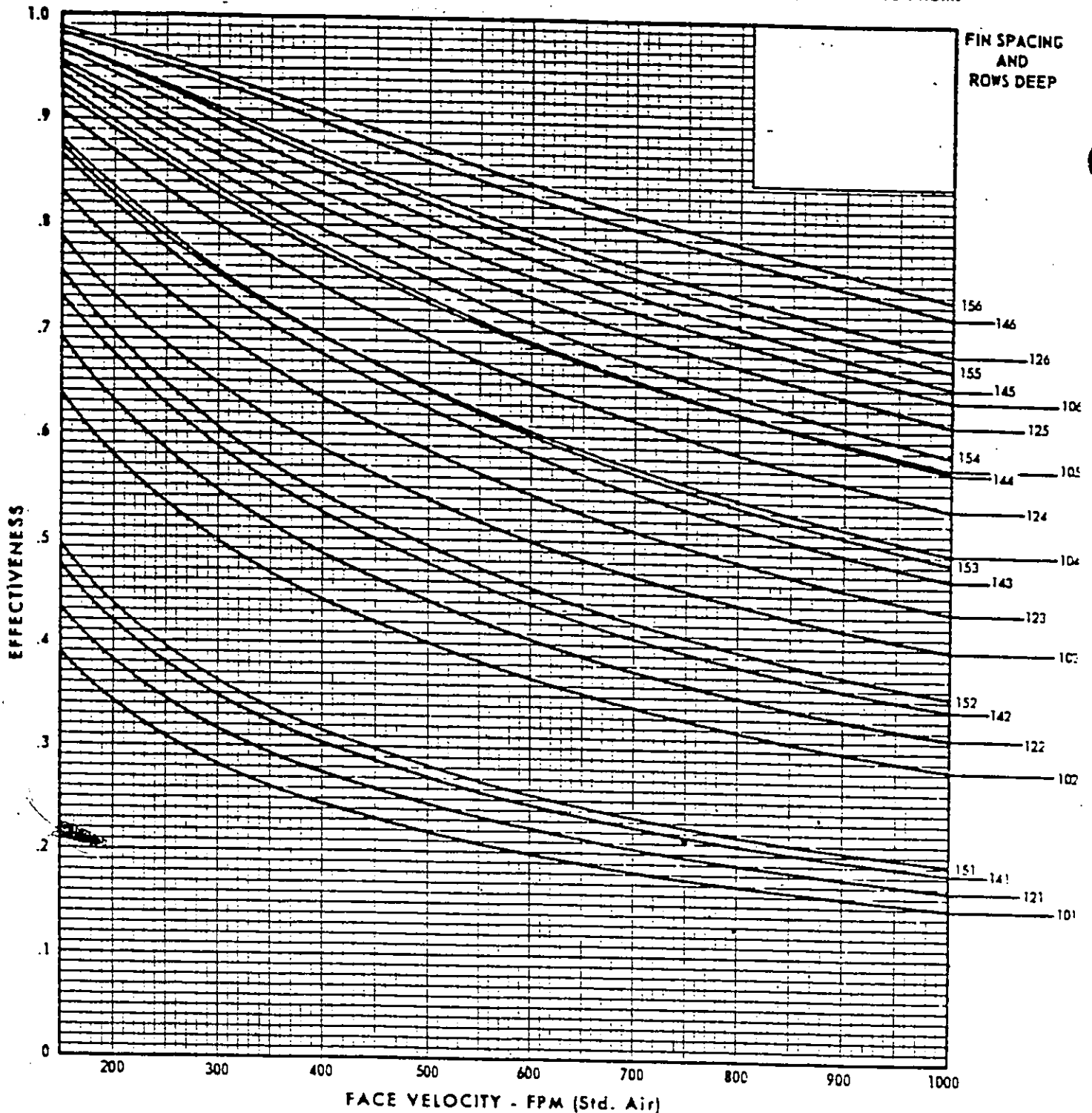


FIGURE F1

APPENDIX G

The friction factors for fully developed laminar and turbulent flow are given by the following relations.

$$\text{Laminar flow} \quad f = 64 / \text{Re} \quad \text{--- (1)}$$

$$\text{Turbulent flow} \quad f = 0.316 / \text{Re}^{1/4} \quad \text{--- (2)}$$

The continuity equation is given by

$$\dot{m} = \rho A V \quad \text{--- (3)}$$

The pressure drop over a distance L, associated with fully developed flow is given by

$$\Delta P = \frac{f L v^2 \rho}{2 g D} \quad \text{--- (4)}$$

The Reynold's number is given by

$$\text{Re} = (\rho V D / \mu) \quad \text{--- (5)}$$

DERIVATION FOR THE LAMINAR FLOW:

Substituting f from equation 1 in equation 4

$$\begin{aligned} \Delta P &= \frac{64}{\text{Re}} \cdot \frac{L v^2 \rho}{2 g D} \\ &= \frac{64}{\left(\frac{\rho V D}{\mu}\right)} \cdot \frac{L v^2 \rho}{2 g D} \\ &= 64 \frac{L V \mu}{2 g D^2} \end{aligned}$$

Substituting for V from equation 3

$$\Delta p = 64 \left[\frac{L \dot{m} \mu}{2 g A D^2 \rho} \right]$$

$$\text{or } \dot{m} = \left[\frac{2 g A D^2 \rho}{64 L \mu} \right] \Delta p$$

for a given state and size of a tube

$$\left[\frac{2 g A D^2 \rho}{64 L \mu} \right] = \text{Constant}$$

Therefore;

$$\dot{m} = \text{Constant} [\Delta p] \quad \text{--- (A)}$$

DERIVATION FOR THE TURBULENT FLOW:

By following the same method as discussed for laminar flow

$$\left[\frac{2 g A D^{5/4} \rho^{1/4}}{0.316 L \mu^{1/4}} \right] = \text{Constant}$$

Therefore;

$$\dot{m} = \text{Constant} [\Delta p]^{4/7} \quad \text{--- (B)}$$

Combining equations A and B

$$\dot{m} = \text{Constant} [\Delta p]^n \quad \text{--- (C)}$$

where $n = 1$ for laminar flow

$n = 0.571$ for turbulent flow

Using equation C

For evaporator first tube row

$$\dot{m}_1 = \text{Constant } \Delta P_1^n \quad \text{--- (D)}$$

For evaporator second tube row

$$\dot{m}_2 = \text{Constant } \Delta P_2^n \quad \text{--- (E)}$$

From equations D & E,

$$\dot{m}_2 = \dot{m}_1 \left[\frac{\Delta P_2}{\Delta P_1} \right]^n \quad \text{--- (F)}$$

Equation F was used in the program to calculate the mass flow rate in the evaporator second and the successive tube rows.

A P P E N D I X H

DETAILS OF THE COMPUTER PROGRAM

1.	PROGRAM NOMENCLATURE	H1
2.	USER'S MANUAL & FIGURE H2	H2
3.	PROGRAM LISTING	H3
4.	PROPERTY SUBROUTINES FOR R-113	H4

APPENDIX H1

PROGRAM NOMENCLATURE

A(1)	- Vapour flow rate for the evaporator (lb/hr)
A(2)	- Recirculation rate for the evaporator (lb/hr)
A	- Power used in the calculation of effective heat transfer coefficient in the dryout region
ACNDEL	- Inside surface area of condenser tube element in the flooded region (sq. ft)
AHTCND	- Inside surface area of non flooded portion of each condenser tube (sq. ft)
AGAM	- GAM (degrees)
ALFMIN	- Minimum angle of inclination of the condenser tubes from the vertical (deg) used in the calculation of CF
ALPA	- ALPHA (degrees)
ALPHA	- Angle of inclination of the plane of the condenser tubes from the vertical (read in degrees then converted to radians) see figure H2
AREAC	- Inside surface area of a condenser tube (sq. ft)
AREAH	- Inside surface area of an evaporator tube element (sq. ft)
AREAP	- Inner surface area of an evaporator tube (sq.ft)
AVEVT(I)	- Average evaporator wall temperature for Ith row (F)
AVTC(K)	- Average condenser wall temperature for Kth row (F)
AVFLTH	- Average film thickness in the condenser (ft)
AVLIQF	- Average liquid fraction for the entire loop
AVTSC	- The local average value of the condenser tube inner and outer surface temperatures (F)
AVTSCS	- Average value of AVTSC for the entire condenser (F)

AVTSH - The local average value of the evaporator tube inner and outer surface temperatures (F)

AVTSHS - Average value of AVTSH for the entire evaporator tube (F)

AVVOID - Average void fraction for the entire loop

AVWATH - Average water side heat transfer coefficient in the annulus (Btu/hr/sq.ft/F)

BETA - Angle of inclination of vapour header from the horizontal (radian) See figure H2

BEX - Exponent in Rohsenow's boiling equation, a function of the fluid-surface combination

BITA - BETA (degrees)

B1 - Total vapour flow rate for an evaporator tube row (lb/hr)

B2 - Total recirculation rate for an evaporator tube row (lb/hr)

BL - Flag used to correct the wall temperature in boiling heat transfer coefficient calculations

CF - Factor used in the condensation heat transfer coefficient, a function of the angle of the condenser tube

CG - Entrance effect factor for the heat transfer coefficient in the turbulent developing flow

CONDQ(K) - Rate of heat transfer for entire Kth row for the condenser (Btu/hr)

CPAIR - Specific heat of air (Btu/lb/F)

CSE - A constant in Rohsenow's boiling equation, a function of fluid-surface combination

DELP	- Pressure drop in a differential element in the condenser (lbf/sq.ft)
DELT	- Difference between the average evaporator and condenser tube temperatures (F)
DENSG	- Density of vapour (lb/ft ³)
DENSM	- Mean density of the two-phase mixture (lb/ft ³)
DENSW	- Water density (lb/ft ³)
DEQVC	- Equivalent diameter of condenser (inch)
DEQVE	- Equivalent diameter of evaporator (inch)
DIST	- Distance of the midpoint of a differential element from the entrance of the evaporator (ft)
DIVCON	- Number of divisions for the condenser, one for the vapour region, the balance for the liquid region
DIVN	- Total number of divisions into which the evaporator is divided for the sequential calculations
DENEVP	- Density of air in the evaporator (lb/ft ³)
DISCND	- Air flow rate approaching the condenser (cu-ft/hr)
DISEVP	- Air flow rate approaching the evaporator (cu-ft/hr)
DOWNST	- Summation of all the source fluid temperatures downstream of each tube element (F)
DLNGTH	- Differential length along the evaporator (ft)
DNSLO	- Density of liquid in condensate return line (lb/ft ³)
DPA	- Acceleration pressure drop along an element (lbf/sq.ft)
DPAC	- Acceleration pressure drop from the tube entrance to the end of the tube element (lbf/sq-ft)
DPCOND	- Hydrostatic head corresponding to the pressure difference between the condenser top and the evap. bottom (ft)
DPDL	- Pressure gradient along a tube (lbf/ft ³)

- DPF - Frictional pressure drop along an element (lbf/sq.ft)
- DPFG - Frictional pressure drop along an element based on the vapour flow (lbf/sq.ft)
- DPH - Hydrostatic pressure drop along an element (lbf/sq.ft)
- DQ - Heat flow rate into a differential element of the evaporator (Btu/hr)
- DT - Local difference between bulk fluid temp. and condenser tube inner surface temperature (F)
- DTA - Local difference between the saturation temperature and the evaporator tube inner surface temperature (F)
- DYNPER - True dynamic charge in the evaporator tube (%) (the ratio of vol of liquid to the total vol of the tube)
- DUCTXS - Cross sectional area of the evaporator duct (sq.ft)
- DX - Change in vapour quality along a differential element of length for the evaporator tube

-
- EFLOP(I) - Effectiveness for I number of loops
- ELEV - Elevation of bottom of condenser tubes relative to the bottom of the evaporator tube (ft). See figure H2
- EVAPST - Average evaporator wall temperature for the present row (F)
- EX - Exponent used in correlating void fraction with X_{tt} , in evaporator and vapour header
- EXTIDC - Exterior diameter of condenser annulus (inch)
- EXTIDE - Exterior diameter of evaporator annulus (inch)

-
- F - Friction factor

- FBL - Distance along the evaporator tube beyond the point where transition to mist flow has occurred (ft)
- FC - Entrance effect factor for laminar flow in the annulus
- FCT - Entrance effect factor for turbulent flow in the annulus
- FLCOUN - If =1, the program uses the read in value for FLRATE; for = 0, the program generates an estimate for FLRATE
- FL(K) - Vapour flow rate for Kth row for the condenser (lb/hr)
- FLAG2 - Counter for number of iterations where the evaporator tube flow rate has been reduced in order that $ZRECRC < ZMAX$
- FLDCND - The length of the condenser which is flooded (ft)
- FLDEV(I) - Length of evaporator recirculation tube flooded for Ith row (ft)
- FLPR - Evaporator tube flow rate in previous iteration (lb/hr)
- FLRATE - Flow rate inside the loop (lb/hr)
- FM - Entrance effect factor for the heat transfer coefficient in a laminar developing flow
- FOULNF - Fouling factor - tube inside surface (hr-sq.ft-F/Btu)
- FOULNW - Fouling factor - tube outside surface (hr-sq.ft-F/Btu)
- FRCPO - Head loss per unit length along the flooded portion of the condenser tube
- FRECRC - Total mass flow rate in the evaporator and condenser recirculation tubes (lb/hr)
- FTEF - Friction factor
-
- GAM - Rotation of the plane of the condenser tubes from the horizontal (read in degrees then converted to radians)
See figure H2.
- GC - Gravitational constant

GF - Mass flux (lb/sq.ft/sec)

HDIAMC - Diameter of vapour header on the condenser (inch)

HDIAME - Diameter of vapour header on the evaporator (inch)

HOON(K) - Condensation coefficient for the Kth row (Btu/hr/sq.ft/F)

HEDVLC - Volume of the vapour header on condenser (ft³)

HEDVLE - Volume of the vapour header on evaporator (ft³)

HIC - Convective heat transfer coefficient for the working fluid
in the condenser (Btu/hr/sq.ft/F)

HLPC - Single-phase liquid heat transfer coefficient for the
working fluid (Btu/hr/sq.ft/F)

H2P1 - Two-phase heat transfer coeff. (Btu/hr/sq.ft/F)

H2P2 - Single-phase vapour heat transfer coeff. (Btu/hr/sq.ft/F)

HEITB - Length of evaporator tubes (ft)

HEITC - Length of condenser tubes (ft)

HF - Enthalpy of saturated liquid at any point around the loop
(Btu/lb)

HFBOT - Initial enthalpy of liquid at the bottom of the evaporator
(Btu/lb)

HFBOTN - The new value of HFBOT calculated by the program (Btu/lb)

HFG - Latent heat of vaporization (Btu/lb)

HFLOLD - Previous enthalpy of the saturated liquid (Btu/lb)

HFOLD - Previous enthalpy of the fluid (Btu/lb)

HFTOP - Enthalpy at the top of the condenser (Btu/lb)

HG - Enthalpy of the saturated vapour at the top of the
condenser (Btu/lb)

HIL - Single-phase liquid heat transfer coefficient for the

working fluid in the evaporator (Btu/hr/sq.ft/F)

- HITP - Effective heat transfer coefficient for the inside surface of an element in the evaporator (Btu/hr/sq.ft/F)
- HITP2 - Effective heat transfer coefficient for the inside surface of an element in the condenser (Btu/hr/sq.ft/F)
- HITPB - Boiling heat transfer coefficient (Btu/hr/sq.ft/F)
- HTCFNT - Condensation coefficient (Btu/hr/sq.ft/F)
- HTFLUX - Heat flux (Btu/hr/sq.ft)
- HW - Water side heat transfer coefficient in evaporator annulus (Btu/hr.sq.ft/F)
- HW2 - Water side heat transfer coefficient in condenser annulus (Btu/hr/sq.ft/F)
-
- ICNDSB - Number of elements in the submerged portion of the condenser
- ICNHOT - If all source fluid temperatures are identical = 1, otherwise = 0
- IDC - Inside diameter of the condenser tube (inch)
- IDH - Inside diameter of the evaporator tube (inch)
- IDLC - Inside diameter of the liquid return line (inch)
- IDUC - Inside diameter of vapour header (inch)
- IHYSTR - A counter set = 0 for increasing temperature differences, set = 1 for decreasing overall temperature differences
- IPCOND - Ith element in the submerged portion of condenser for which printout is required
- IPRI - A flag which if set = 0, suppresses the printing of the transition points in the evaporator
- IPRINT - A flag which if set = 0, suppresses the printing of data for

the vapour header/liquid return line

- KF - Thermal conductivity of fluid (liquid or vapour)
(Btu/hr/ft/F)
- KG - Thermal conductivity of vapour (Btu/hr/ft/F)
- KL - Thermal conductivity of liquid (Btu/hr/ft/F)
- K(I) - Total recirculation rate for evaporator Ith row (lb/hr)
- KW - Thermal conductivity for water (Btu/hr/fr/F)
-
- L - Counter which detects the initiation of subcooled boiling
- LCNDSN - Non flooded length of condenser (ft)
- LEM - Inclination of the condensate return tube from the
horizontal (radians) See figure H2
- LENGTH - Distance to the downstream end of an element in the
evaporator or condenser (ft)
- LFB - A flag which is set equal to 1 when transition to mist
flow occurs in the evaporator (otherwise = 0)
- LH - Length of liquid column in the condensate return tube (ft)
- LHDEVF - Length of evaporator header (ft)
- LHDCND - Length of condenser header (ft)
- LLOXO - Length of the condensate return line (ft)
- LN - New LENGTH calculated in subroutines BULKPT or SBCOOL
which locates where bulk or subcooled boiling starts
- LO - Length to the upstream end of an element (ft)
- LUPXO - Length of the vapour header (ft)
- LQFR(I) - Liquid fraction for the evaporator Ith row

MAIREV - Mass flow rate of source air in the evap. duct (lb/hr)
 MAIRCN - Mass flow rate of sink air in the cond. duct (lb/hr)
 MCLWIR - Mass flow rate for the sink water (lb/hr)
 MHOIW(I) - Mass flow rate for the source water for Ith row (lb/hr)
 MODE - A counter set = 1, for water jacketed evap. and cond. tubes; set = 2, for a prescribed heat flux; set = 3, for air-to-air system
 MWT - Molecular mass of the working fluid (lbm); 137.37 for R-11 and 187.4 for R-113
 MX - A flag which is set equal to 1 at the beginning of all vapour flow (otherwise = 0)

NBEND - Number of bends in the vapour header
 NCC - Number of printouts desired along the length of the condenser
 NCP - Number of printouts desired along the length of the evaporator
 NROWC - Number of rows in the condenser
 NROWE - Number of rows in the evaporator
 NSCB - Exponent used in equation when combined boiling and convective heat transfer takes place in subcooled boiling
 NTUBEC - Number of tubes in a condenser row
 NTUBEE - Number of tubes in an evaporator row
 NTU - Number of transfer units (UA/mcp)
 NVALVE - Number of valves in the condensate line
 NUW - Nusselt number for the water in the evap. annulus
 NUW2 - Nusselt number for the water in the cond. annulus

ODC	- Outside diameter of the condenser (inch)
ODH	- Outside diameter of the evaporator (inch)

P	- A counter which detects the onset of bulk boiling
PAR	- Parachor of working fluid; 193.4 for R-11 and 249.6 for R-113
PEBEST	- Estimated pressure at the entrance to the evaporator tube (psia)
PEB(I)	- Pressure at the bottom of the evaporator Ith tube row (psia)
PET	- Pressure at the top of the evaporator first row (psia)
PMAX	- Maximum pressure possible for the loop (psia)
PMIN	- Minimum pressure possible for the loop (psia)
PERLNG	- Percent of evaporator length flooded under static conditions (%)
PERDY(I)	- True dynamic charge for the evaporator Ith tube row (%)
PEVTOP	- Pressure at the top of the evaporator tube (psia)
PI	- 3.1428571
PO	- Value of PRESS at the upstream end of an element (psia)
POWER	- Power used in the calculations for the mass flow rate
PR	- Prandtl number (either liquid or vapour)
PRCFL	- Percent of evaporator recirculation tube flooded under dynamic conditions (%)
PRESS	- Pressure at a particular location along the loop (psia)
PRESOL	- Pressure at the bottom of the evaporator from the previous iteration (psia)

PRF	- Prandtl number of liquid
PRG	- Prandtl number of vapour
PRLNGO	- Previous value of PERLNG
PROOUN	- If = 1, the program uses the read in value of PEBEST if = 0, the program generates an estimate for PEBEST
PRW	- Prandtl number for water in evaporator annulus
PRW2	- Prandtl number for water in condenser annulus
<hr/>	
Q	- Heat flow rate (Btu/hr)
QA	- Heat flux (Btu/hr/sq.ft)
QAVFLX	- Average heat flux for both evap. and condenser (Btu/hr/sq.ft)
QC(K, KK)	- Rate of heat transfer for KKth element for Kth row for condenser (Btu/hr)
QCFLUX	- Average heat flux for all condenser tubes (Btu/hr/sq.ft)
QCOND	- Heat transfer rate per condenser tube (Btu/hr)
QCS	- Total heat transfer rate in the condenser coil (Btu/hr)
QEFLUX	- Average heat flux for all the evaporator tubes (Btu/hr/sq.ft)
QES	- Total heat transfer rate in the evaporator coil (Btu/hr)
QLOXO	- Heat loss from the condensate tube (Btu/hr)
QO	- Total heat flow rate to the start of current element (Btu/hr)
QROWE	- Heat transfer rate for an evaporator row (Btu/hr)
QQ	- Minimum heat flux necessary to sustain boiling (Btu/hr/sq.ft)
QRECR	- Heat loss from the recirculation tube (Btu/hr)
QUPXO	- Heat loss from the vapour header (Btu/hr)

QULTY(I) - Quality at the top of the Ith evaporator row
 QINPT(I) - Heat transfer rate for the evaporator Ith row (Btu/hr)

RECID - Recirculation tube inside diameter (inch)
 RECRCL - Recirculation tube length (ft)
 REN - Reynold number (either liquid or vapour)
 RENF - Reynold number of liquid
 RENG - Reynold number of vapour
 RENW - Reynold number for water in evaporator annulus
 RENW2 - Reynold number for water in condenser annulus
 RINSID - Inside tube resistance for evaporator or condenser
 (hr.sq.ft.F/Btu)
 RESLO - Resistance per unit length of the recirculation and
 condensate return line (hr.ft.F/Btu)
 RESUP - Resistance per unit length of the vapour header
 (hr.ft.F/Btu)
 ROUT - Outside evaporator or condenser resistance
 (hr.sq.ft.F/Btu)
 RWALL - Tube wall resistance (hr.sq.ft.F/Btu)

SATC - AVTSCS (C)
 SATCON - Saturation temperature of the condenser (F)
 SATH - AVTSHS (C)
 SAAT(I) - Saturation temperature at the bottom of the evaporator Ith
 tube row (F)
 SCB - Distance from the entrance of the evaporator to the point
 where bulk boiling initiates (ft)

SCL	- Distance from the entrance of the evaporator to the point where subcooled boiling initiates (ft)
SDT	- DELT (C)
SELEV	- ELEV (m)
SFLRT	- FLRATE (Kg/sec)
SGMA	- SIGMA (degrees)
SHEITB	- HEITB (m)
SHEITC	- HEITC (m)
SIDC	- IDC (m)
SIDH	- IDH (m)
SIDLC	- IDLC (m)
SIDUC	- IDUC (m)
SIGMA	- Angle of inclination of the plane of each row of evaporator tubes from the vertical (read in degrees then converted to radian) See figure H2
SQA	- QAVFLX (W/sq.m)
SQC	- QCFLUX (W/sq.m)
SQCS	- QCS (W)
SQE	- QEFLUX (W/sq.m)
SQES	- QES (W)
SU	- UEXPR (W/sq.m/K)
SWIDE	- LUPXO (m)
SWIDER	- LLOXO (m)
SYSTVL	- Volume of entire thermosiphon loop (ft ³)
SUEVT(I)	- Sum of the evaporator wall temperatures for all elements for Ith row (F)
<hr/>	
TAIR	- Temperature of the sink air temperature approaching the

- condenser coil (F)
- TEB - Estimated temperature of the fluid in the evaporator liquid header (F)
- TEMRMX - Temperature of the recirculation fluid entering the evaporator liquid header (F)
- TCL - TCOLD, condenser air approach temperature calculated by the program from the previous iteration (F)
- TCNDWL - Condenser tube inner surface temperature (F)
- TCR - Critical temperature of the working fluid (F); 388.4 for R-11 and 417.4 for R-113
- TCWOUT - Condenser tube inner surface temperature (F)
- TEM - Temperature of the bulk fluid at the inlet of an element in the condenser (F)
- TEMP - Temperature of the bulk fluid at the inlet of an element in the evaporator (F)
- TEMPF - Temperature of the bulk fluid (F)
- TEMPLO - Temperature required at the exit of the liquid return line to satisfy the initial assumption (F)
- TEVTOP - Temperature of the fluid at the top of the evaporator (F)
- THETA - Rotation of the evaporator tubes from the horizontal (read in degrees then converted to radian) See figure H2
- THICKC - Thickness of condenser tubes (inch)
- THICKH - Thickness of evaporator tubes (inch)
- THOT(I) - Temperature of hot air approaching the Ith evaporator row (F)
- THIN(I,II) - Source fluid temperature for Ith row and IIth element (F)
- TOUT(I,II) - Source fluid temperature leaving IIth element for Ith row (F)

TCOUT(K, KK) - Sink fluid temperature leaving KKth element for Kth row
(F)

TOTALQ - Total heat input in the evaporator coil (Btu/hr)

TOTALF - Total mass flow rate for the evaporator coil (lb/hr)

TOTALR - Total recirculation rate for the evaporator coil (lb/hr)

TRECR - Temperature of the liquid at the end of the recirculation tube (F)

TSATEB - Saturation temperature at the bottom of evaporator (F)

TSAT1 - Value of TSATEB from the previous iteration (F)

TSATO - Value of TSAT from the previous iteration (F)

TSCF - Local inside surface temperature for the condenser tube
(F)

TSCS - Sum of condenser tube temperatures for all elements (F)

TSCSCN - Local average condenser wall temperature (F)

TSHF - Inside wall temperature of an element in the evaporator
(F)

TSHFO - Previous value of TSHF

TUSPCE - Center-to-center distance between two evaporator tubes in the same row (inch)

TSP - Degree of superheat (F)

TST - Superheat temperature (F)

TWIN - Temperature of heating fluid at the inlet of an element
(F)

TROOM - Room temperature (F)

U - Overall (inside to outside fluid) heat transfer coefficient (Btu/hr/sq.ft/F)

U2 - Overall (inside to outside fluid) heat transfer

- coefficient (Btu/hr/sq.ft/F)
- UOUT - Evaporator outside heat transfer coefficient
(Btu/hr/sq.ft/F)
- UOUTC - Condenser outside heat transfer coefficient
(Btu/hr/sq.ft/F)
- UF - Evaporator tube inside surface heat transfer coefficient
taking into consideration FOULNF (Btu/hr/sq.ft/F)
- UF2 - Condenser tube inside surface heat transfer coefficient
taking into consideration FOULNF (Btu/hr/sq.ft/F)
- UOA - Loop conductance based on average evaporator heat flux and
overall temperature difference between the source and the
sink fluid (Btu/hr/sq.ft/F)

- VDFHDC - Volume of vapour in the condenser outlet header (ft³)
- VDFHDE - Volume of vapour in the evaporator outlet header (ft³)
- VDFR(I) - Void fraction for Ith evaporator tube row
- VF - Velocity of fluid (either liquid or vapour) (ft/s)
- VELAIR - Velocity of air in condenser or evaporator duct (ft/s)
- VFCOP - Liquid velocity in flooded condenser tube (ft/s)
- VG - Vapour velocity in the evaporator (ft/s)
- VISC - Viscosity (either liquid or vapour) (lb/ft/s)
- VL - Liquid velocity in the evaporator (ft/s)
- VODC(K) - Condenser tube void fraction for Kth row

- WATH(I) - Average water side heat transfer coefficient for Ith row
in the annulus (Btu/hr/sq.ft.F)

- X - Vapour quality along the loop
- XACOND - Cross sectional area of the condenser tube (ft²)
- XAEVAP - Cross sectional area of the evaporator tube (ft²)
- XALORX - Cross sectional area of the liquid return tube (ft²)
- XARCRC - Cross sectional area of the recirculation tube (ft²)
- XAUPRX - Cross sectional area of the vapour header tube (ft²)
- XFB - Critical quality
- XP - Lockhart-Martinelli parameter
- XSCB - Vapour quality at the exit from an element
- XXX - A counter if set = 1 prints the input data cards
-
- Y - A counter used to make viscosity corrections in the convective heat transfer equation
- ZCOND - Elevation of the liquid vapour interface in the condensate return line above the bottom of the evap. tubes (ft)
- ZMAX - Elevation of the top of the evaporator above the bottom of the evaporator tubes (ft)
- ZREC(I) - Elevation of the liquid vapour interface in the recirculation tube from the bottom of the evaporator tube for Ith row (ft)
-

APPENDIX - H2

USER MANUAL FOR THE PROGRAM

In order to use this program the reader must provide the following input data:

1st Read statement Format (5,*)

IDH, IDC, THICKH, THICKC, HEITB, HEITC, LUPXO, LLOXO, TW2,
SIGMA, THETA, ALPHA, GAM, ELEV, TUSPCE, NROWE, NROWC, IPRI

2nd Read statement Format (5,*)

FOULNW, FOULNF, TCR, PAR, MWT, CSF, BEX, DIVN, DIVCON

3rd Read statement Format (5,*)

DPUPXO, NTUBEE, NTUBEC, RECRCL, RESUP, RESLO, TAIR, XXX

4th Read statement Format (5,*)

LEQAE, LEQAC, LEQRE

5th Read statement Format (5,*)

PEBEST, FLRATE, PRCOUN, FLCOUN, IHYSTR

6th Read statement Format (5,*)

NCP, NCC, IPRINT, FLAGUP, A, NBEND, NVALVE

7th Read statement Format (5,*)

MCLWTR, SPHTW, EXTIDE, EXTIDC, IDUC, IDLC

8th Read statement Format (5,*)

DUCTXS, MAIREV, MAIRCN, CPAIR, THOTG, RECID, TROOM

9th Read statement Format (5,*)

MODE, QFLUX

10th Read statement Format (5,*)

HDIAME, HDIAMC, LHDEVP, LHDCND, LEM

11th Read statement Format (5,*)

THOT(1)

THOT(I)

12th Read statement Format (5,*)

MHOTW(1)

MHOTW(I)

The following physical properties of the working fluid should be provided in subroutines.

1. Saturation pressure as a function of temperature (psia, F)
SUBROUTINE SAT PR(TEMPF, PSAT)
2. Saturation temperature (F) as a function of pressure (psia)
SUBROUTINE SAT TEM(PRESS, TSAT)
3. Enthalpy (Btu/lb) of saturated liquid as a function of temperature (F)
SUBROUTINE LIQ HF(TEMPF, HFL)
4. Density of liquid (lb/ft³) as a function of temperature (F)
SUBROUTINE LIQ DEN(TEMPF, DENSL)
5. Latent heat (Btu/lb) of vaporization as a function of temperature (F)
SUBROUTINE LAT HT(TCR, TEMPF, HFG)
6. Surface tension (lbf/ft) for the liquid-vapour interface
SUBROUTINE SUR TEN(PAR, DENSL, DENSG, MWT, SURFTN)
7. Viscosity of liquid (lb/ft/s) as a function of temperature (F)
SUBROUTINE LIQ VIS(TEMPF, VISCL)
8. Thermal conductivity (Btu/hr/ft/F) of liquid as a function of temperature (F)
SUBROUTINE TH COND(TEMPF, KL)

9. Specific heat (Btu/lb/F) of liquid as a function of temperature (F)

SUBROUTINE LIQ CP(TEMPF, SPHTL)

10. Viscosity (lb/ft/s) of vapour as a function of temperature (F)

SUBROUTINE GAS VIS(TEMPF, VISC)

11. Specific heat (Btu/lb/F) of vapour as a function of temperature (F)

SUBROUTINE GAS CP(TEMPF, SPHTG)

The following subroutine is provided in the program to calculate the water properties.

Density (lb/ft³), viscosity (lb/ft/s) and thermal conductivity (Btu/hr/ft/F) as a function of temperature (F).

SUBROUTINE WTRPRP (TEMP, DENS, VISC, KW)

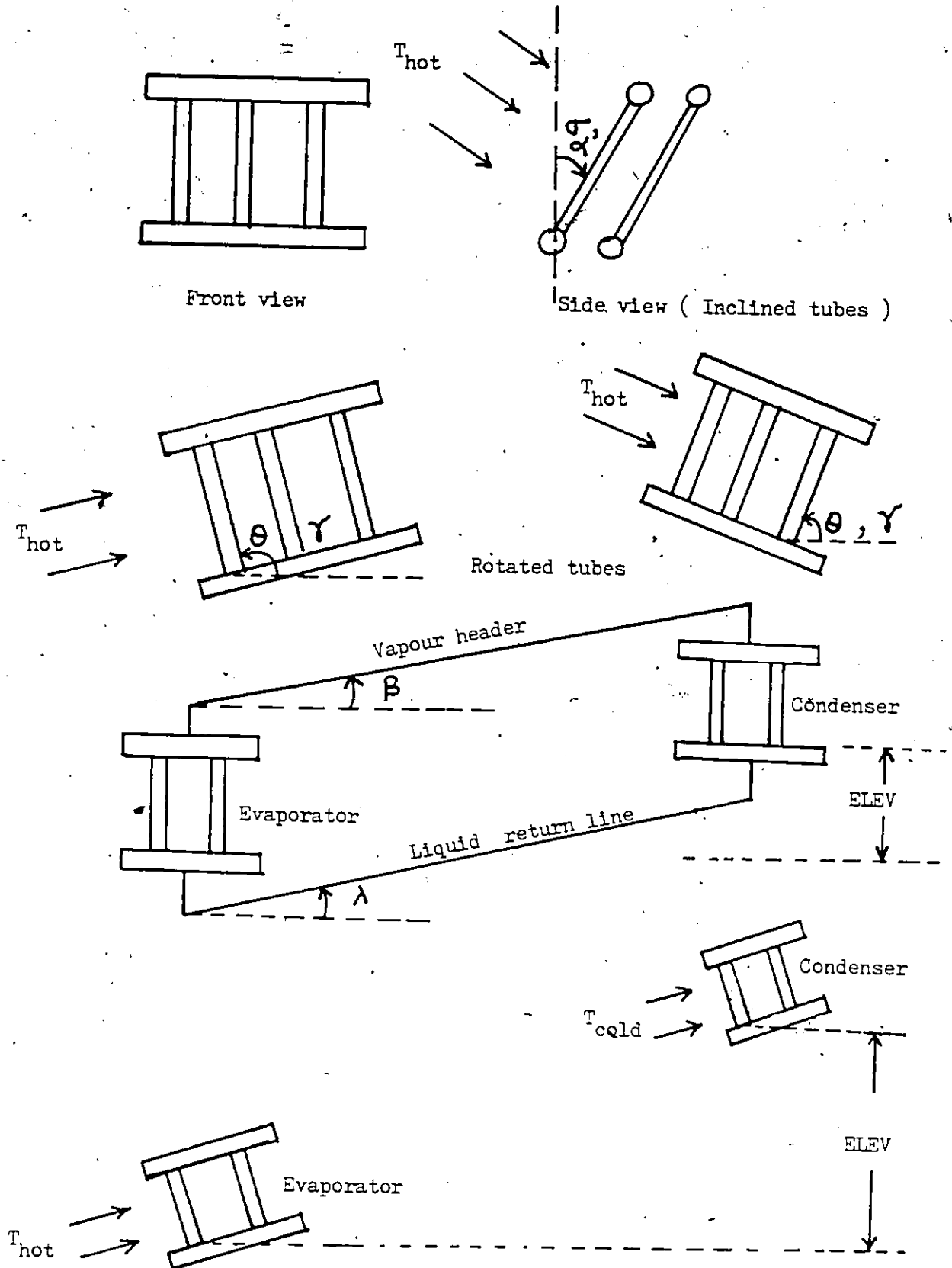
The following subroutine calculates the density (lb/ft³) of air as a function of the air temperature (F).

SUBROUTINE AIRDEN (AIRTEM, DENAIR)

The following subroutine calculates the air side resistance (hr.F/Btu) as a function of the air velocity (ft/s).

SUBROUTINE EFF(DISC, AREA, TAIR, PI, INDIA, OUTDIA, CPAIR, GC, ROUT, TCR, DENAIR, CF, DUCTXS, ALPHA, ALFMIN, TUSPCE, LENGTH, UA)

FIGURE H2



DESCRIPTION OF VARIOUS ANGLES FOR BOTH THE SECTIONS

APPENDIX H3

PROGRAM LISTING

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1. //MATHUR JOB (G271,E35,1,4),'DAS',CLASS=Z,MSGLEVEL=(1,1)
2. // EXEC FORTVCLG,FVLANG='LAGLVL(66)'
3. //FORT.SYSIN DD *
4. *****
5. * PROGRAM BY:- GURSARAN DAS MATHUR *
6. * *****
7. * PROGRAM NAME :- ROUTINE *
8. * *****
9. * THIS COMPUTER PROGRAM IS CAPABLE OF SIMULATING THE OVERALL PERFORMANCE
10. * OF FLUID-TO-FLUID TWO-PHASE THERMOSIPHON LOOP HEAT EXCHANGER SYSTEMS
11. * IN WHICH THE COILS CONSIST OF PARALLEL ROWS OF STRAIGHT TUBES RUNNING
12. * BETWEEN A COMMON LIQUID HEADER AND A COMMON VAPOUR HEADER. THIS
13. * PROGRAM CAN SIMULATE MULTIPLE TUBE EVAPORATOR AND CONDENSER COILS
14. * WITH INDEPENDENT HEATING OF EACH TUBE AND SINGLE-PASS, FINNED-TUBE,
15. * MULTIPLE-ROW EVAPORATOR AND CONDENSER COILS WHERE THE SOURCE AND
16. * SINK FLUID IS A GAS. THE TUBE ROWS MAY BE INCLINED FROM THE VERTICAL
17. * AND MAY BE SUBJECT TO DIFFERENT EXTERNAL FLOW AND THERMAL CONDITIONS.
18. * THIS PROGRAM CAN SIMULATE THE FOLLOWING SITUATIONS:-
19. * (A) AIR-TO-AIR TWO-PHASE THERMOSIPHON LOOP HEAT EXCHANGER
20. * (B) TWO-PHASE THERMOSIPHON LOOP HEAT EXCHANGER WITH WATER JACKETED
21. * EVAPORATOR AND CONDENSER TUBES
22. * (C) CONSTANT HEAT FLUX BOUNDARY CONDITION IMPOSED ON THE EVAPORATOR
23. * TUBE
24. *
25. * TO RUN THE PROGRAM, THE OPERATOR MAY SPECIFY THE NUMBER OF ELEMENTS
26. * INTO WHICH THE EVAPORATOR TUBES ARE TO BE DIVIDED (THE DEFAULT VALUE
27. * IS 10). AND THE WORKING FLUID FLOW RATE IN A TYPICAL TUBE OF THE
28. * 1ST ROW OF THE EVAPORATOR TUBES (A DEFAULT VALUE IS CALCULATED
29. * BASED ON THE OTHER INPUT DATA). AFTER AN INITIAL SOLUTION WHICH
30. * YIELDS A STATIC CHARGE IN THE SYSTEM, THE OPERATOR MAY INCREASE
31. * THE WORKING FLUID FLOW RATE TO YIELD RESULTS FOR LARGER STATIC
32. * CHARGES OR VICE VERSA.
33. *
34. * THE CONVERGENCE ROUTINE IS BASED ON THE SINK FLUID TEMPERATURE. THE
35. * PROGRAM CALCULATES THE SINK FLUID TEMPERATURE REQUIRED TO CONDENSE THE
36. * VAPOUR SUPPLIED TO THE CONDENSER. IF THIS SINK FLUID TEMPERATURE IS NOT
37. * CORRECT, VARY THE STATE OF THE WORKING FLUID ENTERING THE EVAPORATOR UNTIL
38. * CONVERGENCE IS OBTAINED ON THE SINK FLUID TEMPERATURE.
39. *
40. * IN ORDER TO USE THE PROGRAM, THE FOLLOWING DATA IS FED TO THE PROGRAM:
41. * READ STATEMENT # 1:-
42. * INSIDE DIA OF THE EVAPORATOR(INCH), INSIDE DIA OF THE CONDENSER(INCH),
43. * THICKNESS OF EVAPORATOR TUBE(INCH), THICKNESS OF THE CONDENSER TUBE(INCH),
44. * LENGTH OF EVAPORATOR TUBE(FT), LENGTH OF THE CONDENSER TUBE(FT), LENGTH
45. * OF THE UPPER CONNECTING PIPE(FT), LENGTH OF LOWER CONNECTING PIPE(FT),
46. * SINK TEMPERATURE (F), EVAP. ANGLE FROM THE VERTICAL(DEC), EVAP. ROTATION
47. * FROM THE HORIZONTAL(DEC), COND. TILT & ROTN ANGLE(DEC), ELEVATION OF THE COND.
48. * BOTTOM FROM THE BASE OF THE EVAPORATOR(FT), TUBE SPACING(INCH), NUMBER
49. * OF EVAP. AND COND. ROWS, COUNTER WHEN=0 SUPPRESSES THE PRINTING OF
50. * THE TRANSITION POINTS IN THE EVAPORATOR
51. * READ STATEMENT # 2:-
52. * FOULING FACTOR OUTSIDE EVAPORATOR TUBE(FT2-HR-F/BTU), FOULING FACTOR
53. * INSIDE EVAPORATOR TUBE(FT2-HR-F/BTU), TCR- CRITICAL TEMPERATURE OF THE
54. * WORKING FLUID(F), PAR- PARACHOR OF THE WORKING FLUID, MW- MOLECULAR
55. * WEIGHT OF THE WORKING FLUID, CSF- A CONSTANT IN ROHSENOWS BOILING EQUATION
56. * BEX-AN EXPONENT IN ROHSENOWS BOILING EQUATION, NUMBER OF LENGTH ELEMENTS
57. * ALONG THE EVAPORATOR, NUMBER OF LENGTH ELEMENTS ALONG
58. * READ STATEMENT # 3:-
59. * ESTIMATE OF THE FRICTIONAL PRESSURE DROP IN THE UPPER CONNECTING TUBES(PST),
60. * NUMBER OF TUBES IN AN EVAPORATOR ROW, NUMBER OF TUBES IN A CONDENSED ROW,
61. * LENGTH OF THE RECIRCULATION TUBES(FT), SURROUNDING THERMAL RESISTANCE
62. * PER UNIT LENGTH OF THE RECIRCULATION TUBE, VAPOUR HEADER
63. * AND CONDENSATE RETURN LINE, AIR TEMP(F), A COUNTER WHEN=0 PRINTS THE
64. * DATA CARDS
65. * READ STATEMENT # 4:-
66. * EQUIVALENT LENGTH OF FITTINGS, BENDS ETC. IN VAPOUR HEADER, CONDENSER,
67. * AND RECIRCULATION TUBE
68. * READ STATEMENT # 5:-
69. * ESTIMATED PRESSURE AT THE BOTTOM OF THE FIRST ROW(PST), ESTIMATED FLOWRATE IN
70. * THE FIRST ROW OF THE EVAPORATOR (LBM/HR), A COUNTER IF = 1, THE PROGRAM
71. * USES THE READ IN VALUE OF REBEST, A COUNTER IF = 1, USES THE READ IN
72. * VALUE OF FLRATE, A COUNTER IF = 0, FOR INCREASING OVERALL TEMPERATURE
73. * DIFFERENCES AND 1 FOR DECREASING TEMPERATURE DIFFERENCES
74. * READ STATEMENT # 6:-
75. * NUMBER OF PRINTOUTS DESIRED ALONG THE LENGTH OF THE EVAPORATOR, NUMBER
76. * OF PRINTOUTS DESIRED ALONG THE LENGTH OF THE CONDENSER, A FLAG WHICH
77. * IF=0 SUPPRESSES THE PRINTING OF DATA FOR THE VAPOUR HEADER AND THE LIQUID
78. * RETURN LINE, A FLAG WHICH IF SET=1 USE READ IN VALUE OF THE VAPOUR HEADER
79. * DROP DROPXO, POWER USED IN CALCULATION OF THE EFFECTIVE HEAT TRANSFER
80. * COEFF. IN THE DRYOUT REGION, NUMBER OF BENDS IN VAPOUR HEADER, NUMBER

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81. C * OF VALVES IN THE LIQUID RETURN LINE
82. C * READ STATEMENT # 7:-
83. C * SINK WATER FLOW RATE (LBM/HR), SPECIFIC HEAT OF WATER (BTU/LBM/F),
84. C * EXTERIOR DIAMETERS FOR THE EVAPORATOR AND CONDENSER ANNULUS (INCH),
85. C * DIAMETERS OF UPPER AND LOWER CONNECTING PIPES (INCH),
86. C * READ STATEMENT # 8:-
87. C * DUCT X-SECTIONAL AREA (FT2), FLOWRATES OF AIR: IN EVAP. IN COND. (LBM/SEC),
88. C * SP. HEAT OF AIR (BTU/LB-F), HOT AIR TEMP. APPROACHING THE HEAT EXCHANGER,
89. C * DIAMETER OF THE RECIRCULATION TUBE (INCH), SURROUNDING TEMPERATURE (F),
90. C * READ STATEMENT # 9:-
91. C * A COUNTER SET = 1 FOR WATER JACKETED TUBES, 2 FOR PRESCRIBED HEAT FLUX,
92. C * 3 FOR AIR-TO-AIR SYSTEM, HEAT FLUX (BTU/HR/SQ.FT)
93. C * READ STATEMENT # 10:-
94. C * HEADER DIAMETERS FOR EVAPORATOR AND CONDENSER (INCH), LENGTH OF THE
95. C * HEADERS FOR EVAPORATOR AND CONDENSER (INCH)
96. C * READ STATEMENT # 11:-
97. C * SOURCE FLUID TEMPERATURES FOR EACH OF THE TUBE ROWS (F)
98. C * READ STATEMENT # 12:-
99. C * SOURCE FLUID FLOW RATE FOR EACH OF THE TUBE ROWS (LBM/HR)
100. C *****
101. C DIMENSION WATH(4)
102. C DIMENSION TCAIR(4), CONDO(4)
103. C DIMENSION QC(4,10), FL(4), TCOUT(4,10), VDOC(4), HCON(4), AVTC(4)
104. C DIMENSION SUEVT(4), VDFR(4), LDFR(4), EEXT(4), PERDY(4), MHOTW(4)
105. C DIMENSION QULTY(4), QINPT(4), FLRAT(4), THOT(4), PES(4), SAAT(4), TFLUD(
106. C 4), THIN(4,20), TOUT(4,20)
107. C REAL LBE, LUPXO, LLOXO, LEOAE, LEORE, LEOAC, LEOYE, LIOFS, LOFR, K1, K2
108. C REAL IDH, IDC, IDUC, IDLC, NSCB, LCNDSN, LN, LO, LETMCH
109. C REAL LCON, NSENO, NVALVE, KW, KL, KG, KM, KF, LH, LEM, LEMDA, LHDEVP, LHDCND
110. C REAL LENGTH, LIOF, MWT, NTU, MASAIR
111. C REAL LIDHDE, LIDHDC, MHMTWTR, MCLWTR, IDSE, MHOTW, MAIREV, MAIRCN
112. C INTEGER FLAGUP, CIAM, COUNT2, FLAG1, FLAG2, FLAG3, FLAG4, XXX
113. C READ (5,*) IDH, IDC, THICKH, THICKC, HEITB, HEITC, LUPXO, LLOXO, TW2
114. C 9, SIGMA, THETA, ALPHA, GAM, ELEV, TUSPCE, NROWE, NROWC, IPRI
115. C READ (5,*) FOULNW, FOULNF, TCR, PAR, MWT, CSF, BEX, DIVN, DIVCON
116. C READ (5,*) IDUPXO, NTUBES, NTUREC, RECRCL, RESUP, RESLO, RESBE, TAIR, XXX
117. C READ (5,*) LEOAE, LEOAC, LEORE
118. C READ (5,*) PESEST, FLRATE, PRCOJN, FLCOUN, INHYSTR
119. C READ (5,*) NCC, NCC, IPRI, FLAGUP, A, NSENO, NVALVE
120. C READ (5,*) MCLWTR, SPHTW, EXTIDE, EXTIDC, IDUC, IDLC
121. C READ (5,*) IDUCTXS, MAIREV, MAIRCN, CPAIR, THOTG, RECIO, TROOM
122. C READ (5,*) MODE, OFLUX
123. C READ (5,*) HOTAME, HOTAMC, LHDEVP, LHDCND, LEM
124. C NR=NROWE
125. C NR=NROWC
126. C DO 41 I=1,NR
127. C 41 READ (5,*) THOT(I)
128. C DO 43 I=1,NR
129. C 43 READ (5,*) MHOTW(I)
130. C ICNHOT=0
131. C PRINT 277
132. C CALL
133. C 9 SIGMA, THETA, ALPHA, GAM, ELEV, XXX, TAIR)
134. C CALL PPLINT(TAIR, THOTG, PESEST, DENSL, PMIN, PMAX)
135. C IF (PRCOJN.EQ.1) GO TO 98
136. C PESEST=(PMAX+PMIN)/2.0+HEITB*DENS*COS(SIGMA)/144.0
137. C DIVN=10.0
138. C 98 PI=3.141596
139. C GC=32.174
140. C IF (FLCOJN.EQ.0) CALL MASS FL(PI, IDH, HEITB, TEMPF, FLRATE, TCR)
141. C EX=0.21
142. C ICOUNT=0
143. C IF ICOUN=0
144. C IF (MODE.NE.3) GO TO 1966
145. C CALL AIDENY(THOTG, DEHEVP)
146. C DISEVP=MAIREV/DEHEVP
147. C CALL AIDREN(TAIR, DEHCND)
148. C DISCND=MAIRCND/DEHCND
149. C UNHTL1=UNHTL1/12.0
150. C UNHTL2=UNHTL2/12.0
151. C DPFLM1=DPFLM1*(1.4504/10000.0)
152. C DPFLM2=DPFLM2*(1.4504/10000.0)
153. C DPFL1=DPFL1*(1.4505/10000.0)
154. C DPFL2=DPFL2*(1.4505/10000.0)
155. C 1966 FLAG2=0
156. C FLAG3=0
157. C ICNOSB=0 IVCON=1.0
158. C RNCB=NCC
159. C RNC=NCC
160. C IOBF=IORE/12.0

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161. RECID=RECID/12.0
162. LHDEVP=LHDEVP/12.0
163. LHDCND=LHDCND/12.0
164. HEDVLE=(PI*(HDIAME**2.0)/4.0)*LHDEVP
165. HEDVLC=(PI*(HDIAMC**2.0)/4.0)*LHDCND
166. IDH=IDH/12.0
167. IDC=IDC/12.0
168. XAEVAP=(IDH**2)*(PI/4.0)
169. XACOND=(IDC**2)*(PI/4.0)
170. XAUPOX=(IDUC**2)*(PI/4.0)
171. XALOPX=(IDLC**2)*(PI/4.0)
172. RECID=IDLC
173. XARCRC=(RECID**2.0)*(PI/4.0)
174. XABE=(IDBE**2.0)*(PI/4.0)
175. THICKH=THICKH/12.0
176. THICKC=THICKC/12.0
177. ODH=IDH*(2.0*THICKC)
178. ODC=IDC*(2.0*THICKH)
179. AREAP=PI*IDH*HEITS
180. AREAC=PI*IDC*HEITC
181. CNROTN=0
182. IF (THETA.EQ.90.0)CNROTN=1
183. CALL VOL (CNROTN,VSTRAN,THETA)
184. EXTIDE=EXTIDE/12.0
185. EXTIDC=EXTIDC/12.0
186. AANULC=(PI/4.0)*((EXTIDC**2.0)-(ODC**2.0))
187. AANULE=(PI/4.0)*((EXTIDE**2.0)-(ODH**2.0))
188. DEQVE=EXTIDE-ODH
189. DEQVC=EXTIDC-ODC
190. MCLWTR=MCLWTR/3.0
191. GWC=MCLWTR/AANULC/3600.0
192. AGAM=GAM
193. ALFA=ALPHA
194. THTA=THETA
195. SGMA=SIGMA
196. ALPHA=PI*ALPHA/180.0
197. SIGMA=PI*SIGMA/180.0
198. THETA=PI*THETA/180.0
199. GAM=PI*GAM/180.0
200. ALFIN=ATAN(IDC/HEITC)
201. IF (ALPHA.EQ.0.0)CF=0.0
202. IF (ALPHA.EQ.0.0)GO TO 693
203. IF (ALPHA.GE.((PI*50.0)/180.0))CF=0.724
204. IF (ALPHA.GE.((PI*50.0)/180.0))GO TO 693
205. CF=0.724*((COS(((50.0*PI)/180.0)-ALPHA))/((50.0*PI)/180.0))
206. 3 (PI/2.0))**0.25)
207. 693 BETAMARSIN(((ELEV*(HEITC*SIN(GAM)*COS(ALPHA)))-(HEITR*SIN(THETA)*
208. 3COS(SIGMA)))/LUPXD)
209. PRESS=PEBEST
210. CALL SAT TEM(PRESS,TSAT)
211. TEB=TSAT-0.023*(TSAT-TAIR)
212. TA=0.005*(TSAT-TAIR)
213. CONST=0.045
214. IF (MODE.NE.3)GO TO 308
215. TEMPF=TEB
216. DENSG=(PRESS*MWT)/(10.730*(TEMPF+460.0))
217. VELAIR=0.1SEVP/DUCTXS
218. 309 TEMPF=TEB
219. 307 PEB(1)=PRESS
220. PSTART=PEB(1)
221. CALL LIO DEN(TEMPF,DENSL)
222. DO 62 I=1,NR
223. 62 PEB(I+1)=PEB(I)+(I*TUSPCE*DENSL*COS(THETA))/144
224. DO 63 I=1,NR
225. 63 CALL SAT TEM(PEB(I),SAAT(I))
226. IF (ICOUNT.GE.1) GO TO 66
227. IF (THETA)64.65.65
228. 65 TEB=SAAT(I)-CONST*(SAAT(I)-TAIR)
229. GO TO 66
230. 64 TEB=SAAT(NR)-CONST*(SAAT(NR)-TAIR)
231. 66 FLAG1=0
232. TEMPF=TEB
233. CALL LIO HF(TEMPF,HFL)
234. HFBOT=HFL
235. HF=HFBOT
236. PROT=PPRESS
237. CALL HFLS(TSAT,HFSTPT)
238. CALL LIO DEN(TEMPF,DENSL)
239. CALL LIO VIS(TEMPF,VISCL)
240. TSATEB=SAAT(I)

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241. PRINT 277
242. 277 FORMAT('---',
243.
244. PRINT 1957,NROWE,NTUBEE
245. 1987 FORMAT('---',EVAPORATOR: TOTAL NUMBER OF ROWS='11,3X,'NUMBER OF
246. TUBES IN EACH ROW='1,12)
247. PRINT 674,NROWC,NTUBEC
248. 574 FORMAT('---',CONDENSER: TOTAL NUMBER OF ROWS='11,3X,'NUMBER OF TU
249. BLES IN EACH ROW='1,12)
250. PRINT 277
251. PRINT 189,TWOTG
252. 189 FORMAT('---',TEMP. OF HOT AIR APPROACHING THE EVAPORATOR(F)='1,F7.3)
253. PRINT 190,TAIR
254. 190 FORMAT('---',TEMP. OF COLD AIR APPROACHING THE CONDENSER(F)='1,F7.3)
255. PRINT 1989,TROOM
256. 1989 FORMAT('---',ROOM TEMPERATURE (F)='1,F7.3)
257. PRINT 545,MAIREV
258. 545 FORMAT('---',MASS FLOW RATE OF AIR IN THE EVAPORATOR(LBM/SEC)='1,F10
259. 3.7)
260. PRINT 546,MAIRCN
261. 546 FORMAT('---',MASS FLOW RATE OF AIR IN THE CONDENSER(LBM/SEC)='1,F10
262. 3.7)
263. PRINT 277
264. I=1
265. 91 IF(MODE.NE.3.AND.I.EQ.2)GO TO 745
266. IF(I.EQ.2)GO TO 746
267. TWIN=THOT(I)
268. TSAT=SAAT(I)
269. TEMPF=TEB
270. WHTWTR=WHTW(I)
271. GW=WHWTR/AANULE/3600.0
272. GF=FLRATE/XAEVAP/3600.0
273. PLOHR1=PEB(1)+(((2.0/12.0)*DENSL)/144.0)+1.0*(1.0/(GC*2.0))
274. D=((GF/DENSL)*2.0)*(1.0/144.0)
275. GO TO 745
276. 746 CALL DELPBE(I,1,LBE,LEQBE,GC,ZBE,IDBE,DPFVAP,NTUBEE,PI,
277. DENSL,VISCL,XABF,DLPRE,DPOL,UNMTL,DENSG,VISCG,IDUC,NELBOW,
278. DLETMCH,JPNDMM,OPTEE,DELBO,OPEXIT)
279. PRESS=PROTBE-DLQBE-OPFLW2-OPFL2
280. CALL SAT TEM(PRESS,SAAT(2))
281. PEB(2)=PRESS
282. PRESS=PEB(2)
283. TSAT=SAAT(2)
284. PLOHR2=PEB(2)+(((4.0/12.0)*DENSL)/144.0)+1.0*(1.0/(GC*2.0))
285. D=((GF/DENSL)*2.0)*(1.0/144.0)
286. 745 CALL HFLS(TSAT,HFSTR)
287. TEMPF=TEB
288. CALL LIO HF(TEB,HFL)
289. HFBOT=HFL
290. HF=HFROT
291. DEGSUB=TSAT-TEB
292. DLNGTH=HEITR/DIVN
293. GF=FLRATE/XAEVAP/3600.0
294. DOWNST=0.0
295. LENGTH=0.0
296. Q=0.0
297. HWSUM=0.0
298. HW2SUM=0.0
299. HW2SM2=0.0
300. XTOP=0.0
301. VOIDF=0.0
302. VOIDFS=0.0
303. LIOFS=0.0
304. DPA=0.0
305. DPAC=0.0
306. DPACP=0.0
307. HIPC=0.0
308. TSHS=0.0
309. D=0.0
310. HITPR=0.0
311. XP=0.0
312. PH=0.0
313. VG=0.0
314. HITPC=0.0
315. MX=0
316. TSP=0.0
317. LCCN=0.0
318. FRL=0.0
319. H2P1=0.0
320. H2P2=0.0

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321. XSCR=0.0
322. NSCR=0.0
323. UAVEV=0.0
324. SUHITP=0.0
325. ICOUNT=0
326. TWOUT=TWIN
327. LIOF=1.0
328. TSHF=TEMPF
329. L=0
330. LFB=0
331. ICP=1
332. M=DIVN
333. RM=M
334. C *****
335. C * START OF EVAPORATOR ELEMENT LOOP *
336. C *****
337. DO 1 II=1,M
338. AREAHP=PI*IDH*DLNGTH
339. IF(MODE.NE.3)GO TO 1973
340. CALL AIRDEN(THOTG,DENAIR)
341. IF(I.GT.1)CALL AIRDEN(THIN(I-1,II),DENAIR)
342. MASAIR=(DENAIR*VELAIR*3600.0)/(DLNGTH*TUSPCE/12.0)
343. CALL EFF(DISEVP,AREAP,TAIR,PI,IDH,ODH,CPAIR,GC,ROUT,
344. BTCL,DENAIR,CF,DUCTXS,ALPHA,ALFMIN,TUSPCE,DLNGTH,UA)
345. UOUT=((1.0/ROUT)*(1.0/AREAHP))
346. 1973 AREAHP=PI*ODH*DLNGTH
347. HFLD=HFL
348. IF(II.EQ.1)HFLD=HFLSTRT
349. HFLD=HF
350. LO=LENGTH
351. TSHFO=TSHF
352. TSATD=TSAT
353. TEMPFO=TEMPF
354. VOIDFO=VOIDF
355. P0=PRESS
356. IF(H2P1.GT.H1PC)H1PC=0.0
357. GL=0.0
358. IMRB=0
359. IX=0
360. PD=0.0
361. XBOTOM=XTOP
362. X=XBOTOM
363. TEMP=TEMPF
364. IF(X.GE.1.0)H2P1=0.0
365. IF(HITP3.GT.HITPC)HITPC=0.0
366. DENSGL=(PRESS*MWT)/(10.730*(TEMPF+460.0))
367. PRESS=PRESS*144.0
368. DIST=LENGTH*(DLNGTH/2.0)
369. LENGTH=LENGTH+DLNGTH
370. CG=1.0+(7.0*IDH/DIST)
371. FM=((1.0+(5.0/EXP(DIST/(10.0*IDH))))
372. FC=((1.0+(3.5/EXP(DIST/(90.0*DEOVE))))
373. FCT=((1.0+(3.0/EXP(DIST/(30.0*DEOVE))))
374. TWIN=TWOUT
375. CALL WTPRP(TWIN,DENSW,VISCW,KW)
376. RENW=DEOVE*GWM/VISCW
377. PRW=VISCW*SPHTW/KW
378. IF(RENW-1900.0)650.650.689
379. 689 NUW=0.02*(RENW**0.8)*(PRW**0.3333)*((EXTIOE/(ODH))**0.53)
380. HW=NUW*KW*FCT*3600.0/DEOVE
381. GO TO 590
382. 650 HW=((1.86*KW/DEOVE*(RENW*PRW*DEOVE/DIST))*
383. 30.3333)*3600.0*FC
384. 690 CALL LIO DEN(TEMPF,DENSL)
385. CALL LAT HT(TCP,TEMPF,HFG)
386. CALL SUR TEN(DAR,DENSL,DENSG,MWT,SURFTN)
387. VL=GF*(1.0-X)/DENSL
388. IF(X.GT.3.0.AND.X.LT.1.0)VL=GF*(1.0-X)/(DENSL*LIOF)
389. IF(X.GE.1.0)GO TO 555
390. CALL LIO VIS(TEMPF,VISCL)
391. RENF=IDH*VL*DENSL/VISCL
392. POWER=1.0
393. IF(RENF.GE.2000.0)POWER=0.571
394. CALL TH CDND(TEMPF,KL)
395. CALL MT CDND(TEMPF,KM)
396. CALL LIO CP(TEMPF,SPHTL)
397. PPF=SPHTL*VISCL/KL
398. DECLET=RENF*PPF
399. VF=VL
400. DEN=RENF

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401.      OR=PRF
402.      VISC=VISC L
403.      KF=KL
404.      555 CALL      GAS VIS(TEMP,F,VISCG)
405.      CALL      GAS CP(TEMP,SPHTG)
406.      KG=VISC*(SPHTG+2.44/MWT)
407.      VG=GF*X/DENSG
408.      IF(X.GT.0.0)VG=GF*X/(DENSG*VOIDF)
409.      IF(X.GT.0.0)RENG=IDH*VG*DENSG/VISCG
410.      REN=IDH *GF*X/VISCG
411.      PRG=SPHTG*VISC/GK
412.      IF(X.LE.0.0)GO TO 1003
413.      IF(H2O1.GT.HIPC.AND.X.LT.1.0)GO TO 5000
414.      IF(X.LT.1.0)GO TO 1003
415.      REN=RENG
416.      OR=PRG
417.      VISC=VISC G
418.      KF=KG
419.      VF=VG
420.      1003 Y=1.0
421.      AA=IDH
422.      BB=OIST
423.      CC=0.0
424.      HH=HIPC
425.      1031 CALL      SPMASE(AA,BB,CC,HH,Y,REN,KF,PR,FM,VISC,VISWAL,CG)
426.      HIPC=HH
427.      HITP=HH
428.      IF(X.GE.1.0)H2P2=HITP
429.      GO TO 850
430.      5000 CALL      TWORME(X,VISC,VISCL,H2P1,KL,IDH,GF,SPHTL)
431.      IF(X.LE.0.0)GO TO 7300
432.      VAVG=GF/(1/VOIDF+DENS L)
433.      CALL      DTINRO(TSHF,TSAT,PCLET,DITI)
434.      CALL      CRITDY(RENG,RENF,MFG,0,PI,GC,IDH,LENGTH,DLNGTH,FBL,FB,
435.      3DENS L,DENSG,SURFTN,VISCL,L17F,X,IPRI,PRG,HITPC,HITD,GF,KG,
436.      3H2P1,XFB)
437.      IF(X.GT.XFB)H2P2=HITPC
438.      GO TO 525
439.      850 HITPC=HITD
440.      IF(X.GE.1.0)HITP3=0.0
441.      7 IF(MODE.EQ.2)GO TO 1956
442.      XYZ=IDH/ODH/HW
443.      IF(MODE.EQ.3)XYZ=1.0/UDUT
444.      U=1.0/((1.0/HITP +XYZ+((ALOG(ODH /IDH ))*IDH
445.      3/(2.0*KM)) +FOULNW+FOULNF)
446.      IF(MODE.EQ.3)GO TO 1967
447.      DO=U*AREA*(TWIN-TEMPF)
448.      GO TO 1957
449.      1967 XY=(U*AREA)/(MASAIR*CPAIR)
450.      EFFECT=1.0-(EXP(-(XY)))
451.      IF(X.GT.1)GO TO 173
452.      DO=MASAIR*CPAIR*EFFECT*(THOTG-TEMPF)
453.      TOUT(I,II)=THOTG-DO/(MASAIR*CPAIR)
454.      GO TO 1957
455.      173 IF(THIN(I-1,II).LE.TEMP F)GO TO 171
456.      GO TO 174
457.      171 WRITE(6,656)
458.      656 FORMAT(' ***** TEMPERATURE OF INCOMING HOT AIR IS LESS THAN OR
459.      3EQUAL TO TEMPERATURE OF THE FLUID *****')
460.      GO TO 121
461.      174 DO=MASAIR*CPAIR*EFFECT*(THIN(I-1,II)-TEMPF)
462.      TOUT(I,II)=THIN(I-1,II)-DO/(MASAIR*CPAIR)
463.      GO TO 1957
464.      1956 DO=QFLUX*AREA
465.      1957 UF=1.0/((1.0/HITP)+FOULNF)
466.      TSHF=TEMPF+(DO/(UF*AREA))
467.      IF(X.GE.1.0)GO TO 420
468.      IF(BL.GT.0.0)GO TO 5007
469.      GO TO 420
470.      5007 IF(ABS(TSHF-TSHFP)-TA)422,422,136
471.      420 IF(Y.EQ.1.0)GO TO 137
472.      IF(ABS(TSHF-TSHFP)-TA)6000,6000,137
473.      5000 IF(X.GT.0.0.AND.X.LT.1.0)GO TO 5000
474.      GO TO 525
475.      137 IF(X.LT.1.0)GO TO 517
476.      CALL      VISHGW(TSHF,VISWAL)
477.      GO TO 518
478.      517 CALL      VISHLW(TSHF,VISWAL)
479.      518 Y=Y+1.0
480.      TSHFP=TSHF

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481. GO TO 1031
482. 525 REN=RENF
483. IF(X.GE.1.0)REN=RENG
484. IF(REN-2000.0)113,813,814
485. 913 F=64.0/DEN
486. IF(X.GE.1.0)GO TO 422
487. GO TO 560
488. 914 F=4.0*0.079/(REN**0.25)
489. IF(X.GE.1.0)GO TO 422
490. 560 IF(L.EQ.0)GO TO 422
491. 136 IF(TSHF.GT.TSAT)GO TO 5010
492. TSHF=(THDP+TSAT)/2.0
493. THDP=TSHF
494. IHBR=IHBB+1
495. GO TO 5011
496. 5010 IF(IMBB.GT.0)GO TO 422
497. 5011 CALL MPOOLB(DTA,TSHF,TSAT,DITI,HITPB,SPHTL,HFG,CSF,HFLUX,
498. DBEX,VISCL,SURFTN,DENSL,DENSG,TEMPF,PRF,X)
499. IF(X.LE.0.0)GO TO 5009
500. 7300 NSCB=2.0-((TSAT-TEMPF)/DTI)
501. HITP=((HITPB**4.0)+(H2P1**4.0))**((1.0/4.0)
502. IF(H2P1.GT.HIPC)GO TO 7301
503. 7302 HITP=((HITPB**NSCB)+(HIPC**NSCB))**((1.0/NSCB)
504. 7301 IF(LFB.EQ.0)GO TO 6001
505. C1=((X-XFB)/(1.0-XFB))**((1.0/A))
506. C2=1.0-C1
507. IF(X.GT.XFB)HITP=(C1*H2P2)+(C2*HITP)
508. 6001 TSHFP=TSHF
509. SL=SL+1.0
510. GO TO 7
511. 5008 XSCB=HFLUX*7600.0*AREAH/(FLRATE*HFG)
512. IF(HFLUX.EQ.0.0)GO TO 7302
513. VOIDF=1.0/((1.0+(1.0-XSCB)/XSCB)*(DENSG/DENSL))
514. LIOF=1.0-VOIDF
515. DENSM=VOIDF*DENSG+((1.0-VOIDF)*DENSL
516. VG=GF/DENSM
517. PENG=IDH *VG*DENSG/VISCG
518. GO TO 5000
519. 422 TSHW=TSHF+(DO*THICKH/XM/AREAH)
520. AVTSH=(TSHW+TSHF)/2.0
521. TSHS=TSHS+AVTSH
522. IF(MODE.EQ.2)GO TO 1963
523. UAVEV=UAVEV+(U*AREAH)
524. 1963 SUHITP=SUHITP+HITP
525. TWOUT=TWIN-(DO/(MHTWTR*SPHTW))
526. IF(MODE.EQ.3)THIN(I,II)=TOUT(I,II)
527. IF(MODE.EQ.3)DOWNST=DOWNST+TOUT(I,II)
528. Q=Q+DO
529. C FINDING TEMPERATURE AND PRESSURE FOR NEXT HIGHER POINT
530. IHBB=0
531. HF=HFBOT+Q/FLRATE
532. IF(X.GE.1.0)GO TO 931
533. IF(X.GT.0.0)GO TO 185
534. IF(RL.GT.0.0)GO TO 185
535. 186 CALL LIQCP(TEMPF,SPHTL)
536. TEMPF=TEMPF+(HF-HFBOT)/SPHTL
537. 453 CALL LIO HF(TEMPF,HFCK)
538. IF(ABS(HF-HFCK)-0.05)451,451,452
539. 452 CALL LIQCP(TEMPF,SPHTL)
540. TEMPF=TEMPF+(HF-HFCK)/SPHTL
541. GO TO 453
542. 931 TEMPF=TEMPF+QO/(FLRATE*SPHTG)
543. 451 TOF=TEMPF
544. IF(PP.GT.0.0)GO TO 540
545. 195 IF(X.GT.0.0.AND.X.LT.1.0)GO TO 16
546. IF(X.EQ.0.0.OR.X.GE.1.0)GO TO 17
547. 16 CALL PRES2B(OPAC,OPA,OPACP,LIOF,DENSL,VOIDF,DENSG,THETA,
548. DSIGMA,XBOTOM,F,DLNGTH,IDH,GF,GC,FLRATE,PI,X,XFB,RENF,RENG,XP,
549. DVISCL,VISCG,PHI,IX)
550. IF(XBOTOM.LE.0.0)GO TO 901
551. 113 CALL XPCAL(RENF,PENG,VISCL,VISCG,DENSL,DENSG,X,XP)
552. IF(IX.GT.0)GO TO 114
553. 524 CALL PHICAL(RENF,RENG,PHI,XP)
554. GO TO 21
555. 17 CALL SPRESS(DLNGTH,DENSG,THETA,SIGMA,F,IDH,VF,GC,DENSL,X,
556. DDPH,DPE,OPA)
557. 21 PRESS=PRESS-DDPH-DPE-OPA
558. IF(II.EQ.4)OPDL=(DPH+DPE+OPA)/DLNGTH
559. DENSM=(LIOF*DENSL+VOIDF*DENSG)
560. IF(III.EQ.4)PRESS=PRESS-0.5*(VF*VF/2.0/GC)*DENSM

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561. IF(X.GE.1.0)GO TO 801
562. PRESS=PRESS/144.0
563. CALL SAT TEN(PRESS,TSAT)
564. CALL HFLS(TSAT,HFL)
565. QD=Q-QD
566. IF(L.GT.0)GO TO 432
567. IF(IMYSTR.EQ.0)GO TO 864
568. GO TO 865
569. 864 IF((ABS(TSHF-TEMPF).GE.24.0)GO TO 866
570. GO TO 501
571. 865 IF(TSHF.LT.TSAT)GO TO 501
572. IF(TEMPF.GT.TSAT)GO TO 432
573. 866 CALL SCODLS(KL,HFG,SURFTN,TSAT,DENSG,DENSL,PI,IDH,LENGTH,
574. 3REN,VISCL,PRF,VL,TEMPF,Q,QA,QD)
575. IF(QA.GT.QD)GO TO 427
576. GO TO 432
577. 427 DTI=0.0
578. DTA=TSHF-TSAT
579. L=L+1
580. CALL SBCCOL(TSHF,TSAT,TSHFO,TSATO,LO,LENGTH,PO,PRESS,Q,TEMPFO,
581. 3TEMPF,HF,HFL,FLRATE,THPP)
582. CALL CORECT(TSHS,AVTSH,QD,Q,QQ,TSHW,TSHF,THICK,KM,AREAH,
583. 3DLNGTH,HEITB,LENGTH,RM,II,LO,TEMPFO,IDH,PI,UF)
584. MASAIR=(DENAIR*VELAIR*3600.0*(LENGTH-LO)*TJSPCE)/12.0
585. TOUT(I,II)=THOTG-QD/(MASAIR*CPAIR)
586. IF(L.EQ.1)DTI=TSAT-TEMPF
587. GO TO 501
588. C BEGINNING OF TWO PHASE FLOW
589. 432 TST=TSAT+TSP
590. CALL HFLS(TST,HFL)
591. IF(HFL-HF)9,201,201
592. 201 PP=PP+1.0
593. IF(BL.GT.0.0)GO TO 186
594. 540 TEMPF=TDF
595. GO TO 501
596. 9 DENSG=(PRESS*MWT)/(10.730*(TEMP+460.0))
597. CALL GASVIS(TEMPF,VISCG)
598. CALL LIQ DEN(TEMPF,DENSL)
599. TEMPF=TSAT
600. IF(P.GT.0.0)GO TO 630
601. CALL BULKPT(HF,HFL,HFOLD,HFOLD,LO,LENGTH,PO,PRESS,FLRATE,3,TSHFO,
602. 3TSHF,VOIDQ,VOIDF,TSAT)
603. CALL CORECT(TSHS,AVTSH,QD,Q,QQ,TSHW,TSHF,THICK,KM,AREAH,
604. 3DLNGTH,HEITB,LENGTH,RM,II,LO,TEMPFO,IDH,PI,JF)
605. MASAIR=(DENAIR*VELAIR*3600.0*(LENGTH-LO)*TJSPCE)/12.0
606. TOUT(I,II)=THOTG-QD/(MASAIR*CPAIR)
607. TEMPF=TSAT
608. DTI=1.0
609. L=2
610. X=XSCB
611. IF(XSCB.EQ.0.0)X=1E-10
612. GO TO 14
613. 630 CALL HFGTS(TCR,TST,HFGO)
614. XTOP=(HF-HFL)/HFGO
615. DX=XTOP-XBOTM
616. X=XTOP
617. IF(X.LT.1.0)GO TO 516
618. HG=HFL+HFGO
619. LENGTH=LO+((HG-HFL)/(HF-HFL))*DLNGTH
620. IF(II.EQ.4)DLNGTH=HEITB-LENGTH
621. IF(II.EQ.4)GO TO 345
622. DLNGTH=(HEITB-LENGTH)/(RM-II)
623. 345 Q=(FLRATE*(HF-HG))
624. QD=QD-(FLRATE*(HF-HG))
625. HF=HG
626. X=1.0
627. VOIDF=1.0
628. PRESS=PRESS*144.0
629. 516 IF(1.0-X)901,901,14
630. 14 IX=IX+1
631. 199 CALL XPCAL(RENF,RENG,VISCL,VISCG,DENSL,DENSG,X,XP)
632. IF(IX.GT.0)GO TO 114
633. GO TO 199
634. 114 IX=0
635. IF(X.GT.0.0.AND.X.LT.1.0)LIOF=(1.0-(1.0/((1.0+XP**0.8)**0.379)))
636. IF(X.GT.0.0.AND.X.LT.1.0)VOIDF=1.0-LIOF
637. PRESS=PRESS*144.0
638. IF(XROTW.LE.0.0.AND.II.EQ.4)GO TO 524
639. 901 PRESS=PRESS/144.0
640. IF(MODE.NE.3)GO TO 1968

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641. IF (I1.EQ.M) DPEXITE=PRESS
642. IF (I1.EQ.1.AND.I1.EQ.M) PVDHRI=PRESS-OPDL*UNHTL1/144.0
643. IF (I1.EQ.2.AND.I1.EQ.M) PVDHRI2=PRESS-OPDL*UNHTL2/144.0
644. 1968 CALL SAT TEM(PRESS,TSAT)
645. D=P41
646. 501 VOIDFS=VOIDFS+(VOIDF*(LENGTH-LO)*XAEVAP)
647. LIOFS=LIOFS+((1.0-VOIDF)*(LENGTH-LO)*XAEVAP)
648. IF (MODE.NE.3) GO TO 1969
649. IF (I1.EQ.1.AND.I1.EQ.M) VAPV1=(VOIDF*UNHTL1*XAEVAP)
650. IF (I1.EQ.1.AND.I1.EQ.M) LIOV1=((1.0-VOIDF)*UNHTL1*XAEVAP)
651. IF (I1.EQ.2.AND.I1.EQ.M) VAPV2=(VOIDF*UNHTL2*XAEVAP)
652. IF (I1.EQ.2.AND.I1.EQ.M) LIOV2=((1.0-VOIDF)*UNHTL2*XAEVAP)
653. 1969 HWSUM=HWSUM+HW
654. IF (X.GE.1.0) MX=MX+1
655. FLAG=0
656. IF (IPRI.EQ.0) GO TO 120
657. IF (MX.EQ.1) PRINT 255,LENGTH,TSAT,TEMPF
658. 255 FORMAT('1. QUALITY BECAME ONE AT (FT)=',F8.4,3X,'TSAT=',F7.3,3X,
659. 'BULK FLUID TEMPERATURE (F)=',F7.3)
660. 120 XTOP=X
661. IF (X.GE.1.0) H1PC=0.0
662. IF (X.GE.1.0) H2P2=H1P
663. IPEVAP=(ICP*M)/QNPC
664. IF (I1.EQ.IPEVAP) GO TO 109
665. GO TO 110
666. 109 ICP=ICP+1
667. IF (MODE.EQ.3) GO TO 1970
668. IF (ICNHT.EQ.1.AND.CNROTN.EQ.1) GO TO 71
669. IF (I1.EQ.1) GO TO 71
670. IF ((ABS(PRESS-PMCHAN)).LE.0.05) GO TO 71
671. GO TO 72
672. 1970 A1=FLRATE*X
673. IF (I1.EQ.1) GO TO 71
674. CALL DELPBE(A1,LBE,LEQBE,GC,ZBE,IDBE,OPVAP,NTUBEE,PI,
675. 3DENSL,VISCL,XABF,OLPBE,OPDL,UNHTL1,DENSG,VISCG,IDUC,NELBOW,
676. 3LETMCH,OPNOMH,OPTEE,DELBO,DPEXIT)
677. PRESS=OPVAP+OPVAP-((OPTEE+DELBO+DPEXIT)*DENSG/144.0)
678. CALL QLOSS(A1,TEB,TROOM,LBE,RESBE,HFBOT,QBE,HBOTSE)
679. IF ((ABS(PRESS-PMCHAN)).LE.0.05) GO TO 73
680. A1=X*FLRATE
681. GO TO 72
682. 71 CALL DELPBE(A1,LBE,LEQBE,GC,ZBE,IDBE,OPVAP,NTUBEE,PI,
683. 3DENSL,VISCL,XABF,OLPBE,OPDL,UNHTL1,DENSG,VISCG,IDUC,NELBOW,
684. 3LETMCH,OPNOMH,OPTEE,DELBO,DPEXIT)
685. 3BOTBE=PLOHRI+OLPBE+OPFLM1+OPFIL1
686. CALL QLOSS(A1,TEB,TROOM,LBE,RESBE,HFBOT,QBE,HBOTSE)
687. PMCHAN=3VPHRI-OPVAP-((OPTEE+DELBO+DPEXIT)*DENSG/144.0)
688. 73 PRESS=PMCHAN
689. IF (MODE.EQ.3) AVDOWN=DOWNST/I1
690. TFLUD(I)=TEMPF
691. FLRAT(I)=FLRATE
692. QULTY(I)=X
693. QINPT(I)=Q
694. SUEVT(I)=TSHS
695. EVAPST=SUEVT(I)/RM
696. A1=FLRAT(I)*QULTY(I)
697. A2=FLRAT(I)-A1
698. B1=A1*NTUBEE
699. B2=A2*NTUBEE
700. VDFP(I)=VOIDFS
701. LOFP(I)=LIOFS
702. DYNPER=((XAEVAP*HEITB-VDFP(I))/(XAEVAP*HEITB))*100.0
703. DERDY(I)=DYNPER
704. UAVEV=UAVEV/AREAP
705. HWAVG=HWSUM/RM
706. SUHITP=SUHITP/RM
707. WATH(I)=HWAVG
708. HEVAP=(QINPT(I))/(HEITB*IDH*P1*((TSHS/RM)-TSAT))
709. GO TO 74
710. 72 FLRATE=FLRATE*(((PEB(I)-PMCHAN)/(PEB(I)-PRESS))*POWER)
711. GO TO 91
712. 74 FLRAT(I)=FLRATE
713. IF (MODE.NE.3) GO TO 1971
714. IF (I1.EQ.2) GOTO 204
715. PRINT 202,PROTBE
716. 202 FORMAT('1. PRESSURE IN THE CHAMBER BELOW THE EVAPORATOR (PSIA)=',
717. 3F7.3)
718. PRINT 203,PLOHRI
719. 203 FORMAT('1. PRESSURE IN THE LIQUID HEADER FOR EVAPORATOR 203 =',
720. 3PSIA),F7.3)

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721.      GO TO 1971
722.      204 PRINT 204, PLOHR2
723.      206 FORMAT(' ', 'PRESSURE IN THE LIQUID HEADER FOR EVAPORATOR ROW # 2 (
724.      3PSIA)=', F7.3)
725.      1971 PRINT 256.1
726.      256 FORMAT(' ', 'FOR EVAPORATOR LIQUID HEADER TUBE ROW ', I1)
727.      PRINT 221
728.      221 FORMAT(' ', '-----')
729.      IF (MODE.EQ.2) PRINT 1958, QFLUX
730.      GO TO 1959
731.      1958 FORMAT(' ', 'HEAT FLUX PER UNIT AREA (BTU/HR/SQ-F)=', F9.3)
732.      1959 PRINT 15, PE9(1), TER, SAAT(1), FLPAT(1)
733.      18 FORMAT(' ', 'AT EVAP TUBE BOTTOM: PRESSURE(PSIA)=', F7.3, 3X,
734.      3TEMP(F)=', F7.3, 3X, 'TSAT(F)=', F7.3, 3X, 'FLOW RATE(LB/HR)=', F3.3)
735.      PRINT 998, HFBOT, DEGSUR
736.      998 FORMAT(' ', 'ENTHALPY(BTU/LB)=', F7.3, 3X, 'INLET SUBCOOLING(F)=', F6.3
737.      3)
738.      PRINT 257, LENGTH, X, VOIDF
739.      257 FORMAT(' ', 'DISTANCE FROM BOTTOM OF EVAPORATOR(FT)=', F8.4, 3X,
740.      3QUALITY=', F6.4, 3X, 'VOID FRACTION=', F6.4)
741.      PRINT 258, AVTSH, TEMPF, TSAT
742.      258 FORMAT(' ', 'WALL TEMP(F)=', F7.3, 3X, 'BULK FLUID TEMP(F)=', F7.3, 3X,
743.      3SATURATION TEMP(F)=', F7.3)
744.      PRINT 2260, U
745.      2260 FORMAT(' ', 'EFFECTIVE HEAT TRANSFER COEFFICIENT "U" (BTU/HR/FT2/F)
746.      3=', F9.4)
747.      PRINT 260, HIPC
748.      260 FORMAT(' ', '1-PHASE LIQUID CONVECTIVE HEAT TRANSFER COEFFICIENT
749.      3(BTU/HR/FT-FT/F)=', F9.4)
750.      PRINT 261, H2O1, HITR9
751.      261 FORMAT(' ', '2-PHASE CONVECTIVE COEFFICIENT(BTU/HR/FT-FT/F)=', F9.4
752.      3, 3X, 'BOILING COEFFICIENT(BTU/HR/FT-FT/F)=', F9.4)
753.      PRINT 262, H2O2, Q
754.      262 FORMAT(' ', '1-PHASE VAPOUR CONVECTIVE COEFFICIENT(BTU/HR/FT-FT/F)
755.      3=', F9.4, 3X, 'CUMULATIVE HEAT INPUT(BTU/HR)=', F9.4)
756.      PRINT 244, HITP
757.      244 FORMAT(' ', 'ACTUAL INSIDE "H" USED IN THE CALCULATION(BTU/HR/FT-2/
758.      3F)=', F9.3)
759.      PRINT 1984, PEXITE
760.      1988 FORMAT(' ', 'PRESSURE(PSIA)=', F7.3)
761.      IF (MODE.NE.3) GO TO 1972
762.      PRINT 263, TOUT(1, M)
763.      263 FORMAT(' ', 'DOWNSTREAM AIR TEMP.(F)=', F7.3)
764.      PRINT 188, AVDOWN, EVAPST
765.      188 FORMAT(' ', 'AVERAGE DOWNSTREAM AIR TEMP.(F)=', F7.3, 3X, 'AVERAGE VAL
766.      3E TEMP.(F)=', F7.3)
767.      PRINT 900, UOUT
768.      900 FORMAT(' ', 'EVAPORATOR AIR SIDE "U" (BTU/HR/SQ-FT/F)=', F9.4)
769.      IF (I.EQ.1) PRINT 207, PVPHR1
770.      IF (I.EQ.2) PRINT 208, PVPHR2
771.      207 FORMAT(' ', 'PRESSURE IN THE VAPOUR HEADER FOR EVAPORATOR ROW # 1 (
772.      3PSIA)=', F7.3)
773.      GO TO 209
774.      208 FORMAT(' ', 'PRESSURE IN THE VAPOUR HEADER FOR EVAPORATOR ROW # 2 (
775.      3PSIA)=', F7.3)
776.      209 PRINT 22, PMCHAM
777.      22 FORMAT(' ', 'MIXING CHAMBER PRESSURE (PSIA)=', F7.3)
778.      1972 PRINT 289, A2, A1
779.      289 FORMAT(' ', 'RECIRCULATION RATE FOR THIS TUBE(LB/HR)=', F9.3, 3X,
780.      3VAPOUR FLOW RATE(LB/HR)=', F9.3)
781.      PRINT 19, R2, R1
782.      19 FORMAT(' ', 'RECIRCULATION RATE FOR THIS ROW(LB/HR)=', F9.3, 4X,
783.      3VAPOUR FLOW RATE FOR THIS ROW(LB/HR)=', F9.3)
784.      PRINT 176, DYNDPO
785.      176 FORMAT(' ', 'PERCENT OF EVAP. LENGTH FLOODED UNDER DYNAMIC CONDITIO
786.      3NS=', F5.1)
787.      SHEVAP=HEVAP*5.6783
788.      PRINT 187, HEVAP, SHEVAP
789.      187 FORMAT(' ', 'EVAP. HEAT TRANSFER COEFF.(BTU/HR/FT2/F)=', F10.4, 11X,
790.      3IN S.I. UNITS(W/M2/K)=', F10.4)
791.      SUAVEV=UAVEV*5.6783
792.      SSUHT=SUHTD*5.6783
793.      SHWAVG=HWAVG*5.6783
794.      PRINT 789, SUHTD, SSUHT
795.      789 FORMAT(' ', 'AVERAGE ACTUAL "H" IN THE EVAPORATOR(BTU/HR/FT2/F)=',
796.      3F10.3, 3X, 'IN S.I. UNITS(W/M2/K)=', F10.4)
797.      IF (MODE.NE.1) GO TO 110
798.      PRINT 577, TWIN
799.      577 FORMAT(' ', 'INCOMING TEMP. OF THE WATER(F)=', F7.3)
800.      PRINT 191, TWOUT, HW

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901. 191 FORMAT(' ', 'OUTGOING TEMP. OF THE WATER(F)=' , F7.3, 3X, 'WATER SIDE '
902. 3H' (BTU/HR-FT-2/F)=' , F10.3)
903. PRINT 981, HWAVG, SHWAVG
904. 981 FORMAT(' ', 'AVERAGE WATER SIDE "H" FOR THE EVAP.=' , F10.3, 17X,
905. 3'IN S.I. UNITS(W/M2/K)=' , F10.4)
906. PRINT 198, UAVEV, SUAVEV
907. 198 FORMAT(' ', 'AVERAGE EVAPORATOR OVERALL "U"(BTU/HR-FT-FT/F)=' , F10.4
908. 3, 7X, 'IN S.I. UNITS(W/M2/K)=' , F10.4)
909. 110 IF(L.EQ.1)GO TO 410
910. GO TO 412
911. 410 SCL=LENGTH
912. IF(IPP1.EQ.0)GO TO 125
913. PRINT 414, SCL
914. 414 FORMAT(' ', 'INITIATION OF SUBCOOLED BOILING AT HEIGHT FROM BOTTOM
915. 3OF HEATER(FT)=' , F7.3)
916. 125 L=L+1
917. 412 IF(P.EQ.1.0)GO TO 413
918. GO TO 1
919. 413 SCB=LENGTH
920. IF(IPP1.EQ.0)GO TO 126
921. PRINT 415, SCB
922. 415 FORMAT(' ', 'BULK BOILING BEGINS AT HEIGHT FROM BOTTOM OF HEATER
923. 3(FT)=' , F7.3)
924. 126 P=P+1.0
925. CONTINUE
926. IF(ICNROT.EQ.1.AND.CNROT.EQ.1)GO TO 919
927. I=I+1
928. A1=0.5*A1
929. FLRATE=FLRATE*((PEB(1)-PMCHAM)/(PEB(1)-PRESS))**POWER)
930. IF(I.GT.NR) GO TO 78
931. GO TO 91
932. C *****
933. C * END OF EVAPORATOR TUBE CYCLE *
934. C *****
935. 78 TOTALQ=0.0
936. TOTALF=0.0
937. TOTREC=0.0
938. IF(MODE.NE.3)GOTO 1975
939. EFLOP1=((THOTG-AVDOWN)/(THOTG-TAIR))*100.0
940. EFLOP2=((4.0*EFLOP1/100.0)/(1.0+(3.0*EFLOP1/100.0)))*100.0
941. PRINT 287, EFLOP1, EFLOP2
942. 287 FORMAT(' ', 'EFFECTIVENESS OF THERMOSIPHON FOR A SINGLE LOOP (%)=' ,
943. 3F7.3, 3X, 'FOR FOUR LOOPS (%)=' , F7.3)
944. 1975 TEMPF=TFLUD(1)
945. DO 92 I=1, NR
946. 92 TOTALQ=TOTALQ+QINPT(I)*NTUBEE
947. DO 93 I=1, NR
948. 93 TOTALF=TOTALF+FLRAT(I)*NTUBEE
949. DO 94 I=1, NR
950. 94 TOTPEC=TOTREC+FLRAT(I)*NTUBEE*(1-QULTY(I))
951. QIN=TOTALQ
952. PRINT 95, QIN
953. 95 FORMAT(' ', 'TOTAL CUMULATIVE HEAT INPUT(FOR ALL THE TUBES(BTU/HR))
954. 3=' , F14.4)
955. IF(MODE.NE.2)GOTO 1998
956. DO 96 I=1, NR
957. 96 AVWATH=AVWATH+WATH(I)
958. AVWATH=AVWATH/NR
959. 1998 QES=TOTALQ
960. FLRATE=TOTALF
961. FRECRC=TOTREC
962. FLRATE=FLRATE-FRECRC
963. IF(MODE.EQ.3)GO TO 1982
964. VOIDFS=(VOFR(1)+VOFR(2)+VOFR(3))*NTUBEE
965. LIOFS=(LOFR(1)+LOFR(2)+LOFR(3))*NTUBEE
966. AVEVT1=SUEVT(1)/RM
967. AVEVT2=SUEVT(2)/RM
968. GO TO 920
969. 1982 IF(NROWE.EQ.1)GO TO 99
970. VOIDFS=(VOFR(1)+VOFR(2)+VAPV1+VAPV2)*NTUBEE
971. LIOFS=(LOFR(1)+LOFR(2)+LIOV1+LIOV2)*NTUBEE
972. GO TO 99
973. 99 LIOFS=(LOFR(1)+LIOV1)*NTUBEE
974. VOIDFS=(VOFR(1)+VAPV1)*NTUBEE
975. 90 QES=TOTALQ
976. AVEVT1=SUEVT(1)/RM
977. PERDY1=PERDY(1)
978. IF(NROWE.EQ.1)GO TO 211
979. AVEVT2=SUEVT(2)/RM
980. PERDY2=PERDY(2)

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881. 211 DEVTOP=DEXITE
882. GO TO 920
883. 919 QIN=0
884. FLRATE=FLRATE*NTUBEE*NR
885. LIOFS=LIOFS*NTUBEE*NR
886. VOIDFS=VOIDFS*NTUBEE*NR
887. QES=QIN*NR*NTUBEE
888. AVWATH=HWAVG
889. PRINT Q5,QES
890. AVTSHS=TSHS/RM
891. AVEVT1=AVTSHS
892. AVEVT2=AVTSHS
893. AVEVT3=AVTSHS
894. TOTALF=FLRATE
895. FRECRC=FLRATE*(1.0-X)
896. TOTREC=FRECRC
897. FLRATE=FLRATE*X
898. 920 DEVTOP=PRESS
899. TEVTOP=TE4PF
900. TSETOP=TSAT
901. TOTPLE=FLRATE
902. IF (MODE.EQ.3) GOTO 556
903. HAVEVP=QES/(((AVEVT1+AVEVT2)/2.0)-TSAT)*NTUBEE*NR*PI*(IDH
904. D*HEITB)
905. HEVPSI=HAVEVP*5.6783
906. PRINT 988,HAVEVP,HEVPSI
907. 888 FORMAT(' ',AVERAGE HEAT TRANSFER COEFF. FOR THE EVAP.(BTU/40-FT2/
908. 9F)=' ',F9.4,3X,' IN ',S.I. UNITS(W/M2-K)=' ',F10.4)
909. IF (ICNROT.EQ.1.AND.CNROT.EQ.1) GO TO 557
910. ELEFT1=ELEFT(1)
911. ELEFT2=ELEFT(2)
912. ELEFT3=ELEFT(3)
913. PERDY1=PERDY(1)
914. PERDY2=PERDY(2)
915. PERDY3=PERDY(3)
916. GO TO 718
917. 557 ELEFT1=TWOUT
918. ELEFT2=TWOUT
919. ELEFT3=TWOUT
920. PERDY1=DYNPER
921. PERDY2=DYNPER
922. PERDY3=DYNPER
923. IF (CNROT.EQ.1) GO TO 718
924. GO TO 719
925. 718 CALL LIOVOL(FRECRC,VLIOHE)
926. LIOFS=LIOFS+VLIOHE
927. 719 LIOFS=LIOFS+VSTPAN
928. 556 HF=HFBOT+QES/TOTALF
929. *****
930. * START OF SEPARATOR CALCULATIONS
931. *****
932. CALL LIOHF(TEMPF,HFL)
933. CALL LIOCP(TEMPF,SPHTL)
934. HR=HFL
935. CALL GAS CP(TEMPF,SPHTG)
936. HF=(TOTALF*HF-TOTREC*HR)/FLPATE
937. X=1.0
938. VOIDF=1.0
939. *****
940. * VAPOR HEADER PIPE FLOW AND HEAT TRANSFER CALCULATIONS
941. *****
942. CALL FRCSHN(FLRATE,IDUC,DENSG,VISCG,P1,F,VF)
943. RENG=VF*IDUC*DENSG/VISCG
944. IF (FLAGUP.EQ.1) PRESS=(PRESS*144.0)-(DPUPXD*144.0)
945. IF (FLAGUP.EQ.1) GO TO 923
946. DPF=F*(LUPXD+LEDAE)/IDUC*((VF**2.0)/2.0/GC)*DENSG
947. DPH=LUPXD*SIN(BETA)*DENSG
948. PRESS=(PRESS*144.0-DPH-DPF)
949. 923 PRESS=PRESS/144.0
950. CALL SAT TEH(PRESS,TSAT)
951. EFFECT=1.0-(EXP(-(LUPXD/(RESUP*FLRATE*SPHTG))))
952. QUPXD=EFFECT*FLRATE*SPHTG*(TSETOP-TAIR)
953. CALL HFGTS(TCP,TSAT,HFG2)
954. CALL HFLS(TSAT,HF2)
955. HF=HF-(QUPXD/FLRATE)
956. TEMPF=TEMPF-(QUPXD/(FLRATE*SPHTG))
957. HG2=HFG2+HF2
958. IF (HF.GT.HG2) GO TO 710
959. X=((HF-HF2)/HFG2)*X)/2.0
960. TEMPF=TSAT

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961. FLUPXO=FLRATE*(1.0-X)
962. FRECRC=FRECRC+FLUPXO
963. FLRATE=FLRATE-X
964. CALL FPCSHN(FLUPXO, IDUC, DENSL, VISCL, PI, F, VF)
965. RENF=VF*IDUC*DENSL/VISCL
966. CALL XPCAL(RENF, RENGL, VISCL, VISCG, DENSL, DENSG, X, XP)
967. LIOF=(1.0-(1.0/((1.0+XP**0.8)**0.378)))
968. VOIDF=1.0-LIOF
969. 710 VOIDFS=VOIDFS+(VOIDF*FLUPXO*XAPRX)
970. LIOFS=LIOFS+(LIOF*FLUPXO*XAPRX)
971. X=1.0
972. VOIDF=1.0
973. *****
974. *
975. * RECIRCULATION TUBE CALCULATIONS
976. *****
977. IF (FRECRC.LE.0.15) EFFECT=1.0
978. IF (FRECRC.LE.0.15) GO TO 500
979. EFFECT=1.0-(EXP(-(RECRC/(RESLO*FRECRC*SPHTL))))
500. QRECR= EFFECT*FRECRC*SPHTL*(TEVTOPT-TAIR)
980. TRECRC=TEVTOPT-(EFFECT*(TEVTOPT-TAIR))
981. IF (FRECRC.LE.0.01) GO TO 502
982. HR=HR-(QRECR/FRECRC)
983. 502 HELOXO=((TOTAL*HFBOT)-(FRECRC*HR))/FLRATE
984. TEMPO=TEMF
985. 666 CALL LIQHF(TEMPO, HFCK)
986. IF (ABS(HELOXO-HFCK))-0.01) 667.667.668
987. 668 CALL LIQCR(TEMPO, SPHTLO)
988. TEMPO=TEMPO+((HELOXO-HFCK)/SPHTLO)
989. GO TO 666
990. 667 CALL LIQDEN(TEMPO, DNSLO)
991. CALL LIQVIS(TEMPO, VSCL)
992. IF (MODE.EQ.3) GO TO 1983
993. IF (IPRINT.EQ.0) GO TO 641
994. WRITE(6.658) FRECRC
995. WRITE(6.177) FLRATE
996. 658 FORMAT(' ', 'TOTAL RECIRCULATION RATE (FOR ALL TUBES (LB/HR)) =', F9.2)
997. 177 FORMAT(' ', 'TOTAL VAPOUR RATE (FOR ALL TUBES (LB/HR)) =', F9.2)
998. GO TO 641
999.
1000. 1983 K1=FLRAT(1)*(1.0-QULTY(1))*NTUBEE
1001. CALL FRCSHN(K1, RECID, DENSL, VISCL, PI, F, VRECR)
1002. FRIC=((VRECR**2.0)/2.0/GC)*(F/RECID)
1003. RENF=VRECR*RECID*DENSL/VISCL
1004. ZREC1=((PLQHR1-PVPHR1)*144.0/DENSL)-(FRIC*LEGR)/((1.0+FRIC)
1005. ZMAX=(HEITB*CDS(SIGMA))+HDIAME
1006. IF (NRWE.EQ.1) GO TO 212
1007. K2=FLRAT(2)*(1.0-QULTY(2))*NTUBEE
1008. CALL FRCSHN(K2, RECID, DENSL, VISCL, PI, F, VRECR)
1009. FRIC=((VRECR**2.0)/2.0/GC)*(F/RECID)
1010. RENF=VRECR*RECID*DENSL/VISCL
1011. ZREC2=((PLQHR2-PVPHR2)*144.0/DENSL)-(FRIC*LEGR)/((1.0+FRIC)
1012. 212 IF (ZREC1.LE.ZMAX) GO TO 1637
1013. FLRATE=FLRAT(1)*(ZMAX/ZREC1)
1014. WRITE(6.639)
1015. 639 FORMAT(' ', 'HEIGHT OF LIQUID COLUMN TOO LARGE; REDUCING FLOW RATE')
1016. FLAG2=FLAG2+1
1017. TEMF=TEB
1018. PRESS=PEB(1)
1019. GO TO 307
1020. 1637 FLDEV1=ZREC1/COS(SIGMA)
1021. WRITE(6.636) FLDEV1
1022. 636 FORMAT(' ', 'LENGTH OF LIQUID COLUMN IN RECIRCULATION TUBE FOR ROW
1023. 3# 1(FT) =', F7.3)
1024. IF (NRWE.EQ.1) GO TO 213
1025. FLDEV2=ZREC2/COS(SIGMA)
1026. WRITE(6.633) FLDEV2
1027. 633 FORMAT(' ', 'LENGTH OF LIQUID COLUMN IN RECIRCULATION TUBE FOR ROW
1028. 3# 2(FT) =', F7.3)
1029. 213 SIGHT1=((PLQHR1-PVPHR1)*144.0/DENSL)
1030. PSIGT1=((SIGHT1-(2.0/12.0))/HEITB)*100.0
1031. IF (NRWE.EQ.1) GO TO 214
1032. SIGHT2=((PLQHR2-PVPHR2)*144.0/DENSL)
1033. PSIGT2=((SIGHT2-(4.0/12.0))/HEITB)*100.0
1034. 214 CALL VOIHOE(LHOEVP, HDIAME, ZREC1, HEITB, SIGMA, HEDVLE, PI, NTUBEE,
1035. VOIHOE, LIOHDE, FLAG3)
1036. VOIDFS=VOIDFS+VOIHOE*HEDVLC
1037. LIOFS=LIOFS+LIOHDE
1038. IF (NRWE.EQ.1) ZREC2=0.0
1039. VOIDFS=VOIDFS+XARCRC*(NRWE*RECRC-((ZREC1/COS(SIGMA))-(ZREC2/
1040. COS(SIGMA))))
LIOFS=LIOFS+XARCRC*((ZREC1/COS(SIGMA))+(ZREC2/COS(SIGMA)))

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1041.      GO TO 1984
1042.      ZMAX=(HEITB*COS(SIGMA))+HDIAE
1043.      CALL LIQDEN(TEVTP,DENSL)
1044.      CALL LIQVIS(TEVTP,VISCL)
1045.      CALL FRCSHN(FRECR,RECID,DENSL,VISCL,PI,F,VRECR)
1046.      FRIC=((VRECR**2.0)/2.0/GC)*(F/RECID)
1047.      RENF=VRECR*RECID*DENSL/VISCL
1048.      ZRECR=((P90T-TEVTP)*144.0/DENSL)-(FRIC*LEORE))/
1049.      3(1.0+FRIC)
1050.      IF(ZRECR.LE.ZMAX)GO TO 637
1051.      FLRATE=FLRAT(1)*(ZMAX/ZRECR)
1052.      WRITE(6,639)
1053.      FLAG2=FLAG2+1
1054.      TEMPF=TEF
1055.      PRESS=PEB(1)
1056.      GO TO 307
1057.      637 FLOFVP=ZRECR/(COS(SIGMA)*SIN(THETA))
1058.      PRCFL=(FLOFVP/HEITB)*100.0
1059.      WRITE(6,1636)FLOFVP
1060.      1636 FORMAT('1. LENGTH OF LIQUID COLUMN IN RECIRCUL. TUBE(FT)='F7.3)
1061.      CALL VOIHDE(LHDEVP,HOIAME,ZRECR,HEITB,SIGMA,HEOVLE,PI,NTJBE,
1062.      3VDFHDE,LHIDE,FLAG3)
1063.      VOIDFS=VOIDFS+VDFHDE*HEOVLC
1064.      LIOFS=LIOFS+LHIDE
1065.      IF(FLAG3.EQ.0)GO TO 681
1066.      VOIDFS=VOIDFS+(XARCRC*(RECRCL-(ZRECR/COS(SIGMA))))
1067.      LIOFS=LIOFS+(XARCRC*(ZRECR/COS(SIGMA)))
1068.      GO TO 299
1069.      681 LIOFS=LIOFS+(RECRCL*XARCRC)
1070.      1984 OPCDND=(P90T-PRESS)*144.0/DNSLO
1071.      *****
1072.      *          START OF CONDENSER CALCULATIONS          *
1073.      *****
1074.      299 IF(IPRINT.EQ.0)GO TO 124
1075.      PRINT 222,OPCDND
1076.      222 FORMAT('1. HEAT LOSS BETWEEN EVAP. AND COND. VAPOUR HEADER(BTU/HR
1077.      3)='F9.4)
1078.      PRINT 77,PRESS,TEMPF
1079.      77 FORMAT('1. PRESSURE AT THE TOP OF CONDENSER='F9.5,5X,
1080.      3TEMPERATURE AT THE TOP OF CONDENSER='F7.3)
1081.      PRINT 337,X,VOIDF,TSAT
1082.      337 FORMAT('1. QUALITY AT THE TOP OF CONDENSER='F7.4,5X,'VOID
1083.      3FRACTION='F8.5,3X,'SATURATION TEMP='F7.3)
1084.      124 CALL FRCSHN(FLRATE,IDC,DNSLO,VSCLO,PI,FTEF,VTEF)
1085.      FLRATE=FLRATE/(NTUBEC*NP)
1086.      CALL FRCSHN(FLRATE,IDC,DNSLO,VSCLO,PI,FCOP,VFCOP)
1087.      FRCOP=FCOP/IDC*((VFCOP**2.0)/2.0/GC)
1088.      FRTF=FTEF/IDC*((VTEF**2.0)/2.0/GC)
1089.      ZRECR=ZRECR
1090.      IF(ZRECR.LE.ELEV)GO TO 643
1091.      ZCONO=(3PCNO-(FRTF*((LLOXO*COS(LEM))+LEOAC)))/((FRCOP*ELEV)/
1092.      3(SIN(GAM)*COS(ALPHA))))/(1.0+(FRCOP/(SIN(GAM)*COS(ALPHA))))
1093.      IF(ZCONO.GT.ELEV)GO TO 649
1094.      643 ZCONO=(3PCNO-(FRTF*((LLOXO*COS(LEM))+LEOAC)))/(1.0+FRTF)
1095.      649 FLOCND=(ZCONO-ELEV)/(SIN(GAM)*COS(ALPHA))
1096.      FLOCND=0.0
1097.      IF(FLOCND.LE.0.0)FLOCND=0.0
1098.      CALL THCOND(TSAT,KL)
1099.      CALL LIQDEN(TSAT,DENSL)
1100.      CALL LIQCP(TSAT,SPHTL)
1101.      CALL LATHT(TCR,TSAT,HFG)
1102.      CALL LIQVIS(TSAT,VISCL)
1103.      CALL LIQHF(TSAT,HFL)
1104.      LCNDSN=HEITC-FLOCND
1105.      DLNGTH=LCNDSN/10.0
1106.      HFTOP=HF
1107.      ICP=1
1108.      TSCS=0.0
1109.      WRITE(6,265)FLOCND,LCNDSN
1110.      265 FORMAT('1. FLOODED CONDENSER LENGTH(FT)='F7.3,3X,'LENGTH AVAILAB
1111.      3LE FOR CONDENSATION(FT)='F7.3)
1112.      IF(MODE.EQ.3)GO TO 1985
1113.      AHTCND=LCNDSN*PI*IDC
1114.      CALL VTRPP(TW2,DENSW,VISCW,KW)
1115.      RENW2=DEQVC*GWC/VISCW
1116.      RPW2=VISCW*SPHTW/KW
1117.      DO 697 LC=1,10
1118.      697 LC=LC+1
1119.      FCT=((1.0+(3.0/EXP((FLOCND+(DLNGTH*(RLC+0.5)))/(30.0*DEQVC)))))/EN
1120.      3TEAC

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1121. IF (PENW2-1900.0) 691.691.692
1122. 692 4UW=0.02*(REW2**0.3)*(PRW2**0.3333)*((EXT(DC/(ODC))**0.53)
1123. HW2=FCT*KW*NUW*3600.0/DEQVC
1124. GO TO 1567
1125. 691 FC=((1.0+(3.5/EXP((FLDCND+(DLNGTH*(RLC+0.5)))/(90.0*DEQVC))))/ENT
1126. 3FAC
1127. 4W2=((1.95*KW/DEQVC*(PENW2*PRW2*DEQVC/(FLDCND+(DLNGTH*(RLC+0.5))))
1128. 3**0.333)*3600.0*FC
1129. 1567 4W25*2=HW25*2+HW2
1130. 597 CONTINUE
1131. HW2=HW25M2/10.0
1132. CALL CONCOF(KL,DENSL,HFG,GC,VISCL,LCNDSN,ALPHA,ALFMIN,IDC,
1133. 3HF,HFL,FLRATE,TSAT,AHTCND,HW2,HTCFCN,CF,TCNDWL,QCOND,TCOLD)
1134. ZZ=1.0/((ALOG(ODC/IDC)*IDC/(2.0*KM))+FOULNW*(DC/ODC/HW2)
1135. TCOLD=TCNDWL-(QCOND/ZZ/AHTCND)
1136. DT=TSAT-TCNDWL
1137. AVFLTH=1.131*(((VISCL*KL*LCNDSN*DT)/(GC*(DENSL**2)*HFG ))**0.25)
1138. IF (ALPHA.GT.ALFMIN)AVFLTH=(KL*3600.0)/HTCFCN
1139. VOID=((IDC-(2.0*AVFLTH)**2)*(PI/4.0))*LCNDSN
1140. VOIDF=VOID/(LCNDSN*XACOND)
1141. CALL MT CONO(TFMPF,KM)
1142. QCOND=(1.0/HTCFCN+IDC/ODC/HW2+((ALOG(ODC/IDC))*IDC/
1143. 3(2.0*KM)))/(AREAC*NTUBEC*NR)
1144. TEMPF=TSAT
1145. DLNGTH=(HEITC-LCNDSN)/(DIVCON-1.0)
1146. HW2CN=HW2
1147. TW2OUT=TCOLD+(QCOND/(MCL*TR*SPHTW))
1148. LENGTH=LCNDSN
1149. LCON=0.0
1150. Q=QCOND
1151. TCWOUT=TCNDWL-(QCOND*THICKC/KM/AHTCND)
1152. TSCSCN=(TCNDWL+TCWOUT)/2.0
1153. AVTSC=TSCSCN
1154. H1C=0.0
1155. X=0.0
1156. SATCON=TSAT
1157. PCOROT=PRESS
1158. IF (IPRINT.EQ.0) GO TO 264
1159. PRINT 269,LENGTH,X,VOIDF
1160. PRINT 258,AVTSC,TEMPF,TSAT
1161. PRINT 270,HTCFCN
1162. PRINT 271,H1C
1163. PRINT 272,Q,PRESS
1164. GO TO 264
1165. 1985 KK=DIVCON
1166. K=1
1167. 1974 Q=0.0
1168. L=DIVCON
1169. FL(K)=FLRATE
1170. DO 1977 KK=1,L
1171. AHTCND=LCNDSN*PI*IDC
1172. HTCFNT=0.343*3600.0*(((KL**3)*(DENSL**2)*HFG*GC/(VISCL*LCNDSN))
1173. 3**0.25)
1174. IF (ALPHA.GT.ALFMIN)HTCFNT=CF*3600.0*(((KL**3)*(DENSL**2)*HFG*GC/
1175. 3(VISCL*IDC))**0.25)
1176. IF (K.EQ.1)OC(K,KK)=(HF-HFL)*FL(K)
1177. CALL AIROEN(TAIR,DENAIR)
1178. CALL EFF(DISCND,AREAC,TAIR,PI,IDC,ODC,CPAIR,GC,ROUT,
1179. 3TCR,DENAIR,CF,DUCTXS,ALPHA,ALFMIN,TUSPCE,HEITC,UA)
1180. JOUTC=(1.0/ROUT)*(1.0/AREAC)
1181. UATOT=(UA*NTUBEC)
1182. NTU=((UATOT)/(CPAIR*DISCND*DENAIR*3600))
1183. EFFECT=1.0-(1.0/EXP(NTU))
1184. IF (K.EQ.2) GO TO 1990
1185. TCOLD=TSAT-((OC(K,KK)*NTUBEC)/(CPAIR*DISCND*DENAIR*3600.
1186. 3*EFFECT))
1187. TCOUT(K,KK)=TCOLD+((OC(K,KK)*NTUBEC)/(CPAIR*DISCND*DENAIR*3600))
1188. QCC1=QC(K,KK)
1189. GO TO 1979
1190. 1990 QC(K,KK)=EFFECT*CPAIR*DISCND*DENAIR*3600.0*(TSAT-TCOUT(K-1,KK))
1191. TCOUT(K,KK)=TCOUT(K-1,KK)+(QC(K,KK)/(CPAIR*DISCND*DENAIR
1192. 3*3600.0))
1193. QC(K,KK)=QC(K,KK)/NTUBEC
1194. QCC2=QC(K,KK)
1195. 1979 TCNDWL=TSAT-((QC(K,KK)/HTCFNT/AHTCND)**1.33333)
1196. HTCFCN=HTCFNT/((TSAT-TCNDWL)**0.25)
1197. DT=TSAT-TCNDWL
1198. AVFLTH=1.131*(((VISCL*KL*LCNDSN*DT)/(GC*(DENSL**2)*HFG ))**0.25)
1199. IF (ALPHA.GT.ALFMIN)AVFLTH=(KL*3600.0)/HTCFCN
1200. VOID=((IDC-(2.0*AVFLTH)**2)*(PI/4.0))*LCNDSN

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1201. VOIDF=VOID/(LCNOSN*ACONO)
1202. CALL HT COND(TEMPF,KM)
1203. RCOND=(1.0/HTCFEN+(1.0/UOUT)*((IDC/ODC)+((ALOG(ODC/IDC))*IDC/
1204. 3(2.0*KM)))/(AREAC*NTUBEC*NROWC)
1205. TEMPF=TSAT
1206. LENGTH=LCNOSN
1207. LCON=0.0
1208. Q=0+QC(K,KK)
1209. IF(K.EQ.2.AND.KK.EQ.DIVCON)FL(K)=Q/HFG
1210. TCWOUT=TCNDWL-(QC(K,KK)*THICKC/KM/AHTCND)
1211. TSCSCN=(TCNDWL+TCWOUT)/2.0
1212. AVTSC=TSCSCN
1213. HIC=0.0
1214. X=0.0
1215. VODC(K)=VOIDF
1216. HCON(K)=HTCFEN
1217. AVTC(K)=AVTSC
1218. TCAIR(K)=TCOUT(K,KK)
1219. CONDO(K)=QC(K,KK)
1220. SATCON=TSAT
1221. IF(KK.NE.L)GOTO 1977
1222. 1977 CONTINUE
1223. KK=K+1
1224. IF(K.GT.NRIGO TO 566
1225. GO TO 1974
1226. 566 TOTFLC=(FL(1)+FL(2))*NTUBEC
1227. IF(ABS(TOTFLC-TOTFLE).LE.5.5)GO TO 567
1228. IF(TOTFLC.GT.TOTFLE)FLRATE=0.9*FL(1)
1229. IF(TOTFLC.LT.TOTFLE)FLRATE=1.22*FL(1)
1230. K=1
1231. GO TO 1974
1232. 567 J=NROWC
1233. DO 1991 K=1,J
1234. PRINT 1980,K
1235. 1980 FORMAT(' ','FOR CONDENSER TUBE ROW #',I1)
1236. PRINT 221
1237. PRINT 1976,FL(K)
1238. 1976 FORMAT(' ','VAPOUR FLOW RATE/TUBE FOR THIS ROW(LB/HR)='',F9.3)
1239. PRINT 269,LENGTH,X,VODC(K)
1240. PRINT 258,AVTC(K),TEMPF,TSAT
1241. PRINT 270,HCON(K)
1242. PRINT 271,HIC
1243. PRINT 272,CONDO(K),PRESS
1244. PRINT 1979,TCAIR(K)
1245. 1979 FORMAT(' ','DOWNSTREAM AIR TEMPERATURE (F)='',F7.3)
1246. 1991 CONTINUE
1247. PRINT 565,UOUTC
1248. 565 FORMAT(' ','CONDENSER AIR SIDE "U" (BTU/HR/SQ-FT/F)='',F9.4)
1249. CALL CNLQHR(FLRATE,NTUBEC,XACOND,DENSL,VISCL,IDH,GC,LHOCN,VOLUME)
1250. LIOFS=LIOFS+2*VOLUME
1251. 264 IF(FLCOND.EQ.0.0)GO TO 612
1252. J=0
1253. ACNOEL=0+IDC*OLNGTH
1254. 235 J=J+1
1255. Q=0.0
1256. TW2IN=TW2OUT
1257. TEM=TEMPF
1258. LCON=LCON+OLNGTH
1259. LENGTH=LENGTH+OLNGTH
1260. FM=(1.0+(9.0/EXP(LCON/10.0*IDC)))
1261. FC=(1.0+(3.5/EXP((HEITC+(OLNGTH/2.0)-LENGTH)/(90.0*DEQVC))))
1262. FCT=(1.0+(3.0/EXP((HEITC+(OLNGTH/2.0)-LENGTH)/(30.0*DEQVC))))
1263. RENW2=DEQVC*GWC/VISCL
1264. PPW2=VISCL*SPHTW/KW
1265. IF(RENW2-1900.0)694.694.371
1266. 571 NUW=0.02*(RENW2*0.8)*(PRW2**0.3333)*((EXTIDC/ODC)**0.53)
1267. HW2=FCT*KW*NUW*3600.0/DEQVC
1268. GO TO 1001
1269. 694 HW2=(1.36*KW/DEQVC*(RENW2*PPW2*DEQVC/(HEITC+(OLNGTH/2.0)-LENGTH))
1270. 3**0.3333)*3600.0*FC
1271. 1001 CALL LIO DEN(TEM ,DENSL)
1272. CALL LIO VIS(TEM ,VISC)
1273. REN=(IDC *GF)/VISC
1274. CG=3.144*(REN**(-0.078153))*((LCON/IDC)**(-1.93/(REN**0.3)))
1275. VOIDF=0.0
1276. PRESS=PRESS+144.0
1277. VF=GF/DENSL
1278. CALL LIO CP(TEM ,SPHTL)
1279. CALL TH CCND(TEM ,KF)
1280. CALL HT CCND(TEM ,KM)

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1241.      PR=SPHTL*VISC./KF
1242.      CALL LAT HT(TCR,TEMPF,HFG)
1243.      Y=1.0
1244.      AA=IDC
1245.      BB=LCCN
1246.      CC=0.3
1247.      HH=HIC
1248.      141 CALL SPHASE(AA,BB,CC,HH,Y,REN,KF,PR,FM,VISC,VISWAL,CG)
1249.      HIC=HH
1250.      IF(R.EQ.0.0)GO TO 550
1251.      R=R+1.0
1252.      IF(PEN -2000.0)17.317.318
1253.      817 F=64.0/REN
1254.      GO TO 591
1255.      918 F=4.0*0.079/(REN**0.25)
1256.      591 HIL2=HITP2
1257.      GO TO 146
1258.      550 U2=1.0/(1.0/HITP2+(1.0/UOUT)*ACNDEL*(IDC/ODC)+((ALOG(ODC /IDC ))
1259.      3*(IDC/(2.0*KM)) +FOULNW+FOULNF)
1260.      DO=U2*ACNDEL*(TEM-TAIR )
1261.      UF2=1.0/(1.0/HITP2+FOULNF)
1262.      TSCF=TEMPF-DO/(UF2*ACNDEL)
1263.      IF(Y.EQ.1.0)GO TO 147
1264.      IF((ABS(TSCF-TSCFP))-TA)551.551.147
1265.      147 CALL VISCLW(TSCF,VISWAL)
1266.      Y=Y+1.0
1267.      TSCFP=TSCF
1268.      GO TO 141
1269.      146 TSCW=TSCF-DO*THICKC/KW/ACNDEL
1270.      AVTSC=(TSCF+TSCW)/2.0
1271.      TSCS=TSCS+AVTSC
1272.      O=O+DO
1273.      DPF=F*DLNGTH/IDC *((VF **2.0)/2.0/GC)*DENSL
1274.      DPA=0.0
1275.      DPH=DLNGTH*DENS*5IN(GAM )*COS(ALPHA)
1276.      PRESS=PRESS+DPH-DPF+DPA
1277.      PRESS=PRESS/144.0
1278.      PCOROT=PRESS
1279.      CALL SAT TEM(PRESS,TSAT)
1280.      SATCO=TSAT
1281.      PCOND=ICO*(LCNDSB/RNCC)
1282.      IF(J.EQ.1)PCOND)GO TO 1000
1283.      GO TO 30
1284.      1000 ICP=ICP+1
1285.      PRINT 259,LENGTH,X,VOIDF
1286.      269 FORMAT(' ','DISTANCE FROM CONDENSER TOP(FT)=' ,F9.4,JX,'QUALITY=' ,
1287.      3F6.4,JX,'VOID FRACTION=' ,F6.4)
1288.      PRINT 258,AVTSC,TEMPF,TSAT
1289.      PRINT 270,HTCFCH
1290.      270 FORMAT(1X,'INSIDE CONDENSATION COEFFICIENT(BTU/HR/FT-FT/F)=' ,F9.4)
1291.      PRINT 271,HIC
1292.      271 FORMAT(' ','1-PHASE LIQUID CONVECTIVE COEFFICIENT(BTU/HR/FT-FT/F)
1293.      3=' ,F9.4)
1294.      PRINT 272,O,PRESS
1295.      272 FORMAT(' ','HEAT REMOVED(BTU/HR)=' ,F9.4,JX,'PRESSURE(PSIA)=' ,F7.3)
1296.      IF(FLDCND.EQ.0.0)GO TO 612
1297.      30 IF(J.LT.ICNDSB)GO TO 235
1298.      612 QOUT=0
1299.      FLRATE=FLRATE*NTUBEC*NR
1300.      IF(MODE.NE.3)QOUTOT=QOUT*NTUBEC*NRWC
1301.      IF(MODE.EQ.3)QOUTOT=(QOCC1+QOCC2)*NTUBEC
1302.      IF(MODE.EQ.3)FLRATE=TOTFLR
1303.      PRINT 613,QOUTOT
1304.      613 FORMAT(' ','TOTAL HEAT REMOVED(BTU/HR)=' ,F14.4)
1305.      QCS=QOUTOT
1306.      VOIDFS=VOIDFS+(VOID*NTUBEC)
1307.      LIQFS=LIQFS+((FLDCND*XACOND)+((1.0-VOIDF)*XACOND*LCNDSN))*NTUBEC
1308.      HF=HFTOP-(QCS/FLRATE)
1309.      AVTSCS=(TSCSCN*LCNDSN/HEITC)+((TSCS/(O(VCON-1.0)))*(FLDCND/HEITC))
1310.      IF(MODE.EQ.3)GOTO 1992
1311.      HW2AVG=(HW2CN*LCNDSN/HEITC)+((HW2SUM/(O(VCON-1.0)))*(FLDCND/HEITC))
1312.      PRINT 980,HW2AVG
1313.      980 FORMAT(' ','AVERAGE WATER SIDE "4" FOR THE COND.(BTU/HR/FT-FT/F)='
1314.      3,F10.3)
1315.      C *****
1316.      C * START OF CONDENSATE RETURN LINE CALCULATIONS
1317.      C *****
1318.      1992 EFFECT=1.0-(EXP(-(LLOXO/(PESLO*FLRATE*SPHTL))))
1319.      QLOXO=EFFECT*FLRATE*SPHTL*(TEMPF-TAIR)
1320.      HF=HF-(QLOXO/FLRATE)
1321.

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1361.      HFBDTN=(FLRATE*HF)+(FRECRC*HR)/(FLRATE+FRECRC)
1362.      TEMPF=TEB
1363.      TEMPMX=TEMPF
1364.      675 CALL LIOHF(TEMPF,HFCK)
1365.      IF((ABS(HFBDTN-HFCK))-0.01)695,695,698
1366.      698 CALL LIDCO(TEMPF,SPHTL)
1367.      TEMPF=TEMPF+(HFBDTN-HFCK)/SPHTL)
1368.      TEMPMX=TEMPF
1369.      GO TO 675
1370.      695 FLRATE=FLRAT(1)
1371.      *****
1372.      *          START OF CONVERGENCE ROUTINE          *
1373.      *****
1374.      IF(IPRINT.EQ.0)GO TO 642
1375.      WRITE(6,603)TCOLD
1376.      603 FORMAT('-.',INLET COOLING FLUID TEMP(F)='F7.3)
1377.      642 IF(ABS(TCOLD-TW2).LE.0.5)GO TO 677
1378.      IF(IPR1.EQ.0)GO TO 679
1379.      WRITE(6,678)TEMPMX,TEB
1380.      678 FORMAT('-.',NEW INITIAL TEMPERATURE(F)='F7.3,5X,'OLD INITIAL TEM
1381.      PERATURE(F)='F7.3)
1382.      GO TO 679
1383.      677 IF(ABS(HFBDTN-HFBDTN).LE.0.06)GO TO 679
1384.      ICOUNT=ICOUNT+1
1385.      IF(IPR1.EQ.0)GO TO 224
1386.      PRINT 277
1387.      WRITE(6,734)FLRATE,TEMPMX
1388.      734 FORMAT('-.',FLOW RATE='F7.3,5X,'NEW INITIAL TEMPERATURE='
1389.      F7.3)
1390.      WRITE(6,735)TEB,PES(1)
1391.      735 FORMAT('-.',OLD INITIAL TEMPERATURE='F7.3,5X,'OLD PRESSURE='
1392.      F9.5)
1393.      PRINT 277
1394.      224 TEB=TEMPMX
1395.      PEVIN=PEV(1)
1396.      PRESS=PEVIN
1397.      679 IF(FLCOND.GT.0.0)GO TO 1004
1398.      IF(ZCOND.GT.(ELEV-HDIAMC))GO TO 1005
1399.      LIOF=1.0-VOIDF)*(((IDC**2)*NTUBEC)/(IDLC**2))
1400.      VOIDF=1.0-LIOF
1401.      VOIDFS=VOIDFS+(VOIDF*(ELEV-ZCOND-HDIAMC)*XALORX)+HEDVLC
1402.      LIOFS=LIOFS+(((LIOF*(ELEV-ZCOND-HDIAMC))+ZCOND)*XALORX)
1403.      GO TO 359
1404.      1005 CALL VOIDHDC (LHDCND,NTUBEC,HDIAMC,ELEV,ZCOND,PI,HEDVLC,VOIDHDC,
1405.      3LHDCND,NR)
1406.      LIOFS=LIOFS+LHDCND*((ELEV-HDIAMC)*XALORX)
1407.      VOIDFS=VOIDFS+VOIDHDC
1408.      GO TO 359
1409.      1004 LIOFS=LIOFS+HEDVLC*((ELEV-HDIAMC)*XALORX)
1410.      359 IF(IPRINT.EQ.0)GO TO 702
1411.      WRITE(6,1002)ZCOND
1412.      1002 FORMAT('-.',LIQUID HEAD IN LOWER CROSS-OVER(FT)='F9.5)
1413.      PRINT 223,QLQX
1414.      223 FORMAT('-.',HEAT LOSS IN LOWER HEADER(BTU/HR)='F9.4)
1415.      PRINT 277
1416.      702 CONTINUE
1417.      IF(VOIDFS.EQ.0.0)GO TO 959
1418.      SYSTVL=(HEITB*NTUBEC*XAEVAP)+(LUPXO*XAUORX)+(HEITC*NTUBEC*XACOND)
1419.      3*((LLOXO+ELEV)*XALORX)+(RECRCL*XARCRC)+HEDVLE+HEDVLC
1420.      AVVOID=VOIDFS/SYSTVL
1421.      GO TO 957
1422.      959 AV VOID=0.0
1423.      957 AV LIOF=1.0-AV VOID
1424.      CALL PERCNT(XAEVAP,NR,NTUBEC,XARCRC,XALORX,ELEV,LIOFS,PERLV
1425.      3G,SIGMA,HEITS,XACOND,NTUBEC,ZCOND,FLOCD,THETA,ALPHA)
1426.      QEFUX=QES/(AREAP*NTUBEC*NROWE)
1427.      QCFUX=QCS/(AREAC*NTUBEC*NROWC)
1428.      QAVFLX=(QES+QCS)/((AREAC+AREAP)*NTUBEC*NR)
1429.      AVTSHS=(AVEVT1+AVEVT2)/2.0
1430.      IF(MODE.NE.3)GO TO 1986
1431.      IF(NROWE.EQ.1)AVEVT2=0.0
1432.      IF(NROWE.EQ.1)PFROY2=0.0
1433.      AVTSHS=(AVEVT1+AVEVT2)/2.0
1434.      IF(NROWE.EQ.1)AVTSHS=AVEVT1
1435.      1986 DELT=AVTSHS-AVTSCS
1436.      JEXPP=QEFUX/DELT
1437.      BITA=BETA*190.0/PI
1438.      CALL STUNIT(HEITS,IDH,HEITC,IDC,ELEV,LUPXO,LOUXO,LLJXO,
1439.      3IDLC,PES(1),TOTALF,QEFUX,QCFUX,QAVFLX,QES,QCS,AVTSHS,AVTSCS,DELT,
1440.      3AVWATH,JEXPP,UNA,SGMA,THETA,ALFA,AGAM,BITA,AVVOID,AVLIOF,PER,NG,

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1441. 3TEB, QAVFLX, THOT1, THOT2, TW2, AVEVT1, AVEVT2, PRCFL,
1442. 3TW2OUT, MCWTR, EXTINC, EEXT1, EEXT2, EEXT3, DEQVE,
1443. 3DQVC, PERDY1, PERDY2, PERDY3, HW2AVG, MHTW1, MHTW2, CNROTH,
1444. 3MODE, OFLUX, UOUT, UOUTC)
1445. PRINT 277
1446. IF (FLAG4.EQ.0) GO TO 1006
1447. WRITE(6,1007)
1448. 1007 FORMAT(' ', 'STATIC LIQUID LEVEL RESTS IN CONDENSER HEADER.')
1449. 1006 FLAG4=0
1450. XX=TW2-TCOLD
1451. PRINT 277
1452. IF ((ABS(TW2-TCOLD)).LE.0.5) GO TO 3
1453. IICOUN=IICOUN+1
1454. IF (IICOUN.GE.2) GO TO 128
1455. IF (TW2-TCOLD) 2,3,2
1456. C CHANGING SATURATION TEMPERATURE IN THE CONDENSER
1457. 2 CALL SAT PR(SATCON,P1)
1458. TEMPF=SATCON+XX
1459. CALL SAT PR(TEMPF,P2)
1460. DEVINN=PEB(1)+(XX/(RCOND*1500+XX/(P2-P1)))
1461. WRITE(6,122) DEVINN
1462. 122 FORMAT(' ', 'NEW PRESSURE AT THE BOTTOM OF THE EVAPORATOR=',F9.5)
1463. WRITE(6,123) PEB(1)
1464. 123 FORMAT(' ', 'OLD PRESSURE AT THE BOTTOM OF THE EVAPORATOR=',F9.5)
1465. TSAT1=TSATE9
1466. PRESS=DEVINN
1467. TC1=TCOLD
1468. PRINT 277
1469. GO TO 308
1470. 129 TSATEN=TSATE9+((TSAT1-TSATE9)/(TC1-TCOLD))*XX
1471. TEMPF=TSATEN
1472. CALL SAT PR(TEMPF,PSAT)
1473. DEVINN=PSAT
1474. PRINT 122,DEVINN
1475. PRINT 123,PEB(1)
1476. TC1=TCOLD
1477. TSAT1=TSATE9
1478. PRESS=DEVINN
1479. PRINT 277
1480. GO TO 308
1481. 3 WRITE(6,129)
1482. 129 FORMAT(' ', 'CONVERGENCE OBTAINED ON THE COLD FLUID TEMPERATURE')
1483. 121 PRINT 277
1484. STOP
1485. END
1486. SUBROUTINE DTINPO(TSHF,TSAT,PECLET,DITI)
1487. C THIS SUBROUTINE ESTIMATES THE TEMPERATURE DIFFERENCE REQUIRED TO
1488. C MAINTAIN BOILING.
1489. DITI=(TSHF-TSAT)*((1.0/7000.0)*PECLET-(1.0/7.0))
1490. IF (PECLET.LE.11000.0) DITI=0.0
1491. IF (PECLET.GE.18000.0) DITI=(TSHF-TSAT)
1492. RETURN
1493. END
1494. SUBROUTINE FRCSHN(FLRATE,DIAM,DENS,VISC,PI,F,VF)
1495. C THIS SUBROUTINE CALCULATES THE FRICTION FACTOR
1496. IF (FLRATE.GT.0.0) GO TO 1
1497. F=0.0
1498. VF=0.0
1499. GO TO 2
1500. 1 VF=FLRATE**4.0/(PI*(DIAM**2.0)*DENS)/3600.0
1501. RE=VF*DIAM*DENS/VISC
1502. F=0.046**4.0/(RE**0.2)
1503. IF (RE.LT.2000.0) F=64.0/RE
1504. 2 RETURN
1505. END
1506. SUBROUTINE BULKPT(HF,HFL,HFOLD,HFOLD,LO,LN,PC,PRESS,FLRATE,Q,
1507. 3TSHF0,TSHF,VOID0,VOIDF,TSAT)
1508. C THIS SUBROUTINE CALCULATES THE POINT WHERE BULK BOILING HAS STARTED
1509. REAL LO,LN
1510. A=(LN-LO)
1511. Q=HFOLD-(((HF-HFOLD)/A)*LO)
1512. C=HFOLD-(((HFL-HFOLD)/A)*LO)
1513. LN=((B-C)*A)/((HFL-HFOLD)-(HF-HFOLD))
1514. PRESS=P0-(((LN-LO)/A)*(P0-PRESS))
1515. CALL SATTEM(PRESS,TSAT)
1516. CALL HFLS(TSAT,HFL)
1517. HFOLD=HF
1518. HFL=HFL
1519. Q=Q-((HFOLD-HF)*FLRATE)
1520. TSHF=TSHF0+(((LN-LO)/A)*(TSHF-TSHF0))

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1521.      VOIDF=VOID0+(((LN-LO)/A)*(VOIDF-VOID0))
1522.      RETURN
1523.      END
1524.      SUBROUTINE SBCONL(TSHF,TSAT,TSHFO,TSATO,LO,LN,PO,PRESS,J,TEMPFO,
1525.      3TEMPF,HF,HFL,FLRATE,THPP)
1526.      C      THIS SUBROUTINE CALCULATES THE POINT WHERE SUBCOOLED BOILING HAS STARTED
1527.      REAL LO,LN
1528.      A=LN-LO
1529.      B=TSHFO-(((TSAT-TSHFO)/A)*LO)
1530.      C=TSATO-(((TSAT-TSATO)/A)*LO)
1531.      LN=((B-C)*A)/((TSAT-TSATO)-(TSHF-TSHFO))
1532.      TSHF=B+(((TSHF-TSHFO)/A)*LN)
1533.      TSAT=TSHF
1534.      THPP=TSHF
1535.      PRESS=EXP(((-4623.38)/(TSAT+410.6652))+12.2055)
1536.      TEMPF=TEMPF-((1.0-((LN-LO)/A))* (TEMPF-TEMPFO))
1537.      CALL HFLS(TSAT,HFL)
1538.      HFOLD=HF
1539.      CALL LIOHF(TEMPF,HF)
1540.      Q=Q-((HFOLD-HF)*FLRATE)
1541.      RETURN
1542.      END
1543.      SUBROUTINE SAT PR(TEMPF,PSAT)
1544.      C      THIS SUBROUTINE CALCULATES THE SATURATION PRESSURE FOR A
1545.      C      GIVEN TEMPERATURE
1546.      PSAT=EXP((-4623.38/(TEMPF+410.6652))+12.2055)
1547.      RETURN
1548.      END
1549.      SUBROUTINE SAT TEM(PRESS,TSAT)
1550.      C      THIS SUBROUTINE CALCULATES THE SATURATION TEMPERATURE FOR A
1551.      C      GIVEN PRESSURE
1552.      TSAT=(-4623.38/(ALOG(PRESS)-12.2055))-410.6652
1553.      RETURN
1554.      END
1555.      SUBROUTINE LIO HF(TEMPF,HFL)
1556.      C      THIS SUBROUTINE CALCULATES LIQUID ENTHALPY FOR A GIVEN TEMPERATURE
1557.      HFL =9.1634845+0.1993021*TEMPF+0.000091623*(TEMPF**2)+0.00300042
1558.      31026*(TEMPF**3)+0.000000012*(TEMPF**4)
1559.      RETURN
1560.      END
1561.      SUBROUTINE LIO DEN(TEMPF,DENSL)
1562.      C      THIS SUBROUTINE CALCULATES DENSITY OF LIQUID FOR A GIVEN TEMPER
1563.      DENSL=98.216372-0.07441*TEMPF-0.0000660245*(TEMPF**2)+0.000001505
1564.      3*(TEMPF**3)-0.0000000067*(TEMPF**4)
1565.      RETURN
1566.      END
1567.      SUBROUTINE LAT HT(TCR,TEMPF,HFG)
1568.      C      THIS SUBROUTINE CALCULATES LATENT HEAT OF EVAPORATION FOR A
1569.      C      GIVEN TEMPERATURE
1570.      HFG=2.6154541+0.246939*ALOG(TCR-TEMPF)+0.0095585*((ALOG(TCR-TEMPF
1571.      3))**2)
1572.      HFG=EXP(HFG)
1573.      RETURN
1574.      END
1575.      SUBROUTINE SUR TEN(PAR,DENSL,DENSG,MWT,SURFTN)
1576.      C      THIS SUBROUTINE CALCULATES SURFACE TENSION FOR GIVEN PARACHOR,
1577.      C      LIQUID DENSITY,GAS DENSITY,MOLECULAR WEIGHT
1578.      REAL MWT
1579.      SURFTN=((PAR*(DENSL-DENSG)*0.01602/MWT)**4)*(2.2481/((10**6)*0.03
1580.      3251))
1581.      RETURN
1582.      END
1583.      SUBROUTINE LIO VIS(TEMPF,VISCL)
1584.      C      THIS SUBROUTINE FINDS LIQUID VISCOSITY FOR A GIVEN TEMPERATURE
1585.      VISCL=-3.9026924+(1640.191052323/(TEMPF+460.0))
1586.      VISCL=EXP(VISCL)*0.000672
1587.      RETURN
1588.      END
1589.      SUBROUTINE TH COND(TEMPF,KL)
1590.      C      THIS SUBROUTINE CALCULATES THERMAL CONDUCTIVITY FOR A GIVEN TEMP.
1591.      C      TEMPERATURE
1592.      REAL KL
1593.      KL=(0.056-0.000111*(TEMPF-32.0))/3600.0
1594.      RETURN
1595.      END
1596.      SUBROUTINE MT COND(TEMPF,KM)
1597.      C      THIS SUBROUTINE FINDS THERMAL CONDUCTIVITY OF METAL FOR GIVEN
1598.      C      TEMPERATURE
1599.      REAL KM
1600.      KM=224.0-((TEMPF-32.0)/30.0)

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1601. RETURN
1602. END
1603. SUBROUTINE LIO CP(TEMPF,SPHTL)
1604. C THIS SUBROUTINE CALCULATES SP. HEAT OF LIQID FOR A GIVEN TEMP.
1605. SPHTL=-1.219353-(164.748862/(TEMPF+460.0))
1606. SPHTL=EXP(SPHTL)
1607. RETURN
1608. END
1609. SUBROUTINE GAS VIS(TEMPF,VISCG)
1610. C THIS SUBROUTINE CALCULATES GAS VISCOSITY FOR A GIVEN TEMPERATURE
1611. VISCG=-3.679208963-(454.851219/(TEMPF+460.0))
1612. VISCG=EXP(VISCG)*0.000672
1613. RETURN
1614. END
1615. SUBROUTINE GAS CP(TEMPF,SPHTG)
1616. C THIS SUBROUTINE CALCULATES THE SP. HEAT OF GAS FOR A GIVEN TEMP.
1617. SPHTG=0.03433333+0.003272*TEMPF-0.000035341*(TEMPF**2)+0.000000155
1618. 3*(TEMPF**3)-0.00000000023022*(TEMPF**4)
1619. RETURN
1620. END
1621. SUBROUTINE VISHGW(TSHF,VISWAL)
1622. C THIS SUBROUTINE CALCULATES GAS VISCOSITY AT THE WALL TEMPERATURE
1623. C OF THE EVAPORATOR
1624. VISWAL=-3.679208963-(454.851219/(TSHF+460.0))
1625. VISWAL=EXP(VISWAL)*0.000672
1626. RETURN
1627. END
1628. SUBROUTINE VISHLW(TSHF,VISWAL)
1629. C THIS SUBROUTINE CALCULATES LIQ. VISCOSITY AT THE WALL TEMPERATURE
1630. C OF THE EVAPORATOR
1631. VISWAL=-3.9026924+(1640.191052323/(TSHF+460.0))
1632. VISWAL=EXP(VISWAL)*0.000672
1633. RETURN
1634. END
1635. SUBROUTINE HFLS(TST,HFL)
1636. C THIS SUBROUTINE CALCULATES THE SATURATED ENTHALPY
1637. HFL=8.1634845+0.1993021*TS T +0.000091623*(TS T **2)-0.0000004210
1638. 326*(TS T **3)+0.0000000012*(TS T **4)
1639. RETURN
1640. END
1641. SUBROUTINE HFGTS(TCP,TST,HFGO)
1642. C THIS SUBROUTINE CALCULATES THE LATENT HEAT
1643. HFGO=2.6154541+0.246939*ALOG(TCR-TS T )+0.0095585*((ALOG(TC?-TS T )
1644. 9)**2)
1645. HFGO=EXP(HFGO)
1646. RETURN
1647. END
1648. SUBROUTINE VISCLW(TSCF,VISWAL)
1649. C THIS SUBROUTINE CALCULATES LIQ. VISCOSITY AT THE WALL TEMPERATURE
1650. C OF THE CONDENSER
1651. VISWAL=-3.9026924+(1640.191052323/(TSCF+460.0))
1652. VISWAL=EXP(VISWAL)*0.000672
1653. RETURN
1654. END
1655. SUBROUTINE VISGW(TSCF,VISWAL)
1656. C THIS SUBROUTINE CALCULATES GAS VISCOSITY AT THE WALL TEMPERATURE
1657. C OF THE CONDENSER
1658. VISWAL=-3.679208963-(454.851219/(TSCF+460.0))
1659. VISWAL=EXP(VISWAL)*0.000672
1660. RETURN
1661. END
1662. SUBROUTINE VOONDE(LNHOP,HEDIAM,ZRECR,HEITB,SIGMA,HEDVOL,PI,NTUBE
1663. 3E,VDFHDE,LIOHDE,FLAG3)
1664. C THIS SUBROUTINE CALCULATES THE VOLUME OF LIQUID & VAPOUR IN THE
1665. C EVAPORATOR OUTLET HEADER
1666. REAL LIOHDE
1667. INTEGER FLAG3
1668. IF(ZRECR<LE.(HEITB*COS(SIGMA)))GO TO 1
1669. IF(ZRECR<GT.((HEITB*COS(SIGMA))+(HEDIAM/2.0)))GO TO 2
1670. X=(HEDIAM/2.0)-(ZRECR-(HEITB*COS(SIGMA)))
1671. CALL TUBVOL(HEDIAM,LNHOP,NTUBES,X,VOL,PI)
1672. LIOHDE=VOL
1673. VDFHDE=HEDVOL-LIOHDE
1674. GO TO 3
1675. 2 X= ZRECR-(HEITB*(COS(SIGMA)))-(HEDIAM/2.0)
1676. CALL TUBVOL(HEDIAM,LNHOP,NTUBES,X,VOL,PI)
1677. VDFHDE=VOL
1678. LIOHDE=HEDVOL-VDFHDE
1679. GO TO 3
1680. 1 VDFHDE=HEDVOL

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1681. LIQHDE=0.0
1682. FLAG3=1
1683. 3 RETURN
1684. END
1685. SUBROUTINE VODHOC(LNHDR,NTUBEC,HEDIAM,ELEV,ZCOND,PI,HEDVOL,VDFHOC,
1686. 3 LIQHOC,NR)
1687. C THIS SUBROUTINE CALCULATES THE VOLUME OF LIQUID & VAPOUR IN THE
1688. C CONDENSER OUTLET HEADER
1689. REAL LIQHOC
1690. IF(ZCOND.GT.(ELEV-(HEDIAM/2.0)))GO TO 1
1691. X=(ELEV-(HEDIAM/2.0))-ZCOND
1692. CALL TUBVOL(HEDIAM,LNHDR,NTUBEC,X,VOL,PI)
1693. LIQHOC=VOL
1694. VDFHOC=HEDVOL-LIQHOC
1695. GO TO 2
1696. 1 X=ZCOND-(ELEV-(HEDIAM/2.0))
1697. CALL TUBVOL(HEDIAM,LNHDR,NTUBEC,X,VOL,PI)
1698. VDFHOC=VOL
1699. LIQHOC=HEDVOL-VDFHOC
1700. 2 RETURN
1701. END
1702. SUBROUTINE PERCNT(XAEVAP,NR,NTUBEE,XARCRC,XALCRX,ELEV,LIOFS,PERLN
1703. 3 G,SIGMA,HEITB,XACOND,NTUBEC,ZCOND,FLDCND,THETA,ALPHA)
1704. C THIS SUBROUTINE CALCULATES THE X STATIC CHARGE IN THE EVAPORATOR
1705. REAL LIOFS
1706. NROWC=NR
1707. IF(ELEV.EQ.0.0)GO TO 91
1708. IF(FLDCND.EQ.0.0)GO TO 93
1709. GO TO 94
1710. 93 X=LIOFS/(((XAEVAP*NR*NTUBEE)/COS(SIGMA))+2.0*XARCRC+XALCRX)
1711. GO TO 92
1712. 94 X=LIOFS/(((XAEVAP*NR*NTUBEE)/COS(SIGMA))+2.0*XARCRC+((NR*NR*WC*XAC
1713. 3 OND)/COS(ALPHA)))
1714. GO TO 92
1715. 91 X=LIOFS/(((XAEVAP*NR*NTUBEE)/COS(SIGMA))+2.0*XARCRC+((NR*NR*WC*XAC
1716. 3 OND)/COS(ALPHA)))
1717. 92 PERLNG=((X/(COS(SIGMA)*SIN(THETA)))/HEITB)*100.0
1718. RETURN
1719. END
1720. SUBROUTINE TUBVOL(HEDIAM,LNHDR,NTUBE,X,VOL,PI)
1721. C THIS SUBROUTINE CALCULATES THE VOLUME OF LIQUID & VAPOUR
1722. R=HEDIAM/2.0
1723. THETA=ARCOS(X/R)
1724. A=(2.0*THETA*PI)/180.0
1725. B=2.0*THETA
1726. C=SIN(B)
1727. VOL=LNHDR*(R**2.0/2)*(A-C)
1728. RETURN
1729. END
1730. SUBROUTINE WTRPRP(TEMP,DENSW,VISCW,KW)
1731. C THIS SUBROUTINE CALCULATES THE PROPERTIES FOR WATER
1732. REAL KW
1733. DENSW=62.3777+(0.003684*TEMP)-(0.0000746*(TEMP**2.0))
1734. VISCW=(2.2975-(0.02236*TEMP)+(0.00006187*(TEMP**2.0)))*0.000672
1735. KW=(0.3174239+0.0004397*TEMP)/3600.0
1736. RETURN
1737. END
1738. SUBROUTINE LIQVOL(FRECRC,VLIOHE)
1739. C THIS SUBROUTINE CALCULATES THE LIQUID VOLUME IN EVAP VAP. HEADER
1740. VLIOHE=1.4*(FRECRC/210.0)
1741. VLIOHE=VLIOHE/1728.0
1742. RETURN
1743. END
1744. SUBROUTINE VOL(CNROTN,VSTRAN,THETA)
1745. C THIS SUBROUTINE CALCULATES THE LIQUID VOLUME IN THE SEPARATOR
1746. VSTRAN=0.500/1728.0
1747. IF(CNROTN.EQ.0)VSTRAN=VSTRAN*(1-COS(THETA))
1748. RETURN
1749. END
1750. SUBROUTINE CNLQHR(FLRATE,NTUBEC,XACOND,DENSL,VISCL,IDH,GC,FLDCND,
1751. 3 VOLUME)
1752. C THIS SUBROUTINE CALCULATES THE LIQUID VOLUME IN THE CONDENSER LIQUID
1753. C HEADER
1754. REAL IDH,LHDCND
1755. FLRATE=FLRATE*NTUBEC
1756. VEL=((FLRATE/3600.0)/(XACOND*DENSL))
1757. REN=(DENSL*VEL*IDH)/VISCL
1758. F=64.0/REN
1759. IF(REN.GT.2000.0)F=(4.0*0.079)/(REN**0.25)
1760. DELP=((F*1.5*VEL*VEL*DENSL)/(2.0*GC*IDH))
1761.

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1761.      Z=(DELO/DENSL)*12.0
1762.      VOL=Z*(LHOCND*12.0)
1763.      VOLUME=VOL/1728.0
1764.      FLRATE=FLRATE/NTUBEC
1765.      RETURN
1766.      END
1767.      SUBROUTINE PRLINT(TAIR,THOTG,PEBEST,DENSL,PMIN,PMAX)
1768.      C THIS SUBROUTINE CALCULATES THE MAX. & MIN. PRESSURES FOR THE SYSTEM
1769.      TEMPF=TAIR
1770.      CALL SAT PR(TEMPF,PMIN)
1771.      TEMPF=THOTG
1772.      CALL SAT PR(TEMPF,PMAX)
1773.      CALL LID DEN(TEMPF,DENSL)
1774.      PRINT 1,PMIN
1775.      1 FORMAT(' ',MINIMUM PRESSURE LIMIT FOR THE SYSTEM(Psia)=',F9.5)
1776.      PRINT 2,PMAX
1777.      2 FORMAT(' ',MAXIMUM PRESSURE LIMIT FOR THE SYSTEM(Psia)=',F9.5)
1778.      RETURN
1779.      END
1780.      SUBROUTINE MASS FL(PI,IDH,HEITS,TEMPF,FLRATE,TCR)
1781.      C THIS SUBROUTINE CALCULATES THE MASS FLOW RATE OF THE WORKING FLUID
1782.      REAL IDH,KL
1783.      FLRATE=10.0
1784.      X=0.45
1785.      AREAS=PI*IDH*HEITS
1786.      AREASX=(PI/4.0)*(IDH**2.0)
1787.      TWALL=TEMPF+8.0
1788.      CALL GAS VIS (TEMPF,VISCG)
1789.      CALL LID VIS (TEMPF,VISCL)
1790.      CALL TH COND (TEMPF,KL)
1791.      CALL LAT HT(TCR,TEMPF,HFG)
1792.      CALL LID CP(TEMPF,SPHCL)
1793.      2 FL=FLRATE
1794.      VIS2P=1.0/(X/VISCG+(1.0-X)/VISCL)
1795.      H2P=0.033*(KL/IDH)*((IDH*GF/VIS2P)**0.87)*((SPHCL*VISCL/KL)**0.4)
1796.      QIN=H2P*AREAS*(TWALL-TEMPF)
1797.      FLRATE=QIN/(HFG*X)
1798.      IF((ABS(FLRATE-FL))/FL)-0.1)1,1.2
1799.      1 RETURN
1800.      END
1801.      SUBROUTINE AIRDEN(AIRTEM,DENAIR)
1802.      C THIS SUBROUTINE CALCULATES THE DENSITY OF AIR
1803.      R1=0.0860146816
1804.      R2=-1.83631126*(1.0/10000.0)
1805.      R3=2.38149835*(1.0/1000000.0)
1806.      DENAIR=R1+R2*AIRTEM+R3*AIRTEM**2
1807.      RETURN
1808.      END
1809.      SUBROUTINE EFF(DISC,AREA,TAIR,PI,INDIA,OUTDIA,CPAIR,GC,ROUT,
1810.      3TCR,DENAIR,CF,DUCTXS,ALPHA,ALFMIN,TUSPCE,LENGTH,UA)
1811.      C THIS SUBROUTINE CALCULATES THE AIR SIDE RESISTANCE
1812.      REAL INDIA,OUTDIA,KL,KM,MASAIR,LENGTH
1813.      VELAIR=0.1SC/DUCTXS
1814.      TSAT=120.0
1815.      TWALL=110.0
1816.      TEMPF=(TSAT+TWALL)/2.0
1817.      R0=0.59718992
1818.      R1=-0.08137122
1819.      R2=5.8313175*(1.0/1000)
1820.      R3=-1.5054689*(1.0/10000)
1821.      EFFC=R0+R1*VELAIR+R2*(VELAIR**2)+R3*(VELAIR**3)
1822.      MASAIR=(DENAIR*VELAIR*3600.0)*(LENGTH*TUSPCE/12.0)
1823.      UA=(MASAIR*CPAIR)*(ALOG(1.0/(1.0-EFFC)))
1824.      DENSL=72.708
1825.      DENSG=0.29799
1826.      KL=0.04865/3600.0
1827.      HFG=75.461
1828.      VISCL=0.4645/3600.0
1829.      CALL MT COND(TEMPF,KM)
1830.      72 HTCFNT=0.943*3600.0*(((KL**3)*(DENSL**2)*HFG*GC/(VISCL*LENGTH))
1831.      3**0.25)
1832.      IF (ALPHA.GT.ALFMIN)HTCFNT=CF*3600.0*(((KL**3)*(DENSL**2)*HFG*GC/
1833.      3(VISCL*INDIA))**0.25)
1834.      HCOND=HTCFNT/((TSAT-TWALL)**0.25)
1835.      RINSID=1.0/(HCOND*PI*(INDIA*LENGTH))
1836.      RWALL=(ALOG(OUTDIA/INDIA))*((1.0/(2*PI*LENGTH*KM))
1837.      3ROUT=1.0/UA-(RINSID+RWALL)
1838.      Q=MASAIR*CPAIR*EFFC*(TWALL-TAIR)
1839.      TWALLN=TAIR+Q/UA
1840.      IF (ABS(TWALLN-TWALL).LE.0.6)GO TO 77

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1841.      TWALL=TWALLN
1842.      GO TO 72
1843.      77 RETURN
1844.      END
1845.      SUBROUTINE QLOSS(A1,TE3,TROOM,LBE,RESBE,HF90T,QBE,HBOTBE)
1846.      THIS SUBROUTINE CALCULATES THE HEAT LOSS FROM THE TUBES BELOW THE EVAP.
1847.      REAL LBE
1848.      CALL LTOCP(TE3,SPHTL)
1849.      EFFECT=1.0-(EXP(-(LBE/(RESBE*A1*SPHTL))))
1850.      IF(EFFECT.EQ.1.0)TEMPBE=TE3
1851.      IF(EFFECT.EQ.1.0)GO TO 356
1852.      TEMPBE=(TE3-(EFFECT*TROOM))/(1.0-EFFECT)
1853.      356 QBE=EFFECT*A1*SPHTL*(TEMPBE-TROOM)
1854.      HBOTBE=HF90T+(QBE/A1)
1855.      RETURN
1856.      END
1857.      SUBROUTINE DELPBE(FLRATE,LBE,LEQBE,GC,ZBE,IDBE,DPEFVAP,NTUBES,PI,
1858.      DENSL,VISCL,XABE,DLPBE,DPDL,LENGTH,DENSG,VISCG,IDUC,NELROW,
1859.      LETMCH,DPNONH,DPTTE,DPLEBO,DPEXIT)
1860.      THIS SUBROUTINE CALCULATES THE PRESSURE DROP FOR THE TUBES BELOW
1861.      THE EVAPORATOR
1862.      REAL LBE,LEQBE,IDBE,IDUC,LENGTH,LETMCH
1863.      FLRATE=FLRATE*NTUBES
1864.      GFBE=FLRATE/XABE/3600.0
1865.      RENBE=(IDBE*GFBE)/VISCL
1866.      IF(RENBE-2000.0)698,699,699
1867.      698 F=64.0/RENBE
1868.      GO TO 605
1869.      699 F=4.0*0.079/(RENBE**0.25)
1870.      605 DLPBE=((F*(LBE+LEQBE)/IDBE*((GFBE/DENSL)**2.0)/2.0/GC)*DENSL)+
1871.      3(ZBE*DENSL))/144.0
1872.      DPNONH=DPDL*LENGTH/144.0
1873.      XAVAPH=(IDUC**2.0)*(PI/4)
1874.      GFVAPH=FLRATE/XAVAPH/3600.0
1875.      DPTTE=1.4*((GFVAPH/DENSG)**2.0)*(1.0/(2.0*GC))
1876.      DPLEBO=NELROW*0.9*(1.0/(2.0*GC))*((GFVAPH/DENSG)**2.0)
1877.      DPEXIT=1.0*(1.0/(2.0*GC))*((GFVAPH/DENSG)**2.0)
1878.      RENTOP=(IDUC*GFVAPH)/VISCG
1879.      IF(SENTOP-2000.0)98,98,99
1880.      98 F=64.0/RENTOP
1881.      GO TO 7
1882.      99 F=4.0*0.079/(RENTOP**0.25)
1883.      7 DPEFVAP=((F*LETMCH/IDUC*((GFVAPH/DENSG)**2.0)/2.0/GC)*DENSG)/144.0
1884.      RETURN
1885.      END
1886.      SUBROUTINE SCOOB(KL,HFG,SURFTN,TSAT,DENSG,DENSL,PI,IDH,LENGTH,
1887.      REN,VISCL,PRF,VL,TEMPF,Q,QA,QQ)
1888.      THIS SUBROUTINE CALCULATES THE HEAT FLUX REQUIRED TO SUSTAIN
1889.      BOILING
1890.      REAL KL,IDH,LENGTH
1891.      QA=0/(PI*IDH*LENGTH)
1892.      QQ1=QA
1893.      CC=(778.2*KL*3600.0*HFG/(8.0*SURFTN*(TSAT+460.0)*((1./DENSG1)-(1./
1894.      DENSL))))
1895.      117 IF(REN.LT.2000.0)GO TO 118
1896.      CB=((IDH**0.2)*((VISCL/DENSL)**0.8)*((QQ1/3600.0)/(0.023*(DQ=**0
1897.      3.4)*KL*(VL**0.8))))
1898.      GO TO 119
1899.      118 CB=((QQ1/3600.0)/1.86/KL/.667)*((IDH*VISCL*LENGTH/(VL*DENSL*PRF))
1900.      3**0.3333)
1901.      119 QQ=(CC*((CB-(TSAT-TEMPF))**2)*3600.0)
1902.      IF(ABS(QQ-QQ1)-1.0)112,112,111
1903.      111 DQDT=2*3600*CC*(CB-(TSAT-TEMPF))*CB/QQ1
1904.      QQ1=QQ1+(QQ1-QQ)/DQDT
1905.      GO TO 117
1906.      112 RETURN
1907.      END
1908.      SUBROUTINE CRITQY(REN,PF,HFG,Q,PI,GC,IDH,LENGTH,DLNTH,FBL,LFR,
1909.      DENSL,DENSG,SURFTN,VISCL,LIOF,X,IPRI,PRG,HITPC,HITD,GF,KG,
1910.      DH2PI,XFR)
1911.      THIS SUBROUTINE CALCULATES THE CRITICAL QUALITY AND IF XCR>X,
1912.      SINGLE PHASE VAPOUR "M" IS CALCULATED
1913.      REAL IDH,LENGTH,LIOF,KG
1914.      IF(LFR.GT.0)GO TO 54
1915.      WEB=((1.39/(10**7))*REN*((DENSL/DENSG)**0.60))*((IDH/
1916.      30.0608)**(-0.48))
1917.      XFB=SQRT(WEB*NO*GC*SURFTN*DENSG/(GF*VISCL*((REN/LIOF)**0.125)))
1918.      IF(X.LE.XFB)GO TO 51
1919.      54 LFB=LFB+1
1920.      IF(IPRI.E2.0)GO TO 127

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1921. IF(LFB.EQ.1)PRINT 254,XFB,LENGTH
1922. 254 FORMAT('---1. TRANSITION TO MIST FLOW TAKES PLACE AT QUALITY=',F6.4,
1923. '33X, AND AT A HEIGHT(FT)=' ,F8.4)
1924. 127 FBL=FBL+DLNGTH
1925. IF(RENG.LT.2000.0)GO TO 52
1926. H2P2=(0.323*KG/IDH )*(RENG**0.8)*(PRG**0.4)*3600.0
1927. GO TO 51
1928. 52 H2P2=(1.86*KG/IDH )*(RENG*PRG*IDH /FBL)**0.3333)*3600.0
1929. 51 HITPC=H2P1
1930. IF(X.GT.XFB)HITPC=H2P2
1931. RETURN
1932. END
1933. SUBROUTINE SPRESS(DLNGTH,DENSG,THETA,SIGMA,F, IDH,VF,GC,DENS,X,
1934. DPH,DPF,OPA)
1935. C THIS SUBROUTINE CALCULATES PRESSURE DROP FOR SINGLE PHASE REGION
1936. REAL IDH
1937. IF(X.GE.1.0)GO TO 70
1938. OPA=0.0
1939. DPH=DLNGTH*DENSL*SIN(THETA)*COS(SIGMA)
1940. DPF=F*DLNGTH/IDH *((VF**2.0)/2.0/GC)*DENSL
1941. GO TO 71
1942. 70 VOIDF=1.0
1943. LIOF=0.0
1944. OPA=0.0
1945. XP=0.0
1946. PHI=1.0
1947. DPH=DLNGTH*DENSG*SIN(THETA)*COS(SIGMA)
1948. DPF=F*DLNGTH/IDH *((VF**2.0)/2.0/GC)*DENSG
1949. 71 RETURN
1950. END
1951. SUBROUTINE PRES2P(DPAC,OPA,DPACP,LIOF,DENSL,VOIDF,DENSG,THETA,
1952. SIGMA,XBOTM,F,DLNGTH, IDH,GF,GC,FLRATE,PI,X,XFB,RENF,RENG,XP,
1953. VISCL,VISCG,PHI,IX)
1954. REAL IDH,LIOF
1955. C THIS SUBROUTINE CALCULATES TWO-PHASE PRESSURE DROP
1956. 16 DPAC=(GF*GF/(DENSL*GC))*(((1.0-X)**2.0)/(1.0-VOIDF))+(X*X)*DENSL/
1957. 3(DENSG*VOIDF))-1.0)
1958. OPA=DPAC-DPACP
1959. DPACP=DPAC
1960. DENSM=(LIOF*DENSL+VOIDF*DENSG)
1961. DPH=DENSM*DLNGTH*SIN(THETA)*COS(SIGMA)
1962. C USING LOCKHART & MARTINELLI METHOD TO FIND TWO-PHASE PRESS DROP
1963. CALL XPCAL(RENF,PENG,VISCL,VISCG,DENSG,X,XP)
1964. IF(IX.GT.0)GO TO 529
1965. CALL PHICAL(RENF,RENG,PHI,XP)
1966. IF(XBOTM.LE.0.0)GO TO 529
1967. DPF=F*DLNGTH/IDH*(((GF/DENSL)**2.0)/2.0/GC)*DENSL
1968. DPF=DPF*(PHI**2.0)
1969. RENGL=4.0*FLRATE*X/(PI*IDH*VISCG)/3600.0
1970. VG1=(RENG1*VISCG)/(IDH*DENSG)
1971. F=0.046**0.0/(RENG1**0.2)
1972. IF(PENG1.LT.2000.0)F=64.0/RENG1
1973. DPF=GF*DLNGTH/IDH*((VG1**2.0)/2.0/GC)*DENSG
1974. IF(X.GT.XFB)DPF=(((X-XFB)*DPFG)+((1.0-X)*DPF))/((1.0-XFB)
1975. 529 RETURN
1976. END
1977. SUBROUTINE TWOPHE(X,VISCG,VISCL,H2P1,KL,IDH,GF,SPHTL)
1978. C THIS SUBROUTINE CALCULATES THE TWO PHASE HEAT TRANSFER COEFF.
1979. REAL IDH,KL
1980. VIS2P=1.0/(X/VISCG+(1-X)/VISCL)
1981. H2P1=0.33*(KL/IDH)*(((IDH*GF/VIS2P)**0.87)*((SPHTL*VISCL/KL)**0.4)
1982. 3*3600.0
1983. RETURN
1984. END
1985. SUBROUTINE CONCOF(KL,DENSL,HFG,GC,VISCL,LCNDSN,ALPHA,ALFMIN,IDC,
1986. DHF,HFL,FLRATE,TSAT,AHTCND,HW2,HTCFN,CF,TCNDWL,QCONO,TCOLO)
1987. C THIS SUBROUTINE CALCULATES THE CONDENSATION COEFFICIENT IN THE CONDENSED.
1988. REAL KL,LCNDSN,IDC
1989. HTCFNT=0.943*3600.0*(((KL**3)*(DENSL**2)*HFG*GC/(VISCL*LCNDSN))
1990. 3**0.25)
1991. IF(ALPHA.GT.ALFMIN)HTCFNT=CF*3600.0*(((KL**3)*(DENSL**2)*HFG*GC/
1992. 3(VISCL*IDC))**0.25)
1993. QCONO=(HFL-HFL)*FLRATE
1994. TCNDWL=TSAT-(QCONO /HTCFNT/AHTCND)**1.33333)
1995. HTCFN=HTCFNT/((TSAT-TCNDWL)**0.25)
1996. RETURN
1997. END
1998. SUBROUTINE CORRECT(TSHS,AVTSH,DO,DOO,TSHW,TSHF,THICK,KM,AREAH,
1999. DLNGTH,HEIT9,LENGTH,RM,II,LO,TEMPFO,IDH,PI,UF)
2000. C THIS SUBROUTINE MAKES CORRECTION FOR Q,TWALL IN THE SURCOOL S

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2001. C BULK BOILING SUBROUTINES, AFTER THE POINT WHERE SUBCOOLED/
2002. C BULK BOILING HAS BEEN LOCATED
2003. REAL KM,LENGTH,LO,IDH
2004. TSHS=TSMS-AVTS
2005. DD=0-70
2006. TSHF=TEMPD+(DD/(UF*(LENGTH-LO)*(DH*PI)))
2007. TSHW=TSHF+(DD*THICKH/(KM*(LENGTH-LO)*(DH*PI)))
2008. AVTS=(TSHW+TSHF)/2.0
2009. TSMS=TSMS+AVTS
2010. DLNTH=(HEITS-LENGTH)/(RM-II)
2011. RETURN
2012. END
2013. SUBROUTINE HPNOLS(OTA,TSHF,TSAT,DITI,HITPB,SPHTL,HFG,CSF,HTFLUX,
2014. 3REX,VISCL,SURFTN,DENSL,DENSG,TEMPF,PRF,X)
2015. C THIS SUBROUTINE CALCULATES THE HEAT FLUX & THE BOILING HEAT
2016. C TRANSFER COEFFICIENT
2017. OTA=TSHF-TSAT
2018. IF(DITI.GT.OTA)HITPB=0.0
2019. IF(DITI.GT.OTA)GO TO 7025
2020. DTB=OTA-DITI
2021. HTFLUX=((SPHTL*DTB)/(HFG*CSF*(PRF**1.7)))*BEX)*((VISCL
2022. 3HFG)/((SURFTN/(DENSL-DENSG))**0.5))
2023. HITPB=(HTFLUX/(TSHF-TEMPF))*3600.0
2024. 7025 IF(X.LE.0.0)GO TO 508
2025. 508 RETURN
2026. END
2027. SUBROUTINE SPHASE(AA,BB,CC,HH,Y,REN,KF,PR,FM,VISC,VISWAL,CG)
2028. C THIS SUBROUTINE CALCULATES THE SINGLE PHASE HEAT TRANSFER COEFF.
2029. C IN EVAPORATOR & CONDENSER
2030. REAL KF,AA
2031. IF(Y.GT.1.0)GO TO 131
2032. 131 IF(REN-2000.0)5.6.6
2033. 5 IF(Y.GT.1.0)GO TO 132
2034. 7D=(1.86*KF/AA)*((REN*PR*AA/BB)**0.333)*3600.0
2035. EE=DD*0.6667*FM
2036. GO TO 416
2037. 132 DD=(1.86*KF/AA)*((REN*PR*AA/BB)**0.333)*((VISC/VISW
2038. 3AL)**0.14)*3600.0*0.6667
2039. EE=DD*FM
2040. GO TO 416
2041. 6 IF(Y.GT.1.0)GO TO 134
2042. DD=(0.027*KF/AA)*((REN**0.8)*(PR**CC))*3600.0
2043. EE=DD*CG
2044. GO TO 416
2045. 134 DD=(0.027*KF/AA)*((REN**0.8)*(PR**CC)*((VISC/VISWAL)**0.14
2046. 3))*3600.0
2047. EE=DD*CG
2048. 416 HH=EE
2049. RETURN
2050. END
2051. SUBROUTINE XPCAL(RENF,RENG,VISCL,VISCG,DENSL,DENSG,X,XP)
2052. C THIS SUBROUTINE CALCULATES THE LOCKHART-MARTINELLI PARAMETER
2053. C "XP" AT THE EXIT QUALITY
2054. IF(RENF.GT.2000.0.AND.RENG.GT.2000.0)XP=((VISCL/VISCG)**0.1)*((
2055. 3DENSG/DENSL)**0.5)*(((1.0-X)/X)**0.9)
2056. IF(RENF.LE.2000.0.AND.RENG.GT.2000.0)XP=(RENG**(-0.4))*(((15.0/0.
2057. 3046)*(1.0-X)/X)*(DENSG/DENSL)*(VISCL/VISCG))**0.5)
2058. IF(RENF.GT.2000.0.AND.RENG.LE.2000.0)XP=(RENF**0.4)*(((0.0+5/16.0)
2059. 3*((1.0-X)/X)*(DENSG/DENSL)*(VISCL/VISCG))**0.5)
2060. IF(RENF.LE.2000.0.AND.RENG.LE.2000.0)XP=((1.0-X)/X)*(DENSG/DENSL)
2061. 3*(VISCL/VISCG)**0.5
2062. RETURN
2063. END
2064. SUBROUTINE PHICAL(RENF,RENG,PHI,XP)
2065. C THIS SUBROUTINE CALCULATES THE LOCKHART-MARTINELLI PARAMETER "PHI"
2066. IF(RENF.GT.2000.0.AND.RENG.GT.2000.0)PHI=((1.0+XP**(-2.0*0.2433975
2067. 3))**((1.0/0.2433975))**0.5)
2068. IF(RENF.LE.2000.0.AND.RENG.GT.2000.0)PHI=((1.0+XP**(-2.0*0.2775379
2069. 3))**((1.0/0.2775379))**0.5)
2070. IF(RENF.GT.2000.0.AND.RENG.LE.2000.0)PHI=((1.0+XP**(-2.0*0.2920067
2071. 3))**((1.0/0.2920067))**0.5)
2072. IF(RENF.LE.2000.0.AND.RENG.LE.2000.0)PHI=((1.0+XP**(-2.0*0.3495068
2073. 3))**((1.0/0.3495068))**0.5)
2074. RETURN
2075. END
2076. SUBROUTINE DATA(IDH,IDC,THICKH,THICKC,HEITS,HEITC,LUPXO,LLOXO,
2077. C SIGMA,THETA,ALPHA,GAM,ELEV,XXX,TAIR)
2078. REAL IDH,IDC,LUPXO,LLOXO
2079. REAL IDH,IDC,ALPHA,GAM,ELEV,XXX,TAIR)
2080. INTEGER XXX

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2081. PRINT 9020
2082. 9020 FORMAT('1',-----)
2083.
2084. WRITE(6,9019)
2085. 9019 FORMAT('1',-----) DATA CARDS FOR THIS RUN
2086.
2087. PRINT 9022
2088. 9022 FORMAT('1',-----)
2089.
2090. IF(XXX.EQ.0)GO TO 7575
2091. QIDH=(IDH*30.48)/12.0
2092. QHEITB=HEITB*30.48
2093. QTHIK=(THICKH*30.48)/12.0
2094. QIDC=(IDC*30.48)/12.0
2095. QHEITC=HEITC*30.48
2096. QTHIC=(THICKC*30.48)/12.0
2097. QLUPXO=LUPXO*30.48
2098. QLLOXO=LLOXO*30.48
2099. QELEV=ELEV*30.48
2100. QTAIR=(TAIR-32.0)/1.8
2101. WRITE(6,9000) QIDH,IDH
2102. 9000 FORMAT('1', 'EVAPORATOR: I.D. OF TUBES(CM)=' ,F9.5,3X,
2103. 3 'I.D. OF TUBES(INCHS)=' ,F7.3)
2104. WRITE(6,9001) QHEITB,HEITB
2105. 9001 FORMAT('1', 'EVAPORATOR: LENGTH OF TUBES(CM)=' ,F7.2,3X,
2106. 3 'LENGTH OF TUBES(Feet)=' ,F7.3)
2107. WRITE(6,9002) QTHIK,THICKH
2108. 9002 FORMAT('1', 'EVAPORATOR: THICKNESS OF TUBES(CM)=' ,F9.5,3X,
2109. 3 'THICKNESS OF TUBES(INCHS)=' ,F7.3)
2110. WRITE(6,9003) QIDC,IDC
2111. 9003 FORMAT('1', 'CONDENSER: I.D. OF TUBES(CM)=' ,F9.5,3X,
2112. 3 'I.D. OF TUBES(INCHS)=' ,F7.3)
2113. WRITE(6,9004) QHEITC,HEITC
2114. 9004 FORMAT('1', 'CONDENSER: LENGTH OF TUBES(CM)=' ,F7.2,3X,
2115. 3 'LENGTH OF TUBES(Feet)=' ,F7.3)
2116. WRITE(6,9005) QTHIC,THICKC
2117. 9005 FORMAT('1', 'CONDENSER: THICKNESS OF TUBES(CM)=' ,F9.5,3X,
2118. 3 'THICKNESS OF TUBES(INCHS)=' ,F7.3)
2119. WRITE(6,9006) QLUPXO,LUPXO
2120. 9006 FORMAT('1', 'LENGTH OF UPPER CROSS OVER TUBE(CM)=' ,F7.2,3X,
2121. 3 'LENGTH OF UPPER CROSS OVER TUBE(Feet)=' ,F7.2)
2122. WRITE(6,9007) QLLOXO,LLOXO
2123. 9007 FORMAT('1', 'LENGTH OF LOWER CROSS OVER TUBE(CM)=' ,F7.2,3X,
2124. 3 'LENGTH OF LOWER CROSS OVER TUBE(Feet)=' ,F7.2)
2125. WRITE(6,9008) QELEV,ELEV
2126. 9008 FORMAT('1', 'CONDENSER ELEVATION(CM)=' ,F7.2,3X,
2127. 3 'CONDENSER ELEVATION(Feet)=' ,F7.3)
2128. WRITE(6,9023) SIGMA,THETA
2129. 9023 FORMAT('1', 'EVAPORATOR: INCLINATION(DEG)=' ,F6.2,3X,
2130. 3 'ROTATION(DEG)=' ,F6.2)
2131. WRITE(6,9024) ALPHA,GAM
2132. 9024 FORMAT('1', 'CONDENSER: INCLINATION(DEG)=' ,F6.2,3X,
2133. 3 'ROTATION(DEG)=' ,F6.2)
2134. WRITE(6,9029) QTAIR,TAIR
2135. 9029 FORMAT('1', 'INCOMING TEMPERATURE OF AIR(C)=' ,F7.2,3X,
2136. 3 'INCOMING TEMPERATURE OF AIR(F)=' ,F7.2)
2137. GO TO 9018
2138. 7575 QIDH=(IDH/30.48)*12.0
2139. QHEITB=HEITB/30.48
2140. QTHIK=(THICKH/30.48)*12.0
2141. QIDC=(IDC/30.48)*12.0
2142. QHEITC=HEITC/30.48
2143. QTHIC=(THICKC/30.48)*12.0
2144. QLUPXO=LUPXO/30.48
2145. QLLOXO=LLOXO/30.48
2146. QELEV=ELEV/30.48
2147. QTAIR=32.0+TAIR*1.8
2148. *WRITE(6,9000) IDH,QIDH
2149. *WRITE(6,9001) HEITB,QHEITB
2150. *WRITE(6,9002) THICKH,QTHIK
2151. *WRITE(6,9003) IDC,QIDC
2152. *WRITE(6,9004) HEITC,QHEITC
2153. *WRITE(6,9005) THICKC,QTHIC
2154. *WRITE(6,9006) LUPXO,QLUPXO
2155. *WRITE(6,9007) LLOXO,QLLOXO
2156. *WRITE(6,9009) ELEV,QELEV
2157. *WRITE(6,9023) SIGMA,THETA
2158. *WRITE(6,9024) ALPHA,GAM
2159. *WRITE(6,9029) QTAIR,TAIR
2160. IDH=QIDH

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2161.      HEITB=QHEITB
2162.      THICKH=QTHIK
2163.      IDC=QIDC
2164.      HEITC=QHEITC
2165.      THICKC=QTHIC
2166.      LUPXO=QLUPXO
2167.      LLOXO=QLLOXO
2168.      ELEV=QELEV
2169.      TAIR=QTAIR
2170. 9018 PRINT 9021
2171. 9021 FORMAT(' ',
2172. -----
2173.      RETURN
2174.      END
2175.      SUBROUTINE SIUNIT(HEITB, IDH, HEITC, IDC, ELEV, LUPXO, IDUC, LLOXO,
2176. 3IDLC, PSTART, TOTALF, QEF, QFLUX, QCF, QAVFL, QES, QCS, AVTSHS, AVTSCS, DELT,
2177. 3AVWATH, JEXPR, UGA, SGMA, THTA, XLCFA, AGAM, BITA, AVV, CID, AVL, IQF, PER, L, NG,
2178. 3TEB, QAVFLX, THOT1, THOT2, TW2, AVEVT1, AVEVT2, PRCF,
2179. 3TW2OUT, MCLWTR, EXTIDE, EXTIDC, EEXT1, EEXT2, EEXT3, DEQVE,
2180. 3DEQVC, PERDY1, PERDY2, PERDY3, HW2AVG, MHOTW1, MHOTW2, CNROT,
2181. 3MODE, QFLUX, UOUT, UOUTC)
2182. C THIS SUBROUTINE CONVERTS ALL DATA IN S.I. UNITS
2183. REAL IDH, IDC, LUPXO, IDUC, LLOXO, IDLC, MHTWTR, MCLWTR
2184. SHEITB=HEITB*30.48
2185. SIDH=IDH*30.48
2186. SHEITC=HEITC*30.48
2187. SIDC=IDC*30.48
2188. SDEQVE=DEQVE*30.48
2189. SDEQVC=DEQVC*30.48
2190. SELEV=ELEV*30.48
2191. SWIDE=LUPXO*30.48
2192. SIDUC=IDUC*30.48
2193. IF(MODE.EQ.3)GOTO 919
2194. SEEXT1=(EEXT1-32.0)/1.8
2195. SEEXT2=(EEXT2-32.0)/1.8
2196. STW2OT=(TW2OUT-32.0)/1.8
2197. SMHT1=(MHOTW1*0.00755988)/60.0
2198. SMHT2=(MHOTW2*0.00755988)/60.0
2199. SMCL=(MCLWTR*0.00755988*3.0)/60.0
2200. SEXIDE=EXTIDE*30.48
2201. SEXIDC=EXTIDC*30.48
2202. SHW=AVWATH*5.6783
2203. SHW2=HW2AVG*5.6783
2204. 919 SWIDER=LLOXO*30.48
2205. SIDLC=IDLC*30.48
2206. STEB=(TEB-32.0)/1.8
2207. STHOT1=(THOT1-32.0)/1.8
2208. STHOT2=(THOT2-32.0)/1.8
2209. STCOWA=(TW2-32.0)/1.8
2210. SPC=(PSTART*6894.8)
2211. SFLRT=(TOTALF*0.00755988)/60.0
2212. SQE=QEF*3.152481
2213. SQC=QCF*3.152481
2214. SQA=QAVFLX*3.152481
2215. SQES=QES*0.292875
2216. SQCS=QCS*0.292875
2217. SATH1=(AVEVT1-32.0)/1.8
2218. SATH2=(AVEVT2-32.0)/1.8
2219. SATC=(AVTSCS-32.0)/1.8
2220. SATH=(AVTSHS-32.0)/1.8
2221. SDT=DELT/1.8
2222. PRINT 920
2223. 920 FORMAT(' ',
2224. -----
2225.      WRITE(6,279)
2226. 279 FORMAT(' ', 'RESULTS IN SI UNITS')
2227.      PRINT 922
2228. 922 FORMAT(' ',
2229. -----
2230.      WRITE(6,822)SHEITB, SIDH, SGMA, THTA
2231. 922 FORMAT(' ', 'EVAPORATOR: LENGTH(CM)=', F7.2, 3X, '1.0. OF TUBES(CM)=',
2232. 3F9.5, 3X, 'INCLINATION( DEG)=', F6.2, 3X, 'ROTATION( DEG)=', F6.2)
2233.      WRITE(6,680)SHEITC, SIDC, ALFA, AGAM
2234. 680 FORMAT(' ', 'CONDENSER: LENGTH(CM)=', F7.2, 3X, '1.0. OF TUBES(CM)=',
2235. 3F9.5, 3X, 'INCLINATION( DEG)=', F6.2, 3X, 'ROTATION( DEG)=', F6.2)
2236.      IF(MODE.EQ.2.OR.MODE.EQ.3)GO TO 1961
2237.      PRINT 968, SDEQVC
2238. 968 FORMAT(' ', 'EQUIVALENT DIA. OF WATER JACKET OF THE EVAPORATOR(CM)=',
2239. 3F9.3)
2240.      PRINT 969, SDEQVE

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2241. 969 FORMAT(' ', 'EQUIVALENT DIA. OF WATER JACKET OF THE CONDENSER(CM)='
2242. 3,F9.3)
2243. 1961 WRITE(6,824)SELEV
2244. 824 FORMAT(' ', 'CONDENSER ELEVATION(CM)=' ,F7.2)
2245. WRITE(6,682)SWIDE,SIDUC,BITA
2246. 682 FORMAT(' ', 'LENGTH OF UPPER CROSS-OVER(CM)=' ,F7.2,3X, 'I.D. OF TUBE
2247. 3(CM)=' ,F9.5,3X, 'INCLINATION( DEG)=' ,F6.2)
2248. WRITE(6,826)SWIDER,SIDLC
2249. 826 FORMAT(' ', 'LENGTH OF LOWER CROSS-OVER(CM)=' ,F7.2,3X, 'I.D. OF TUBE
2250. 3(CM)=' ,F9.5)
2251. WRITE(6,602)STER
2252. 602 FORMAT(' ', 'FINAL FREON TEMPERATURE(C)=' ,F7.3)
2253. WRITE(6,946)SPC
2254. 946 FORMAT(' ', 'FINAL FREON PRESSURE(NEWTON/M2)=' ,F10.2)
2255. WRITE(6,807)SFLRT
2256. 807 FORMAT(' ', 'FINAL FREON FLOW RATE(KGM/SEC)=' ,F10.7)
2257. WRITE(6,966)SOA
2258. 966 FORMAT(' ', 'FINAL HEAT FLUX(WATT/M2)=' ,F11.3)
2259. IF(MODE.EQ.2)GO TO 1962
2260. IF(MODE.EQ.3)GO TO 221
2261. WRITE(6,299)SMHT1
2262. 299 FORMAT(' ', 'MASS FLOW RATE OF HEATING FLUID ROW # 1(KGM/SEC)=' ,F10
2263. 3,7)
2264. WRITE(6,699)SMHT2
2265. 699 FORMAT(' ', 'MASS FLOW RATE OF HEATING FLUID ROW # 2(KGM/SEC)=' ,F10
2266. 3,7)
2267. 1962 WRITE(6,230)SMCL
2268. 230 FORMAT(' ', 'MASS FLOW RATE OF COOLING FLUID(KGM/SEC)=' ,F10.7)
2269. IF(MODE.EQ.2) GO TO 1964
2270. WRITE(6,828)STHOT1,SLEXT1
2271. 828 FORMAT(' ', 'INCOMING TEMP OF HEATING FLUID FOR ROW # 1(C)=' ,F7.3,
2272. 3X, 'OUTGOING TEMP OF HEATING FLUID(C)=' ,F7.3)
2273. WRITE(6,830)STHOT2,SLEXT2
2274. 830 FORMAT(' ', 'INCOMING TEMP OF HEATING FLUID FOR ROW # 2(C)=' ,F7.3,
2275. 3X, 'OUTGOING TEMP OF HEATING FLUID(C)=' ,F7.3)
2276. 1964 WRITE(6,832)STCWA,STW2OT
2277. 832 FORMAT(' ', 'INCOMING TEMP OF COOLING FLUID (C)=' ,F7.3,3X, 'OUTGOING
2278. 3 TEMP OF COOLING FLUID(C)=' ,F7.3)
2279. 921 WRITE(6,834)SATH1
2280. 834 FORMAT(' ', 'AVERAGE WALL TEMP FOR ROW # 1(C)=' ,F9.3)
2281. WRITE(6,835)SATH2
2282. 835 FORMAT(' ', 'AVERAGE WALL TEMP FOR ROW # 2(C)=' ,F9.3)
2283. WRITE(6,811)SATC,SATC
2284. 811 FORMAT(' ', 'AVERAGE HEATER SIDE WALL TEMP (C)=' ,F9.3,3X, 'AVERAGE C
2285. 3 CONDENSER SIDE WALL TEMP(C)=' ,F9.3)
2286. WRITE(6,945)SDT
2287. 945 FORMAT(' ', 'DIFFERENCE BETWEEN AVERAGE HEATER AND CONDENSER WALL T
2288. 3 TEMPERATURES(C)=' ,F9.3)
2289. IF(MODE.EQ.2)GO TO 1965
2290. IF(MODE.EQ.3)SHW=UOUT*5.6783
2291. IF(MODE.EQ.3)SHW2=UOUTC*5.6783
2292. WRITE(6,600)SHW
2293. 600 FORMAT(' ', 'AVG. OUTSIDE EVAP. HEAT TRANSFER COEF.(WATT/M2/C)=' ,
2294. 3F10.2)
2295. WRITE(6,575)SHW2
2296. 575 FORMAT(' ', 'AVG. OUTSIDE COND. HEAT TRANSFER COEF.(WATT/M2/C)=' ,
2297. 3F10.2)
2298. 1965 WRITE(6,855)SOES,SQCS
2299. 855 FORMAT(' ', 'QIN EVAP.(WATT)=' ,F9.2,3X, 'QOUT COND.(WATT)=' ,F9.2)
2300. WRITE(6,827)SQE,SQC,SOA
2301. 827 FORMAT(' ', 'EVAP. FLUX(WATT/M2)=' ,F10.3,3X, 'COND. FLUX(WATT/M2)=' ,
2302. 3F10.3,3X, 'AVG. FLUX(WATT/M2)=' ,F10.3)
2303. WRITE(6,941)AVVOID,AVL1QF
2304. 941 FORMAT(' ', 'AVERAGE VOID FRACTION FOR ENTIRE LOOP=' ,F6.4,5X,
2305. 3 'AVERAGE LIQUID FRACTION FOR ENTIRE LOOP=' ,F6.4)
2306. IF(THTA.NE.90.0)GO TO 711
2307. WRITE(6,570)PERLNG
2308. 570 FORMAT(' ', 'PERCENT OF EVAP. LENGTH FLOODED UNDER: STATIC CONDITIO
2309. 3 N=' ,F5.1)
2310. GO TO 713
2311. 711 WRITE(6,712)PERLNG
2312. 712 FORMAT(' ', 'AVERAGE STATIC CHARGE IN THE EVAPORATOR=' ,F5.1)
2313. 713 WRITE(6,970)PERDY1
2314. 970 FORMAT(' ', 'PERCENT DYNAMIC CHARGE IN ROW # 1=' ,F5.1)
2315. WRITE(6,971)PERDY2
2316. 971 FORMAT(' ', 'PERCENT DYNAMIC CHARGE IN ROW # 2=' ,F5.1)
2317. IF(CNROT4.EQ.0)GO TO 967
2318. IF(MODE.EQ.3)GO TO 267
2319. WRITE(6,572)PRCCL
2320. 572 FORMAT(' ', 'APPARENT % DYNAMIC CHARGE(RECIRCULATION TUBE)=' ,F5.1)

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2321. 967 RETURN
2322. END
2323. /*
2324. //GO.SYSIN ON *
2325. 0.311 0.311 0.032 0.032 3.00 1.00 13.00 20.0 102.191
2326. 0.0 70.0 45.0 90.0 4.00 1.0 2 2 0
2327. 0.0 0.0 388.4 193.4 137.36854 0.01 1.80 1000 6.0
2328. 0.50 12 36 3.00 30.0 70.0 1000 102.191 1
2329. 1.0 1.0 1.0
2330. 50.333 27.0 1 1 1
2331. 1
2332. 447.6 1.0 1 0.545 1 5.0 2.0 0.0
2333. 3.0 0.8739 0.8739 0.24 1.0 0.311
2334. 3 4000.0 275.00 0.311 79.08
2335. 1.0 1.0 16.0 40.0 30.0
2336. 104.0
2337. 122.0
2338. 196.6
2339. 196.6
2340. //

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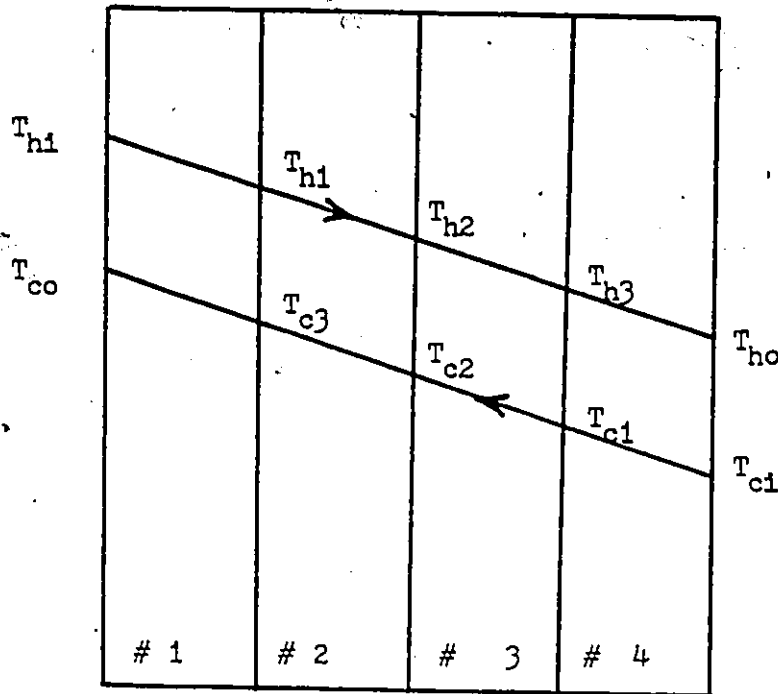

APPENDIX - H4

PROPERTY SUBROUTINES FOR R-113

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SUBROUTINE SAT PR(TEMPF,PSAT)
  PSAT=EXP((( -5084.637)/(TEMPF+406.3816))+12.3902)
  RETURN
END
SUBROUTINE SAT TEM(PRESS,TSAT)
  TSAT=(-5084.637/(ALOG(PRESS)-12.3902))-406.3816
  RETURN
END
SUBROUTINE LIQ HF(TEMPF,HFL)
  HFL = 7.1943386+0.2217*TEMPF-0.0000349*(TEMPF**2)+0.00000012914
  * (TEMPF**3)+0.0000000026402*(TEMPF**4)
  RETURN
END
SUBROUTINE LIQ DEN(TEMPF,DENSL)
  DENSL=101.09509+0.02066*TEMPF-0.00109*(TEMPF**2)+0.0000043665
  * (TEMPF**3)-0.0000000062323*(TEMPF**4)
  RETURN
END
SUBROUTINE LAT HT(TCR,TEMPF,HFG)
  HFG=0.9198105+0.78294*ALOG(TCR-TEMPF)-0.038114*((ALOG(TCR-TEMPF
  ))**2)
  HFG=EXP(HFG)
  RETURN
END
SUBROUTINE SUR TEN(PAR,DENSL,DENSG,MWT,SURFTN)
  REAL MWT
  SURFTN=((PAR*(DENSL-DENSG)*0.01602/MWT)**4)*(2.2481/((10**6)*0.03
  281))
  RETURN
END
SUBROUTINE LIQ VIS(TEMPF,VISCL)
  VISCL=-4.19153+(2030.8041/(TEMPF+460.0))
  VISCL=EXP(VISCL)*0.000672
  RETURN
END
SUBROUTINE TH COND(TEMPF,KL)
  REAL KL
  KL=(0.04775-0.0000625*TEMPF)/3600.0
  RETURN
END
SUBROUTINE LIQ CP(TEMPF,SPHTL)
  SPHTL=-0.8126-(366.0732/(TEMPF+460.0))
  SPHTL=EXP(SPHTL)
  RETURN
END
SUBROUTINE GAS VIS(TEMPF,VISCG)
  VISCG=(-0.9990031-(1992.131/(TEMPF+460.0)))
  VISCG=EXP(VISCG)*0.000672
  RETURN
END
SUBROUTINE GAS CP(TEMPF,SPHTG)
  SPHTG=0.16+(0.01/100)*(TEMPF-100)
  RETURN
END
SUBROUTINE VISHGW(TSHF,VISWAL)
  VISWAL=(-0.9990031-(1992.131/(TSHF+460.0)))
  VISWAL=EXP(VISWAL)*0.000672
  RETURN
END
SUBROUTINE VISHLW(TSHF,VISWAL)
  VISWAL=-4.19153+(2030.8041/(TSHF+460.0))
  VISWAL=EXP(VISWAL)*0.000672
  RETURN
END

```

APPENDIX I

Assumption: Same temperature drop across each loop ($\dot{m}_h = \dot{m}_c$)

Temperature drop across evaporator for loop # 1 $\Delta T_1 = T_{h1} - T_{c1}$

Overall temperature difference for loop # 1 $OTD_1 = T_{h1} - T_{c3}$

Effectiveness for one loop $E_1 = \Delta T_1 / OTD_1$... (1)

Effectiveness for "n" loops $E_n = \frac{n \Delta T_1}{OTD_n} = \frac{n \Delta T_1}{OTD_1 + (n - 1) \Delta T_1}$

$$E_n = \frac{n E_1}{1 + (n - 1) E_1} \quad \dots (2)$$

and

$$\begin{aligned} OTD_1 &= OTD_n - (n - 1) \Delta T_1 \\ &= OTD_n - (n - 1) E_1 OTD_1 \end{aligned}$$

or $OTD_n = OTD_1 [1 + (n - 1) E_1] \quad \dots (3)$

Using equation for the single loop, the effectiveness E_n and the overall temperature difference OTD_n for "n" loops can be predicted in terms of E_1 and OTD_1 from equations 2 and 3 respectively.

PUBLICATIONS WITH DR. T.W. McDONALD

PAPERS

1. Mathur, G.D. and McDonald, T.W. Simulation program for a two-phase thermosiphon-loop heat exchanger. ASHRAE Trans., Vol 92, Pt. 2, paper # 3005, June 1986.
2. Stauder, F.A., Mathur, G.D. & McDonald, T.W. Experimental and computer simulation study of an air-to-air two-phase thermosiphon-loop heat-exchanger. ASME paper # 85-WA/HT-15.
3. Mathur, G.D. & McDonald, T.W. Validation of simulation program for two-phase coil loop thermosiphon heat exchanger systems. Presented at world congress on heating ventilating and air conditioning, CLIMA 2000, Copenhagen, Denmark, Vol 6, pp 557-562, Aug. 1985.

ABSTRACT

1. Mathur, G.D. & McDonald, T.W. Simulation program for a two-phase thermosiphon loop heat exchanger. Presented at 10th Canadian congress of applied mechanics, London, Ontario, June 2-7, 1985.

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